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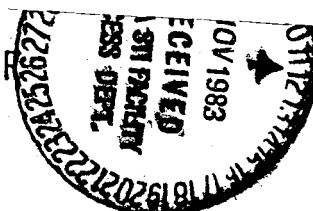
(NASA-CR-166069) IMPROVEMENTS TO THE FATOLA
COMPUTER PROGRAM INCLUDING ADDED ACTIVITY
CONTROLLED LANDING GEAR SUBROUTINES Final
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IMPROVEMENTS TO THE FATOLA COMPUTER
PROGRAM INCLUDING ADDED ACTIVELY
CONTROLLED LANDING GEAR SUBROUTINES



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INTRODUCTION

Program TOLA (Take-Off and Landing Analysis) provides a non real time simulation of the dynamics of conventional aircraft during takeoffs and landings. The program models the performance of an aircraft during a takeoff roll or during the glide slope, flare, impact, and rollout of a landing. It includes the effects of a number of external and internal conditions such as wind shears, rough runway, engine failure, ground effect, etc. Extensive documentation of the TOLA program--its capabilities, problem formulation, and user and programmer guides--have been written (Lynch 1972; Lynch and Dueweke 1974 a & b; Young and Dueweke 1975).

TOLA has been modified to include a flexible airframe option (Dick and Benda 1975) and is identified as program FATOLA at NASA Langley. Following validation of the flexible airframe analysis and some additional modifications to the program to improve its capabilities (Carden and McGehee 1977), a provision for actively controlled landing gear has been incorporated. The active control code simulates dynamic load control during impact and rollout, and during takeoff roll on rough runways. Additionally, a program restart capability has been added as well as other program enhancements.

This report includes a brief description of the added capabilities, a detailed description of specific program changes, and includes information required for a user to exercise the new options. A complete listing of the modifications to the FATOLA program is included as an appendix.

1. ADDED PROGRAM CAPABILITIES

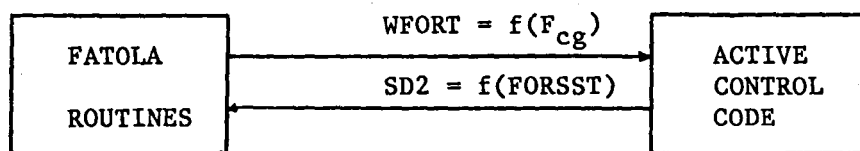
The FATOLA program has been modified to provide for actively controlled landing gears, metering pin area as a function of strut stroke, and continuously varying aerodynamic coefficients C_{A_0} and C_{N_0} . A restart capability has been added to the program.

1.1 Active Control Landing Gear - The active gear limits the force applied at the gear-airframe interface by limiting the shock strut force (FORSST) with a closed loop series-hydraulic control. An impact limit force (WLFOR) is determined from the value of the gear-airframe interface force (WFORT) very shortly after aircraft touchdown. This determination is made when the work potential of the strut (energy dissipation potential) exceeds the aircraft energy at touchdown apportioned among the main gears. The gear-airframe interface force is compared with the impact limit force, the shock strut force then being adjusted to bring the gear-airframe interface force within specified limits about the control impact limit force.

After the initial impact energy has been dissipated and the aircraft is in the rollout phase of the landing simulation, the control limit force is reduced to a value where the shock strut force is controlled to support the aircraft weight. The transition from impact limit force (WLFOR) to rollout limit force (WLFORR) is

carried out smoothly by the use of a ramp function from impact to rollout. Active control performs similarly during a takeoff roll by using the rollout limit force value to control the interface force experienced when accelerating on a rough runway. A detailed description of the active control model is presented in a technical note by McGehee and Carden (1976).

1.2 Implementation of the Active Control Code - In FATOLA, the landing gear dynamics are modeled in subroutines LGEAR1 and LGEA3C. The active control subroutines replace the passive shock strut calculations in LGEAR1 (see Figs. 1 & 2). However, passive gear calculations can be made using the active gear subroutines. The control code is integrated into FATOLA by defining the gear-airframe interface force in terms of the total force applied to the aircraft center of gravity, which is computed in FATOLA. The active code returns a value for the shock strut force which is a term used in the FATOLA calculation of the shock strut acceleration:



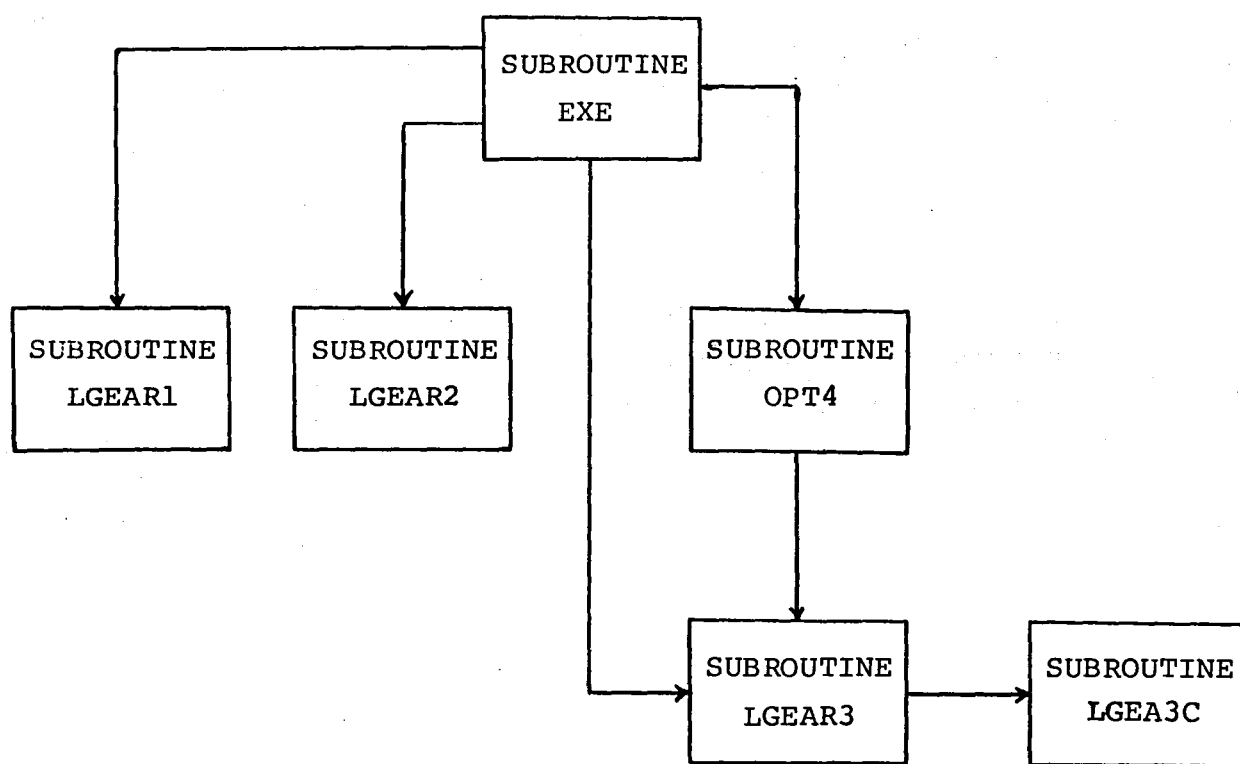


FIGURE 1: SUBROUTINE LINKAGE FOR CALCULATION OF PASSIVE
LANDING GEAR DYNAMICS (ORIGINAL FATOLA ROUTINES)

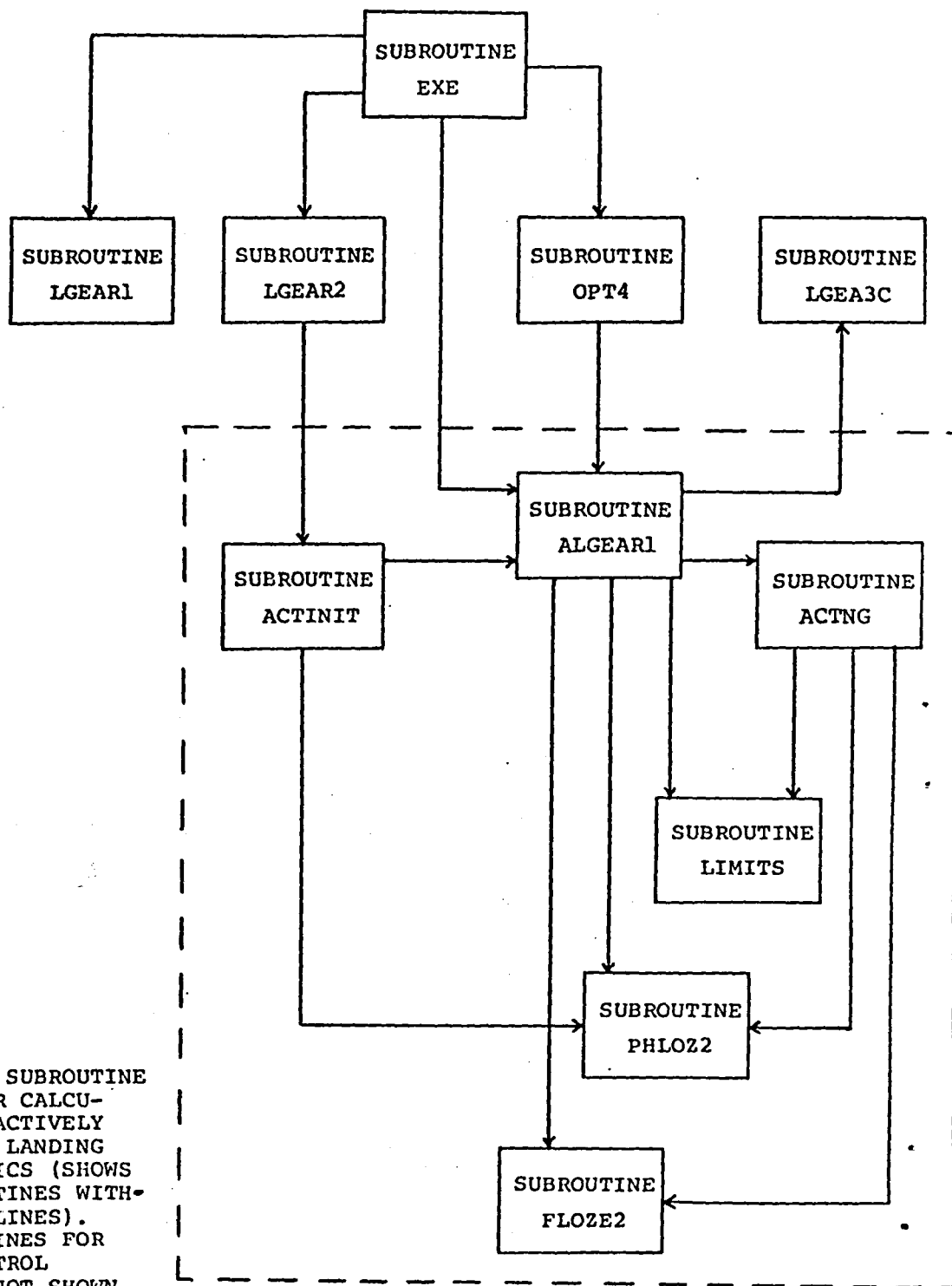


FIGURE 2: SUBROUTINE LINKAGE FOR CALCULATION OF ACTIVELY CONTROLLED LANDING GEAR DYNAMICS (SHOWS NEW SUBROUTINES WITHIN DASHED LINES). INPUT ROUTINES FOR ACTIVE CONTROL VARIABLES NOT SHOWN.

The active control feature has been incorporated in FATOLA with eight new subroutines:

DIRACT--block data input variables

ACTIN--reads active control input variables

ACTINIT--initializes the active control variables

ALGEAR1--performs the main active control calculations

ACTNG--performs the active nose gear calculations

PHLOZ2--calculates initial servovalve hydraulic variables

FLOZE2--calculates servovalve hydraulic variables

LIMITS--sets limits on servovalve power spool motion and displacements

The active control variables have all been blocked into labeled common, including the additional input variables. This common block is named ACTIVE and has been added to FATOLA subroutines EXE, OPT1, LGEA3C, and SDFLGP, as well as the active subroutines ACTINIT, ACTIN, and ALGEAR1. See Tables 1 and 2 for lists of the added input variables and other key program variables.

The active control mode is enabled by setting the landing gear type indicator switch (INDLG) to -3 in the input data set. Tests are performed on this indicator whenever gear calculations are needed, leading to calls to the active subroutines rather than to the LGEAR3 entry point. To perform passive gear calculations, using the active gear subroutines, the switch IMODE(I) must be set to zero

for each gear. Set IMODE(I) to one for each gear for active gear calculations. These tests take place in subroutines EXE and OPT1. Subroutine SDFLGP has been extensively modified to output values of selected active control variables to a listing and to a plot data file.

TABLE 1
INPUT VARIABLES

| VARIABLE NAME | TYPE | UNITS | DESCRIPTION |
|------------------|-------|-----------------------------|--|
| AMUH | | LBF- SEC/FT ² | Dynamic viscosity of hydraulic fluid |
| APINT | ARRAY | FT ² | Area of metering pin as function of strut stroke |
| AREA1 | ARRAY | FT ² | Area of strut hydraulic chamber (piston) |
| AREA2 | ARRAY | FT ² | Area of strut pneumatic chamber (cylinder) |
| AREA3 | ARRAY | FT ² | Area of chamber between piston and cylinder |
| AREMO | ARRAY | FT ² | Area of strut main orifice |
| AREO3 | ARRAY | FT ² | Orifice area of chamber between piston and cylinder |
| BETA | | LBF/IN ² | Bulk modulus of hydraulic fluid |
| BLMU | ARRAY | -- | Coefficient of friction for strut lower bearing |

| | | | |
|--------|-------|---------|---|
| BUMU | ARRAY | — | Coefficient of friction for strut upper bearing |
| CDMOC | ARRAY | — | Strut main orifice compres- sion discharge coefficient |
| CDMOE | ARRAY | — | Strut main orifice exten- sion discharge coefficient |
| CDSV | ARRAY | — | Servovalve orifice discharge coefficient |
| CD3 | ARRAY | — | Discharge coefficient for chamber between piston and cylinder |
| CFFOR | ARRAY | LBF | Coulomb friction force between piston and cylinder bearings |
| DIOTA | | DEGREES | Angle between body x-axis and wing main chord |
| DSV | ARRAY | INCHES | Diameter of servovalve spool |
| EPSILO | ARRAY | LBF | Tolerance about impact control limit force |
| EPSROL | ARRAY | LBF | Tolerance about rollout control limit force |
| EPSSLP | | LBF/SEC | Time rate of change of tolerance about transi- tion control limit force |
| ETASV | | 1/SEC | Damping coefficient in servovalve transfer function |
| FWORK | ARRAY | — | Coefficient for selecting value of impact limit force |

| | | | |
|--------|------------------|--|---|
| GAMAH | ARRAY | LBF/FT ³ | Specific weight of hydraulic fluid |
| GNR | | -- | Constant -- set to 0.0 |
| IMODE | INTEGER ARRAY | -- 0=Off 1=On | Switch to select active control for each gear |
| IRST | | -- 0=None 1=Write 2=Read 3=Read/ Write | Restart indicator |
| KAPT | INTEGER ARRAY | -- 0=No Pin 1=Constant Area 2=Variable Area | Indicator for metering pin |
| LTAB10 | REAL ARRAY | -- | C _N limiting value 0 |
| LTAB80 | REAL ARRAY | -- | C _A limiting value 0 |
| OMRUN | | DEGREES | Runway slope |
| PATM | | LBF/FT ² | Atmospheric pressure |
| PERCNT | ARRAY | -- | Constant |
| PGAHAC | ARRAY | LBF/FT ² | Pressure in high pressure control reservoir |
| PGALAC | ARRAY | LBF/FT ² | Pressure in low pressure control accumulator |
| PGA1I | ARRAY | LBF/FT ² | Strut hydraulic charging pressure |
| PGA2I | ARRAY | LBF/FT ² | Strut pneumatic charging pressure |
| PGA3I | ARRAY | LBF/FT ² | Hydraulic pressure in chamber between strut piston and cylinder |

| | | | |
|--------|-------|-----------------------|--|
| PINM | ARRAY | FT ² | Main gear metering pin area table |
| PINN | ARRAY | FT ² | Nose gear metering pin area table |
| QPUMPS | ARRAY | FT ³ /SEC | Maximum hydraulic pump flow rate |
| RCLSV | ARRAY | INCHES | Radial clearance between spool and sleeve of servovalve |
| RHOH | | SLUGS/FT ³ | Mass density of hydraulic fluid |
| RTAB10 | ARRAY | | C _N ₀ rate of change |
| RTAB80 | ARRAY | | C _A ₀ rate of change |
| STROM | ARRAY | FT | Main gear stroke table for pin area |
| STNON | ARRAY | FT | Nose gear stroke table for pin area |
| TAUF | | SEC | Time constant in strut position feedback loop |
| TC1 | | SEC | Time constant of electronic compensation network |
| TC2 | | SEC | Same definition as TC1 |
| TC3 | | SEC | " " " " |
| TC4 | | SEC | " " " " |
| VOLACI | ARRAY | FT ³ | Total volume of high pressure accumulator |
| VOLANI | ARRAY | FT ³ | Initial volume of charging nitrogen of high pressure accumulator |

| | | | |
|--------|-------|---------------------|---|
| VOL1I | ARRAY | FT ³ | Initial volume of hydraulic fluid in shock strut piston |
| VOL2I | ARRAY | FT ³ | Pneumatic volume of charged strut |
| VOL3I | ARRAY | FT ³ | Volume between strut piston and cylinder |
| WC | | SEC ⁻¹ | Corner frequency in active control servovalve transfer function |
| WC1 | | SEC ⁻¹ | Natural frequency in electronic compensation network |
| WLFOR | ARRAY | LBF | Control limit force for impact |
| WLFORR | | LBF | Control limit force for rollout |
| WSV | | SEC ⁻¹ | Natural frequency in active control servovalve transfer function |
| WSV1 | | IN | Window width of servovalve orifice for high pressure |
| WSV3 | | IN | Window width of servovalve orifice for low pressure |
| XBIAS | ARRAY | IN | Servovalve spool displacement for controlling strut charging pressure |
| XDDMAX | ARRAY | IN/SEC ² | Maximum positive acceleration of servovalve spool |
| XDDMIN | ARRAY | IN/SEC ² | Maximum negative acceleration of servovalve spool |

| | | | |
|--------|-------|-----------|---|
| XKA | ARRAY | AMPS/VOLT | Amplifier gain in active control loop |
| XKF | ARRAY | VOLT/IN | Position feedback gain in strut position control loop |
| XKSV | ARRAY | IN/AMP | Position gain of servovalve in active control loop |
| XLPSV1 | ARRAY | IN | Overlap or underlap between spool and sleeve at null for flow Q1 |
| XLPSV3 | ARRAY | IN | Overlap or underlap between spool and sleeve at null for flow Q3 |
| XSCOM | ARRAY | IN | Commanded (static) position of shock strut |
| XSTHR | | FT | Elec cont. function not used in program Threshold strut stroke for determining takeoff or landing mode |
| XSVDNM | ARRAY | IN/SEC | Maximum negative velocity of servovalve spool. |
| XSVDMX | ARRAY | IN/SEC | Maximum positive velocity of servovalve spool |
| XSVMAX | ARRAY | IN | Maximum positive displacement of servovalve spool |
| XSVMIN | ARRAY | IN | Maximum negative displacement of servovalve spool |
| ZETAC1 | | | Damping coefficient in electronic compensation network |
| ZETAC2 | | | Damping coefficient in electronic compensation network |

TABLE 2
OTHER VARIABLES

| VARIABLE NAME | TYPE | UNITS | INITIAL VALUE | DESCRIPTION |
|------------------|-------|---|------------------|--|
| ACON | | SEC ³ | | $1./W_{sv}^{**2}*W_c$ |
| AIC | ARRAY | | 0.0 | Switch for regulating flow between strut piston and cylinder |
| AP2TO | ARRAY | LBF/FT ² | | Absolute value of pneumatic charging pressure in strut |
| BCON | | SEC ² | | $1./W_{sv}^{**2}+2.*\eta_{sv}/(W_{sv}*W_c)$ |
| BLFORT | ARRAY | LBF | | Normal force between strut piston and cylinder at lower bearing |
| BUFORT | ARRAY | LBF | | Normal force between strut piston and cylinder at upper bearing |
| CCON | | SEC | | $2.*\eta_{sv}/W_{sv}+1./W_c$ |
| CMASNG | | SLUGS | | Effective mass at nose gear root |
| COEFO | ARRAY | FT ² / LBF-SEC ² | | Strut main orifice flow coefficient modified for $d\gamma_h/d\rho$ |
| COEF1 | ARRAY | IN ² / SEC ² LBF | | Coefficient of flow from high-pressure accumulator to strut piston |
| COEF3 | ARRAY | IN ² / SEC ² LBF | | Coefficient of flow from strut piston to low-pressure reservoir |
| COPA | | -- | | $\cos (\phi+\alpha)$ |

| | | | | |
|--------|-------|------------------------------|------|--|
| CSV1 | ARRAY | $\frac{FT^3}{SEC\sqrt{LBF}}$ | | Servovalve orifice coefficient from accumulator to strut piston |
| CSV3 | ARRAY | $\frac{FT^3}{SEC\sqrt{LBF}}$ | | Servovalve orifice coefficient from strut piston to reservoir |
| DCON | | | 1.0 | |
| DELTX | ARRAY | INCHES | 0.0 | Difference between strut stroke and command stroke |
| DELTX1 | ARRAY | VOLTS | | Product of DELTX and position feedback gain in piston control loop |
| DF | ARRAY | LBF | 0.0 | Variation of gear-airframe interface force about control force |
| DLTX1D | ARRAY | VOLTS/SEC | 1.0 | Time rate of change of DELTX1 |
| DMD1 | ARRAY | | | Restart data block |
| DMTANH | ARRAY | — | | Function of hyperbolic tangent |
| DP1 | ARRAY | $\frac{LBF-SEC}{FT^2}$ | 0.0 | Time rate of change of hydraulic pressure in strut piston |
| DSTOP | | FT | .004 | Stopping distance for strut extension stop |
| DWFORT | | LBF | | Earth axes force at aircraft center of gravity |
| ENCG | | FT-LBF | | Kinetic energy at aircraft center of gravity |
| ENUP | ARRAY | FT-LBF | 0.0 | Kinetic energy of aircraft at landing gear root |
| FFORT | ARRAY | LBF | 0.0 | Strut axial friction force |

| | | | | |
|--------|-------|--------|-----|---|
| FONHST | ARRAY | LBF | 0.0 | Ground force normal to strut axis |
| FORCHT | ARRAY | LBF | 0.0 | Charging force in fully extended strut |
| FORSST | ARRAY | LBF | 0.0 | Strut force along strut axis |
| FSTOP | ARRAY | LBF | 0.0 | Strut extension stop force |
| FSTOPK | | LBF/FT | 0.0 | Strut extension stop spring rate |
| HMM | ARRAY | -- | 0.0 | Active control operation indicator 0=off, 1=on |
| IA1 | ARRAY | -- | 0 | Indicator for defining the return of notch-network compensation current to zero when gear is fully-extended during rebound; =0, non-zero current; =1, zero current |
| IA2 | ARRAY | -- | 0 | Same as IA1 |
| IA3 | ARRAY | -- | 0 | Same as IA1 |
| IA4 | ARRAY | -- | 0 | Same as IA1 |
| IA6 | ARRAY | -- | 0 | Indicator for defining the return of first lead-lag network compensation current to zero when gear is fully-extended during rebound: =0, non-zero current; =1, zero current |
| IA7 | ARRAY | -- | 0 | Same as IA6 |

| | | | | |
|--------|------------------|---|---|--|
| IA9 | ARRAY | — | 0 | Indicator for defining the return of second lead-lag network compensation current to zero when gear is fully-extended during rebound: =0, non-zero current; =1, zero current |
| IA10 | ARRAY | — | 0 | Same as IA9 |
| ICOSV | INTEGER ARRAY | — | 0 | Servo valve control indicator for return to null position |
| ICU | ARRAY | — | 0 | Indicator for defining when the fluid volume in the strut piston has returned to the fully-extended strut value during rebound: =0, has not returned; =1, returned |
| IFR | INTEGER ARRAY | — | 0 | Indicator for selection of tanh function |
| IFSTOP | INTEGER ARRAY | — | 0 | Indicator for strut extension stopping force |
| IGE | ARRAY | — | 0 | Indicator for defining fully-extended gear during rebound: =0, gear compressed; =1, gear fully-extended |
| IGO | ARRAY | — | 0 | Indicator for defining accelerated return of servo-controller parameters to fully-extended gear values: =0, inactive; =1, active |
| IIXSVH | INTEGER ARRAY | — | 0 | Indicator for servo valve returning to null from high pressure |

| | | | | |
|---------|------------------|----|---|---|
| IIXSVL | INTEGER ARRAY | -- | 0 | Indicator for servovalve returning to null from low pressure . |
| ITRIP | ARRAY | -- | 0 | Indicator for defining the return of gear and servo- valve parameters to fully- extended gear values: =0, inactive; =1, active |
| INDEACT | INTEGER ARRAY | -- | 0 | Indicator for control regime 0=off, 1=impact, 2=rollout |
| INITSW | INTEGER | -- | 1 | Switch indicating first pass through program 1=first pass |
| IOPCO | INTEGER ARRAY | -- | 1 | Indicator for control operation during return to null position |
| IPASS | INTEGER ARRAY | -- | 0 | " |
| IPSTOP | INTEGER | -- | 0 | " |
| IQCU | ARRAY | -- | 0 | Indicator for defining when the cumulative fluid volume supplied by the control system has returned to zero when the strut is fully-extended during rebound: =0, non-zero; =1, zero |
| ISSET | INTEGER ARRAY | -- | 1 | See IOPCO |
| ISTROK | INTEGER ARRAY | -- | 0 | Indicator for strut stroke initialization |

| | | | | |
|--------|------------------|----------------------|-----|--|
| IXS | ARRAY | -- | 0 | Indicator defining the condition of servovalve spool displacement equal to bias value for the fully-extended strut during rebound: =0, unequal =1, equal |
| IXSVH | INTEGER ARRAY | -- | 0 | Indicator for control operation during return to null position |
| IXSVL | INTEGER ARRAY | -- | 0 | " |
| NAC | INTEGER ARRAY | -- | 0 | Fluid flow indicator 0=off; 1=from strut; 2=to strut |
| NITER | INTEGER | -- | | Number of iterations required to converge to solution |
| PGA1T | ARRAY | LBF/FT ² | | Gauge pressure of hydraulic fluid in strut piston |
| PGA2T | ARRAY | LBF/FT ² | | Gauge pressure of nitrogen in strut cylinder |
| PGA3T | ARRAY | LBF/FT ² | | Gauge pressure of hydraulic fluid in volume 3 |
| PGA1T1 | ARRAY | LBF/FT ² | | See PGA1T |
| PR | ARRAY | LBF/IN ² | | Servovalve return pressure |
| PS | ARRAY | LBF/IN ² | | Servovalve supply pressure |
| P1 | ARRAY | LBF/IN ² | | Strut piston pressure |
| QC | ARRAY | IN ³ /SEC | 0.0 | Flow rate to the load |
| QO | ARRAY | FT ³ /SEC | 0.0 | Flow rate through strut main orifice |

| | | | | |
|---------|-------|----------------------|-------|---|
| QS1 | ARRAY | IN ³ /SEC | | Flow rate from accumulator to strut piston |
| QS3 | ARRAY | IN ³ /SEC | | Flow rate from strut piston to reservoir |
| QSV | ARRAY | FT ³ /SEC | | Fluid flow rate through servovalve, positive from supply to strut |
| QSVCU | ARRAY | FT ³ | 0.0 | Cumulative volume of fluid added to or removed from strut |
| QSVN | ARRAY | FT ³ /SEC | 0.0 | Volume expansion or compression rate of nitrogen in high pressure accumulator |
| QSV1 | ARRAY | | | See QS1 |
| QSV3 | ARRAY | | | See QS3 |
| QTOLER | | IN ³ /SEC | .0001 | Tolerance allowed in calculating flow rates |
| REDSLP | ARRAY | LBF/SEC | 1.E5 | Slope for reducing control limit force from impact value to rollout value |
| RESA | ARRAY | | 0.0 | Control activation switch |
| SA | ARRAY | | 0.0 | " |
| SBFOT | | | 0.0 | Strut elastic bending force |
| SIPA | | | | $\sin(\phi + \alpha)$ |
| UNSPRNG | | SLUGS | | Mass of strut piston and gear components attached |
| VCUM | ARRAY | FT ³ | 0.0 | Cumulative volume of fluid flowed from piston to cylinder |
| VELDEC | ARRAY | FT/SEC | 0.0 | Velocity for initiating transition from impact limit force to rollout value |

| | | | | |
|------------|-------|-----------------|-----|--|
| VMAS | ARRAY | SLUGS | | Vehicle mass |
| VOLAHT | ARRAY | FT ³ | | Volume of hydraulic fluid in high pressure accumu- lator |
| VOLANT | ARRAY | FT ³ | | Volume of nitrogen in high pressure accumulator |
| VOL1T | ARRAY | FT ³ | | Volume of hydraulic fluid in strut piston |
| VOL2T | ARRAY | FT ³ | | Pneumatic volume in strut cylinder |
| VOL3T | ARRAY | FT ³ | | Volume between piston and cylinder |
| WFORT | ARRAY | LBF | 0.0 | Gear-airframe interface force |
| XMA | ARRAY | AMPERES | | Input signal to electronic compensation networks |
| XMA1 ... 4 | ARRAY | " | 0.0 | modification of input signal to the notch and lead-lag networks |
| XMA5 | ARRAY | " | | " |
| XMA6,7 | ARRAY | " | 0.0 | " |
| XMA8 | ARRAY | AMPERES | | " |
| XMA9,10 | ARRAY | AMPERES | | " |
| XMA11 | ARRAY | AMPERES | | Output signal from electronic compensation networks to servovalve |
| XMU | | CENTIPOISE | | Viscosity of hydraulic fluid |
| XSTOT | ARRAY | FT | | Instantaneous stroke required to dissipate remaining kinetic energy at simultaneous force level |

| | | | | |
|--------|-------|---------------------|-----|--|
| XSV | ARRAY | INCHES | | Servo valve spool displacement |
| XSVDD | ARRAY | IN/SEC ² | 0.0 | Acceleration of servo valve spool |
| XSVDDD | ARRAY | IN/SEC ³ | 0.0 | Jerk on servo valve spool |
| XSVDOT | ARRAY | IN/SEC | 0.0 | Velocity of servo valve spool |
| XVALVE | ARRAY | IN | | Analytically controlled servo valve spool displacement |
| ZDANT | | FT/SEC | | Velocity of nose gear root |
| ZSSC | ARRAY | FT | | Allowable strut stroke for activating control |

1.3 Stroke Dependent Metering Pin Area - Following implementation of the active control landing gear, an additional capability was required to model the F4 aircraft landing gears. The metering pin area was allowed to have either a constant value (the original design) or to be functionally dependent on the strut stroke. No new subroutines were required and modifications were limited to the active input routines, ACTINIT, and ALGEAR. Input variables are presented and defined in Table 1.

1.4 Variable Aerodynamic Coefficients - The use of staging to introduce changes in the aerodynamic coefficients C_{A_0} and C_{N_0} was a cumbersome process and led to discontinuities in program results. New program variables were introduced to allow the specification of a rate of change for each of these coefficients providing a much smoother transition. This feature required two new subroutines:

AERO4--initializes and updates coefficients

AEROIN--reads rates and limiting values

the new variables have been stored in a new common block, AEROCO.

Input variables are included in Table 1.

1.5 Restart Capability

Recent attempts to model the F4 gear have shown that at the time of touchdown extremely large changes occur in several variables resulting in a small integration step size. The final result is a large amount of computer time. The problem was compounded by having to rerun the entire simulation whenever a program failure was experienced.

The introduction of a restart capability eliminated the problem of total reruns. Whenever any data is staged into the program, a disk file is generated which contains all the information required to restart the program from that point in time. Thus when program failure occurs, it is only necessary to rerun from the most recent stage. This technique has been found extremely useful in the analysis of the effects of various runway conditions. Rather than running the entire landing simulation for each runway to be considered, it is now only necessary to begin the simulation at that point in time where the runway bump is encountered.

No new subroutines were required, but many existing routines were modified by relocating their local variables into labeled common blocks. With all program variables in common blocks declared in the main program, their storage locations are contiguous and a single read or write transfers the entire block of restart data.

New program variables required for the restart option are included in Tables 1 and 2.

2. CHRONOLOGICAL HISTORY OF PROGRAM MODIFICATIONS

Numerous program changes have been made to the FATOLA program since the last formal documentation (Reference 3) in 1975. A total of 52 UPDATE corrections sets have been applied since that time. Numerous modifications were performed by Structures and Dynamics Division (SDD) personnel and were reported in References (1) and (2). The remaining coding changes were performed by Computer Sciences Corporation (CSC) personnel. For the sake of completeness, all SDD and CSC modifications are included in this history.

2.1 Structures and Dynamics Division Modifications - Beginning in November 1975 and ending in November 1978, several improvements were made to the FATOLA program to enhance its simulation capabilities. The net effect of the three correction sets C82477, C11778, and SDDMODS is reported in References (1) and (2) with the following exceptions.

In program TOLA, the call to FTNBIN was commented out when the Integrated Computer Operating System (ICOPS) was replaced with the Network Operating System (NOS).

Dummy space was not added to the end of the DIRCOM common block in subroutine TFFS9 as reported in Reference (2). This addition is not necessary for proper program execution.

A call to subroutine FLEX5 was inserted in subroutine OPT1 immediately before the call to FLEX6. The reason for this addition is not apparent since it is a do-nothing entry point.

The function ATAN2 was declared EXTERNAL immediately following the SUBROUTINE LGEA3C statement. Again, there is no apparent reason for this addition.

In an attempt to prevent over-extension or compression of the struts, coding changes were made to two subroutines. In subroutine LGDET, the FORMAT statements labeled 49 and 53 were replaced by the following statements

```
49 FORMAT(58X,4H-ES(,I1,19H) EXCEEDED IN LGDET)
53 FORMAT(58X,4H ES(,I1,20H) EXCEEDED IN LGDET /
```

The following lines of code were inserted before statement number 50,

```
Y(J)=-.5*ES(I)
Y(JJ)=-1.0E-10
P(JJ)=-1.0E-10
```

and the statements between and including statement numbers 51 and 21 were replaced with the following.

```
51 IF(SD2(1,I).LT.0.)GO TO 30
   IF(SD1(1,I).LT.0.)SD1(1,I)=0.
   GO TO 55
30 SD2(1,I)=0.
   SD1(1,I)=0.
```

In subroutine LGEAR1, the formats were similarly replaced as follows.

```
49 FORMAT(58X,4H-ES(,I1,20H) EXCEEDED IN LGEAR1/  
(53 FORMAT(58X,4H ES(,I1,20H) EXCEEDED IN LGEAR1/
```

the following code was inserted before statement number 50

```
S(1,I)=-0.5*ES(I)  
SD1(1,I)=-1.0E-10  
SD2(1,I)=-1.0E-10
```

and the two lines beginning with statement number 51 were replaced with the following.

```
51 IF(SD2(1,I).LT.0.)GO TO 30  
   IF(SD1(1,I).LT.0.)SD1(1,I)=0.  
   GO TO 55  
30 SD2(1,I)=0.
```

Additionally, the following code was inserted immediately before statement number 20.

C RE-CHECK SHOCK STRUT FORCE FOR RE-CONDITIONED FULLY EXTENDED STATE
IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(SF(I)))SF(I)=-FT(I)

2.2 Computer Sciences Corporation Modifications - In early 1978, CSC was requested to support a major modification to the FATOLA program, by introducing the capability of an active control landing gear. A mathematical model of an active control landing gear

(ACOLAG) had been developed and programmed (Reference 7), however this program did not include airframe elastic effects. The goal of the program modifications was to incorporate the series-hydraulic active control logic and equations for all landing gear oleo-pneumatic shock struts into the FATOLA computer program.

The correction idents ACOBLK, CSCMODS, ACTINIT, ALGEAR, PHLOZ2, FLOZE2, and LIMITS contained the code required for a first cut at implementing the active control. A comdeck, ACOBLK, was inserted following the deck TOLA to establish a common block containing the active gear variables. The original contents of this common block are shown below.

```
C***** ACTIVE VARIABLES *****
COMMON /ACTIVE/ AMUH ,ACON ,BCON ,CCON ,DCON
1,APINT(5) ,AP2TO(5) ,AREA1(5) ,AREA2(5) ,AREA3(5)
2,AREMO(5) ,AREO3(5) ,BLFORT(5) ,BLMU(5) ,BUFORT(5) ,BUMU(5)
3,BETA ,CDMOC(5) ,CDMOE(5) ,CDSV(5) ,CFFOR(5) ,CD3(5)
4,COEF ,COEFO(5) ,COEF3(5) ,COPA
5,CSV1(5) ,CSV3(5) ,DELT ,DELTX(5) ,DELTX1(5) ,DLTX1D(5)
6,DF(5) ,DP1(5) ,DSV(5) ,DSTOP ,DIOTA ,DMTANH(5)
7,ENUP(5) ,EPSILO(5) ,EPSROL(5) ,EPSSLP ,ETASV ,FSTOPK
8,FFORT(5) ,FOAHST(5) ,FONHST(5) ,FORCHT(5) ,FORSST(5) ,FSTOP(5)
9,GAMAH(5) ,GAMAN ,GNR ,HMM(5) ,ICOSV(5) ,IDEACT
*,IFRI(5) ,IIXSVH(5) ,IIXSVL(5) ,IOPCO(5) ,IPASS(5) ,ISET(5)
1,IXSVH(5) ,IXSVL(5) ,IFSTOP(5) ,ISTROK(5) ,KAPT(5)
2,NAC(5) ,NITER ,OMRUN ,PATM
3,PERCNT(5) ,PGAHAC(5) ,PGALAC(5) ,PGA1I(5) ,PGA1T(5)
4,PGA2I(5) ,PGA2T(5) ,PGA3I(5) ,PGA3T(5)
5,PR(5) ,PS(5) ,P1(5) ,QC(5) ,QD(5) ,QPUMPS(5)
6,QSV(5) ,QSVCU(5) ,QSVN(5) ,QSV1(5) ,QSV3(5) ,QS1(5)
7,QS3(5) ,QTOLER ,RCLSV(5) ,REDSLP(5) ,RHOH ,SBFOT
8,SIPA ,SA(5) ,RESA(5) ,TAUF ,TC1 ,TC2
COMMON /ACTIVE/ TC3 ,TC4 ,VCUM(5) ,VELDEC ,VOL1I(5)
1,VOL1T(5) ,VOL2I(5) ,VOL2T(5) ,VOL3I(5) ,VOL3T(5) ,WC
2,WC1 ,WFORT(5) ,WLFOR(5) ,WLFORR ,XBIAS(5) ,XVALVE(5)
3,XKA(5) ,XKF(5) ,XKSV(5) ,XLPSV1(5) ,XLPSV3(5) ,XSTHR
4,XMA(5) ,XMA1(5) ,XMA2(5) ,XMA3(5) ,XMA4(5) ,XMA5(5)
5,XMA6(5) ,XMA7(5) ,XMA8(5) ,XMA9(5) ,XMA10(5) ,XMA11(5)
6,XMU ,XSCOM(5) ,XSTOT(5) ,XSV(5) ,XSVDOT(5) ,XSVDD(5)
```

```

7,XSVDDD(5),XSVDNM(5),XSVDHX(5),XSVMAX(5),XSVMIN(5),XDDMAX(5)
8,XDDMIN(5),VOLACI(5),VOLAH(5),VOLANI(5),VOLANT(5)
9,WSV,WSV1,WSV3,ZETAC1,ZETAC2,ZSSC
C *****

```

New decks ACTINIT, for initializing the active gear variables, and ALGEAR, PHLOZ2, FLOZE2, and LIMITS, containing the active gear logic, were added after the LGEA3C deck. These new decks are provided below.

SUBROUTINE ACTINIT

```

C
C ***** FATOLA VARIABLES *****
COMMON/DIRCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,
CDCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FXB7P,
CDUM4(3),FYB7P(4),FZB7P,DM5(17),GXB7F,DM6(8),GZB7F,
CDM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,
CINDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),
CQI77R(2),QI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),
CXG77F(2),XG77F1(12),YG77F(2),YG77F1(12),
CZG77F(2),ZG77F1(2),DUM13(52),
CNSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEG,RGR,
CNTIRES(5),RZERO(5),W(5),DELTAM(5),MOMENT(5),
CRF(5),VZ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),
CA2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),
CCC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),RLT,NDELTA,
CES(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)
COMMON/DIRCOM/
CSQ11(2),SD12(2),SD13(2),SD14(2),SD15(2),
CS1(2),SS2(2),S3(2),S4(2),S5(2),
CS2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),
CS2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),
CS21(2),S22(2),S23(2),S24(2),S25(2),
COMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),
COMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),
CAI(5),BI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,
CDELTA1,DDELTA2,DDELTA3,DDELTA4,DDELTA5,ISTAGE,
CPRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,
CDM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLX,
*NMODE,DM18(40),SXMOD(100),SYMOD(100),SZMOD(100),DM19(1686),
*GQD2(20),DDM20(20)
*,DDM21(20),SLEN1(5),SLEN2(5),DUM15(13),INDNWS,DDM22(10),ETADES,
*DDM23(5),AH(5),PH(5),DDM24(30)
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
C
*CALL ACOBLK

```

DIMENSION CD3(5), SD1(2,5)
EQUIVALENCE(SD11(1),SD1(1,1)), (DM5(16),GREFF)

DATA STATEMENTS TO SET VALUES OF INPUT VARIABLES

DATA AMUH/0.00018/, BETA/14400000./
DATA BLMU,BUMU/10*0.15/
DATA AREA1,AREA2,AREA3/5*0.21139,5*0.32167,5*0.00698/
DATA AREMO,AREO3,APINT/5*0.0066,5*0.19635,5*0.0/
DATA CDMOC,CDMOE,CD3/5*0.9,5*0.14197,5*0.6/, CFFOR/5*50.0/
DATA CDSV,DELT,DIOTA/5*0.62,0.0001,0.0/
DATA EPSILO,EPSSLP,EPSROL,ETASV/5*200.0,0.0,5*2000.0,0.436/
DATA GAMAN,GAMAH,GNR/1.06,5*52.36,0.0/
DATA ICOSV,IFSTOP/10*0/
DATA IDEACT,II,KAPT,OMRUN,PERCNT/7*0,-0.07575,5*1.66115/
DATA PATH,PGAHAC,PGALAC/2116.8,5*432000.0,5*0.0/
DATA PGA1I,PGA2I,PGA3I/15*40320.0/
DATA TAUF,TC1,TC2,TC3,TC4/0.1,0.281,0.141,0.001,0.0001/
DATA VOL1I,VOL2I,VOL3I/5*0.36379,5*0.47164,5*0.0/
DATA RHQH,WLFOR,WLFORR/1.626,5*500.0,0.0/
DATA DSV,RCLSV,WSV1,WSV3/5*1.1875,5*0.0000475,2*3.480/
DATA WC,WC1,WSV,XKSV,XSTHR/1263.0,251.3,655.5,5*0.00250,0.020833/
DATA XKA,XKF,XLPSV1,XLPSV3/5*0.04,5*60.0,5*0.0,5*0.0/
DATA XBIAS,XSCOM,XSVDMN,XSVDMX/5*-0.00014,5*1.206,5*-30.1,5*30.1/
DATA XDDMAX,XDDMIN/5*1.0E20,5*-1.0E20/
DATA XSVMAX,XSVMIN,ZETAC1,ZETAC2/5*0.100,5*-0.100,5.1,0.1/
DATA VOLACI,VOLANI,QPUMPS/5*1.33333,5*0.26667,5*0.02005/

C*****

IF (TIME .GT. DELT) RETURN

ACON=1./(WSV*WSV*WC)
BCON=1./(WSV*WSV)+2*ETASV/(WSV*WC)
CCON=2.*ETASV/WSV+1./WC
DCON=1.
XMU=AMUH/.0000209
QTOLER=.0001
SBFOT=0.0
VELDEC=0.
OMRUN=OMRUN*0.01745329
II = 0

DO 100 I=1,NSTRUT
REDSLP(I) = 100000.0
ISTROK(I)=0
NAC(I)=0
VOLANT(I) = VOLANI(I)
VOL1T(I)=VOL1I(I)
VOL2T(I)=VOL2I(I)
VOL3T(I)=VOL3I(I)
QSVCU(I)=0.0
DF(I)=0.

```

DELTX(I)=0.
DELTX1(I)=DELTX(I)*XKF(I)
DLTX1D(I)=0.
XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
XMA5(I)=XMA(I)
XMA8(I)=XMA5(I)
XMA11(I)=XMA8(I)
XMA1(I)=0.
XMA2(I)=0.
XMA3(I)=0.
XMA4(I)=0.
XMA6(I)=0.
XMA7(I)=0.
XMA9(I)=0.
XMA10(I)=0.
XSVDD(I)=0.
XSVDDD(I)=0.
XSV(I)=XKSV(I)*XMA11(I)+XBIAS(I)
XSVDDOT(I)=0.0
DP1(I)=0.
COEF3(I)=CD3(I)*SQRT(2.*GREFF/GAMAH(I))

```

NOTE: SUBROUTINE 'PHLOZ2' COMPUTES INITIAL PRESSURES AND FLOWS IN
UNITS OF INCHES.

```

COEF=CDSV(I)*SQRT(2.*GREFF/GAMAH(I))*144.
CSV1(I)=COEF*WSV1
CSV3(I)=COEF*WSV3
PS(I)=PGAHAC(I)/144.
PR(I)=PGALAC(I)/144.
QC(I)=0.

```

```

CALL PHLOZ2(PS(I),PR(I),XSV(I),QC(I),XLPSV1(I),XLPSV3(I),RCLSV,DSV
& ,CSV1(I),CSV3(I),XMU,QTOLR,NITER,P1(I),QS1(I),QS3(I))

```

```

PGA1I(I)=P1(I)*144.
PGA2I(I)=PGA1I(I)
PGA3I(I)=PGA2I(I)
QSV1(I)=QS1(I)/1728.
QSV3(I)=QS3(I)/1728.

```

```

PGA1T(I)=PGA1I(I)
PGA2T(I)=PGA2I(I)
PGA3T(I)=PGA3I(I)
QO(I)=0.
QSV(I)=QSV1(I)-QSV3(I)
VCUM(I)=0.

```

```

AP2TO(I)=PGA2I(I)+PATM
FFORT(I) = 0.0
FORSST(I)=0.0
WFORT(I)=0.0
FONHST(I)=0.0
FORCHT(I)=0.0
XVALVE(I)=XSV(I)

```

100 CONTINUE

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OF POOR QUALITY

C
CALL SETUP
RETURN
END
*IDENT ALGEAR
*INSERT ACTINIT.145
*DECK ALGEAR
SUBROUTINE ALGEAR1

C

***** FATOLA VARIABLES *****
COMMON/DIRCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,
CDCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FXB7P,
CDUM4(3),FYB7P(4),FZB7P,DM5(17),GXB7F,DM6(8),GZB7F,
CDM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,
CINDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),
COI77R(2),OI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),
CXG77F(2),XG77F1(12),YG77F(2),YG77F1(12),
CZG77F(2),ZG77F1(2),DUM13(52),
CNSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEG,RGR,
CNTIRES(5),RZERO(5),W(5),DELTAM(5),MOMENT(5),
CRF(5),VZ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),
CA2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),
CCC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),RLT,NDELTA,
CFS(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)
COMMON/DIRCOM/
CSD11(2),SD12(2),SD13(2),SD14(2),SD15(2),
CS1(2),SS2(2),S3(2),S4(2),S5(2),
CS2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),
CS2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),
CS21(2),S22(2),S23(2),S24(2),S25(2),
COMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),
COMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),
CAI(5),BI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,
CDELTA1,DDELTA2,DDELTA3,DDELTA4,DDELTA5,ISTAGE,
CPRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,
CDM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLX,
*NMDD,DM18(40),SXMOD(100),SYMCD(100),SZMOD(100),DM19(1686),
*GQD2(20),DDM20(20)
*,DDM21(20),SLEN1(5),SLEN2(5),DUM15(13),INDNWS,DDM22(10),ETADES,
*DDM23(5),AH(5),PH(5),DDM24(30)
REAL MASS,MOMENT,MASS2,MUS,NTIRES,MR
DIMENSION DLGAUT(14),DLGDE(47),DLG(7),DLGE(299)
COMMON/LGAUTS/ARG11,ARG13,ARG31,ARG33,AMA(5),VAXLE(5)
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
COMMON/LG/FXM,FYM,FZM,LM,MM,NM,EPSLO2
COMMON/LGE/A11(5),A13(5),A31(5),A33(5),RRCGX(5),
CRL(3,3),RI(3,3,5),RAX(5),RAY(5),RAZ(5),TMP(3),ZZERO(5),
CXR(5),YR(5),EPSLON(5),PA(5),FDELTA(5),
CFTRZ(5),RDX(5),ROY(5),RDZ(5),RDYG(5),RDY(5),RDZG(5),
CVTX(5),VTY(5),VTZ(5),GZ(5),VGPT(5),FTRX(5),FTRY(5),
CDX(5),DY(5),DZ(5),FT(5),FDX(5),FDY(5),FF(5),AA(5),C2(5),
CSR(5),SF(5),PSKD(5),MUVP(5),MTRX(5),MTRY(5),

```

CMTRZ(5),MA(5),RG11,RG13,RG31,RG33,IPRT,
CMTX,MTY,MTZ,SFTRX,SFTRY,SFTRZ,FTRA,
CFTRB,FTRC,SMTRX,SMTRY,SMTRZ
COMMON/FLXOP/GFORC2(100),GFORC3(100),GFORC4(100),BH1(308)
EQUIVALENCE (DLGAUT(1),ARG11),(DLGDE(1),LA(1)),
*(DLG(1),FXM),(DLGE(1),A11(1))
COMMON/TABSRC/DUMM1(103),LOC(7)
COMMON/HTCOM/HT,HT1,HT2

C
*CALL ACOBLK
C
REAL MUVP,MTRX,MTRY,MTRZ,MA,
CMTX,MTY,MTZ,LM,MM,NM
DIMENSION DELTA(5),DDELTA(5),P(5),
C      SD2(2,5),SD1(2,5),S(2,5),S2D2(2,5),S2D1(2,5),S2(2,5),
COMETD1(2,5),OMET(2,5)
EQUIVALENCE (P(1),PRES(1))
EQUIVALENCE (DELTA(1),DELTA1),(DDELTA(1),DDELT1)
EQUIVALENCE
C(SD21(1),SD2(1,1)),(SD11(1),SD1(1,1)),(S1(1),S(1,1)),
C(S2D21(1),S2D2(1,1)),(S2D11(1),S2D1(1,1)),(S21(1),S2(1,1)),
C(OMTD11(1),OMETD1(1,1)),(OMT1(1),OMET(1,1))

C
C*****ACTIVE CODE*****
EQUIVALENCE (DM5(16),GREFF)
DATA IFRI/5*0/,FSTOPK/0.0/,FSTOP/5*0.0/,DSTOP/0.004/
DATA SA,RESA,HMM,IXSVL,IXSVH,IIXSVL,IIXSVH,IPASS
1      /15*0.,25*0/
DATA ENUP/5*0.0/
DATA ISET,IDPCQ,QSVN/10*1,5*0.0/
Q1(T1,T2)=SIGN(1.,(T1-T2))*SQRT(ABS(T1-T2))
C*****
C
DATA RADDEG,DEGRAD/57.2957795,.01745329/

C
C MAIN COMPUTATIONAL AREA
C RL MATRIX ELEMENTS
C RL(1,1)=DCL1*RG11+DCL3*RG13
RL(1,1)=DCL1*RG11+DCL3*RG13
RL(1,2)=DCL2
RL(1,3)=DCL1*RG31+DCL3*RG33
RL(2,1)=DCM1*RG11+DCM3*RG13
RL(2,2)=DCM2
RL(2,3)=DCM1*RG31+DCM3*RG33
RL(3,1)=DCN1*RG11+DCN3*RG13
RL(3,2)=DCN2
RL(3,3)=DCN1*RG31+DCN3*RG33
CALL LGEA3C

C
C***** ACTIVE CODE *****
C START ACTIVE GEAR CALCULATIONS
COPA = COS((THTPD+DIOTA)*DEGRAD)
SIPA = SIN((THTPD+DIOTA)*DEGRAD)
IF(TIME .GT. DELT)GO TO 18

```



```

15 WRITE(6,1013)
1013 FORMAT (1H020H ACTIVE CONTROL GEAR)
18 IF(IDEACT.EQ.1) GO TO 56
19 IF(IDEACT.EQ.2) GO TO 80
   IF(HMM(I) .EQ. 0.) GO TO 90
   IF(OMRUN .GT. 0.0)GO TO 25
   IF(ZG77F1(1) .GE. VELDEC) GO TO 90
   GO TO 40
25 IF(ZG77F1(1) .GE. VELDEC+XG77F1(1)*TAN(OMPUN))GO TO 90
40 WRITE(6,1014)TIME
1014 FORMAT (1H036H REDUCE CONTROL LIMIT FORCE AT TIME=,E16.8)
   IDEACT=1
56 WLFOR(I)=WLFOR(I)-REDSLP(I)*DELT
   EPSILO(I)=EPSILO(I)+EPSSLP*DELT
   IF(WLFOR(I) .GT. WLFORR) GO TO 90
60 WRITE(6,1015)TIME
1015 FORMAT (1H027H CONTROL AT WLFORR AT TIME=,E16.8)
   IDEACT=2
80 WLFOR(I)=WLFORR
   EPSILO(I)=EPSROL(I)
90 CONTINUE
C*****
DO 65 I=1,NSTRUT
WFOR(I) = (SQRT(FXB7P*FXB7P+FYB7P*FYB7P+FZB7P*FZB7P))/(NSTRUT-1)
E      +(FORST(I)*COPA)
C***** ACTIVE CODE *****
IF(KAPT(I).NE.0)GO TO 210
APINT(I)=0.0
210 CONTINUE
IF(PGA1T(I) .LE. -1600.0) PGA1T(I) = -1600.0
VOL1T(I)=VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
VOL3T(I)=VOL3I(I)+APEA3(I)*S(1,I)
VOL2T(I)=VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)+(VOL3T(I)
X -VOL3I(I))-VCUM(I)
PGA2T(I)=AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMAN)-PATM
IF(SD1(1,I) .EQ. 0.0)GO TO 104
PGA3T(I)=((COEF3(I)*AREO3(I))*2*PGA2T(I)-SD1(1,I)/ABS(SD1(1,I))
X *(SD1(1,I)*AREA3(I))*2)
S/((COEF3(I)*AREO3(I))*2)
GO TO 105
104 PGA3T(I)=PGA2T(I)
105 IF(PGA1T(I) .GE. PGA2T(I))GO TO 106
GO TO 107
106 GAMAH(I)=RHOH*GREFF*(1.0+(PGA1T(I)*3.04E-08)-
* (PGA1T(I)**2*2.72E-15))
GO TO 108
107 GAMAH(I)=RHOH*GREFF*(1.0+(PGA2T(I)*3.04E-08)-
* (PGA2T(I)**2*2.72E-15))
108 IF(PGA1T(I) .GE. PGA2T(I))COEFO(I)=
* CDMOC(I)*SORT(ABS(2.*GREFF/GAMAH(I)))
IF(PGA2T(I) .GT. PGA1T(I))COEFO(I)=
* CDMOE(I)*SORT(ABS(2.*GREFF/GAMAH(I)))

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      IF(SD1(1,I).LE.0.0) GO TO 109
      QD(I)=CDEFD(I)*(AREMD(I)-APINT(I))*Q1(PGA1T(I),PGA2T(I))
109  IF(PGA2T(I) .LE. -1600.0)PGA2T(I)=-1600.0
      IF(PGA3T(I) .LE. -1600.0)PGA3T(I)=-1600.0
100  IF(DELTA(I) .LE. 0.0 .AND. TIME .GT. DELT)GO TO 101
      GO TO 110
101  FFORT(I)=0.0
      GO TO 140
110  CONTINUE
C    COMPUTE STRUT AXIAL BINDING FRICTION FORCE
      BLFORT(I)=FONHST(I)*((SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))+1.0)
      BUFORT(I)=FONHST(I)*(SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))
      FFORT(I)=BUMU(I)*ABS(BUFORT(I))+BLMU(I)*ABS(BLFORT(I))
140  CONTINUE
C    COMPUTE SHOCK STRUT CHARGING FORCE
      IF(S(1,I) .GT. 0.0)GO TO 142
141  FORCHT(I)=PGA1T(I)*AREA1(I)+PGA2T(I)*(AREA2(I)-AREA1(I))-PGA3T(I)
      X *AREA3(I)+FFORT(I)+CFFOP(I)
C    COMPUTE NORMAL AND AXIAL HUB TO SHOCK STRUT FORCES AT HUB
142  FONHST(I)=SQRT(FDX(I)**2+FDY(I)**2)-MASS(I)*GREFF*SIPA+SBFOT
      IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 150
      GO TO 801
150  CONTINUE
      FORSST(I)=FT(I)
      SD1(1,I)=0.0
      IF(I.EQ.1) GO TO 450
      ISTROK(I)=1
C***** 289 CHANGED TO 295 FOR DEBUGGING PURPOSES:
      GO TO 295
C    COMPRESSION VELOCITY OF SHOCK STRUT IS POSITIVE
901  IF(SD1(1,I) .LE. 0.8 .AND. IFRI(I) .EQ. 0)GO TO 2
      GO TO 3
2    DMTANH(I)=1.0
      GO TO 284
3    DMTANH(I)=ABS(TANH(2.0*SD1(1,I)))
      IFRI(I)=1
284  IF(S(1,I) .LE. DSTOP .AND. SD1(1,I) .LT. 0.0)GO TO 900
      GO TO 902
900  IF(S(1,I) .LE. 0.0001)GO TO 903
      GO TO 904
903  SD2(1,I)=0.0
      SD1(1,I)=0.0
      S(1,I)=0.0
      DP1(I)=0.0
      FFORT(I)=0.0
      VCUM(I)=0.0
      QD(I)=0.0
      II=0
C    CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C    1      ELT,EPRVAL,DERSUB,CHSUP,ITEXT)
      GO TO 902
904  CONTINUE

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IF(IFSTOP(I) .NE. 0)GO TO 906
905 DSTOP=S(1,I)
FSTOPK=2.0*MASS(I)*SD1(1,I)**2/DSTOP**2
906 IF(S(1,I) .LE. DSTOP/2.0)GO TO 908
907 FSTOP(I)=-FSTOPK*(DSTOP-S(1,I))
GO TO 909
908 FSTOP(I)=-FSTOPK*S(1,I)
909 IFSTOP(I)=1
IFRI(I)=0
GO TO 901
902 FSTOP(I)=0.0
901 IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 500
IF(SD1(1,I) .LT. 0.0)GO TO 470
GO TO 471
470 FFORT(I)=-FFORT(I)
CFFOR(I)=-CFFOR(I)
471 FORSST(I)=-((PGA1T(I)-PGA2T(I))*(AREA1(I)-APINT(I))
& +PGA2T(I)*AREA2(I)
X -PGA3T(I)*AREA3(I)+(FFORT(I)
1 +CFFOR(I))*DMTANH(I)+FSTOP(I))
500 IF(INDFLX.GE. 1)GO TO 295
IF(I.EQ.1) GO TO 450
ISTROK(I)=1
C***** ***** BRANCH FOR DEBUGGING PURPOSES:
GO TO 295
289 IF(S(1,I) .LE. 0.0)290,295
290 IF(IOPCO(I) .EQ. 1)GO TO 295
IF((PGA1I(I)-1000.0).LT.PGA1T(I).AND.
& PGA1T(I).LT.(PGA1I(I)+1000.0))299,298
299 IF(XVALVE(I) .NE. 0.0)GO TO 311
IF(IPASS(I) .EQ. 1)GO TO 296
XVALVE(I)=XKSV(I)*XMA11(I)+XBIAS(I)
IPASS(I)=1
GO TO 294
298 IF(ICOSV(I) .EQ. 1)GO TO 291
IOPCO(I)=0
IF(XSV(I) .LT. 0.002 .AND. XSV(I) .GT. -0.002)291,295
291 IF(SD2(1,I) .LE. 0.0 .AND. ICOSV(I) .EQ. 1)GO TO 311
IF(IOPCO(I) .EQ. 1)GO TO 295
IF(PGA1T(I) .GT. PGA1I(I))292,293
292 IF(IXSVL(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDMN(I)*DELT*PERCNT(I)
IF(XVALVE(I) .LE. -0.1)300,294
300 XVALVE(I)=-0.1
IXSVL(I)=1
GO TO 294
293 IF(IXSVH(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDMX(I)*DELT*PERCNT(I)
IF(XVALVE(I) .GE. 0.1)302,294
302 XVALVE(I)=0.1
IXSVH(I)=1
294 CONTINUE

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      II=0
C     CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C     1     ELT,ERRVAL,DESUB,CHSUB,ITEXT)
      DLTX1D(I)=0.0
      ICOSV(I)=1
296   IF(WFORT(I) .GT. 0.0 .AND. S(1,I) .LE. 0.0)GO TO 410
311   IF(NAC(I) .EQ. 1)GO TO 307
      IF(II XSVH(I) .EQ. 1)GO TO 305
      XVALVE(I)=XVALVE(I)+XSVDMM(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .LE. 0.0)305,400
305   XVALVE(I)=0.0
      IIXSVH(I)=1
      GO TO 400
307   IF(II XSVL(I) .EQ. 1)GO TO 308
      XVALVE(I)=XVALVE(I)+XSVDMM(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .GE. 0.0)308,400
308   XVALVE(I)=0.0
      IIXSVL(I)=1
400   II=0
C     CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,FLF2,
C     1     ELT,ERRVAL,DESUB,CHSUB,ITEXT)
410   IF(XVALVE(I) .NE. 0.0)GO TO 295
      ICOSV(I)=0
      DELTX1(I)=DELT X(I)*XKF(I)
      XMA(I)=(DF(I)+DELT X1(I))*XKA(I)
      XMA11(I)=XMA(I)
      XSV(I)=XKSV(I)*XMA11(I)+XBIAS(I)
      CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
      IPASS(I)=0
      IXSVL(I)=0
      IXSVH(I)=0
      IIXSVL(I)=0
      IIXSVH(I)=0
      IOPCO(I)=1
      CALL PHLOC22(PS(I),PR(I),XSV(I),OC(I),XLPSV1(I),XLPSV3(I),RCLSV(I),
      6 DSV(I),CSV1(I),CSV3(I),XMU,OTOLER,NITER,P1(I),QS1(I),QS3(I))
C
295   IF(ISTROK(I) .EQ. 1 .AND. S(1,I) .GT. 0.0)IOPCO(I)=0
C
      ENUP(I)=.5*AMASS*ZG77F1(1)**2
      ENUP(I)=ENUP(I)/(NSTPUT-1)
      IF(HMM(I) .EQ. 1.0)GO TO 130
      SA(I)=0.
      IF(WFORT(I).LT.0.) XSTOT(I)=ENUP(I)/((-WFORT(I))*COPA)
      IF(WFORT(I).GE.0.) XSTOT(I)=1.E20
C     ZSSC IS A PERCENTAGE OF SP(I) FOR ACTIVATING CONTROL-CDMOC(I) IS USE
      ZSSC=0.6*CDMOC(I)*SR(I)
      IF(XSTOT(I) .LE. (ZSSC-S(1,I)) .OR. RESA(I) .EQ. 1.0)SA(I)=1.0
      PESA(I)=SA(I)
      IF(SA(I).EQ.0. .OR. HMM(I).EQ.1.) GO TO 130
      WLFOR(I)=-WFORT(I)
      VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/(AMASS*RFDSLPI(I))

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WRITE(6,121)TIME,WLFOR(I),VELDEC
121  FORMAT(50H ACTIVE CONTROL INITIATED...TIME, WLFOR, VELDEC = ,
1      3E13.5)
HMM(I)=1.
130  IF(S(1,I) .GT. 0.0) ISET(I)=0
      IF(HMM(I).EQ.0.) GO TO 451
      IF(-WFOR(I).GT.(WLFOR(I)+EPSILO(I))) DF(I)=(WLFOR(I)+
& EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LT.(WLFOR(I)-EPSILO(I))) DF(I)=(WLFOR(I)-
& EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LE.(WLFOR(I)+EPSILO(I)).AND.
& -WFOR(I).GE.(WLFOR(I)-EPSILO(I)))
      : 457,456
457  IF(S(1,I) .LE. 0.0) GO TO 456
453  IF(WFOR(I) .GT. 0.0 .AND. OSVCU(I) .LT. 0.0) 454,455
454  DF(I)=WLFOR(I)-(-WFOR(I))
      GO TO 456
455  DF(I)=0.0
456  DELTX(I)=S(1,I)-XSCOM(I)
      IF(S(1,I) .LE. 0.0 .AND. ISET(I) .EQ. 0) GO TO 451
      GO TO 452
451  DF(I)=0.
      DELTX(I)=0.
452  XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
      IF(GNR.EQ.1. .AND. XMA(I).GT.0.) XMA(I)=
      : XMA(I)*SQRT((PGA1T(I)-PGALAC(I))
      X /(PGAHAC(I)-PGA1T(I)))
C
C      NOTE: SUBROUTINE 'FLOZE2' COMPUTES THE FLOWS FROM THE PRESSURES
C      IN UNITS OF INCHES.
C      P1(I)=PGA1T(I)/144.
C      COMPUTATION OF HIGH PRESSURE ACCUMULATOR NITROGEN VOLUME
C      AND ACCUMULATOR PRESSURE
      VOLANT(I)=VOLANT(I)+OSVN(I)*DELT-QPUMPS(I)*DELT
      PS(I)=(((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMAN)-PATM)/144.0
      IF(PS(I) .GE. 3000.0) 464,465
464  PS(I)=3000.0
      VOLANT(I)=VOLANI(I)
465  VOLAHT(I)=VOLACI(I)-VOLANT(I)
      IF(VOLAHT(I) .LE. 0.0) 466,467
466  WRITE(6,1050)TIME
1050  FORMAT(1H0//45H ACCUMULATOR OIL VOLUME INSUFFICIENT AT TIME=,E16.8
1      //)
      CALL LGFAR6
      STOP 500
467  CONTINUE
      CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
& XSV(I),QS1(I),QS3(I),CSV1(I),CSV3(I),XMU)
      QSV1(I)=QS1(I)/1728.
      QSV3(I)=QS3(I)/1728.

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450 CONTINUE
    OSV1(1) = 0.0
    OSV3(1) = 0.0
    OSV(I)=OSV1(I)-OSV3(I)
    IF(OSV(I) .LT. 0.0)NAC(I)=1
    IF(OSV(I) .GT. 0.0)NAC(I)=2
    IF(NAC(I) .EQ. 2)461,462
461 OSVN(I)=OSV(I)
    GO TO 463
462 OSVN(I)=0.0
463 IF(SD1(1,I).LT. 0.0 .AND. PGA1T(I).LE. -1600.0)PGA1T(I)=-1600.0
274 CONTINUE
    I1 = I+3*NSTRUT
    I2 = I+4*NSTRUT
    I3 = I+5*NSTRUT
    CALL INTEG(LA(I1),DP1(I))
    CALL INTEG(LA(I2),QD(I))
    CALL INTEG(LA(I3),OSV(I))
C*****
    IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(FORSST(I)))FORSST(I)=-FT(I)
    AA(I) = (FT(I)+FORSST(I))/MASS(I)
    SD2(1,I) = SR(I)+AA(I)-GZ(I)
    HT1=HT
    IF(SD1(1,I))76,77,78
76 TTIME=S(1,I)/ABS(SD1(1,I))
79 IF(TTIME.GE.HT)GO TO 77
    HT1=TTIME
    GO TO 77
78 TTIME=(SR(I)-S(1,I))/SD1(1,I)
    GO TO 79
77 CONTINUE
    IF(S(1,I).GT.(-ES(I)))GO TO 50
    WRITE(6,49)I,I,S(1,I)
49 FORMAT(58X,4H-ES(,I1,20H) EXCEEDED IN ALGEAR/
    C58X,2HS(,I1,4H) = E15.7)
    S(1,I)=-0.5*ES(I)
    SD1(1,I)=-1.0E-10
    SD2(1,I)=-1.0E-10
50 IF(S(1,I).LE.ES(I))GO TO 51
    IF(S(1,I).LE.(SB(I)-ES(I)))GO TO 55
    IF(S(1,I).LE.(SR(I)+ES(I)))GO TO 52
    WRITE(6,53)I,I,S(1,I)
53 FORMAT(58X,4H ES(,I1,20H) EXCEEDED IN ALGEAR/
    C58X,2HS(,I1,4H) = E15.7)
    S(1,I)=0.5*ES(I)
52 IF(SD1(1,I).GT.0.)SD1(1,I)=0.
    IF(SD2(1,I).LT.0.)GO TO 55
    SD2(1,I)=0.
    GO TO 55
51 IF(SD2(1,I).LT.0.)GO TO 30
    IF(SD1(1,I).LT.0.)SD1(1,I)=0.
    GO TO 55

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30 SD2(1,I)=0.
   SD1(1,I)=0.
55 CONTINUE
   I2=2*I+NSTRUT-1
   I1=I2+1
   CALL INTEG(LA(I2),SD2(1,I))
   CALL INTEG(LA(I1),SD1(1,I))
C   RE-CHECK SHOCK STRUT FORCE FOR RE-CONDITIONED FULLY EXTENDED STATE
   IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(FORSST(I)))FORSST(I)=-FT(I)
   TMP(1)=RZERO(I)-DELTA(I)
   IF(CASK(I).GT.1.E-10)GO TO 200
   IF(INDNWS.EQ.1.AND.I.EQ.1)GO TO 301
   MA(I)=-FTRY(I)*TMP(1)*RI(2,1,I)+FTRX(I)*TMP(1)
C*RI(2,2,I)
   GO TO 201
301 MA(I)=-FTRY(I)*TMP(1)*SIN((PSIPD+ETADES)*DEGRAD)+
1   FTRX(I)*TMP(1)*COS((PSIPD+ETADES)*DEGRAD)
   GO TO 201
200 MA(I)=TMP(1)*SQRT(FTRY(I)*FTRY(I)+FTRX(I)*FTRX(I))
   MA(I)=SIGN(MA(I),-VAXLE(I)-OMET(1,I)*TMP(1))
201 AMA(I)=MA(I)
   IF(IB(I).NE.(-1))GO TO 48
   OMETD1(1,I)=0.
   OMET(1,I)=0.
   GO TO 21
48 TMP(1)=0.
   IF(OMET(1,I).NE.0.)TMP(1)=OMET(1,I)/ABS(OMET(1,I))
   OMETD1(1,I)=(MA(I)-MP(I)*TMP(1))/(NTIRES(I)*MOMENT(I))
21 CALL INTEG(LA(I),OMETD1(1,I))
C*****
65 CONTINUE
C   CALCULATION OF FTRA, FTRB, AND FTRC
   SFTRX=0.
   SFTRY=0.
   SFTRZ=0.
   DO 70 I=1,NSTRUT
     SFTRX=SFTRX+FTRX(I)
     SFTRY=SFTRY+FTRY(I)
70   SFTRZ=SFTRZ+FTRZ(I)
   FTRA=RL(1,1)*SFTRX+RL(1,2)*SFTRY+RL(1,3)*SFTRZ
   FTRB=RL(2,1)*SFTRX+RL(2,2)*SFTRY+RL(2,3)*SFTRZ
   FTRC=RL(3,1)*SFTRX+RL(3,2)*SFTRY+RL(3,3)*SFTRZ
C   CALCULATION OF MTX, MTY, AND MTZ
   SMTRX=0.
   SMTRY=0.
   SMTRZ=0.
   DO 75 I=1,NSTRUT
     SMTRX=SMTRX+MTRX(I)
     SMTRY=SMTRY+MTRY(I)
75   SMTRZ=SMTRZ+MTRZ(I)
   MTX=RL(1,1)*SMTRX+RL(1,2)*SMTRY+RL(1,3)*SMTRZ

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C

MTY=RL(2,1)*SMTRX+RL(2,2)*SMTRY+RL(2,3)*SMTRZ
MT7=RL(3,1)*SMTRX+RL(3,2)*SMTRY+RL(3,3)*SMTRZ
CALCULATION OF FXM, FYM, FZM, LM, MM, AND NM

FYM=0.

FZM=0.

FXM=0.

LM=0.

MM=0.

NM=0.

BFX=0.

BFY=0.

BFZ=0.

BLM=0.

BMM=0.

BNM=0.

DO 82 I=1,NSTRUT

DFXM=0.

DFYM=0.

DFZM=0.

DLM=0.

DMM=0.

DNM=0.

DO 14 IL4=1,NMODE

NBB=(IL4-1)*NSTRUT+I

DFXM=DFXM-MASS(I)*SXMOD(NBB)*GQD2(IL4)

DFYM=DFYM-MASS(I)*SYMOD(NBB)*GQD2(IL4)

DFZM=DFZM-MASS(I)*SZMOD(NBB)*GQD2(IL4)

NBH=(I-1)*NMODE+IL4

DLM=DLM-MASS(I)*GFQRC2(NBH)*GQD2(IL4)

DMM=DMM-MASS(I)*GFQRC3(NBH)*GQD2(IL4)

14 DNM=DNM-MASS(I)*GFQPC4(NBH)*GQD2(IL4)

BFX=BFX+DFXM

BFY=BFY+DFYM

BFZ=BFZ+DFZM

BLM=BLM+DLM

BMM=BMM+DMM

BNM=BNM+DNM

TMP(1)=MASS(I)*SD2(1,I)

FXM=FXM+TMP(1)*A31(I)

FYM=FYM

FZM=FZM+TMP(1)*A33(I)

LM=LM+TMP(1)*A11(I)*RY(I)

MM=MM-TMP(1)*RRCGX(I)

82 NM=NM+TMP(1)*A13(I)*RY(I)

FXM=FXM+BFX+FTPA

FYM=FYM+BFY+FTRB

FZM=FZM+BFZ+FTRC

LM=LM+BLM+MTX

MM=MM+BMM+MTY

NM=NM+BNM+MTZ

C

ENTRY SETUP


```

C DO 28 I=1,NSTRUT
C IF(SD1(1,I) .LE. 0.0) GO TO 2C
C DP1(I)=(-OD(I)+OSV1(I)-OSV3(I)+(AREA1(I)-APINT(I))*SD1(1,I))
C *BETA/VOL1T(I)
C IF(PGA1T(I) .LE. -1600.0) 10,20
10 PGA1T(I) = -1600.0
C IF (DP1(I) .LT. 0.0) DP1(I)=0.0
20 CONTINUE
C II=0
C CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C 1 ELT,ERRVAL,DESUB,CHSUB,ITEXT)
C IF(ICOSV(I) .EQ. 1)GO TO 26
C XMA5(I)=XMA(I)+(2.*ZETAC2*WC1)*XMA1(I)+(WC1**2)*XMA2(I)
C - (2.*ZETAC1*WC1)*XMA3(I)-(WC1**2)*XMA4(I)
C XMA8(I)=(TC1/TC2)*XMA5(I)+(1./TC2)*XMA6(I)-(1./TC2)*XMA7(I)
C XMA11(I)=(TC3/TC4)*XMA8(I)+(1./TC4)*XMA9(I)-(1./TC4)*XMA10(I)
C IF(S(1,I).LE.XSTHR) XMA11(I)=0.
C XSVDDD(I)=(XKSV(I)*XMA11(I)-BCON*XSVDD(I)-CCON*XSVDDOT(I)-
C DCON*(XSV(I)-XBIAS(I)))/ACON
C DLTX1D(I)=(XKF(I)*DELTX(I)-DELTX1(I))/TAUF
26 CONTINUE
C CALL LIMITS(XSV(I),XSVDDOT(I),XSVMAX(I),XSVMIN(I))
C CALL LIMITS(XSVDDOT(I),XSVDD(I),XSVDMX(I),XSVDMN(I))
C CALL LIMITS(XSVDD(I),XSVDDD(I),XDDMAX(I),XDDMIN(I))
C I1 = 3*I+6*NSTRUT-2
C I2 = I1+1
C I3 = I1+2
C CALL INTEG(LA(I1),XSVDDD(I))
C CALL INTEG(LA(I2),XSVDD(I))
C CALL INTEG(LA(I3),XSVDDOT(I))
C I1 = I+9*NSTRUT
C CALL INTEG(LA(I1),DLTX1D(I))
28 CONTINUE
C RETURN
C END

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SUBROUTINE PHLOZ2(PS,PR,X,OC,LAP1,LAP3,RCL,D,COEF1,COEF3,MU,QTOLER,
8,NITER,P1,Q1,Q3)

'PHLOZ2'.....R. D. EDSON

THIS SUBROUTINE CALCULATES THE STEADY-STATE CHAMBER PRESSURE (P1)
AND FLOW RATES (Q1 & Q3) FOR A TWO-WAY NONSYMMETRICAL SPool VALVE
WITH RECTANGULAR WINDOW SLOTS, GIVEN THE STROKE (X) AND THE LOAD
FLOW (OC). THE PARAMETERS REQUIRED IN THE 'CALL' STATEMENT ARE
THE SAME AS DESCRIBED IN SUBROUTINE 'FLOZE2', WITH THE FOLLOWING
ADDITIONAL PARAMETERS:

OC = THE FLOW RATE TO THE LOAD

QTOLER = THE TOLERANCE ALLOWED IN CALCULATING FLOW RATES, FOR
DETERMINING WHETHER OR NOT THE SOLUTION HAS CONVERGED
(.0001 IS TYPICAL)

NITER = THE NUMBER OF ITERATIONS REQUIRED TO CONVERGE TO A
SOLUTION (INTEGER)

```

C
C
  IMPLICIT REAL(L,M)
  PFN(X1,X2,Y1,Y2,Y3,Y4)=X1+(X2-X1)*(Y2-Y1)/(-Y1+Y2-Y3+Y4)
  NITER=0.
  FLAG=-1.
  P1FLAG=-1.

C
  P1A=PR
  P1B=PS
  CALL FLOZE2(PS,PR,P1A,LAP1,LAP3,RCL,D,X,Q1A,Q3A,COEF1,COEF3,MU)
  CALL FLOZE2(PS,PR,P1B,LAP1,LAP3,RCL,D,X,Q1B,Q3B,COEF1,COEF3,MU)
  IF(Q1A.EQ.0. .AND. Q3B.EQ.0.) GO TO 51
  Q3A=Q3A+QC
  Q3B=Q3B+QC
  GO TO 50
51  P1=(PS+PR)/2.
     Q1=0.
     Q3=0.
     GO TO 400

C
50  P1=PFN(P1A,P1B,Q3A,Q1A,Q1B,Q3B)
     CALL FLOZE2(PS,PR,P1,LAP1,LAP3,RCL,D,X,Q1,Q3,COEF1,COEF3,MU)
     Q3=Q3+QC
     IF(FLAG.LT.0.) GO TO 55
     IF(P1.EQ.P1I) GO TO 100
55  P1I=P1
     IF(Q1.EQ.0. .AND. Q3.EQ.0.) GO TO 100
     IF(ABS(Q1) .GE. ABS(Q3)) QDEN=Q1
     IF(ABS(Q3) .GT. ABS(Q1)) QDEN=Q3
     IF(ABS((Q1-Q3)/QDEN) .LT. QTOLER) GO TO 100
     IF(Q1.LT.Q3) GO TO 90
     P1A=P1
     Q1A=Q1
     Q3A=Q3
     GO TO 150
90  P1B=P1
     Q1B=Q1
     Q3B=Q3
     GO TO 150
100 P1FLAG=1.
150 FLAG=1.
     NITER=NITER+1
     IF(P1FLAG.GT.0.) GO TO 300
     GO TO 50
300 CONTINUE
     Q3=Q3-QC
400 RETURN
     END

```

SUBROUTINE FLOZE2(PS,PR,P1,LAP1,LAP3,RCL,D,X,Q1,Q3,COEF1,COEF3,MU)
'FLOZE2'.....R. D. EDSON

THIS SUBROUTINE CALCULATES THE STEADY-STATE FLOW RATES (Q1 AND Q3) FOR A TWO-WAY NONSYMMETRICAL SPOOL VALVE WITH RECTANGULAR WINDOW SLOTS, GIVEN THE LOAD CHAMBER PRESSURE (P1) AND STROKE (X). THE PARAMETERS REQUIRED IN THE 'CALL' STATEMENT ARE AS FOLLOWS:

X = VALVE STROKE
P1 = PRESSURE IN CHAMBER 1 (TO LOAD)
Q1 = FLOW RATE FROM SUPPLY LINE TO CHAMBER 1
Q3 = FLOW RATE FROM CHAMBER 1 TO RETURN LINE
PS = SUPPLY PRESSURE
PR = RETURN PRESSURE
LAP1 = OVERLAPPED OR UNDERLAPPED LENGTH BETWEEN THE SPOOL AND SLEEVE AT NULL, FOR FLOW Q1. A POSITIVE NUMBER IS USED FOR OVERLAP, A NEGATIVE NUMBER FOR UNDERLAP.
LAP3 = OVERLAPPED OR UNDERLAPPED LENGTH BETWEEN THE SPOOL AND SLEEVE AT NULL, FOR FLOW Q3. A POSITIVE NUMBER IS USED FOR OVERLAP, A NEGATIVE NUMBER FOR UNDERLAP.
RCL = RADIAL CLEARANCE BETWEEN THE SPOOL AND SLEEVE
D = DIAMETER OF SPOOL
COEF1 = FLOW COEFFICIENT OF ORIFICE 1 (SUPPLY TO CHAMBER 1)
 = $CD \cdot W1 \cdot \sqrt{2 \cdot GC / RHO}$
COEF3 = FLOW COEFFICIENT OF ORIFICE 3 (CHAMBER 1 TO RETURN)
 = $CD \cdot W3 \cdot \sqrt{2 \cdot GC / RHO}$
 WHERE W1 = TOTAL WINDOW WIDTH OF ORIFICE 1
 W3 = TOTAL WINDOW WIDTH OF ORIFICE 3
 CD = DISCHARGE COEFFICIENT
 GC = GRAVITATIONAL ACCELERATION CONSTANT
 RHO = DENSITY OF HYDRAULIC FLUID
MU = VISCOSITY OF HYDRAULIC FLUID, CENTIPOISE

THE METHOD OF SOLUTION UTILIZES THE TURBULENT ORIFICE EQUATION AND THE EQUATION FOR FULLY-DEVELOPED LAMINAR FLOW THROUGH AN ANNULUS, WITH FULL ECCENTRICITY ASSUMED. FOR ORIFICE OPENINGS WHERE SOME OVERLAPPED LENGTH EXISTS, THE PROCEDURE IS TO CALC-

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ULATE THE FLOW RATE BY BOTH EQUATIONS, AND THEN USE THE ONE
THAT GIVES THE SMALLEST ABSOLUTE VALUE AS THE ANSWER. FOR
OPENINGS WHERE NO OVERLAPPED LENGTH EXISTS, ONLY THE TURBULENT
ORIFICE EQUATION APPLIES.

IMPLICIT REAL(L,M)
Q12(T1,T2)=SIGN(1.,(T1-T2))*SQRT(ABS(T1-T2))
Q34(T3,T4)=4.5E06*(T3-T4)*D*RCL**3/MU

***** CALCULATE Q1 *****

X2=LAP1-X
X4=SQRT(X2**2+RCL**2)
IF(LAP1 .LE. 0.) GO TO 99

POSITIVE LAPS:
IF(X .GE. LAP1) GO TO 65
Q1=RCL*COEF1*Q12(PS,P1)
Q1L=Q34(PS,P1)/X2
IF(ABS(Q1L) .LT. ABS(Q1)) Q1=Q1L
GO TO 20
65 Q1=X4*COEF1*Q12(PS,P1)
GO TO 20

NEGATIVE LAPS:
99 IF(X .LT. LAP1) GO TO 10
Q1=X4*COEF1*Q12(PS,P1)
GO TO 20
10 Q1=RCL*COEF1*Q12(PS,P1)
Q1L=Q34(PS,P1)/X2
IF(ABS(Q1L) .LT. ABS(Q1)) Q1=Q1L

***** CALCULATE Q3 *****

20 X1=LAP3+X
X3=SQRT(X1**2+RCL**2)
IF(LAP3 .LE. 0.) GO TO 199

C POSITIVE LAPS:
IF(X .LE. -LAP3) GO TO 165
Q3=RCL*COEF3*Q12(P1,PR)
Q3L=Q34(P1,PR)/X1
IF(ABS(Q3L) .LT. ABS(Q3)) Q3=Q3L
GO TO 120
165 Q3=X3*COEF3*Q12(P1,PR)
GO TO 120

C
C NEGATIVE LAPS:
199 IF(X .GT. -LAP3) GO TO 110
Q3=X3*COEF3*Q12(P1,PR)
GO TO 120

110 Q3=RCL*COEF3*Q12(P1,PR)
Q3L=Q34(P1,PR)/X1
IF(ABS(Q3L) .LT. ABS(Q3)) Q3=Q3L
120 RETURN
END

SUBROUTINE LIMITS(X,XDOT,XMAX,XMIN)

C
C STATEMENTS FOR THIS SUPROUTINE OBTAINED BY PHONE BY RD020177
C JOHN R. MCGEEHEE ON 2/1/77. RD020177
C
IF(X .GE. XMAX) GO TO 10
IF(X .LE. XMIN) GO TO 20
GO TO 30

```
10 X=XMAX  
   IF(XDOT .GT. 0.0) XDOT = 0.0  
   GO TO 30  
20 X=XMIN  
   IF(XDOT .LT. 0.0) XDOT = 0.0  
30 RETURN  
   END
```

The ident CSCMODS contains the code to interface the active routines with the remainder of the program. In subroutine EXE, a call to the comdeck ACOBLK was inserted before the LOGICAL statement and the call to the landing gear routine was changed as follows.

```
CALL LGFAR2  
IF (IABS(INDLG).NE.3) GO TO 594  
CALL ALGEAR1  
GO TO 596  
594 CONTINUE  
CALL LGEAR3  
596 CONTINUE  
CALL FLEX2
```

The following code was added to subroutine INUPD as the first executable statements to monitor changes in the number of integration variables.

```
      NNUM = NUM+N  
      WRITE(6,600) NNUM  
600 FORMAT(5X,32HNUMBER OF INTEGRATED VARIABLES =,I3)
```

To allow for a greater number of integration variables the common block LGDE was expanded in LGDET, LGEAR1, LGEA3C, and SDFLGP as follows.

```
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
```

Subroutine LINES was changed as follows since 51 lines are too many for 8 1/2 inch paper.

```
IF(LONG.LE.41)RETURN
```

Subroutine OPT1 was changed by calling the active common block before the REAL statement, by adding the following EQUIVALENCE statement

```
EQUIVALENCE (DM155(133), INDLG)
```

to make the variable INDLG available and the call to the landing gear code was changed as follows.

```
2062 CALL SACS3
      IF (IABS(INDLG).NE.3) GO TO 2070
      CALL ALGEAR1
      GO TO 2075
2070 CONTINUE
2125 CALL LGEAR3
2075 CONTINUE
      CALL FLEX3
```

In subroutine LGEAR1, the number of integration variables was increased by adding the following line of code before the call to INUPD, and

```
IF (IABS(INDLG) .EQ. 3) NDEQ=10*NSTRUT
```

the active control initialization is accomplished by adding the following immediately before the statement labeled 3.

IF (IABS(INDLG).EQ.3) CALL ACTINIT

In subroutine LGEA3C, the following EQUIVALENCE

EQUIVALENCE (DM14(155),INDLG)

statement was added to make the variable INDLG available. The following test was added after

IF (IABS(INDLG).EQ.3) GO TO 31

the statement labeled 27 and the similar test below

IF (IABS(INDLG).EQ.3) GO TO 46

was added after the fifth line below the statement labeled 4. A call to the active gear common block was added to SDFLGP. Program output capabilities were expanded to include the active control parameters by including the following DIMENSION and DATA statements in subroutine SDFLGP.

```
DIMENSION ACOVAR1(8), ACOVAR2(8), ACOVAR3(8), ACOVAR4(8),
* ACOVAR5(8), ACOVAR6(8), ACOVAR7(8), ACOVAR8(8), ACOVAR9(8)
DATA(ACOVAR1(I), I=1,8) /5HVOL1T,5HVOL2T,5HVOL3T,5HPGA1T,5HPGA2T,
* 5HPGA3T,5HGAMAH,5HCOEFO/
DATA(ACOVAR2(I), I=1,8) /2HQO,4HVCUM,6HBLFORT,6HBUFORT,5HFFORT,
* 6HFORCHT,6HFOAHST,6HFONHST/
DATA(ACOVAR3(I), I=1,8) /5HCFFOR,5HFSTOP,6HFORSST,6HXVALVE,3HXS,
* 5HDELTX,6HDELTX1,6HDLTX1D/
DATA(ACOVAR4(I), I=1,8) /2HDF,3HDP1,2HPS,2HP1,3HQSV,4HQSVN,4HQSV1,
* 4HQSV3/
DATA(ACOVAR5(I), I=1,8) /6HEPSILO,3HHMM,6HDMTANH,5HXSTOT,5HICQSV,
* 4HENUP,5HWLFOR,5HIOPCO/
DATA(ACOVAR6(I), I=1,8) /6HIIXSVH,6HIIXSVL,5HIXSVH,5HIXSVL,5HIPASS
* ,3HNAC,4HRESA,2HSA/
DATA(ACOVAR7(I), I=1,8) /6HVELDEC,6HWLFORR,6HIDEACT,4HZSSC,4HCOPA,
* 4HSIPA,5HDSTOP,6HFSTOPK/
DATA(ACOVAR8(I), I=1,8) /5HWFORT,6HVOLAHT,6HVOLANT,5HQSVCU,3HXMA,
```


* 4HXMA5,4HXMA8,5HXMA11/
DATA(ACOVAR9(I), I=1,8) /6HXSVD00,5HXSVD0,6HXSVD0T,2HPR,4HIFRI,
* 6HIFSTOP,6HISTROK,4HISET/

The actual output is accomplished by the following code which was
added three lines below the statement labeled 32.

```

IF(IABS(INDLG).NE.3) GO TO 50
CALL STFL(2,8,ACOVAR1)
DO 33 I=1,NSTRUT
CALL STVAR(8,VOL1T(I),VOL2T(I),VOL3T(I),PGA1T(I),PGA2T(I),
* PGA3T(I),GAMAH(I),CDEF0(I))
33 CONTINUE
CALL STFL(2,8,ACOVAR2)
DO 34 I=1,NSTRUT
CALL STVAR(8,OD(I),VCUM(I),BLFORT(I),RUFORT(I),FFORT(I),
* FORCHT(I),FOAHST(I),FONHST(I))
34 CONTINUE
CALL STFL(2,8,ACOVAR3)
DO 35 I=1,NSTRUT
CALL STVAR(8,CFFOR(I),FSTOP(I),FORSST(I),XVALVE(I),XSV(I),
* DELTX(I),DELTX1(I),DLTX1D(I))
35 CONTINUE
CALL STFL(2,8,ACOVAR4)
DO 36 I=1,NSTRUT
CALL STVAR(8,DF(I),DP1(I),PS(I),P1(I),OSV(I),OSVN(I),OSV1(I),
* OSV3(I))
36 CONTINUE
CALL STFL(2,8,ACOVAR8)
DO 37 I=1,NSTRUT
CALL STVAR(8,WFORT(I),VOLAHT(I),VOLANT(I),QSVCU(I),XMA(I),
* XMA5(I),XMA8(I),XMA11(I))
37 CONTINUE
CALL STFL(2,8,ACOVAR9)
DO 38 I=1,NSTRUT
CALL STVAR(8,XSVDD0(I),XSVDD(I),XSVDD0T(I),PP(I),FLOAT(IFRI(I)),
* FLOAT(IFSTOP(I)),FLOAT(ISTROK(I)),FLOAT(ISET(I)))
38 CONTINUE
CALL STFL(2,8,ACOVAR5)
DO 39 I=1,NSTRUT
CALL STVAR(8,EPSILO(I),HMM(I),DMTANH(I),XSTOT(I),FLOAT(ICOSV(I)),
* ENUP(I),WLFOR(I),FLOAT(IOPCO(I)))
39 CONTINUE
CALL STFL(2,8,ACOVAR6)
DO 40 I=1,NSTRUT
CALL STVAR(8,FLOAT(IIXSVM(I)),FLOAT(IIXSVM(I)),FLOAT(IXSMV(I)),

```

```
* FLOAT(IXSVL(I)),FLOAT(IPASS(I)),
* FLOAT(NAC(I)),RESA(I),SA(I))
40 CONTINUE
  CALL STFL(2,8,ACOVAR7)
  CALL STOVAR(8,VFLDEC,WLFQPR,FLOAT(IDEACT),ZSSC,COPA,SIPA,DSTOP,
* FSTOPK)
50 CONTINUE
```

The active control variables are placed in the proper arrays for subsequent numerical integration by the following code which was inserted in SDFLGP three lines below the statement labeled 5.

```
IF(IABS(INDLG) .NE. 3) GO TO 6
I1 = I+3*NSTRUT
I2 = I+4*NSTRUT
I3 = I+5*NSTRUT
CALL UPDAT(1,LA(I1),PGA1T(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I2),VCUM(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I3),OSVCU(I),DU,DU,DU,DU)
I1 = 3*I+6*NSTRUT-2
I2 = I+9*NSTRUT
CALL UPDAT(3,LA(I1),XSVDD(I),XSVDOT(I),XSV(I),DU,DU)
CALL UPDAT(1,LA(I2),DELTXT1(I),DU,DU,DU,DU)
```

The following code was added immediately after the statement labeled 16 in subroutine READ to correct an initialization problem revealed by an operating system release which does not preset memory to zero.

```
SLTSYM = 0.0
IBC = 0
INXQ = 0
JBC = 0
```

For the same reason, the following code was added below the statement labeled 2 in subroutine AUTS.

```
DELPI = 0.0
TR = 0.0
```

The next phase in the development of the active control capability was the interface of the active code with the input and graphic output modules of the FATOLA program. This was accomplished with the correction sets GMMODS, JMMODS, DIRACT, and ACTIN. A new deck ACTIN was inserted after the deck READ and a new deck DIRACT was added following the deck DIR3DA. The contents of these new decks is shown below.

```

SUBROUTINE ACTIN
COMMON/ACTDIR/XNAME(71),LOC(71)
COMMON/ACTIVE/DATA(646)
DIMENSION IRA(55),MSG(58),IDATA(646)
EQUIVALENCE (MSG(1),SYM),(MSG(2),OP)
EQUIVALENCE (MSG(3),IRA(1)),(MSG(58),INC)
EQUIVALENCE (DATA(1),IDATA(1))
INTEGER COMMA,POINT,E,BLANK
DATA REMARK,COMMA,BLANK,POINT,E,ENDACT,MINUS,AINT/
1 3HREM,1R,,1R ,1R.,1RE,6HENDACT,1R-,3HINT/
100 CONTINUE
1  FORMAT(A6,1X,A3,1X,55R1,I1)
2  FORMAT(18X,A6,1X,A3,1X,55R1,I6)
3  FORMAT(20HOERROR.THE SYMBOL **,A6,
1 26H** IS NOT IN THE DIRECTORY/1H )
4  FORMAT(44HOERROR.ILLEGAL CHARACTER IN NUMERIC FIELD **,1R1,2H**/)
READ(5,1) SYM,OP,IRA,INC
CALL LINES(1)
J = 58
IF(INC.LE.0) J = 57
WRITE(6,2) (MSG(I),I=1,J)
IF(SYM.EQ.REMARK) GO TO 100
IF(SYM.EQ.ENDACT) GO TO 999
DO 110 I=1,71
IF(SYM.EQ.XNAME(I)) GO TO 120
110 CONTINUE
CALL LINES(3)
WRITE(6,3) SYM
GO TO 100
120 CONTINUE
INDEX = LOC(I)
IF(INC.EQ.0) INC = 1
INDEX = INDEX + INC - 1
NUMEXP = 0
NEXP = 0
IEXP = 0
NL = 0
NR = 0

```

```

NUML = 0
NUMR = 0
ISIGN = 0
JSIGN = 0
LEFT = 1
DO 210 I=1,56
IF(I.EQ.56) GO TO 140
IF(IRA(I).EQ.BLANK) GO TO 210
IF(IRA(I).EQ.COMMA) GO TO 140
IF(IRA(I).EQ.POINT) GO TO 170
IF(IRA(I).EQ.E) GO TO 180
IF(IRA(I).EQ.MINUS) GO TO 200
IF(IRA(I).GT.36) GO TO 130
IF(IRA(I).LT.27) GO TO 130
NUM = IRA(I) - 27
IF(IEXP.EQ.1) GO TO 190
IF(LEFT.GT.0) NUML = 10*NUML + NUM
IF(LEFT.GT.0) NL = NL + 1
IF(LEFT.LT.0) NUMR = 10*NUMR + NUM
IF(LEFT.LT.0) NR = NR + 1
GO TO 210
130 CALL LINES(3)
WRITE(6,4) IRA(I)
GO TO 210
140 CONTINUE
IF(NL.EQ.0.AND.NR.EQ.0) GO TO 210
IF(NR.EQ.0) GO TO 160
X = FLOAT(NUML) + FLOAT(NUMR)/10.**NR
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
DATA(INDEX) = X
150 NUML = 0
NUMR = 0
NL = 0
NR = 0
LEFT = 1
ISIGN = 0
JSIGN = 0
IEXP = 0
NEXP = 0
NUMEXP = 0
INDEX = INDEX + 1
GO TO 210
160 CONTINUE
X = NUML
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
NUML = X
IF(OP.EQ.AINT) IDATA(INDEX) = NUML

```

```

      IF(OP.NE.AINT) DATA(INDEX) = X
      GO TO 150
170  CONTINUE
      LEFT = -1
      GO TO 210
180  CONTINUE
      IEXP = 1
      GO TO 210
190  CONTINUE
      NUMEXP = 10*NUMEXP + NUM
      NEXP = NEXP + 1
      GO TO 210
200  CONTINUE
      IF(IEXP.EQ.0) ISIGN = 1
      IF(IEXP.NE.0) JSIGN = 1
210  CONTINUE
      GO TO 100
999  RETURN
      END

```

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```

BLOCK DATA DIRACT
COMMON/ACTDIR/NAME(71),LOC(71)
DATA NAME/ 6HAMUH , 6HAPINT , 6HAREA1 , 6HAREA2 ,
1 6HAREA3 , 6HAREMO , 6HAREO3 , 6HBETA , 6HBLMU ,
2 6HBUMU , 6HCDMOC , 6HCDMOE , 6HCDSV , 6HCD3 ,
3 6HCFFOR , 6HDELT , 6HDIOTA , 6HDSV , 6HEPSILO ,
4 6HEPSROL , 6HEPSSLP , 6HETASV , 6HGAMAH , 6HGAMAN ,
5 6HGNR , 6HIDEACT , 6HKAPT , 6HOMRUN , 6HPATH ,
6 6HPGAHAC , 6HPGALAC , 6HPGA1I , 6HPGA2I , 6HPGA3I ,
7 6HPERCNT , 6HOPUMPS , 6HRCLSV , 6HRHOH , 6HTAUF ,
8 6HTC1 , 6HTC2 , 6HTC3 , 6HTC4 , 6HVOLACI ,
9 6HVOLANI , 6HVOL1I , 6HVOL2I , 6HVOL3I , 6HWC ,
1 6HWC1 , 6HWSV , 6HWSV1 , 6HWSV3 , 6HWLFOR ,
2 6HWLFORR , 6HXBIAS , 6HXDDMAX , 6HXDDMIN , 6HXKA ,
3 6HXKF , 6HXKSV , 6HXLPSV1 , 6HXLPSV3 , 6HXSCOM ,
4 6HXSTHR , 6HXSVDN , 6HXSVDN , 6HXSVMAX , 6HXSVMIN ,
5 6HZETAC1 , 6HZETAC2/
DATA LOC/ 1, 6, 16, 21, 26, 31, 36, 61, 46,
1 56, 62, 67, 72, 82, 77, 109, 141, 135, 152,
2 157, 162, 163, 195, 200, 201, 212, 263, 274, 275,
3 281, 286, 291, 301, 311, 276, 346, 387, 397, 410,
4 411, 412, 413, 414, 621, 631, 421, 431, 441, 451,
5 452, 641, 642, 643, 458, 463, 464, 611, 616, 474,
6 479, 484, 489, 494, 561, 499, 591, 596, 601, 606,
7 644, 645/
END

```

The GMMODS correction set expanded the dimensions of the TITLE, BUF, TBUF, and BMM arrays in the PLTDAT program as illustrated below.

```
DIMENSION TITLE(20),BUF(550),NDIL(28),TBUF(550)
1 BMM(2,550),CMPDS(6),DEPVAR(5),LINE(7),NDVA(5)
```

The JMMODS correction set interfaced the new decks with the FATOLA program. All DATA statements in ACTINIT were replaced with the single DATA statement

```
DATA ICOSV,IFSTOP,II/11*C/
```

Subroutine SDFLGP was modified to output active control parameters to TAPE13, the disk file which communicates with PLTDAT to generate graphic output. The DATA statement

```
DATA N19/19/
```

was inserted in this routine and the write statement below the statement labeled 115 was replaced with:

```
IF(ISUM1.NE.0) WRITE(13) N19,ISUM1,DAT3,OP17,ACQVAR8(1),
1 ACQVAR8(4),ACQVAR4(5),ACQVAR1(4),ACQVAR1(5)
```

Additionally, the final continuation card in each of the tests on ISTPL1, ISTPL2, ISTPL3, ISTPL4, and ISTPL5 were replaced with the following statements respectively.

```
1 DMET(1,1),WFORT(1),OSVCU(1),OSV(1),PGA1T(1),PGA2T(1)
1 DMET(1,2),WFORT(2),OSVCU(2),OSV(2),PGA1T(2),PGA2T(2)
1 DMET(1,3),WFORT(3),OSVCU(3),OSV(3),PGA1T(3),PGA2T(3)
```

```
1 OMET(1,4),WFORT(4),OSVCU(4),OSV(4),PGA1T(4),PGA2T(4)
1 OMET(1,5),WFORT(5),OSVCU(5),OSV(5),PGA1T(5),PGA2T(5)
```

The data statement

DATA ACTIVE/6HACTIVE/

was added to subroutine READ and statement 19 was replaced with the following.

```
19 IF(SYM.EQ.ACTIVE) GO TO 805
   CALL DIPLAC(RA1,INC,BLANK)
```

The following code was inserted after the STOP 25

```
805 CALL ACTIN
     GO TO 100
810 CALL ACTIN
     GO TO 802
```

and statement 26 was replaced with the following.

```
26 IF(SYM.EQ.ACTIVE) GO TO 810
   IF(SYM.EQ.STCASE) GO TO 21
```

The CSCMOD2 correction set contained general program corrections and enhancements with emphasis on the active control code. In the ACOBLK comdeck, the ACTIVE common block was modified by replacing FOAHST(5) with FWORK(5), replacing PGA1T(5) with PGA1T1(5), replacing ZZSSC with ZSSC(5), and by adding IMODE(5), CMASNG, and VMAS(5). The common block HTCOM was expanded to

include INDINT(5), in EXE, MIMIN, LGDET, LGEAR1, ACTINIT, ALGEAR,
and FLEX1. Subroutine EXE was further modified by adding

STOP "FLIGHT TIME LIMIT"

to lines below the statement labeled 743. Three lines of code

C
DO 10 I=1,5
10 INDINT(I) = 1

were inserted as the first executable statements in subroutine
MIMIN. In subroutine LGEAR1, the following three lines of code were
added as the last executable statements in the LGEAR1 entry point,
(pre data initialization).

DO 6 I=1,5
P(I) = 0.0
6 P2(I) = 0.0

The single DATA statement in ACTINIT (inserted by JMMODS) was
replaced with the following code,

EQUIVALENCE (DM15(16),GREFF), (DM1(37),AIYYRS)
DATA DSTOP,FSTOPK /0.004,0.0/
DATA ENUP,FSTOP,HMM,OSVN,RESA,SA /30*0.0/
DATA ICOSV,IFRI,IFSTOP,IPASS /20*0/
DATA IOPCO,ISET /10*1/
DATA IXSVH,IXSVL,IIXSVH,IIXSVL /20*0/
C

the definition of OMRUN was changed to,

OMRUN=ERDEG*0.01745329

and the following definitions were inserted in the 100 loop.

INDINT(I) = 1
VMAS(I)=AMASS

Additionally, the line below statement number 80 was changed to

(PGA1T1(I)=PGA1I(I

In order to make the variable GAMA available to the subroutine
ALGEAR1, the variable DUM15(13) in the DIRCOM common block was
replaced by GAMA, DUM15(12). The following lines of code were
removed

```
EQUIVALENCE (DM5(16),GREFF)
DATA IFRI/5*0/,FSTOPK/0.0/,FSTOP/5*0.0/,DSTOP/0.004/
DATA SA,RESA,HMM,IXSVL,IXSVH,IIXSVL,IIXSVH,IPASS
1  /15*0.,25*0/
DATA ENUP/5*0.0/
DATA ISET,IOPCO,OSVN/10*1,5*0.0/
```

and were replaced with

```
EQUIVALENCE (DM5(16),GREFF), (DM2(27),AXP7F), (DM2(28),AX77F),
*(DM2(29),AYP7F), (DM2(30),AY77F), (DM2(31),AZP7F), (DM2(33),AZ77F),
*(DM1(37),AIYYBS)
DIMENSION INDEACT(5),IPSTOP(5),AIC(5),PGA1T(5)
DATA AIC,INDEACT,IPSTOP /5*0.0,10*0/
```

The following new code was inserted beneath statement number 760.

C
CMASNG = AIYYBS/AMASS+AMASS*PX(1)*PX(1)/AIYYBS
CMASNG = 7.9677
VMASNG(1) = AMASS/CMASNG
ENCG = 0.5*AMASS*ZG77F1(1)*ZG77F1(1)
ZDANT = ZG77F1(1)-OI77R*PX(1)

The following two lines of code

18 IF(IDEACT.EQ.1) GO TO 56
19 IF(IDEACT.EQ.2) GO TO 80

were replaced with

C
18 DO 90 I=1,NSTRUT
IF(INDEACT(I).EQ.1) GO TO 56
IF(INDEACT(I).EQ.2) GO TO 80

The following lines of code were added after the test on OMRUN and after statement number 25, respectively.

IF(I.EQ.1 .AND. ZDANT.GE.VLDEC) GO TO 90
IF(I.EQ.1 .AND. ZDANT.GE.VELDEC+XG77F1(1)*TAN(OMRUN)) GO TO 90

The reduction of the control limit force was modified by replacing the three lines of code centered about statement number 56 with the following.

INDEACT(I)=1
56 IF(INDINT(I).EQ.0) GO TO 58
WLFOR(I)=WLFOR(I)-REDSLP(I)*HT
EPSILO(I)=EPSILO(I)+EPSSLP*HT
58 CONTINUE
INDINT(I)=0

In the line of code IDEACT = 2, IDEACT was changed to INDEACT(I).
The four lines of code beginning the 65 loop were replaced with the following.

```
C*** CALCULATION OF THE WING-GEAR INTERFACE FORCE (WFORT)
      UNSPRNG = 0.0
      DO 66 J=1,NSTRUT
        IF(OMETD1(J).NE. 0.0) UNSPRNG = UNSPRNG+MASS(J)
66  CONTINUE
      DWFORT = (-SORT(AXP7F*AXP7F+AYP7F*AYP7F+
&      AZP7F*AZP7F)+GREFF)*(AMASS-UNSPRNG)
      WFORT(1) = DWFORT+FORSSST(1)
      DO 67 I=2,NSTRUT
        WFORT(I) = DWFORT/(NSTRUT-1)
67  CONTINUE
C
      DO 65 I=1,NSTRUT
C
C***
```

The line below the statement labeled 210 was replaced with

```
PGA1T(I)=PGA1T1(I)
IF(PGA1T1(I).LE.-1600.0) PGA1T(I)=-1600.0
```

and GAMAN was replaced with GAMA in the definition of PGA2T(I). The two lines of code ending with the definition of QO were replaced with the following.

```
IF(IMODE(I).EQ.0 .AND. DDELTA(I).LE.0.0) 112,113
112 QO(I) = 0.0
    GO TO 109
C
113 QO(I)=COEFO(I)*(AREMO(I)-APINT(I))*Q1(PGA1T1(I),PGA2T(I))
    IF(QO(I).LT.0.0 .AND. VCUM(I).LE.0.0) GO TO 102
    AIC(I)=0.0
    GO TO 103
102 IF(PGA1T1(I).LT.PGA2T(I)) GO TO 103
    GO TO 111
111 QO(I)=0.0
    VCUM(I)=0.0
    AIC(I)=1.0
103 IF(QO(I).GT.0.0) AIC(I)=0.0
```

The test on I below statement number 150 was changed to

```
IF(IMODE(I).EQ.0) GO TO 297
```

and the debugging change to 295 below this was changed back to 289.

Statement number 284 was replaced with the following.

```
284 IF(S(1,I).LT.0.0) GO TO 160  
    GO TO 161  
160 IPSTOP(I)=1  
161 IF(S(1,I).LE.DSTOP .AND. IPSTOP(I).EQ.1) GO TO 900
```

In statement number 900, the limiting value for the stroke was changed from 0.0001 to 0.005 and the five lines of code from DP1(I)=0.0 to II=0 were removed. Statement number 901 was replaced with the following code.

```
901 IF(PGA1T(I).LE.(PGA1I(I)+500.0) .AND.  
    : PGA1T(I).GT.(PGA1I(I)-500.0)) GO TO 158  
    GO TO 159  
158 IF(ABS(FT(I)).LE.FORCHT(I) .AND. S(1,I).EQ.0.0) GO TO 500  
159 IF(S(1,I).GE.0.0) GO TO 470
```

The test on I below statement 500 was changed to

```
IF(IMODE(I).EQ.0) GO TO 297
```

and the branch to 295 for debugging purposes was removed. The following code was inserted after statement number 295.

```

297 I2 = 2*I+NSTRUT-1
    I1 = I2+1
    CALL INTEG(LA(I2),SD2(1,I))
    CALL INTEG(LA(I1),SD1(1,I))
C
    IF(IMODE(I).EQ.0) GO TO 450

```

The two-line definition of ENUP(I) was expanded to

```

    IF(I.NE.1) GO TO 119
    ENUP(1) = 0.5*AIYYBS*QI77R*QI77R+(ENCG/CMASNG)*
*           (ZG77F1(1)/ABS(ZG77F1(1)))
    GO TO 120
119 ENUP(I) = ENCG/(NSTRUT-1)
120 CONTINUE

```

and the second test on WFORT(I) below statement 120 was changed to the following.

```

IF(WFORT(I).GE.0.0 .OR. DDELTA(I).LT.0.0) XSTOT(I)=1.E20

```

The two lines of code beginning with the definition of ZSSC, the definition of VELDEC, and the definition of PS(I) were changed as follows.

```

ZSSC(I)=FWORK(I)*SB(I)
IF(XSTOT(I).LE.(ZSSC(I)-S(1,I)) .OR. RESA(I).EQ.1.0) SA(I)=1.0
VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/
E      (AMASS*REDSLP(I))
PS(I)=((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMA)-PATM/144.0

```

The definitions of QSV1(1) and QSV3(1) below statement number 450 were removed and the following new code was inserted in place of the

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six lines below statement 274.

```

I4 = I+4*NSTRUT
I5 = I+5*NSTRUT
CALL INTEG(LA(I4),QO(I))
CALL INTEG(LA(I5),QSV(I))
I9 = I+9*NSTRUT
CALL INTEG(LA(I9),DLTX1D(I))

```

The four lines of code below statement 55 were removed to make up for the insertion at statement number 297. The test on the strut velocity at the top of the 28 loop was removed and the test on PGA1T(I) and the following two lines were replaced with the following.

```

IF(IMODE(I).EQ.0 .AND. DDELTA(I).LE.0.0) GO TO 19
IF(S(1,I).NE.0.0 .OR. AIC(I).NE.1.0) GO TO 20
PGA1T1(I)=PGA1I(I)
PGA1T(I) =PGA1I(I)
19 DP1(I)=0.0

```

The eight lines of code below the last call to LIMITS were replaced with the following code.

```

I3 = I+3*NSTRUT
CALL INTEG(LA(I3),DP1(I))
I9 = I +9*NSTRUT
CALL INTEG(LA(I9),DLTX1D(I))
I6 = 3*I+6*NSTRUT-2
I7 = I6+1
I8 = I6+2
CALL INTEG(LA(I6),XSVDDD(I))
CALL INTEG(LA(I7),XSVDD(I))
CALL INTEG(LA(I8),XSVDGT(I))

```

In subroutine SDFLGP, the DATA statements for ACOVAR1 and ACOVAR2 were modified by substituting PGA1T1 for PGA1T and VMASST for FOAHST,

respectively. The five WRITE statements to unit 13, the call to STOVAR, and the call to UPDAT were also changed by substituting PAG1T1 for PGA1T, and the call to STOVAR for FOAHST was changed to VMASS. In subroutine ACTIN the size of the ACTIVE common block and the IDATA array were expanded from 646 to 656. In subroutine PACK, the size of the I1 array was changed from 1 to 6, a change which has no impact on program execution. Lastly, in BLOCK DATA DIRACT, the 4th, 5th, and 14th continuation lines in the DATA statement for NAME were changed to the following.

```
4 6HEPSROL, 6HEPSSLP, 6HETASV , 6HFWORK ,6HGAMAH ,  
5 6HGNR , 6HKAPT , 6HOMRUN , 6HPATM ,  
5 64ZETAC1, 6H7ETAC2, 6HIMODE /
```

The 2nd and 7th continuation lines in the DATA statement for LOC were accordingly changed as follows.

```
2 157, 162, 163, 170, 195, 201, 263, 274, 275,  
7 644, 645, 651/
```

The EOR correction set is empty and does nothing.

The EORPL correction set inserted a new deck, EORPL, after subroutine CTENGL. This deck does nothing except write a record mark on the COMPILE file such that those decks following it, PLTDAT and FIND, are not ordinarily processed by the compiler.

The CSCMOD3 correction set made several modifications to the program, primarily to the active code. In the comdeck ACOBLK, the variable ZDANT was added to the end of the ACTIVE common block. The labeled common blocks TABDIR, READ1, UPDCAL, LGDE, STGT, TABCOM, CLEAUP, STORA, LGE, and ACTDIR were added to subroutine EXE, although the reason for these additions is unclear. None of the variables in these common blocks is required by EXE. In subroutine DEF, the variable NCASE was set to blanks with a DATA statement preceeding the FORMAT. This was done to clean up the printout on the first page of output which is printed before NCASE is defined. A DATA statement was added to subroutine LGEAR1 to provide initial values of zero to the arrays PGA1T1, PGA2T, QSV, QSVCU, and WFORT. A call to the comdeck ACOBLK was added to subroutine LGEA3C to make the ACTIVE common block available. In subroutine ALGEAR, the definitions of CMASNG, provided in CSCMOD2, were changed to the following.

```
CMASNG = 1+(AMASS*RX(1)*RX(1))/AIYYRS
```

Immediately prior to the definition of UNSPRNG, the variable UNSPR was set to zero, and the test inside the 66 loop, introduced by CSCMOD2, was replaced with the following.

```
IF(J.EQ.1 .AND. OMETD1(1).NE.0.0) UNSPR = MASS(1)
IF(J.GT.1 .AND. OMETD1(J).NE.0.0) UNSPRNG = UNSPRNG+MASS(J)
```

Below this, the definition of WFORT(1) was changed as follows.

WFORT(1) = SR(1)*(VMASS(1)-UNSPR)

In the continuation line of the definition of VELDEC, also provided by CSCMOD2, the variable AMASS was replaced with VMASS(1). In subroutine SDFLGP, the DATA statement for ACOVAR7, introduced by CSCMODS, was changed by replacing the 6HIDEACT with 5HZDANT and the first DATA statement for N1, N15, and N14 was removed. The following code was added beneath the write of N18, N1, and DAT2 to unit 13,

```

      IF(IABS(INDLG).NE.3) 110,115
110  IF(ISUM1.NE.0) WRITE(13) N14,ISUM1,DAT3,OP17
      GO TO 120
115  CONTINUE

```

and a CONTINUE statement labeled 120 was added after the write of ACOVAR3(3). The following code was inserted after the write of ETADES,

```

210  CONTINUE
      IF(ISTPL1.NE.0) WRITE(13) FT(1),SF(1),DELTA(1),P(1),P2(1),MA(1),
      *SD2(1,1),SD1(1,1),S(1,1),S2D2(1,1),S2D1(1,1),S2(1,1),OMETD1(1,1),
      *OMET(1,1)
      IF(ISTPL2.NE.0) WRITE(13) FT(2),SF(2),DELTA(2),P(2),P2(2),MA(2),
      *SD2(1,2),SD1(1,2),S(1,2),S2D2(1,2),S2D1(1,2),S2(1,2),OMETD1(1,2),
      *OMET(1,2)
      IF(ISTPL3.NE.0) WRITE(13) FT(3),SF(3),DELTA(3),P(3),P2(3),MA(3),
      *SD2(1,3),SD1(1,3),S(1,3),S2D2(1,3),S2D1(1,3),S2(1,3),OMETD1(1,3),
      *OMET(1,3)
      IF(ISTPL4.NE.0) WRITE(13) FT(4),SF(4),DELTA(4),P(4),P2(4),MA(4),
      *SD2(1,4),SD1(1,4),S(1,4),S2D2(1,4),S2D1(1,4),S2(1,4),OMETD1(1,4),
      *OMET(1,4)
      IF(ISTPL5.NE.0) WRITE(13) FT(5),SF(5),DELTA(5),P(5),P2(5),MA(5),
      *SD2(1,5),SD1(1,5),S(1,5),S2D2(1,5),S2D1(1,5),S2(1,5),OMETD1(1,5),
      *OMET(1,5)
      GO TO 230
220  CONTINUE

```

and a CONTINUE statement labeled 230 was added after the write of FORSST(5). Finally, in the call to STOVAR, introduced by CSCMODS, the variable FLOAT(IDEACT) was replaced by ZDANT.

The CSCMOD4 correction set, combined with a new deck, ACTNG, removed the nose gear active code from ALGEAR so that the nose gear could be treated independently from the main gears. Other minor changes were made to the active code as well as other parts of the program. The new deck was added after the ALGEAR deck and contained the following.

SUBROUTINE ACTNG

C ***** FATOLA VARIABLES *****
C

COMMON/DIRCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,
C DCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FYB7P,
C DUM4(3),FYB7P(4),FZB7P,DM5(17),GXB7F,DM6(8),GZB7F,
C DM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,
C INDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),
C QI77R(2),QI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),
C XG77F(2),XG77F1(12),YG77F(2),YG77F1(12),
C ZG77F(2),ZG77F1(2),DUM13(52),
C NSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEG,PGR,
C NTIRES(5),RZERO(5),V(5),DELTAM(5),MOMENT(5),
C RF(5),VZ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),
C A2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),
C CC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),PLT,NDELTA,
C ES(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)

C COMMON/DIPCOM/

C SD11(2),SD12(2),SD13(2),SD14(2),SD15(2),
C S1(2),SS2(2),S3(2),S4(2),S5(2),
C S2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),
C S2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),
C S21(2),S22(2),S23(2),S24(2),S25(2),
C OMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),
C OMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),
C AI(5),PI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,
C DDELT1,DDELT2,DDELT3,CDELT4,DDELT5,ISTAGE,
C PRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,
C DM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLY,

```
* NMODE,DM18(40),SXMOD(100),SYM0D(100),SZMOD(100),DM19(1686),
* GOD2(20),DDM20(20),DDM21(20),SLEN1(5),SLEN2(5),
* GAMA,DUM15(12),INDNWS,DDM22(10),ETADES,
* DDM23(5),AH(5),PH(5),DDM24(30)
```

```
C
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
```

```
C
COMMON/LGF/A11(5),A13(5),A31(5),A33(5),RRCGX(5),
C RL(3,3),RI(3,3,5),RAX(5),RAY(5),RAZ(5),TMP(3),ZZERO(5),
C XR(5),YR(5),EPSLON(5),PA(5),FDELTA(5),
C FTRZ(5),RDX(5),RDY(5),RDZ(5),RDXG(5),RDYG(5),RDZG(5),
C VTX(5),VTY(5),VTZ(5),GZ(5),          VGPT(5),FTRX(5),FTRY(5),
C DX(5),DY(5),DZ(5),FT(5),FDX(5),FDY(5),FF(5),AA(5),C2(5),
C SR(5),SF(5),PSKD(5),MUV(5),MTRX(5),MTRY(5),
C MTRZ(5),MA(5),RG11,RG13,RG31,RG33,IPRT,
C MTX,MTY,MTZ,SFTRX,SFTRY,SFTRZ,FTRA,
C FTRB,FTRC,SMTRX,SMTRY,SMTRZ
```

```
C
COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
```

```
C
*CALL AC0BLK
```

```
C
REAL MASS,MOMENT,MASS2,MUS,NTIRES,M8
REAL MUV(5),MTRX,MTRY,MTRZ,MA,MTX,MTY,MTZ
```

```
C
DIMENSION IPSTOP(5),AIC(5),PGA1T(5)
DIMENSION DELTA(5),DDELTA(5),DLGDE(47),
* SD2(2,5),SD1(2,5),S(2,5)
```

```
C
EQUIVALENCE (DLGDE(1),LA(1))
EQUIVALENCE (DELTA(1),DELTA1),(DDELTA(1),DDELTA1)
EQUIVALENCE (SD21(1),SD2(1,1)),(SD11(1),SD1(1,1)),(S1(1),S(1,1))
EQUIVALENCE (DM15(1),ITO)
EQUIVALENCE (DM5(16),GPEFF),
* (DM1(37),AIYYRS)
```

```
C
DATA AIC,IPSTOP /5*0.0,5*0/
```

```
C
O1(T1,T2)=SIGN(1.,(T1-T2))*SORT(ABS(T1-T2))
```

```
C*****
```

```
C
I = 1
```

```
C
IF(INDEACT(I).EQ.1) GO TO 56
IF(INDEACT(I).EQ.2) GO TO 80
IF(HMM(I) .EQ. 0.) GO TO 90
IF(OMRUN .GT. 0.0)GO TO 25
IF(ZDANT.GE.VELDEC) 90,40
```

```
25 IF(ZDANT.GE.VELDEC+XG77F1(1)*TAN(OMRUN)) GO TO 90
```

```
.40 WRITE(6,1014)TIME
```

```
1014 FORMAT (1H036H REDUCE CONTROL LIMIT FORCE AT TIME=,E16.8)
INDEACT(I)=1
```

```
56 IF(INDINT(I).EQ.0) GO TO 58
WLFOR(I)=WLFOR(I)-PESLPL(I)*HT
EPSILO(I)=EPSILO(I)+EPSLPL*HT
```

```

58 CONTINUE
   INDINT(I)=0
   IF(WLFOP(I) .GT. WLFORR) GO TO 90
   WRITE(6,1015)TIME
1015 FORMAT (1H027H CONTROL AT WLFORR AT TIME=,F16.8)
   INDEACT(I)=2
80  WLFOR(I)=WLFORR
   EPSILO(I)=EPSROL(I)
90  CONTINUE
C*****
C
   IF(KAPT(I).NE.0)GO TO 210
   APINT(I)=0.0
210 CONTINUE
   PGA1T(I)=PGA1T1(I)
   IF(PGA1T1(I).LE.-1600.0) PGA1T(I)=-1600.0
   VOL1T(I)=VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
   VOL3T(I)=VOL3I(I)+AREA3(I)*S(1,I)
   VOL2T(I)=VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)+(VOL3T(I)
X -VOL3I(I))-VCUM(I)
   PGA2T(I)=AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMA)-PATM
   IF(SD1(1,I) .EQ. 0.0)GO TO 104
   PGA3T(I)=((COEF3(I)*AREO3(I))*2*PGA2T(I)-SD1(1,I)/ABS(SD1(1,I))
X *(SD1(1,I)*AREA3(I))*2)
E/((COEF3(I)*AREO3(I))*2)
   GO TO 105
104 PGA3T(I)=PGA2T(I)
105 IF(PGA1T(I) .GE. PGA2T(I))GO TO 106
   GO TO 107
106 GAMAH(I)=RHOH*GREFF*(1.0+(PGA1T(I)*3.04E-08)-
X (PGA1T(I)**2*2.72E-15))
   GO TO 108
107 GAMAH(I)=RHOH*GREFF*(1.0+(PGA2T(I)*3.04E-08)-
X (PGA2T(I)**2*2.72E-15))
108 IF(PGA1T(I) .GE. PGA2T(I))COEFO(I)=
X CDMOC(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
   IF(PGA2T(I) .GT. PGA1T(I))COEFO(I)=
X CDMOE(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
   IF(IMODE(I).NE.0 .OR. DDELTA(I).GT.0.0) GO TO 113
   QO(I) = 0.0
   GO TO 109
C
113 QO(I)=COEFO(I)*(AREMD(I)-APINT(I))*Q1(PGA1T1(I),PGA2T(I))
   IF(QO(I).LT.0.0 .AND. VCUM(I).LE.0.0) GO TO 102
   AIC(I)=0.0
   GO TO 103
102 IF(PGA1T1(I).LT.PGA2T(I)) GO TO 103
   QO(I)=0.0
   VCUM(I)=0.0
   AIC(I)=1.0
103 IF(QO(I).GT.0.0) AIC(I)=0.0
109 IF(PGA2T(I) .LE. -1600.0)PGA2T(I)=-1600.0
   IF(PGA3T(I) .LE. -1600.0)PGA3T(I)=-1600.0
   IF(DELTA(I) .LE. 0.0 .AND. TIME .GT. DELT)GO TO 101
   GO TO 110

```

```

101  FFORT(I)=0.0
    GO TO 140
110  CONTINUE
C    COMPUTE STRUT AXIAL BINDING FRICTION FORCE
    BLFORT(I)=FONHST(I)*((SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))+1.0)
    BUFORT(I)=FONHST(I)*(SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))
    FFORT(I)=BUMU(I)*ABS(BUFORT(I))+BLMU(I)*ABS(BLFORT(I))
140  CONTINUE
C    COMPUTE SHOCK STRUT CHARGING FORCE
    IF(S(1,I) .GT. 0.0)GO TO 142
    FORCHT(I)=PGA1T(I)*AREA1(I)+PGA2T(I)*(AREA2(I)-AREA1(I))-PGA3T(I)
X   *AREA3(I)+FFORT(I)+CFFOR(I)
C    COMPUTE NORMAL AND AXIAL HUB TO SHOCK STRUT FORCES AT HUB
142  FONHST(I)=SQRT(FDX(I)**2+FDY(I)**2)-MASS(I)*GREFF*SIPA+SBFOT
    IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 150
    GO TO 801
150  CONTINUE
    FORSST(I)=FT(I)
    SD1(1,I)=0.0
    IF(IMODE(I).EQ.0) GO TO 297
    ISTROK(I)=1
    GO TO 289
C    COMPRESSION VELOCITY OF SHOCK STRUT IS POSITIVE
801  1 (SD1(1,I).LE.0.8 .AND. IFR(I).EQ.0) GO TO 2
    GO TO 3
    2  DMTANH(I)=1.0
    GO TO 284
    3  DMTANH(I)=ABS(TANH(2.0*SD1(1,I)))
    IFR(I) = 1
284  IF(S(1,I).LT.0.0) GO TO 160
    GO TO 161
160  IPSTOP(I)=1
161  IF(S(1,I).LE.DSTOP .AND. IPSTOP(I).EQ.1) GO TO 900
    GO TO 902
900  IF(S(1,I).LE.0.005) GO TO 903
    GO TO 904
903  SD2(1,I)=0.0
    SD1(1,I)=0.0
    S(1,I)=0.0
    IPSTOP(I)=0
    GO TO 902
904  CONTINUE
    IF(IFSTOP(I) .NE. 0)GO TO 906
    DSTOP=S(1,I)
    FSTOPK=2.0*MASS(I)*SD1(1,I)**2/DSTOP**2
906  IF(S(1,I) .LE. DSTOP/2.0)GO TO 908
    FSTOP(I)=-FSTOPK*(DSTOP-S(1,I))
    GO TO 909
908  FSTOP(I)=-FSTOPK*S(1,I)
909  IFSTOP(I)=1
    IFR(I) = 0
    GO TO 901
902  FSTOP(I)=0.0

```

```

901 IF(PGA1T(I).LE.(PGA1I(I)+500.0) .AND.
: PGA1T(I).GT.(PGA1I(I)-500.0)) GO TO 158
GO TO 159
158 IF(ABS(FT(I)).LE.FORCHT(I) .AND. S(1,I).EQ.0.0) GO TO 500
159 IF(S(1,I).GE.0.0) GO TO 470
IF(SD1(1,I) .LT. 0.0)GO TO 470
GO TO 471
470 FFORT(I)=-FFORT(I)
CFFOR(I)=-CFFOR(I)
471 FORSST(I)=-((PGA1T(I)-PGA2T(I))*(AREA1(I)-APINT(I))
& +PGA2T(I)*AREA2(I)
X -PGA3T(I)*AREA3(I)+(FFORT(I)
1 +CFFOR(I))*DMTANH(I)+FSTOP(I))
500 IF(INDFLX.GE. 1)GO TO 295
IF(IMODE(I).EQ.0) GO TO 297
ISTROK(I)=1
289 IF(S(1,I) .GT. 0.0) GO TO 295
IF(IOPCO(I) .EQ. 1) GO TO 295
IF((PGA1I(I)-1000.0).LT.PGA1T(I).AND.
& PGA1T(I).LT.(PGA1I(I)+1000.0))299,298
299 IF(XVALVE(I) .NE. 0.0)GO TO 311
IF(IPASS(I) .EQ. 1)GO TO 296
XVALVE(I)=XKSV(I)*XMA11(I)+XBIAS(I)
IPASS(I)=1
GO TO 294
298 IF(ICOSV(I) .EQ. 1)GO TO 291
IOPCO(I)=0
IF(XSV(I) .LT. 0.002 .AND. XSV(I) .GT. -0.002)291,295
291 IF(SD2(1,I) .LE. 0.0 .AND. ICOSV(I) .EQ. 1)GO TO 311
IF(IOPCO(I) .EQ. 1)GO TO 295
IF(PGA1T(I) .LE. PGA1I(I)) GO TO 293
IF(IXSVL(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDMMN(I)*DELT*PERCNT(I)
IF(XVALVE(I) .GT. -0.1) GO TO 294
XVALVE(I)=-0.1
IXSVL(I)=1
GO TO 294
293 IF(IXSVH(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDMMX(I)*DELT*PERCNT(I)
IF(XVALVE(I) .LT. 0.1) GO TO 294
XVALVE(I)=0.1
IXSVH(I)=1
294 CONTINUE
DLTX1D(I)=0.0
ICOSV(I)=1
296 IF(WFORT(I) .GT. 0.0 .AND. S(1,I) .LE. 0.0)GO TO 410
311 IF(NAC(I) .EQ. 1)GO TO 307
IF(IIIXSVH(I) .EQ. 1)GO TO 305
XVALVE(I)=XVALVE(I)+XSVDMMN(I)*DELT*PERCNT(I)

```

```

305 IF(XVALVE(I) .LE. 0.0)305,400
    XVALVE(I)=0.0
    IIXSVH(I)=1
    GO TO 400
307 IF(IIXSVL(I) .EQ. 1)GO TO 308
    XVALVE(I)=XVALVE(I)+XSVDX(I)*DELT*PERCNT(I)
    IF(XVALVE(I) .GE. 0.0)308,400
308 XVALVE(I)=0.0
    IIXSVL(I)=1
400 CONTINUE
410 IF(XVALVE(I) .NE. 0.0)GO TO 295
    ICOSV(I)=0
    DELTX1(I)=DELT*(I)*XKF(I)
    XMA(I)=(DF(I)+DELT*(I))*XKA(I)
    XMA11(I)=XMA(I)
    XSV(I)=XKSV(I)*XMA11(I)+XBIAS(I)
    CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
    IPASS(I)=0
    ICSV(I)=0
    IIXSVL(I)=0
    IIXSVH(I)=0
    IIXSVL(I)=0
    IIXSVH(I)=0
    IOPCO(I)=1
    CALL PHLOZ2(PS(I),PR(I),XSV(I),QC(I),XLPSV1(I),XLPSV3(I),PCLSV(I),
    & CSV(I),CSV1(I),CSV3(I),XMU,QTOLER,NITER,P1(I),QS1(I),QS3(I))
C
295 IF(ISTROK(I) .EQ. 1 .AND. S(1,I) .GT. 0.0)IOPCO(I)=0
.297 I2 = 2*I+NSTRUT-1
    I1 = I2+1
    CALL INTEG(LA(I2),SD2(1,I))
    CALL INTEG(LA(I1),SD1(1,I))
C
    IF(IMODE(I).EQ.0) GO TO 450
C
    FNUP(1) = 0.5*AIYYBS*QI77R(1)*QI77R(1)+(ENCG/CMASNG)*
    * (ZG77F1(1)/ABS(ZG77F1(1)))
    IF(HMM(I) .EQ. 1.0)GO TO 130
    SA(I)=0.
    IF(WFORT(I).LT.0.) XSTOT(I)=ENUP(I)/((-WFORT(I))*COPA)
    IF(WFORT(I).GE.0.0 .OR. DDELTA(I).LT.0.0) XSTOT(I)=1.E20
C
    ZSSC IS A PERCENTAGE OF SB(I) FOR ACTIVATING CONTROL-CDMOC(I) IS U
    ZSSC(I)=FWORK(I)*SB(I)
    IF(XSTOT(I).LE.(ZSSC(I)-S(1,I)) .OR. PESA(I).EQ.1.0) SA(I)=1.0
    RESA(I)=SA(I)
    IF(SA(I).EQ.0. .OR. HMM(I).EQ.1.) GO TO 130
    WLFOR(I)=-WFORT(I)
C
    VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/
    & (VMASS(1)*REDSLP(I))

```

```

WRITE(6,121)TIME,WLFOR(I),VELDEC
121  FORMAT(50H ACTIVE CONTROL INITIATED...TIME, WLFOR, VELDEC = ,
1      3E13.5)
HMM(I)=1.
130  IF(S(1,I) .GT. 0.0) ISET(I)=0
      IF(HMM(I).EQ.0.) GO TO 451
      IF(-WFOR(I).GT.(WLFOR(I)+EPSILO(I))) DF(I)=(WLFOR(I)+
& EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LT.(WLFOR(I)-EPSILO(I))) DF(I)=(WLFOR(I)-
& EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LE.(WLFOR(I)+EPSILO(I)).AND.
& -WFOR(I).GE.(WLFOR(I)-EPSILO(I)))
: 457,456
457  IF(S(1,I) .LE. 0.0) GO TO 456
      IF(WFOR(I) .GT. 0.0 .AND. QSVCU(I) .LT. 0.0) 454,455
454  DF(I)=WLFOR(I)-(-WFOR(I))
      GO TO 456
455  DF(I)=0.0
456  DELTX(I)=S(1,I)-XSCOM(I)
      IF(S(1,I) .LE. 0.0 .AND. ISET(I) .EQ. 0) GO TO 451
      GO TO 452
451  DF(I)=0.
      DELTX(I)=0.
452  XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
      IF(GNR.EQ.1. .AND. XMA(I).GT.0.) XMA(I)=
: XMA(I)*SQRT((PGA1T(I)-PGALAC(I))
X /(PGAHAC(I)-PGA1T(I)))
C
C      NOTE: SUBROUTINE 'FLOZE2' COMPUTES THE FLOWS FROM THE PRESSURES
C      IN UNITS OF INCHES.
C      P1(I)=PGA1T(I)/144.
C      COMPUTATION OF HIGH PRESSURE ACCUMULATOR NITROGEN VOLUME
C      AND ACCUMULATOR PRESSURE
VOLANT(I)=VOLANT(I)+OSVN(I)*DELT-OPUMPS(I)*DELT
PS(I)=(((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMA)-PATM)/144.0
IF(PS(I) .GE. 3000.0) 464,465
464  PS(I)=3000.0
      VOLANT(I)=VOLANI(I)
465  VOLAHT(I)=VOLACI(I)-VOLANT(I)
      IF(VOLAHT(I) .LE. 0.0) 466,467
466  WRITE(6,1050)TIME
1050  FORMAT(1H0//45H ACCUMULATOR OIL VOLUME INSUFFICIENT AT TIME=,E16.8
1      //)
      CALL LGEAR6
      STOP 500
467  CONTINUE
      CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
& YSV(I),QS1(I),QS3(I),CSV1(I),CSV3(I),XMU)
      OSV1(I)=QS1(I)/1728.

```


OSV3(I)=OS3(I)/1728.

```

C
450 CONTINUE
   OSV(I)=OSV1(I)-OSV3(I)
   IF(OSV(I) .LT. 0.0)NAC(I)=1
   IF(OSV(I) .GT. 0.0)NAC(I)=2
   IF(NAC(I) .NE. 2) GO TO 462
   QSVN(I)=OSV(I)
   GO TO 463
462 QSVN(I)=0.0
463 IF(SD1(1,I).LT. 0.0 .AND. PGA1T(I).LE. -1600.0)PGA1T(I)=-1600.0
C
   RETURN
   END

```

In the comdeck ACOBLK, the variable IDEACT in the ACTIVE common block was changed to INITSW, the array IFRI was changed to IFR, the variable VELDEC was changed to ENCG, and the array INDEACT(5) and the simple variable VELDEC were added at the end. In subroutine EXE, the variable TPD was changed to TIME in the test below statement 412 and the return was changed to

STOP "EXECUTIVE ROUTINE"

The following declaration statements were added to subroutine ACTINIT to make the variables S

```

DIMENSION S(2,5)
EQUIVALENCE (S1(1),S(1,1))
EQUIVALENCE (DM15(1),ITO)

```

and ITO available, and the DATA statement introduced by CSCMOD2 was changed by substituting IFR for IFRI. The variable INITSW was

initialized to unity following the definition of OMRUN, and the following code was inserted after the definition of VMAS(I).

```
INDEACT(I) = 0
IF(ITO.EQ.1) INDEACT(I) = 2
```

The call to SETUP was changed to a call to ALGEAR. In subroutine ALGEAR1 the following declaration statement was added to make the variable ITO available.

```
EQUIVALENCE (DM15(1),ITO)
```

The call to LGEA3C was replaced with the following code,

C

```
IF(INITSW.EQ.1) GO TO 16
```

and the following code was inserted after the definition of SIPA.

C

```
CALL LGEA3C
```

C

Statement number 18, which had been modified by CSCMOD2, was changed as follows.

```
18 CALL ACTNG
```

C

```
DO 274 I=2,NSTRUT
```

The two tests on I and ZDANT introduced by CSCMOD2 were removed as

were the lines of code running from the definition of UNSPR through statement number 67. The DO 65 statement introduced by CSCMOD2 was removed and in statement number 801, IFR was substituted for IFRI, with a similar substitution in the statement below the statement labeled 3. The four calls to VIRK4 were removed and IFR was substituted for another IFRI below statement 909. The first four lines of code to modify ENUP(I), which had been introduced by CSCMOD2, were removed, as was statement number 120. The continuation line in the definition of VELDEC(I), which had been modified by CSCMOD3, was further modified as follows.

```
6   VMASS(I)*REDSLP(I))
```

The following code was inserted below statement number 274,

```
C   DO 65 I=1,NSTRUT
```

the SETUP entry point was replaced with the following,

```
16 CONTINUE
   INITSW = 0
```

and the definition of II below statement number 20 was removed. In subroutine SDFLGP, the DATA statement for ACOVAR9 was changed by replacing 4HIFRI with 3HIFR. The DATA statement for N19 was replaced with the following,

```
DATA N20 /20/
```

and the write to unit 13 was changed by replacing N19 with N20 and by adding ACOVAR3(3) to the list. The five write statements to unit 13 below statement 220 were modified by adding the appropriate element of FORSST to each. Finally, the call to STOVAR was changed by substituting IFR for IFRI.

The CSCMOD5 correction set contained the final corrections necessary to make the active code operational. In the ACOBLK common deck the ACTIVE common block was modified by replacing the variables COEF and GAMAN with ALGDUM1 and ALGDUM2, respectively, replacing the simple variable VELDEC with the array VELDEC(5), and by adding the two arrays COEF1(5) and LMODE(5). In subroutine EXE, the variable TIME was changed back to TPD in the test below statement 412, cancelling the effect of the CSCMOD4 change. In subroutine MIMIN a problem with the flight time limit was corrected by changing the test below statement 45 to the following.

```
IF((XF-XO).GT.1.E-10) GO TO 211
```

The DATA statement for NCASE in subroutine DEF was removed. In subroutine ACTINIT the 14th continuation line in the second declaration statement for the common block DIRCOM was changed to the following,

```
*,DDM21(20),SLENDUM(10),GAMA,DUM15(12),INDNWS,DDM22(10),ETADES,
```

and the LGDE common block was removed. The DATA statement for IOPCO was changed to the following.

```
DATA IOPCO,ISET,ISTROK,NAC /10*1,10*0/
```

The following new DATA statements were added

```
DATA INDEACT,INDINT,INITSW /5*0,5*1,1/
DATA FFORT,FONHST,FORCHT,FORSST,WFORT /25*0.0/
DATA QC,QO,QSV1,QSV3,QSVCU,QTOLER /25*0.,0.0001/
DATA DCON,DMTANH,DF,DP1,DELTX,DELTX1,DLTX1D /6*1.0,25*0.0/
DATA XSVDDT,XSVDD,XSVDDD /15*0.0/
DATA REDSLP,SBFOT,VCUM,VELDEC /5*100000.,0.,10*0.0/
DATA XMA1,XMA2,XMA3,XMA4,XMA6,XMA7,XMA9,XMA10 /40*0.0/
DATA XMA,XMA5,XMA8,XMA11 /20*0.0/
```

Assignment statements for the variables DCON, QTOLER, SBFOT, VELDEC, INITSW, REDSLP(I), INDINT(I), ISTROK(I), NAC(I), QSVCU(I), DF(I), DELTX(I), DELTX1(I), DLTX1D(I), XMA(I), XMA5(I), XMA8(I), XMA11(I), XMA1(I), XMA2(I), XMA3(I), XMA4(I), XMA6(I), XMA7(I), XMA9(I), XMA10(I), XSVDD(I), XSVDDD(I), XSVDDT(I), DP1(I), COEF, CSV1(I), and CSV3(I) were removed. The following code was added after the definition of COEF3(I),

```
COEF1(I) = CDSV(I)*SORT(2.*GREFF/GAMAH(I))*144.
CSV1(I) = COEF1(I)*WSV1
CSV3(I) = COEF1(I)*WSV3
```

and the definition of QC(I) was changed to the following.

```
QSV1(I) = 0.0
QSV3(I) = 0.0
IF(ITO.EQ.1) GO TO 60
IF(IMODE(I).EQ.0) GO TO 80
```

The following new code was added beneath the second definition of QSV3(I).

GO TO 80

C

```

60 CONTINUE
  HMM(I) = 1.0
  INDEACT(I) = 2
  LMODE(I) = IMODE(I)
  IMODE(I) = 0
  AP2TO(I) = PGA2I(I)+PATM
  VOL1T(I) = VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
  VOL3T(I) = VOL3I(I)+AREA3(I)*S(1,I)
  VOL2T(I) = VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)
  + (VOL3T(I)-VOL3I(I))-VCUM(I)
  PGA2I(I) = AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMA)-PATM
  PGA1I(I) = PGA2I(I)
  PGA3I(I) = PGA2I(I)
  FORSST(I) = -(PGA2I(I)*AREA2(I)-PGA3I(I)*AREA3(I)
  + DMTANH(I)*((FFORT(I)+CFFOR(I))+FSTOP(I)))
  WFORT(I) = FORSST(I)

```

C

80 CONTINUE

Assignment statements for the variables QO(I), VCUM(I), FFORT(I), FORSST(I), WFORT(I), FONHST(I), FORCHT(I), and INDEACT(I) were removed. The test on ITO was removed, and the call to ALGEAR, from CSCMOD4, was changed to a call to ALGEAR1. In subroutine ALGEAR1, the variable XCGRF was made available by the following statement.

```
EQUIVALENCE (DM8(79),XCGRF)
```

The DIMENSION and DATA statements involving the variable IPSTOP were changed as follows.

```

DIMENSION IPSTOP(5),AIC(5),PGA1T(5)
DATA AIC,IPSTOP /5*0.0,5*0/

```

The definition of CMASNG was changed to the following.

```
PXCG1 = PX(1)-XCGRF  
CMASNG = 1+(AMASS*PXCG1*PXCG1)/AIYYBS
```

and in the definition of ZDANT, the variable QI77R was changed to the array element QI77R(1). The statement labeled 18 was changed to the following.

```
UNSPRNG = 0.0  
DO 22 I=1,NSTRUT  
  IF(ITC.EQ.1 .AND. TIME.GE.2.0) IMODE(I) = LMODE(I)  
  IF(OMETD1(I).NE.0.0) UNSPRNG = UNSPRNG+MASS(I)  
22 CONTINUE  
C  
  DWFORT = (-SQRT(AXP7F*AXP7F+AYP7F*AYP7F+AZP7F*AZP7F)+GREFF)*  
  (AMASS-UNSPRNG)  
C  
  CALL ACTNG  
C
```

The following code was added at the top of the 274 loop.

```
C  
C  
WFORT(I) = DWFORT/(NSTRUT-1)
```

The four occurrences of VELDEC were changed to VELDEC(I), and the two definitions of II, left over from earlier VIRK4 calls, were removed. Statement 296 was changed by modifying the branch from 410 to 400, and the statement label of 410 was eliminated. The test on IMODE below statement 297 was changed as follows.

```
IF(IMODE(I).NE.0.0) GO TO 119  
OSV1(I) = 0.0  
OSV3(I) = 0.0  
GO TO 450
```

The continuation line in the definition of VELDEC(I) was corrected as follows.

• (VMASS(I)*REDSLP(I))

In subroutine ACTNG, the DIMENSION statement for SD2 was expanded to include OMETD1(2,5), and OMETD1 was equivalenced to OMTD11. The four occurrences of VELDEC were changed to VELDEC(I), and the following code was inserted below statement 90.

```
UNSPRNG = 0.0  
IF(OMETD1(1).NE.0.0) UNSPRNG = MASS(1)  
WFORT(1) = SR(1)*(VMASS(1)-UNSPRNG)
```

The branch in statement 296 was changed from 410 to 400 and the 410 label was eliminated. The test on IMODE below the statement labeled 297 was changed to the following.

```
IF(IMODE(I).NE.0.0) GO TO 119  
QSV1(I) = 0.0  
CSV3(I) = 0.0  
GO TO 450  
119 CONTINUE
```

In subroutine DECOMP, the array IPS was moved from blank common to labeled common, IPSCOM, with an identical change in subroutine SOLVE. Finally, extensive comments were added to subroutine ACTIN, but no changes were made to the executable code.

The correction set PINARY was primarily intended to introduce logic which would allow the area of the metering pin to vary as a function of strut stroke. In the comdeck ACOBLK, the ACTIVE common block was expanded to include the four arrays PINN(30), PINM(30), STRON(30), and STROM(30), and in subroutine EXE, the arrays NAME and LOC in the ACTDIR common block were expanded from 71 to 75. In subroutine ACTINIT, the following new code was inserted after the initialization of QSV3.

```
IF(KAPT(I).EQ.1) GO TO 50
APINT(I) = 0.
IF(KAPT(1).EQ.0) GO TO 50
IF(I.GT.1) GO TO 25
STROK = -1.
DO 10 J=1,29
  IF(STRON(J).LT.STROK) GO TO 15
  STROK = STRON(J)
  IF(S(1,I).GE.STRON(J).AND.S(1,I).LE.STRON(J+1)) GO TO 20
10 CONTINUE
15 J = J - 1
20 APINT(I) = PINN(J)
  GO TO 50
25 STROK = -1.
  DO 30 J=1,29
    IF(STROM(J).LT.STROK) GO TO 35
    STROK = STROM(J)
    IF(S(1,I).GE.STROM(J).AND.S(1,I).LE.STROM(J+1)) GO TO 40
30 CONTINUE
35 J = J - 1
40 APINT(I) = PINM(J)
50 CONTINUE
```

In both ALGEAR1 and ACTNG the three lines of code beginning with a test on KAPT(I) were removed. In subroutine ACTIN, the arrays XNAME and LOC in the ACTDIR common block were expanded from 71 to 75, and the arrays DATA and IDATA, representing the ACTIVE common block were expanded from 656 to 802. The following code was inserted as the

first executable statements.

```
C
C      ZERO OUT PIN AND STROKE ARRAYS
C
      DO 10 J=683,802
      DATA(J) = 0.
10 CONTINUE
```

The FORMAT statement labeled 1 was changed by replacing I1 with I2, and the upper limit on the 110 loop was changed from 71 to 75. In the DIRACT block data, the NAME and LOC arrays were enlarged from 71 to 75. The final continuation card in the DATA statement for NAME was changed to the following,

```
5 6HZETAC1, 6HZETAC2, 6HIMODE , 6HPINN , 6HPINM ,
6 6HSTRON , 6HSTROM /
```

and the final continuation card in the DATA statement was changed as follows.

```
7      644, 645, 651, 683, 713, 743, 773/
```

The KLUGEZ correction set was intended to prohibit secondary piston calculations for the nose gear. The following code was inserted at the top of the 100 loop in subroutine LGEAR1.

```
C
C*****
C
C***** TEMPORARY FIX TO FREEZE NOSE SECONDARY
C
      IF(I.EQ.1) GO TO 59
C
C*****
C
```

The correction set SECFIX was meant to provide constraints prohibiting over-extension or over-compression of the secondary piston in a manner similar to earlier corrections made by SDD for the main piston. In subroutine LGEAR1, the following code was added after statement 57.

```
S2(1,I) = -0.5*ES2(I)
S2D1(1,I) = -1.E-10
S2D2(1,I) = -1.E-10
```

The two lines of code beginning with statement 61 were replaced with the following,

```
61 IF(S2D2(1,I).LT.0.) GO TO 140
   IF(S2D1(1,I).LT.0.) S2D1(1,I) = 0.
```

and the following line was added below the statement labeled 140.

```
S2D1(1,I) = 0.
```

For debugging purposes, the following code was inserted in subroutine LGEA3C below the calculation of TMP(1).

```
IF(TMP(1).LT.0.) WRITE(6,1234) I,S2(1,I),S2D1(1,I),
1 S2D2(1,I)
1234 FORMAT(1X,7H*--*--*,I5,3E16.8)
IF(TMP(1).LT.0.) GO TO 31
```

The TABFIX correction set corrected a table look-up problem associated with the secondary piston. In subroutine LGEA3C, the

last two calls to HIHO were changed by substituting S2(1,I) for S(1,I).

The correction set CSCMOD6 was intended to introduce a new variable, HT3, to be used from anywhere in the program to limit the integration step size in critical conditions and to correct a problem with the reduction of the control limit force. The HTCOM common block was modified in subroutines EXE, MIMIN, LGEAR1, and ALGEAR1 as follows.

```
COMMON/HTCOM/HT,HT1,HT2,HT3,INDINT(5),H
```

The ACTDIR common block was removed from subroutine EXE where it was not needed. The initialization of INDINT in subroutine MIMIN was changed from a 10 loop to a 445 loop and moved to beneath statement 40. In subroutine LGEAR1, the following code was introduced beneath statement 4,

```
INTFLAG = 0
```

and the call to INUPD was changed to the following.

```
IF(INTFLAG.EQ.0) CALL INUPD(NDEQ,LA)  
INTFLAG = 1
```

In subroutine ALGEAR1, the definition of WLFOR(I) above statement 56 was changed by substituting H for HT.

The SECFIX1 correction set made a minor addition to the SECFIX changes to control the secondary piston over-compression. The following statement was added to subroutine LGEAR1 below statement 140

S2(1,I) = 0.

Additional modifications to provide for a stroke dependent metering pin area were provided in the PINARYX correction set. the following new code was inserted in ALGEAR1 below the initialization of the RL array.

```

DO 760 I=1,NSTRUT
IF(KAPT(I).EQ.1) GO TO 750
APINT(I) = 0.
IF(KAPT(I).EQ.0) GO TO 750
IF(I.GT.1) GO TO 725
STROK = -1.
DO 710 J=1,29
IF(STRON(J).LT.STROK) GO TO 715
STROK = STRON(J)
IF(S(1,I).GE.STRON(J).AND.S(1,I).LE.STRON(J+1)) GO TO 720
710 CONTINUE
715 J = J - 1
720 APINT(I) = PINN(J)
GO TO 750
725 STROK = -1.
DO 730 J=1,29
IF(STROM(J).LT.STROK) GO TO 735
STROK = STROM(J)
IF(S(1,I).GE.STROM(J).AND.S(1,I).LE.STROM(J+1)) GO TO 740
730 CONTINUE
735 J = J - 1
740 APINT(I) = PINM(J)
750 CONTINUE
760 CONTINUE

```

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The CSCMOD6 correction set was refined with the addition of the HTRY mods. In subroutine EXE, the new variable HT3 was initialized to HT above statement 302. In subroutine MIMIN, HT3 was added to the list in the AMIN1 statement labeled 30 and the statement was moved to below statement 40.

A restart capability was added to the program with the REST correction set which placed most all local variables into labeled common blocks and declared all common blocks in the main program to assure contiguous storage of all program variables. The unit TAPE7 is used to hold the restart information. In program TOLA, TAPE7 was added to the program card. The following new code was added below the READ1 common declaration,

```
COMMON/STOPIT/DM2(2)
COMMON/DIRCOM/DM3(4059)
COMMON/TABSRC/DM4(110)
COMMON/EXEAUT/DM5(9)
COMMON/LG/DM6(7)
COMMON/AUTSC/DM7(40)
COMMON/AUTPRC/DM8(63)
COMMON/LGAUTS/DM9(14)
COMMON/FLXOP/DM10(608)
COMMON/AUTSAC/DM11(6)
COMMON/HTCOM/DM12(10)
COMMON/CONTRD/DM13(4)
COMMON/UPDCAL/DM14(181)
COMMON/LGDE/DM15(72)
COMMON/STGT/DM16(10)
COMMON/TABCOM/DM17(230)
COMMON/CLEAUP/DM18(3)
COMMON/STORA/DM19(67)
COMMON/LGE/DM20(299)
COMMON/ACTIVE/DM21(802)
COMMON/IPSCOM/DM23(20)
COMMON/XEXE/DM24(3),BLANK,STCOM2,STCOM3,DM25(5)
COMMON/XMIMIN/DM26(908)
COMMON/XARRAY/DM27
COMMON/XTFFS8/DM28(2)
COMMON/XSACS1/DM29(106)
COMMON/XCPT1/DM30(97)
COMMON/XLGER1/DM31(21)
```

```
COMMON/XLGE3C/DM32(12),IFRI(5),DM33(19)
COMMON/XALGEA/AIC(5),DM34(14),IPSTOP(5),DM34X(10)
COMMON/XACTNG/AICX(5),IPSTOX(5),DM35(6)
COMMON/XFLEX1/DM36(2077)
COMMON/XSDFLG/DM37(179)
COMMON/XAUTS/DM38(8)
COMMON/XFLARE/DM39(20)
COMMON/XAUTPR/DM40(4)
COMMON/RESTR/IRST
```

and the following code was inserted just above the first executable statement.

```
DATA STCOM2,STCOM3,BLANK/4HTMAX,5HSTAGE,6H
DATA IFRI/0,0,0,0,0/
DATA AIC,IPSTOP/5*0.0,5*0/
DATA DM34Y,DM34Z,DM34A/5*0.,5*31000.,65*0./
DATA AICX,IPSTOX/5*0.0,5*0/
```

The following read sequence was placed immediately before the call
to EXE.

```
READ(5,2) IPST
2 FORMAT(I1)
```

In subroutine EXE, the following declaration statements were added,

```
COMMON/XEXE/END,SWT2,SWT3,BLANK,STCOM2,STCOM3,
1 MIM,NDEFS,TIMEA,TIMEP,TPD
COMMON/RESTR/IRST
DIMENSION DMD1(1)
EQUIVALENCE(DMD1(1),TABLE(1))
```

and the DATA statement was removed. The FORMAT labeled 7, which was not referenced, was replaced with the following.

```
5 FORMAT(1H0,16X,*JOB RESTARTED AT*,E16.8)
7 FORMAT(1H0,16X,*RESTART TAPE WRITTEN AT*,E16.8)
```

The following statement was inserted as the first executable statement,

```
IF(IRST.GT.1) GO TO 800
```

and the new code below was inserted at the top of the staging logic below statment 416.

```
IF(IRST.EQ.1.OR.IRST.EQ.3) GO TO 810
GO TO 511
800 CONTINUE
READ(7) (DMD1(IJ),IJ=1,11002)
CALL LINES(2)
WRITE(6,5) TIMES
GO TO 511
810 CONTINUE
WRITE(7) (DMD1(IJ),IJ=1,11002)
CALL LINES(2)
WRITE(6,7) TIMES
```

Local variables in subroutine MIMIN were transferred to a labeled common block, XMIMIN, as follows.

```
COMMON/XMIMIN/ACH,ERR,J,K,KF,PO,R,S,XF,XK,XO,
1 YMAX,YP,YO,Y1,Z
```

The variable NMAX in subroutine ARRAY was transferred to common XARRAY, and in subroutine TFFS8, the variables TH1 and TH2 were put in common block XTFFS8. The local variables in SACS1 and OPT1 were relocated to common blocks XSACS1 and XOPT1 as follows.

```
COMMON/XSACS1/AERO2,BETADE,HG,IGO1,IGO2,IGO3,
1 TC,TMP,CN
COMMON/XOPT1/ALPHD1,B,BETAD1,C,D,DRAGC,E,F,FDC,
1 G,H,INDER,J,K,L,LA,LTHTRR,M,MDX,MDY,MDZ,P,THTRP,
2 THTRR1,TMP
```


In subroutine LGEAR1, local variables were moved to XLGER1,

```
COMMON/XLGER1/BFX,BFY,BFZ,BLM,BMM,BNM,DFXM,DFYM,  
1 DFZM,DLM,DMM,ERR,IL4,INTFLAG,I1,I2,NBB,NBH,NDEQ,  
2 THETAR,TTIME  
3 TEMP3,TMPETA,YAWPMH,YAWPRM,DELNM
```

and the DATA statement for PGA1T1 was removed. Local variables in
LGEA3C were moved to XLGE3C,

```
COMMON/XLGE3C/CRNPRM,CRNPWR,DELMN,DFTRX,DFTRXM,  
1 DFTRY,DFTRYM,ETAVE,ETAVEM,FGPYM,FI,HYPTAN,IFRI,  
2 ILP,IL2,IL3,ND,NDO,NOD,SIDEMU,SIDMUM,TEMP1,TEMP2.
```

and the DATA statement for IFRI was removed. In ALGEAR1 local
variables were relocated to XALGEA,

```
COMMON/XALGEA/AIC,BFX,BFY,BFZ,BLM,BMM,BNM,DFXM,  
1 DFYM,DFZM,DLM,DMM,DNM,DWFOR,IL4,IPSTOP,NBB,NBH,  
2 PGA1T,RXCGL,TTIME,UNSPRNG
```

and in ACTNG they were moved to XACTNG

```
COMMON/XACTNG/AIC,IPSTOP,UNSPRNG,NIN,TIMEL
```

and the DATA statement for AIC was removed. In subroutines FLEX1,
SDFLGP, AUTS, FLARE1, and AUTPR1, the following common blocks were
created, respectively.

```
COMMON/XFLEX1/COEF,COPMAS,CTMP1,DIFF,FDC,GF,  
1 GFORC1,GMASS,GSMOD,GTF,HDR,IG,II,LA,OMXD1M,  
2 OMYD1M,OM7D1M,PTN,OS,OS1,RKSY,RVAR,RXA13,RZA11,  
3 SDO,SD1,SD2,SMASS,SUM1,SUM3,TITL,VARY1  
COMMON/XSDFLG/ACOVAR1,ACOVAR2,ACOVAR3,ACOVAR4,  
1 ACOVAR5,ACOVAR6,ACOVAR7,ACOVAR8,ACOVAR9,DAT1,
```

```
2 DAT2,DAT3,DAT4,DAT5,ISUM1,ISUM2,LASTPT,NHL,N1,  
3 N14,N15,N18,N20,OP16,OP17,OP18,OP19,OP20,OP21  
COMMON/XAUTS/DELPI,DELRDI,GAMPPR,SWT2,TMP1,  
1 TMP2,TMP3,TMP5  
COMMON/XFLARE/AE,AH,AX,D1,ERRAD,GAMAPD,GAMAPR,  
1 GAMERR,GAMMAD,L1,RSP,SWT1,TAE,TMP1,TMP2,TMP3,  
2 TMP4,TMP5,TMP6,TTH
```

The correction set PGAPRT changed the printed output. In subroutine SDFLGP, the array PGA1T was made available with the following declaration.

```
COMMON/XALGFA/DMX(26),PGA1T(5),DMXX(3)
```

Additionally, the seven occurrences of PGA1T1, introduced by CSCMOD2, were changed back to PGA1T.

The DELTFIX mods were intended to once again address a problem with the reduction of the control limit force. In subroutine EXE, the variable H was initialized to 0.0001 above statement 302. In ALGEAR1, the test on TIME above statement 15 was changed by changing DELT to 0., and the following code was added at the top of the 274 loop.

```
DELT = 0.  
IF(INDINT(I).NE.0) DELT = H  
INDINT(I) = 0
```

The two statements beginning with statement 56 were replaced by a CONTINUE, and the next three lines were replaced with the following.

```
WLFOR(I)=WLFOR(I)-REDSLP(I)*DELT  
EPSILO(I)=EPSILO(I)+EPSSLP*DELT
```

In statement 100, DELT was replaced by 0.0001. In subroutine ACTNG, the new HTCOM common block introduced by CSCMOD6 replaced the original declaration and PGA1T was moved from XACTNG to XALGEA. The following code was inserted at the top of the subroutine,

```
DELT = 0.
IF(INDINT(I).NE.0) DELT = H
INDINT(I) = 0
```

and the five lines of code beginning with statement 56 were replaced with the following.

```
56 CONTINUE
WLFOR(I) = WLFOR(I) - REDSLP(I)*DELT
EPSILO(I) = EPSILO(I) - EPSSLP*DELT
```

The test on TIME below statement 109 was changed by substituting 0.0001 for DELT.

A deficiency was noted in the active code in that the input signal and its modifications to the electronic compensation networks were not being treated as integration variables. The XMAFIX correction set was intended to correct this deficiency. In program TOLA, the size of the LGDE common block was increased from 72 to 112 elements, and the ACTIVE common block was increased from 802 to 842 elements. In the ACOBLK comdeck, the final two continuation lines of the COMMON declaration were replaced with the following.

```
1,COEF1(5) ,LMODE(5) ,PINN(30) ,PINM(30) ,STRON(30) ,STROM(30)
2,XMA1DT(5) ,XMA2DT(5) ,XMA3DT(5) ,XMA4DT(5) ,XMA6DT(5) ,XMA7DT(5)
3,XMA9DT(5) ,XMA10D(5)
```

In subroutine EXE, the size of the LA array in the LGDE common block was increased from 50 to 90, the number of words being read from and written to TAPE7 was increased from 11002 to 11077, and an ENDFILE 7 was added after the WRITE to unit 7. The size of the LA array in the LGDE common block was also increased from 50 to 90 in LGDET, LGEAR1, LGEA3C, ALGEAR1, ACTNG, and SDFLGP. In subroutine LGEAR1, the second test on INDLG was changed as follows,

```
IF (IABS(INDLG) .EQ. 3) NDEQ=10*NSTRUT
```

and in ACTINIT the following DATA statement was added.

```
DATA XMA1DT,XMA2DT,XMA3DT,XMA4DT /20*0.0/,  
      XMA6DT,XMA7DT,XMA9DT,XMA10D /20*0.0/
```

In subroutine ALGEAR1, the following code was inserted after the definition of XMA5(I),

```
XMA1DT(I) = XMA(I)  
XMA2DT(I) = XMA1(I)  
XMA3DT(I) = XMA5(I)  
XMA4DT(I) = XMA3(I)
```

the following was added after the definition of XMA8(I),

```
XMA6DT(I) = XMA5(I)  
XMA7DT(I) = XMA8(I)
```

and the following was added after the definition of XMA11(I).

```
XMA9DT(I) = XMA8(I)  
XMA10D(I) = XMA11(I)
```

The following new code was inserted after the last call to INTEG.

```
I10 = I+10*NSTRUT
I11 = I+11*NSTRUT
I12 = I+12*NSTRUT
I13 = I+13*NSTRUT
I14 = I+14*NSTRUT
I15 = I+15*NSTRUT
I16 = I+16*NSTRUT
I17 = I+17*NSTRUT
CALL INTEG(LA(I10),XMA1DT(I))
CALL INTEG(LA(I11),XMA2DT(I))
CALL INTEG(LA(I12),XMA3DT(I))
CALL INTEG(LA(I13),XMA4DT(I))
CALL INTEG(LA(I14),XMA6DT(I))
CALL INTEG(LA(I15),XMA7DT(I))
CALL INTEG(LA(I16),XMA9DT(I))
CALL INTEG(LA(I17),XMA10D(I))
```

In subroutine SDFLGP, the following new code was added after the last call to UPDAT.

```
I10 = I+10*NSTRUT
I11 = I+11*NSTRUT
I12 = I+12*NSTRUT
I13 = I+13*NSTRUT
I14 = I+14*NSTRUT
I15 = I+15*NSTRUT
I16 = I+16*NSTRUT
I17 = I+17*NSTRUT
CALL UPDAT(1,LA(I10),XMA1(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I11),XMA2(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I12),XMA3(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I13),XMA4(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I14),XMA6(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I15),XMA7(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I16),XMA9(I),DU,DU,DU,DU)
CALL UPDAT(1,LA(I17),XMA10(I),DU,DU,DU,DU)
```

It was found that portions of the active code were referencing the controlled servovalve spool displacement when they should have

referenced the servovalve spool displacement. This problem was addressed with the XSVMOD correction set. Identical corrections were made to both ALGEAR1 and ACTNG. Statement number 500 was replaced with a CONTINUE. The following statement was introduced below statement 290,

```
XVALVE(I) = XSV(I)
```

the following code replaced statement 299,

```
299 CONTINUE  
IF(XVALVE(I).NE.0.) GO TO 311
```

and the following was inserted after statement 400.

```
XSV(I) = XVALVE(I)
```

The GENFIX mods made several small general changes to the program. The XACTNG common block was corrected in both TOLA and ACTNG. In TOLA, the size was reduced to reflect the removal of PGA1T in the REST correction set, and in ACTNG the name was corrected from ACTNG (an error in REST) to XACTNG. Two new elements, NIN and TIMEL, were added for later use. In subroutine EXE, the variable NCASE was made available by changing the third continuation line of the DIRCOM common declaration to the following,

```
*      DM43      ,INDVPC      ,DM44 ( 7),NCASE      ,DM45      ,
```

and NCASE was initialized to be blank. The upper limit on the 35 loop was changed from 4036 to 4059 to reflect the correct size of the DIRCOM common block. A rewind of unit 7 was added after the read from 7 to prepare the unit to receive new restart information. In subroutine MIMIN, the 45 loop was moved to below statement 205.

The DELTFXX mods represented another attempt to resolve the control limit force reduction. New variables NIN and TIMEL were initialized with a DATA statement to zero in both ALGEAR1 and ACTNG. In subroutine ALGEAR1, the following code was inserted as the first executable statements,

```
IF(NIN.EQ.0) TIMEL = TIME
NIN = 1
DELT = TIME - TIMEL
TIMEL = TIME
```

and the three lines at the top of the 274 loop were removed. In ACTNG, the 2nd through 4th executable statements were replaced with code identical to that above.

The MOD282 correction set finalized solution of the control limit force problem. The common block XALGEA was expanded in both TOLA and LGEAR1 to include NIN and TIMEL. The READ and WRITE statements to unit 7 in EXE were altered to transfer 11081 words. In both ALGEAR1 and ACTNG, the three statements beginning with statement 159 were replaced with the following.

```
159 IF(SD1(1,I).LT.0.) GO TO 471
```

The MOD296 correction set changed the active code. In subroutine ALGEAR1, the statement above statement 112 was changed to the following,

```
IF(IMODE(I).EQ.0.AND.S(1,I).EQ.0.) 112,113
```

and the test on IMODE above statement 19 was removed. In subroutine ACTNG, the test on IMODE below statement 108 was changed as follows.

```
IF(S(1,I).GT.0..OR.IMODE(I).EQ.1) GO TO 113
```

The REST1 mods corrected a problem with the program restart. The common block XAUTS was expanded in TOLA and AUTS to accommodate three new variables, DELTS, ERROR, and IPR. The size of the data array transferred to or from TAPE7 in EXE was increased by three words to 11084.

The primary purpose of the MOD329 correction set was to rearrange some logic flow in the active code. Additionally, the AMIN1 evaluation in subroutine MIMIN which had been moved by HTRY was restored to its original position. In subroutine ALGEAR1, the definition of I9 and subsequent call to INTEG were moved to inside the 28 loop. The definition of I3 and the subsequent call to INTEG were removed from the 28 loop and the following code was inserted after statement 21.


```

    DP1(I) = (-QD(I)+QSV1(I)-QSV3(I)+(AREA1(I) - APINT(I))
1  *SD1(1,I))*BETA/VOLLT(I)
    IF(S(1,I).NE.0..OR.AIC(I).NE.1.) GO TO 64
    DP1(I) = 0.
64 CONTINUE
    I3 = I + 3*NSTRUT
    CALL INTEG(LA(I3),DP1(I))

```

The preexisting definition of DP1 was removed and the following statement was added just inside the 28 loop

```

IF(IMODE(I).EQ.0) GO TO 28

```

In subroutine SDFLGP, the following statement was added after the call to UPDAT for QSVCU.

```

IF(IMODE(I).EQ.0) GO TO 6

```

The MOD351 mods simply added the following statements above statement 64 in ALGEAR1.

```

PGA1T1(I) = PGA1I(I)
PGA1T(I) = PGA1I(I)

```

Following the addition of 40 new integration variables by XMAFIX, it was necessary to increase the hard-wired maximum limit of such variables from 90 to 100. This was accomplished with the MOD1029 correction set. In program TOLA, the size of the UPDCAL common block was increased from 181 to 201, and the XMIMIN common

block was increased from 908 to 1008. The P and Y arrays were each increased in size from 90 to 100 in common block UPDCAL in subroutines EXE, INUPD, LNUPD, INPUZ, INTEG, UPDAT, MIMIN, and LGDET. In EXE, the size of the data block of information transferred to and from TAPE7 was increased from 11084 to 11204. In subroutine INUPD, both the test on NUM+N and the FORMAT were modified to reflect the new maximum limit. In subroutine INPUZ, the upper limit on the loop was increased to 100. Finally, in subroutine MIMIN the DIMENSION statement for YMAX was changed as follows.

```
DIMENSION YMAX(100),YO(100),PO(100),S(100),YP(100),Y1(100),
1 Z(100),XK(100,3)
```

The MOD1040 correction set corrected a problem with the pitch autopilot by inserting the following statement below statement 93 in subroutine AUTS.

```
DELQI = DELQDE
```

Since it is the variable H and not HT which represents the integration step size, a change was made to MIMIN by the MODMIM correction set to enhance the program output. The second write of HT was changed as follows.

```
WRITE(6,701) HT,H
701 FORMAT(* INTEG RTN.      HT = *,E15.8,* H = *,E15.8)
```

A units problem in the active code was corrected by MOD1048. In both ALGEAR1 and ACTNG the value of the stroke in statement 4456 was converted from feet to inches by multiplying by 12.

The MOD1103 correction set changed the LGEAR1 code to allow for a special feature of the F4 gear design. The following new code was added after the test on IL.

```

C  MODIFICATION TO ACCOMMODATE SECONDARY PISTON OF F4 MAIN GEAR
    IF(S(1,I).LE.0.) GO TO 59
    IF(S(1,I).GE.SB(I)-(S2T(I)-S2(1,I))) 83,85
C  SECONDARY PISTON IN CONTACT WITH ORIFICE TUBE
83 IF(SD1(1,I)+1.E-4.GE.S2D1(1,I)) 84,85
84 S2D1(1,I) = SD1(1,I)
    S2D2(1,I) = SD2(1,I)
    GO TO 60
85 CONTINUE

```

A method for slowly and continuously varying the aerodynamic coefficients C_{A0} and C_{N0} was introduced with the AERAT correction set. Two new common blocks AEROCO and XAERO of lengths 8 and 2, respectively, were added to program TOLA. In EXE, the length of the restart data block transferred to and from TAPE7 was increased from 11204 to 11214. A call to AERO4 was added to subroutine OPT1 below statement 621, and the following new subroutine was added after

OPT1.

```

SUBROUTINE AERO4
COMMON/DIRCOM/DM1(2),X
COMMON/TABCOM/LOCS(115),ST(115)
COMMON/TABDIR/TABLE(800)
COMMON/AEROCO/RTAB10(2),RTAB80(2),LTAB10(2),LTAB80(2)
COMMON/XAERO/NIN,TIMEL
REAL LTAB10,LTAB80
DATA TIMEL,NIN/0.,0/
DATA RTAB10,RTAB80,LTAB10,LTAB80/8*0./
IF(NIN.EQ.0) TIMEL = X

```

```

NIN = 1
DELT = X - TIMEL
TIMEL = X
IND1 = LOCS(45)
IND2 = LOCS(114)
ATAB11 = TABLE(IND1)
ATAB12 = TABLE(IND1+1)
ATAB81 = TABLE(IND2)
ATAB82 = TABLE(IND2+1)
ATAB11 = ATAB11 + RTAB10(1)*DELT
ATAB12 = ATAB12 + RTAB10(2)*DELT
ATAB81 = ATAB81 + RTAB80(1)*DELT
ATAB82 = ATAB82 + RTAB80(2)*DELT
IF(RTAB10(1).GT.0..AND.ATAB11.GE.LTAB10(1)) GO TO 40
IF(RTAB10(1).LT.0..AND.ATAB11.LE.LTAB10(1)) GO TO 40
10 IF(RTAB10(2).GT.0..AND.ATAB12.GE.LTAB10(2)) GO TO 50
IF(RTAB10(2).LT.0..AND.ATAB12.LE.LTAB10(2)) GO TO 50
20 IF(RTAB80(1).GT.0..AND.ATAB81.GE.LTAB80(1)) GO TO 60
IF(RTAB80(1).LT.0..AND.ATAB81.LE.LTAB80(1)) GO TO 60
30 IF(RTAB80(2).GT.0..AND.ATAB82.GE.LTAB80(2)) GO TO 70
IF(RTAB80(2).LT.0..AND.ATAB82.LE.LTAB80(2)) GO TO 70
GO TO 80
40 ATAB11 = LTAB10(1)
RTAB10(1) = 0.
GO TO 10
50 ATAB12 = LTAB10(2)
RTAB10(2) = 0.
GO TO 20
60 ATAB81 = LTAB80(1)
RTAB80(1) = 0.
GO TO 30
70 ATAB82 = LTAB80(2)
RTAB80(2) = 0.
80 TABLE(IND1) = ATAB11
TABLE(IND1+1) = ATAB12
TABLE(IND2) = ATAB81
TABLE(IND2+1) = ATAB82
RETURN
END

```

In subroutine READ the following code was inserted after statement 19,

```

IF(SYM.EQ.6HRTAB10) GO TO 905
IF(SYM.EQ.6HRTAB80) GO TO 905
IF(SYM.EQ.6HLTAB10) GO TO 905
IF(SYM.EQ.6HLTAB80) GO TO 905

```

and the following statements were added below statement 810.

```
905 CALL AERDIN(SYM,RA)
    GO TO 100
```

The following new subroutine was added after subroutine READ.

```
SUBROUTINE AERDIN(SYM,RA)
DIMENSION RA(55)
COMMON/AERDCO/DATAX(8)
DATA DATAX/8*0./
CALL LINES(1)
WRITE(6,1) SYM,RA
1 FORMAT(18X,A6,5X,55A1)
2 FORMAT(*OERROR.ILLEGAL CHARACTER IN NUMERIC FIELD*,1R1,2H**/)
IF(SYM.EJ.6HRTAB10) INDEX = 1
IF(SYM.EJ.6HRTAB80) INDEX = 3
IF(SYM.EQ.6HLTAB10) INDEX = 5
IF(SYM.EQ.6HLTAB80) INDEX = 7
NUMEXP = 0
NEXP = 0
EXP = 0
NL = 0
NR = 0
NUML = 0
NUMR = 0
ISIGN = 0
JSIGN = 0
LEFT = 1
DO 210 I=1,56
IF(I.EQ.56) GO TO 140
IF(RA(I).EQ.1H ) GO TO 210
IF(RA(I).EQ.1H,) GO TO 140
IF(RA(I).EQ.1H.) GO TO 170
IF(RA(I).EQ.1HE) GO TO 180
IF(RA(I).EQ.1H-) GO TO 200
NUM = SHIFT(RA(I),6)
NUM = NUM.AND.0000000000000000000778
IF(NUM.GT.36) GO TO 130
IF(NUM.LT.27) GO TO 130
NUM = NUM - 27
IF(IEXP.EQ.1) GO TO 190
IF(LEFT.GT.0) NUML = 10*NUML + NUM
IF(LEFT.GT.0) NL = NL + 1
```

```
IF(LEFT.LT.0) NUMR = 10*NUMR + NUM
IF(LEFT.LT.0) NR = NR + 1
GO TO 210
130 CALL LINES(3)
WRITE(6,2) RA(I)
GO TO 210
140 IF(NL.EQ.0.AND.NR.EQ.0) GO TO 210
IF(NR.EQ.0) GO TO 160
X = FLOAT(NUML) + FLOAT(NUMR)/(10.)**NR
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
150 DATAX(INDEX) = X
NUML = 0
NUMR = 0
NL = 0
NR = 0
LEFT = 1
ISIGN = 0
JSIGN = 0
IEXP = 0
NEXP = 0
NUMEXP = 0
INDEX = INDEX + 1
GO TO 210
160 X = NUML
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
GO TO 150
170 LEFT = -1
GO TO 210
180 IEXP = 1
GO TO 210
190 NUMEXP = 10*NUMEXP + NUM
NEXP = NEXP + 1
GO TO 210
200 IF(IEXP.EQ.0) ISIGN = 1
IF(IEXP.NE.0) JSIGN = 1
210 CONTINUE
RETURN
END
```

The MOD2056 correction set corrected two problems with the program. The DATA statement for initializing the AEROCO common block was moved from subroutine AEROIN to program TOLA. In subroutine ACTINIT, the EQUIVALENCE statement for GREFF was corrected by replacing DM15(16) with DM5(16). The size of the ACTDIR common block was increased from 75 to 77 and the variables REDSLP and DSTOP were added to the list of active input variables.

The MOD2075 correction set remedied a sign problem with the strut force. In subroutine ALGEAR1, the statement label 159 was replaced with the following code.

```
159 SIGX = 0.
   IF(SD1(1,I).EQ.0.) GO TO 471
   SIGX = SD1(1,I)/ABS(SD1(1,I))
```

The final continuation line in the definition of FORSST was replaced with the following.

```
1 + CFFOR(I))*SIGX*DMTANH(I) + FSTOP(I))
```

Identical changes were made in the ACTNG routine.

During conduct of the experimental program reported in reference 9, it became apparent that the original control philosophy (see reference 10) was not adequate to control the gear during more realistic landing simulations as opposed to restrained vertical drop testing. For example, if the airplane rebounded from the initial touchdown impact and the gear shock strut fully extended, the

original control laws would permit the control to add fluid to the strut and result in the development of excessive strut pressure. Consequently the control laws were modified to deactivate the control if the gear should become fully extended. As a result the logic and equations programmed in the active gear, flexible airframe takeoff and landing analysis computer program had to be modified to control the servovalve in order to return the strut and servovalve parameters to initial conditions to accommodate subsequent impacts. The MOD2203 and MOD2235 correction sets incorporated the new logic into the program. The size of the XALGEA and XACTNG common blocks were increased to provide for several logic control flags and the size of the restart common block was increased to 11364. The following code replaced the definition of ISTROK below statement 500.

```

287 IF(S(1,I).GT.ES(I))287,289
287 ISTROK(I)=1
    ICU(I)=0
    IQCU(I)=0
    IXS(I)=0
C    THE FOLLOWING LOGIC RETURNS THE GEARS,DURING REBOUND, TO INITIAL
C    CONDITIONS IN THE EVENT THE GEAR CONTACTS THE SURFACE BEFORE
C    THE LOGIC BETWEEN STATEMENTS 226 AND 421 IS FULLY EFFECTIVE
289 IF(ISTROK(I).EQ.1.AND.DDELTA(I).LT.0.0)IGO(I)=1
    IF(IGE(I).EQ.1.OR.ITRIP(I).EQ.1)227,297
227 IF(IGO(I).EQ.0)GO TO 226
    IF(DELTA(I).GT.0.0.AND.WLFOR(I).EQ.0.0)220,226
220 IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)221,222
221 QO(I)=-VCUM(I)/DSTOP
    GO TO 226
222 PGA2T(I)=PGA1I(I)
    QO(I)=0.0
    IGO(I)=0
226 IF(IGE(I).EQ.1)290,297

```


The code between statements 299 and 400 was replaced with the following.

```

312 IF(ICOSV(I).NE.1)313,314
314 IF(XVALVE(I).NE.XBIAS(I))GO TO 311
298 IF(ICOSV(I).EQ.1)GO TO 311
313 IF(PGA1T(I).GT.PGA1I(I)+2000.)292,293
292 IF(IXSVL(I).EQ.1)GO TO 311
XSVDOT(I)=XSVDNM(I)*PERCNT(I)
IPASS(I)=1
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
IF(XVALVE(I).LE.-0.13)300,303
303 XSV(I)=XVALVE(I)
GO TO 297
300 XVALVE(I)=-0.13
XSVDOT(I)=0.0
IXSVL(I)=1
GO TO 294
293 IF(PGA1T(I).LT.PGA1I(I)-2000.)295,294
295 IF(IXSVH(I).EQ.1)GO TO 311
XSVDOT(I)=XSVDNM(I)*PERCNT(I)
IPASS(I)=2
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
IF(XVALVE(I).GE.0.13)302,304
304 XSV(I)=XVALVE(I)
GO TO 297
302 XVALVE(I)=0.13
IXSVH(I)=1
294 CONTINUE
XSVDOT(I)=0.0
ICOSV(I)=1
311 IF(PGA2T(I).GT.PGA1I(I)+4000.0)316,315
316 IF(QSV(I).LT.QD(I).AND.ICOSV(I).EQ.1)317,318
317 XSVDOT(I)=0.0

XSV(I)=XVALVE(I)
GO TO 297
318 XSVDOT(I)=XSVDNM(I)*PERCNT(I)
XSV(I)=XVALVE(I)
GO TO 297
315 CONTINUE
IF(NAC(I).EQ.1)GO TO 307
320 IF(IIIXSVH(I).EQ.1)GO TO 305
XSVDOT(I)=XSVDNM(I)
IPASS(I)=3

```

```

XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT
IF(XVALVE(I).LE.XBIAS(I))305,306
306 XSV(I)=XVALVE(I)
GO TO 297
305 XVALVE(I)=XBIAS(I)
XSVDOT(I)=0.0
IIXSVH(I)=1
GO TO 400
307 IF(IIXSVL(I).EQ.1)GO TO 308
XSVDOT(I)=XSVDNM(I)
IPASS(I)=4
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT
IF(XVALVE(I).GE.XBIAS(I))308,309
309 XSV(I)=XVALVE(I)
GO TO 297
308 XVALVE(I)=XBIAS(I)
XSVDOT(I)=0.0
IIXSVL(I)=1

```

The following new code was added near the call to LIMITS.

```

DELTX1(I)=0.0
ICOSV(I)=0
QD(I)=0.0
QSVCU(I)=0.0
QSV1(I)=0.0
QSV3(I)=0.0
VCUM(I)=0.0
VOL1T(I)=VOL1I(I)
VOL2T(I)=VOL2I(I)
VOL3T(I)=VOL3I(I)
ITRIP(I)=1
PGA1T1(I)=PGA1I(I)
PGA1T(I)=PGA1I(I)
PGA2T(I)=PGA1I(I)
PGA3T(I)=PGA1I(I)

```

The 5 lines of code beginning with the call to PHLOZ2 were replaced

with the following.

```

297 IF(ITRIP(I).EQ.1)430,421
430 IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)GO TO 431
    QD(I)=0.0
    ICU(I)=1
    GO TO 440
431 IF(VCUM(I).LT.-0.00001)GO TO 432
    QD(I)=-VCUM(I)/DSTOP
    GO TO 440
432 QD(I)=-VCUM(I)/DSTOP
440 IF(QSVCU(I).GT.0.00001.OR.QSVCU(I).LT.-0.00001)GO TO 433
    QSV(I)=0.0
    IQCU(I)=1
    GO TO 420
433 IF(QSVCU(I).LT.-0.00001)GO TO 434
    QSV(I)=-QSVCU(I)/DSTOP
    GO TO 420
434 QSV(I)=-QSVCU(I)/DSTOP
420 IF(XSV(I).GT.XBIAS(I)+.000001)422,423
422 IF(XSV(I).GT. 0.0)XSVDOT(I)=-XSV(I)/DSTOP
    IF(XSV(I).LT. 0.0)XSVDOT(I)=XSV(I)/DSTOP
    GO TO 600
423 IF(XSV(I).LT.XBIAS(I)-.000001)424,425
424 XSVDOT(I)=-XSV(I)/DSTOP
    GO TO 600
425 XSVDOT(I)=0.0
    IXS(I)=1
600 IF(XMA1(I).GT.0.00001 .OR.XMA1(I).LT.-0.00001)GO TO 601
    XMA1DT(I)=0.0
    IA1(I)=1
    GO TO 602
601 XMA1DT(I)=-XMA1(I)/DSTOP
602 IF(XMA2(I).GT.0.00001 .OR.XMA2(I).LT.-0.00001)GO TO 603
    XMA2DT(I)=0.0
    IA2(I)=1
    GO TO 604
603 XMA2DT(I)=-XMA2(I)/DSTOP
604 IF(XMA3(I).GT.0.00001 .OR.XMA3(I).LT.-0.00001)GO TO 605
    XMA3DT(I)=0.0
    IA3(I)=1
    GO TO 606
605 XMA3DT(I)=-XMA3(I)/DSTOP
606 IF(XMA4(I).GT.0.00001 .OR.XMA4(I).LT.-0.00001)GO TO 607
    XMA4DT(I)=0.0
    IA4(I)=1
    GO TO 608
607 XMA4DT(I)=-XMA4(I)/DSTOP
608 IF(XMA6(I).GT.0.00001 .OR.XMA6(I).LT.-0.00001)GO TO 609
    XMA6DT(I)=0.0
    IA6(I)=1
    GO TO 610
609 XMA6DT(I)=-XMA6(I)/DSTOP

```

```

610 IF(XMA7(I).GT.0.00001 .OR.XMA7(I).LT.-0.00001)GO TO 611
    XMA7DT(I)=0.0
    IA7(I)=1
    GO TO 612
611 XMA7DT(I)=-XMA7(I)/DSTOP
612 IF(XMA9(I).GT.0.00001 .OR.XMA9(I).LT.-0.00001)GO TO 613
    XMA9DT(I)=0.0
    IA9(I)=1
    GO TO 614
613 XMA9DT(I)=-XMA9(I)/DSTOP
614 IF(XMA10(I).GT.0.00001 .OR.XMA10(I).LT.-0.00001)GO TO 615
    XMA10D(I)=0.0
    IA10(I)=1
    GO TO 421
615 XMA10D(I)=-XMA10(I)/DSTOP
C   THESE SWITCHES ARE EITHER ZERO OR ONE
421 IF(ICU(I)+IQU(I)+IXS(I)+IA1(I)+IA2(I)+IA3(I)+IA4(I)+IA6(I)
    1   +IA7(I)+IA9(I)+IA10(I) .EQ. 11)ISTROK(I)=0
    I2=2*I+NSTRUT-1

```

The test on HMM(I) below the definition of VELDEC(I) was replaced with the following code.

```

IF(IGE(I).EQ.0)GO TO 131
WLFOR(I)=WLFORR
INDEACT(I)=2
EPSILO(I)=EPSROL(I)
GO TO 451
131 IF(S(1,I).LE.ES(I))GO TO 451
    IF(HMM(I).EQ.0)GO TO 451

```

The following test was inserted before statement 452,

```
IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)GO TO 458
```

and the following code was added after the definition of P1(I).

```
458 P1(I)=PGA1T(I)/144.
459 VOLANT(I)=VOLANT(I)+QSVN(I)*DELT-QPUMPS(I)*DELT
```

The first line of the call to FLOZE2 was replaced with the following.

```
410 IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)462,410
CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
```

The following initializations were added below statement 50,

```
IGE(I)=0
ITRIP(I)=0
```

and the following tests were inserted above statement 55.

```
IF(IMODE(I).EQ.0)GO TO 55
IF(ISTROK(I).EQ.1)IGE(I)=1
```

The following calls were inserted below the call to INTEG for XSVDOT(I).

```
CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
CALL LIMITS(XSVDOT(I),XSVD(I),XSVDMX(I),XSVDMN(I))
CALL LIMITS(XSVD(I),XSVD(I),XDDMAX(I),XDDMIN(I))
```

Similar modifications were required in the ACTING subroutine.

The MOD2235 correction set corrected an error in the MOD2203 logic by replacing the definition of IGO(I) below statement 222 with the following.

```
IF(DELTA(I).GT.0.) IGO(I) = 0
```

3. USER INFORMATION

In order to exercise the new program options, the user needs to be aware of the new data preparation requirements and operating instructions. These two areas are discussed below.

3.1 Data Preparation - The overall input data format is changed by the introduction of a new data card which must precede all other data. The first card of each data deck controls the restart option and must contain a single integer in the first card column. This card must be present even if the restart capability is not selected.

allowed values of the integer restart flag are presented below:

| <u>Value</u> | <u>Action</u> |
|--------------|---|
| 0 | Reject the restart option - This run will read all input data from cards and a subsequent restart will not be possible. |
| 1 | This run will read all input data from cards and will create a file such that the job may be restarted from any point at which data is staged. |
| 2 | This run is a restart of an earlier run. All program variables will be initialized from the restart file to the values corresponding to the desired stage. Subsequent data cards on the input file are limited to stage data only. This run may not be restarted at subsequent data stages. |

- 3 This run is the same as for a value of 2 except this run will create a file such that the job may be restarted again from any subsequent point at which data is staged.

Active control simulations are run simply by inserting active control variables into the input file and setting the landing gear mode indicator switch (INDLG) to -3. The input variables are described in Table 1 and a sample listing of these inputs are given in Figure 3. The active control variable list must be preceded by the word ACTIVE starting in column 1. All following variables are entered according to the same format:

Column Number

1-6 VARIABLE NAME

8-10 DATA TYPE - DEC or blank = REAL

 INT = INTEGER

12-66 DATA VALUES, LEFT JUSTIFIED, SEPARATED BY
 COMMAS, NO EMBEDDED BLANKS

Unlike other program data, the integer counter for continuation of array data must be right justified in column 68 instead of left justified in column 67. The arrays PINM, PINN, STROM, and STRON are only required if KAPT = 2. If included, these arrays must be placed in the active variable list and are each allowed a maximum length of 30. The independent variable (strut stroke)


```

1
STCASE  TAR
ATAR01    2
ATAR02    2
ATAR05    2
ATAR06    2
ATAR08    2
ATAR10    2
ATAR11    2
ATAR12    2
ATAR15    2
ATAR16    2
ATAR51    2

ATAR52    2
ATAR53    2
ATAR56    2
ATAR57    2
ATAR59    2
ATAR80    2
CTAR01    207
CTAR02    207
CTAR03    11
CTAR04    11
ETAR02    99
ETAR03    25
TTAR10    41
VTAR01     5
VTAR02     5
VTAR03     5
VTAR04     5
VTAR06     5

TRA
STCASE
REM      RCD 4F4 ACTIVE GEAR PROGRAM
NCASE    RCD 1LRPG01
REM      RCD 4INTEGRATION INFORMATION
TVARRH   0
TMAX     .31
AMINER   .00005
PRTMIN   0.
AMAXEP   .001
DELT5    .01
PRINT    .01
REM      RCD 4DATA REQD FOR SDF-2
AMASS    1115.7
HGC7F    8.23743
THTRD    7.1
GAM7D    -1.14576
VG77F    250.0

```

Figure 3-a. Full Input Deck

| | |
|--------|----------------------------------|
| OJ77R | 0. |
| XC77F | 0. |
| PWHCP | 0. |
| TNDAPC | 1 |
| TNDADD | 1 |
| TNDEPA | 1 |
| TNDPIA | 1 |
| TNDACM | 1 |
| TNDGRT | 1 |
| TNDWGT | 0 |
| REM | BCD 5VEHICLE PHYSICAL PROP. DATA |
| TNDVPC | 1 |
| XCGRF | 0. |
| VTAB01 | 2,0.,0.,5000.,0. |
| VTAB02 | 2,0.,24873.,1200.,24873. |
| VTAB03 | 2,0.,144516.,1200.,144516. |
| VTAB04 | 2,0.,154897.,1200.,154897. |
| TNDX7S | 1 |
| VTAB06 | 2,C.,3361.,1200.,3361. |
| REM | BCD 4AERODYNAMIC INPUT DATA |
| TNDAFP | 1 |
| ARFF | 547.87 |
| D1RFF | 16.042 |
| D2RFF | 38.667 |
| INDA01 | 1 |
| ATAR01 | -0.000709,-0.000709 |
| INDA02 | 1 |
| ATAR02 | 0.000423,0.000423 |
| INDA05 | 1 |
| ATAR05 | -0.001027,-0.001027 |
| INDA06 | 1 |
| ATAR06 | 0.000136,0.000136 |
| INDA08 | 0 |
| ATAR08 | 0.,0. |
| INDA10 | 1 |
| ATAR10 | .379202,.379202 |
| INDA11 | 1 |
| ATAR11 | .063627,.063627 |
| INDA12 | 1 |
| ATAR12 | -.00018,-.00018 |
| INDA15 | 1 |
| ATAR15 | .0016,.0016 |
| INDA16 | 1 |
| ATAR16 | -.0002,-.0002 |
| INDA51 | 1 |
| ATAR51 | -.083516,-.083516 |
| INDA52 | 1 |
| ATAR52 | -.005969,-.005969 |
| INDA53 | 1 |

| | | |
|--------|--|-----|
| ATARS3 | - .000216, -.000216 | |
| INDAS6 | 1 | |
| ATARS6 | - .010317, -.010317 | |
| INDAS7 | 1 | |
| ATAB57 | - .000005, -.000005 | |
| INDAS9 | 0 | |
| ATARS0 | .0, .0 | |
| INDAR0 | 1 | |
| ATARS0 | 0.141245, 0.141245 | |
| REM | RCD 3ENGINE THRUST DATA | |
| INDTFE | 3 | |
| INDTS0 | 1 | |
| IT10Y | 6 | |
| IT10W | 5 | |
| TTAR10 | -2., -1., 0., 1., 2. | |
| TTAR10 | 0., .05, .1, .15, .2, .25 | 6 |
| TTAR10 | 0., 0., 0., 470., 17600. | 12 |
| TTAR10 | 0., 0., 0., 300., 17300. | 17 |
| TTAR10 | 0., 0., 0., 220., 17200. | 22 |
| TTAR10 | 0., 0., 0., 160., 17080. | 27 |
| TTAR10 | 0., 0., 0., 0., 17080. | 32 |
| TTAR10 | 0., 0., 0., 0., 17100. | 37 |
| REM | RCD 3LANDING GEAR DATA | |
| ACTIVE | | |
| AMUH | .00018 | |
| STRON | 0.0, .1, .18333, .26667, .35, .43333, .51667, .6, .68333, .76667 | |
| STRON | .85, .93333, 1.01667, 1.1, 1.18333, 1.26667, 1.35, 1.43333 | 11 |
| STRON | 1.51667, 1.6, 1.68333, 1.76667, 1.85, 1.93333, 2.0 | 19 |
| PINN | .00231, .00231, .00231, .00204, .00242, .003, .00339, .00352 | |
| PINN | .00396, .00435, .00454, .00473, .00492, .00511, .00511, .00511 | 9 |
| PINN | .00531, .00531, .0055, .00569, .00601, .00627, .00646, .00665 | 17 |
| PINN | .00665 | 25 |
| STRON | 0.0, .06604, .10771, .14938, .19104, .23271, .27438, .31604 | |
| STRON | .35771, .39938, .44104, .48271, .52438, .56604, .60771, .64938 | 9 |
| STRON | .69104, .73271, .77438, .81604, .85771, .89938, .94104, .98271 | 117 |
| STRON | 1.02438, 1.06604, 1.10771, 1.14938, 1.32333 | 25 |
| PINN | .00603, .00595, .00572, .0055, .00518, .00492, .00458, .00424 | |
| PINN | .0039, .00373, .00356, .00366, .00375, .00409, .00443, .0046 | 9 |
| PINN | .00486, .00511, .00528, .00554, .00571, .00588, .00614, .00631 | 117 |
| PINN | .00648, .00665, .00674, .00682, .00682 | 25 |
| AREA1 | .08705, .1134, .1134 | |
| AREA2 | .12962, .16082, .16082 | |
| AREA3 | .01651, .00819, .00819 | |
| AREMO | .00684, .00684, .00684 | |
| AREFO2 | .000021, .000668, .000668 | |
| RETA | 14400000. | |
| RLMU | .15, .15, .15 | |
| RUMU | .15, .15, .15 | |
| CDMOC | .9, .9, .0 | |

Figure 3-a. Cont'd

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| | |
|--------|----------------------------|
| CDMDF | .9,.9,.9 |
| CDSV | .62,.62,.62 |
| CD3 | .6,.6,.6 |
| CEFOR | 50.,50.,50. |
| DELT | .0001 |
| DIOTA | 1. |
| DSV | 1.1875,1.1875,1.1875 |
| EPSILO | 200.,200.,200. |
| FPSROL | 2000.,2000.,2000. |
| EPSSLP | 0. |
| FTASV | .436 |
| FWOPK | .9,.9,.9 |
| GAMAH | 52.36,52.36,52.36 |
| GNP | 0. |
| KAPT | INT 2,2,2 |
| QMPUN | 0. |
| PATM | 2116.8 |
| PGAHAC | 432000.,432000.,432000. |
| PGALAC | 0.,0.,0. |
| PGA1I | 16272.,28800.,28800. |
| PGA2I | 16272.,28800.,28800. |
| PGA3I | 16272.,28800.,28800. |
| PERCNT | 1.66115,1.66115,1.66115 |
| OPUMPS | 0.05793,0.05793,0.05793 |
| RCLSV | 4.75E-5,4.75E-5,4.75E-5 |
| RHOH | 1.626 |
| TAUF | 0.1 |
| TC1 | .281 |
| TC2 | .141 |
| TC3 | .001 |
| TC4 | .0001 |
| VOLACI | 1.33333,1.33333,1.33333 |
| VOLANT | .26667,.26667,.26667 |
| VOL1I | .11733,.11507,.11507 |
| VOL2I | .28923,.26239,.26239 |
| VOL3I | 0.,0.,0. |
| WC | 1263. |
| WC1 | 251.3 |
| WSV | 655.5 |
| WSV1 | 3.48 |
| WSV3 | 3.48 |
| WLFOR | 500.,500.,500. |
| WLFORP | 0.,0.,0. |
| XRIAS | -.000227,-.000171,-.000171 |
| XDDMAX | 1.0E20,1.0E20,1.0E20 |
| XDDMIN | -1.0E20,-1.0E20,-1.0E20 |
| XKA | .04,.04,.04 |
| XKE | 60.,60.,60. |
| XKSV | .0025,.0025,.0025 |

Figure 3-a. Cont'd

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| | |
|-----------|----------------------------------|
| YLPSV1 | 0.,0.,0. |
| XLPSV2 | 0.,0.,0. |
| YSCDM | 1.54667,1.15667,1.15667 |
| YSTHR | .02083,.02083,.02083 |
| XSVDMN | -30.1,-30.1,-30.1 |
| XSVDMY | 30.1,30.1,30.1 |
| XSVMAX | .1,.1,.1 |
| XSVMIN | -.1,-.1,-.1 |
| ZETAC1 | 5.1 |
| ZETAC2 | .1 |
| IMODE INT | 0,1,1 |
| ENDACT | |
| NSTRUT | 3 |
| MASS | 4.28971,13.98819,13.98819 |
| RX | 19.57503,-4.34308,-4.34308 |
| RY | 0.,-8.95333,8.95333 |
| RZ | 1.40617,1.41425,1.41425 |
| THETAD | 0.,4.29137,4.29137 |
| ERDEC | 0.0 |
| CONEM | 23228.5 |
| CTWDM | 69283.P |
| CTHREM | 3180.1 |
| CEOURM | 6581.4 |
| CONFEM | .1683 |
| DELPPM | .4173 |
| VPQWRM | 0.0 |
| AH | .08705,.11020,.11020 |
| RGR | 0. |
| NTIRES | 2.,1.,1. |
| RZERN | .73333,1.21875,1.21875 |
| W | .46042,.9375,.9375 |
| DELTAM | .21666,.36875,.36875 |
| RLT | 1.5E+4 |
| IFD | 1 |
| AT | 1.06148E+5,1.94542E+5,1.94542E+5 |
| RT | 1.3,1.2,1.2 |
| SLEN1 | 1.04037,.78723,.78723 |
| SLEN2 | 3.0122,3.095,3.095 |
| INDATM | 1 |
| GAMA | 1.06 |
| CONC | 6483.9 |
| CTWO | 32406.4 |
| CTHREE | 534.12 |
| CEOUR | 1837.1 |
| CONFRI | .1683 |
| DELPWR | .4173 |
| DTIRE | 1.95208 |
| ETAMAX | 25. |
| ETAMIN | -25. |

| | | |
|--------|---|-----|
| FA | 30. | |
| ETANDS | 0.0 | |
| ETAPT1 | 20. | |
| TNDNWS | 0 | |
| KANWS | 0 | |
| KPNWS | 0 | |
| ISTER | 0 | |
| TRIUD | 0 | |
| TNWPUD | 0 | |
| PCTETA | 1 | |
| PSIDES | 0.0 | |
| RPSI | 0.0 | |
| RYP | 10. | |
| VPOWP | 0.0 | |
| YRDES | 0.0 | |
| FTAR02 | 2,0.,0.,1.5E+5,0. | |
| FTAR03 | 11,0.,.0005,.03,.16,.06,.32,.12,.64,.15,.8 | |
| FTAR03 | .2,.8,.4,.72,.6,.52,.8,.36,1.,.336,100.,.336 | 12 |
| MOMENT | .53703,8.019,8.019 | |
| MR | 0.,0.,0. | |
| OF | 6.0108,5.1525,5.1525 | |
| V7 | 0. | |
| P7FRD | 16272.,28800.,28800. | |
| V7FRD | .09343,.22793,.22793 | |
| A | .08705,.11020,.11020 | |
| TL | 1 | |
| INDC01 | 1 | |
| NSMAIN | 51 | |
| CTAR01 | .0,.066,.1,.108,.149,.183,.191,.233,.267,.274,.316 | |
| CTAR01 | .35,.358,.399,.433,.441,.483,.517,.524,.566,.6,.608 | 12 |
| CTAR01 | .649,.683,.691,.733,.767,.774,.816,.85,.858,.899,.933 | 23 |
| CTAR01 | .941,.983,1.017,1.024,1.066,1.1,1.108,1.149,1.183 | 34 |
| CTAR01 | 1.267,1.323,1.351,1.433,1.517,1.6,1.683,1.767,2.0 | 43 |
| CTAR01 | 1.,2.,3.,32.3,32.3,29.1,29.1,29.1,27.4,27.4 | 52 |
| CTAR01 | 27.4,28.1,28.1,28.1,33.9,33.9,33.9,44.9,44.9,44.9,55.4 | 62 |
| CTAR01 | 55.4,55.4,62.1,62.1,62.1,79.2,79.2,79.2,106.2,106.2 | 73 |
| CTAR01 | 106.2,124.7,124.7,124.7,148.4,148.4,148.4,179.5,179.5 | 83 |
| CTAR01 | 179.5,221.6,221.6,221.6,221.6,221.6,221.6,280.5,280.5 | 92 |
| CTAR01 | 366.3,498.6,1121.9,1994.5,17950.5 | 101 |
| CTAR01 | 2090.,1690.,1690.,1184.,773.,773.,482.,366.,366.,264. | 106 |
| CTAR01 | 199.,199.,155.,139.,139.,125.,133.,133.,141.,172.,172. | 116 |
| CTAR01 | 231.,290.,290.,342.,451.,451.,555.,795.,795.,1054. | 127 |
| CTAR01 | 1463.,1463.,2725.,4756.,4756.,10320.,37168.,37168. | 137 |
| CTAR01 | 122381.,3559184.,3559184.,3559184.,3559184.,3559184. | 145 |
| CTAR01 | 3559184.,0.,0.,0.,0.,0.,2090.,1690.,1690.,1184. | 151 |
| CTAR01 | 773.,773.,482.,366.,366.,264.,199.,199.,155.,139.,139. | 161 |
| CTAR01 | 125.,133.,133.,141.,172.,172.,231.,290.,290.,342.,451. | 172 |
| CTAR01 | 451.,555.,795.,795.,1054.,1463.,1463.,2725.,4756.,4756. | 183 |
| CTAR01 | 10320.,37168.,37168.,122381.,3559184.,3559184.,3559184. | 193 |

Figure 3-a. Cont'd

| | | |
|--------|---|-----|
| CTAR01 | 3559184.,3559184.,3559184.,0.,0.,0.,0.,0. | 200 |
| INDC02 | 1 | |
| CTAR02 | .0.,.066,.1.,.108,.149,.183,.191,.233,.267,.274,.316 | |
| CTAR02 | .35,.358,.399,.433,.441,.483,.517,.524,.566,.6,.608 | 12 |
| CTAR02 | .649,.683,.691,.733,.767,.774,.816,.85,.858,.899,.933 | 23 |
| CTAR02 | .941,.983,1.017,1.024,1.066,1.1,1.108,1.149,1.183 | 34 |
| CTAR02 | 1.267,1.323,1.351,1.433,1.517,1.6,1.683,1.767,2.0 | 43 |
| CTAR02 | 1.,2.,3.,32.3,32.3,29.1,29.1,29.1,27.4,27.4 | 52 |
| CTAR02 | 27.4,28.1,28.1,28.1,33.9,33.9,33.9,44.9,44.9,44.9,55.4 | 62 |
| CTAR02 | 55.4,55.4,62.1,62.1,62.1,79.2,79.2,79.2,106.2,106.2 | 73 |
| CTAR02 | 106.2,124.7,124.7,124.7,148.4,148.4,148.4,179.5,179.5 | 83 |
| CTAR02 | 179.5,221.6,221.6,221.6,221.6,221.6,221.6,280.5,280.5 | 92 |
| CTAR02 | 366.3,498.6,1121.9,1994.5,17950.5 | 101 |
| CTAR02 | 2090.,1690.,1690.,1184.,773.,773.,482.,366.,366.,264. | 106 |
| CTAR02 | 199.,199.,155.,139.,139.,125.,133.,133.,141.,172.,172. | 116 |
| CTAR02 | 231.,290.,290.,342.,451.,451.,555.,795.,795.,1054. | 127 |
| CTAR02 | 1463.,1463.,2725.,4756.,4756.,10320.,37168.,37168. | 137 |
| CTAR02 | 122381.,3559184.,3559184.,3559184.,3559184.,3559184. | 145 |
| CTAR02 | 3559184.,0.,0.,0.,0.,0.,2090.,1690.,1690.,1184. | 151 |
| CTAR02 | 773.,773.,482.,366.,366.,264.,199.,199.,155.,139.,139. | 161 |
| CTAR02 | 125.,133.,133.,141.,172.,172.,231.,290.,290.,342.,451. | 172 |
| CTAR02 | 451.,555.,795.,795.,1054.,1463.,1463.,2725.,4756.,4756. | 183 |
| CTAR02 | 10320.,37168.,37168.,122381.,3559184.,3559184.,3559184. | 193 |
| CTAR02 | 3559184.,3559184.,3559184.,0.,0.,0.,0.,0. | 200 |
| INDC03 | 1 | |
| NS2NDY | 2 | |
| CTAR03 | 0.,.1542,1.,2.,3. | |
| CTAR03 | 0.,0.,28.3,28.3,28.3,28.3 | 6 |
| INDC04 | 1 | |
| CTAR04 | 0.,.1542,1.,2.,3. | |
| CTAR04 | 0.,0.,28.3,28.3,28.3,28.3 | 6 |
| MASS? | .0001,.0986,.0986 | |
| ES | .02,.013,.013 | |
| MIS | .15,.15,.15 | |
| SR | 2.,1.323,1.323 | |
| PEM | RCD 4FLEXIBLE AIRFRAME DATA | |
| INDFLY | 0 | |
| NMODE | 5 | |
| GMASS1 | 22.68,2.88,14.28,5.88,19.02 | |
| GFREQ | 50.868,64.684,80.384,116.180,133.764 | |
| GDAMP | 28.8574,4.6597,27.3655,17.0862,9.5505 | |
| S7MOD | .535,.347,.347,-.333,.333,.333,-.411,.115,.115 | |
| S7MOD | .294,-.282,-.282,.443,.117,.117 | 10 |
| ARMODE | 0.,0.,.05,0.,0.,0.,0.,0.,.037,0.,0.,0. | |
| ARMODE | 0.,0.,-.075,0.,0.,0.,0.,0.,-.050,0.,0.,0. | 13 |
| ARMODE | 0.,0.,-.001,0.,0.,0. | 25 |
| DCMODE | 0.,0.,.987,0.,0.,-.88,0.,0.,-.964 | |
| DCMODE | 0.,0.,.541,0.,0.,-.647 | 10 |
| PF | 0.,0.,0.,0.,0.,0.,0.,0.,0.,0.,0.,0. | |

Figure 3-a. Cont'd

Figure 3-a. Cont'd

| | | | |
|--------|-----|-------|----------------------------|
| KE | INT | 0,0 | |
| PM | | 10.0 | |
| REM | RCD | 5H. | LANDING ROLL MANEUVER DATA |
| TSD | | .5 | |
| TRV | | .1 | |
| TCH | | 4. | |
| TRK | | 3. | |
| TSS | | 0. | |
| TLR | | 0 | |
| TRS | | 0 | |
| REM | RCD | 5I. | ENGINE FAILURE STAGE DATA |
| TC | INT | 1,1 | |
| YRF1 | | 0. | |
| IT1 | INT | 1,1 | |
| YRF2 | | 0. | |
| IT2 | INT | 1,1 | |
| H1 | | 0. | |
| IH1 | INT | 1,1 | |
| H2 | | 0. | |
| IH2 | INT | 1,1 | |
| HR1 | | 0. | |
| IHR1 | INT | 1,1 | |
| HR2 | | 0. | |
| IHR2 | INT | 1,1 | |
| TR1 | | 0. | |
| ITR1 | INT | 1,1 | |
| TR2 | | 0. | |
| ITR2 | INT | 1,1 | |
| REM | RCD | 5J. | PRAKE COND.STAGE DATA |
| TB | INT | 0,0,0 | |
| TRK1 | | 100. | |
| TRK1 | INT | 0,1,1 | |
| TRK2 | | 100. | |
| TRK2 | INT | 0,1,1 | |
| REM | RCD | 4K. | PITCH AUTOPILOT DATA |
| TST | | 0. | |
| ALPDL | | 2. | |
| REALPH | | 0. | |
| DELALA | | .01 | |
| PSH | | 0. | |
| PSH2 | | 0. | |
| REALP2 | | 0. | |
| DELQF | | 0. | |
| DELQTO | | 0. | |
| DELQO | | 7. | |
| DELQOL | | -21. | |
| DELQO1 | | 0. | |
| REM | RCD | 4L. | YAW AUTOPILOT DATA |
| REP | | .5 | |

| | |
|--------|---------------------------------|
| DELRA | .05 |
| PSP | 10. |
| DPSTA | .05 |
| REFSI | .5 |
| PSPSI | 0. |
| DELRL | -30. |
| DELRL | 30. |
| REM | BCD 4M. ROLL AUTOPILOT DATA |
| REFHI | .5 |
| DPHIA | .05 |
| PSA | -10. |
| DELPL | 20. |
| DELPII | -1. |
| REM | BCD 5N. THROTTLE AUTOPILOT DATA |
| NDF | 1.,1. |
| TF | TNT 1,1 |
| TR | INT 0,0 |
| NR | 0.,0. |
| NLR | 0.,0. |
| NTD | 2.,2. |
| K2 | .5 |
| DEM | BCD 40. BRAKE AUTOPILOT DATA |
| MRC | 0.,0.,0. |
| PD | .155 |
| DELTAW | .1 |
| OMECD1 | 20. |
| MRL | 0.,0.,0. |
| MRU | 1.0E+4,1.0E+4,1.0E+4 |
| REM | BCD 5P. CONTROL RESPONSE DATA |
| DELHS | 25. |
| DELPPD | 30. |
| DELA | 120. |
| NED1 | .1 |
| REM | BCD 40. INITIALIZATION |
| TAP | 4 |
| HR | 8.23743 |
| DELON | -12. |
| DELODE | -12. |
| DELOD | -12. |
| DELOU | 7. |
| DELOL | -21. |
| DELOF | -21.0 |
| DELPD | 0. |
| DELRD | 0. |
| MANLOG | 1 |
| PITCHP | 1 |
| ROLLAP | 1 |
| THROAP | 1 |
| CONTRP | 1 |

Figure 3-a. Cont'd

```

AUXICP      1
REM      RCD 2STAGING DATA
REM      RCD 4A.GEARS INTO PROGRAM
INDLG      -3
AINCPS RCD 1TIME
STEST      .005
          TPA

STCASE
DELED1      20.
AINCPS RCD 1TIME
STEST      .02
          TRA

STCASE
PRINT      .001
AINCPS RCD 1TIME
STEST      .04
          TRA

STCASE
PRINT      .01
AINCPS RCD 1TIME
STEST      .3
          TRA

STCASE
PRINT      .01
AINCPS RCD 1DELTA1
STEST      .05
          TRA

STCASE
ICS         1
DELOF      0.
DELED1      25.
AINCPS RCD 1TIME
STEST      .60
          TPA

STCASE
ATAB10     .35,.35
ATAB20     .15,.15
AINCPS RCD 1TIME
STEST      .70
          TPA

STCASE
AMAXER     .0001
ATAB10     .325,.325
ATAB20     .16,.16
AINCPS RCD 1TIME
STEST      .80
          TRA

STCASE
ATAB10     .30,.30

```

| | |
|------------|-----------|
| ATARPO | .17,.17 |
| ATNCPS RCD | 1TIME |
| STEST | .90 |
| TRA | |
| STCASE | |
| ATARPO | .275,.275 |
| ATARPO | .18,.18 |
| ATNCPS RCD | 1TIME |
| STEST | 1.0 |
| TRA | |
| STCASE | |
| ATARPO | .25,.25 |
| ATARPO | .19,.19 |
| ATNCPS RCD | 1TIME |
| STEST | 1.1 |
| TRA | |
| STCASE | |
| ATARPO | .20,.20 |
| ATNCPS RCD | 1TIME |
| STEST | 1.2 |
| TRA | |
| STCASE | |
| ATARPO | .21,.21 |
| ATNCPS RCD | 1TIME |
| STEST | 1.3 |
| TRA | |
| STCASE | |
| ATARPO | .22,.22 |
| ATNCPS RCD | 1TIME |
| STEST | 1.4 |
| TRA | |
| STCASE | |
| ATARPO | .23,.23 |
| ATNCPS RCD | 1TIME |
| STEST | 1.5 |
| TRA | |
| STCASE | |
| ATARPO | .24,.24 |
| ATNCPS RCD | 1TIME |
| STEST | 1.6 |
| TRA | |
| STCASE | |
| ATARPO | .25,.25 |
| ATNCPS RCD | 1TIME |
| STEST | 4.0 |
| TRA | |
| STCASE | |
| AMINER | .00005 |
| AMAXER | .001 |

Figure 3-a. Cont'd

```

3
STCASE
AMAXER      .0005
ATARIO      .35,.35
ATABRO      .15,.15
AINCRS RCD 1TIME
STEST       .70
          TRA

STCASE
AMAXER      .0001
ATARIO      .325,.325
ATABRO      .16,.16
AINCRS RCD 1TIME
STEST       .80
          TRA

STCASE
ATARIO      .30,.30
ATABRO      .17,.17
AINCRS RCD 1TIME
STEST       .90
          TRA

STCASE
ATARIO      .275,.275
ATABRO      .18,.18
AINCRS RCD 1TIME
STEST       1.0
          TRA

STCASE
ATARIO      .25,.25
ATABRO      .19,.19
AINCRS RCD 1TIME
STEST       1.1
          TRA

STCASE
ATABRO      .20,.20
AINCRS RCD 1TIME
STEST       1.2
          TRA

```

Figure 3-b. Short Input Staging Deck for a restart at 0.60 second.

ORIGINAL PAGE IS
OF POOR QUALITY

STCASE
ATARPO .21,.21
AINCRS RCD 1TIME
STEST 1.3
TRA

STCASE
ATARPO .22,.22
AINCRS RCD 1TIME
STEST 1.4
TRA

STCASE
ATARPO .23,.23
AINCRS RCD 1TIME
STEST 1.5
TRA

STCASE
ATARPO .24,.24
AINCRS RCD 1TIME
STEST 1.6
TRA

STCASE
ATARPO .25,.25
AINCRS RCD 1TIME
STEST 4.0
TRA

STCASE
AMTNER .00005
AMAYER .001
AINCRS RCD 1TIME
STEST 10.
TRA

for the nose and main gears are stored in STRON and STROM, respectively. The dependent variable (metering pin area) for the nose and main gears are stored in PINN and PINM, respectively. The active variable list must be terminated with the word ENDACT, which also begins in card column 1.

The INDLG switch is located at the beginning of the staging data section and follows the same format as outlined above. It is also recommended that the minimum integration interval (AMINER) and the maximum integration interval (AMAXER) be set to .00005 and .001 respectively to permit accurate numerical integration during ground contact stages. These variables are located in the integration information section near the beginning of the data set.

In order to specify time rates of change for the aerodynamic coefficients C_{A_0} and C_{N_0} , four new symbols have been added to the program: RTAB10, RTAB80, LTAB10, and LTAB80. These symbols, unlike ATAB10 and ATAB80, should not be included in table size data but may appear anywhere else in the input deck. As with other aerodynamic tables, the new symbols have a size of two. The first data point is for full ground effect and the second is for no ground effect. The use of these symbols is illustrated by an example.

ATAB10 .379202,.379202

.

.

ATAB80 .141245,.141245

.

.

AINCRS BCD 1TIME

STEST .005

TRA

STCASE

RTAB10 -.01,-.01

LTAB10 .36,.36

RTAB80 .005,.005

LTAB80 .15,.15

.

.

.

In this example, the initial values of C_{N_0} and C_{A_0} are .379202 and .141245, respectively. At time $t=.005$, rates of change and limiting values for the parameters are specified. The value of C_{N_0} (ATAB10) will decrease at a rate of .01 (RTAB10) until it has reached a value of .36 (LTAB10). At this time (1.9209 seconds after the stage) the rate of change will be set to zero and C_{N_0} will remain at .36 until it is redefined by a new ATAB10 card or by new values of RTAB10 and LTAB10. Similarly, the value of C_{A_0} (ATAB80)

will increase at a rate of .005 until it has reached a value of .15 (LTAB80). At this time (1.751 seconds after the stage) the rate of change will be set to zero and C_{A_0} will remain at .15 until it is redefined.

3.2 Operating Instructions - The program resides as an UPDATE program library on an indirect access file named ACTOLA. The relocatable binaries are on a file named BTOLAR. The plot program source and binaries are on files NEWPLOT and NAPBN, respectively.

The user is responsible for maintaining the files necessary for restart runs and for plotting the cumulative results of several restart runs. The program files of concern to the user are TAPE7, TAPE16, and TAPE13. The TAPE7 file contains the restart data with one record for each restart generated (i.e., one record for each time data is staged). In subsequent runs, it is the responsibility of the user to correctly position TAPE7. The TAPE16 file also contains restart information, but contains a single record. The file TAPE13 contains data for the plot postprocessor and contains a single record. The use, storage, and manipulation of these files is best illustrated by a set of example runs.

Figure 4-a illustrates the simplest use of the FATOLA program. This run will execute the program without the restart option and no plots will be generated.

```

JOB,...
USER,...
CHARGE,...
GET,BTOLAR/UN=585787N.
MAP,OFF.
BTOLAR.
---EOR
    Full data deck (similar to Figure 3-a)
---EOF

```

Figure 4-a. Deck setup for simple execution of the program.

```

JOB,...
USER,...
CHARGE,...
GET,BTOLAR/UN=585787N.
MAP,OFF.
BTOLAR.
GET,NAPBN/UN=585787N.
ATTACH,LRCGOSF/UN=LIBRARY.
REWIND,TAPE13.
COPYEI,TAPE13,TAPE3.
RETURN,TAPE13.
REWIND,TAPE3.
LDSET,LIB=LRCGOSF.
NAPBN.
PLOT.VARIAN
---EOR
    Full data deck (similar to Figure 3-a)
---EOR
    Plot instructions
---EOF

```

Figure 4-b. Deck setup for simple execution of the program with plotting.

```

JOB,...
USER,...
CHARGE,...
GET,OLDPL=ACTOLA/UN=585787N.
UPDATE.
FTN,I.
REWIND,LGO.
GET,OLDBIN=BTOLAR/UN=585787N.
COPYL,OLDBIN,LGO,BTOLAR.
MAP,OFF.
BTOLAR.
REPLACE,TAPE13=T134C.
---EOR
    UPDATE correction set
---EOR
    Full data deck (similar to Figure 3-a)
---EOF

```

```

JOB,...
USER,...
CHARGE,...
GET,TAPE3=T134C.
GET,NAPBN/UN=585787N.
ATTACH,LRCGOSF/UN=LIBRARY.
LDSET,LIB=LRCGOSF,MAP=SBEX/OTT.
NAPBN.
PLOT.VARIAN
---EOR
    Plot instructions
---EOF.

```

Figure 4-c. Decks for executing the program with corrections and plotting the results at a later time.

```

JOB,...
USER,...
CHARGE,...
GET,BTOLAR/UN=585787N.
MAP,OFF.
BTOLAR.
REPLACE,TAPE7=T74D.
REPLACE,TAPE16=T164D.
REPLACE,TAPE13=T134D.
EXIT.
REPLACE,TAPE7=T74DF.
REPLACE,TAPE16=T164DF.
REPLACE,TAPE13=T134DF.
---EOR
    Full data deck (similar to Figure 3-a with a restart flag of 1.)
---EOF

```

Figure 4-d. Deck setup for an initialization run with the restart option.

```

JOB,...
USER,...
CHARGE,...
GET,BTOLAR/UN=585787N.
MAP,OFF.
GET,TAPE16=T164DF.
GET,TAPE7=T74DF.
SKIPR,TAPE7,5.
BTOLAR.
REPLACE,TAPE7=T74E.
REPLACE,TAPE16=T164E.
REPLACE,TAPE13=T134E.
EXIT.
REPLACE,TAPE7=T74EF.
REPLACE,TAPE16=T164EF.
REPLACE,TAPE13=T134EF.
---EOR
    Short data deck (similar to Figure 3-b with a restart flag of 3.)
---EOF

```

Figure 4-e. Deck setup for restarting example from Figure 4-d at time = 0.60 sec.

```

JOB,...
USER,...
CHARGE,...
GET,NAPBN/UN=585787N.
ATTACH,LRCGOSF/UN=LIBRARY.
MAP,OFF.
GET,T134DF.
GET,T134E.
COPYBR,T134DF,TAPE3.
COPYBR,T134E,TAPE3.
REWIND,TAPE3.
LDSET,LIB=LRCGOSF.
NAPBN.
PLOT.VARIAN
---EOR
    Plot instructions
---EOF

```

Figure 4-f. Plotting cumulative results of restart runs.

When plots are desired, they can either be generated at the same time as the program is executed, or they can be generated at a later time after the user has examined the printed output. Figure 4-b shows the deck setup for executing the program and generating plots at the same time. Figure 4-c contains the deck setups for executing the program with UPDATE modifications and plotting the output at a later time.

When the restart option is selected, the first run is known as the initialization run and the first data card should contain a 1 in the first card column. The user should save the files TAPE7 and TAPE16, and if plots will be desired TAPE13. Figure 4-d contains the deck setup for an initialization run. Note that important files are saved if the program terminates normally, and these files are also saved if the program fails. Assume, as an example, the program fails at time = 0.65 seconds, using the data of Figure 3-a. An analysis of the program output suggests that if a smaller integration step were being used, the failure might have been avoided.

To correct the problem, it is first necessary to determine the TAPE7 record structure. From our sample data or from the program output it can be determined that the first restart record was written at time = 0.005, the second was written at time = 0.02, etc. Thus, the record written at

0.60 was the sixth restart record generated. Figure 4-e illustrates a deck setup for restarting this run. Note that 5 records are skipped on TAPE7, positioning the file to the sixth record. Note also the following points:

- o It is not required to get TAPE13 before executing the program.
- o The permanent file names T164DF and T74DF correspond to the file names under which the files TAPE16 and TAPE7 were replaced (after the EXIT.) in Figure 4-d
- o Following execution, the restart and plot files are saved with new (unique) names. In subsequent runs, T74DF could be used to restart the program prior to time = 0.60 and T74E (or T74EF) could be used to restart the program after time = 0.60.

Assume, to continue our example, the restart run terminates normally and we desire to plot the cumulative results of the two runs. Figure 4-f presents a deck setup to plot the results of our example. Notice that T134DF (after the exit) and T134E (before the exit) correspond to the file names under which TAPE13 was saved in Figures 4-d and 4-e. This example is easily extended to as many restart runs as necessary to complete the simulation. A GET is required for each TAPE13 saved and a COPYBR is also required, with the COPYBR's in the same order in which the files were created.

REFERENCES

1. Carden, Huey D.; and McGehee, John R.: Validation of a Flexible Aircraft Takeoff and Landing Analysis (FATOLA). NASA TP-1025, 1977.
2. Carden, Huey D.; and McGehee, John R.: Improvements to the FATOLA Computer Program Including Nosewheel Steering. NASA TM-78768, 1978.
3. Dick, J. W.; and Benda, B. J.: Addition of Flexible Body Option to the TOLA Computer Program, Part I - Final Report; Part II - User Programmer Documentation. NASA CR-132732-1, Part I and NASA CR-132732-2, Part II, 1975.
4. Lynch, Urban H. D.: Takeoff and Landing Analysis (TOLA) Computer Program. Part I: Capabilities of the Takeoff and Landing Analysis Computer Program. AFFDL-TR-71-155 Part 1. 1972.
5. Lynch, Urban H. D.; and Dueweke, John J.: Takeoff and Landing Analysis (TOLA) Computer Program. Part II: Problem Formulation, Technical Report AFFDL-TR-71-155, May, 1974.
6. ———: Takeoff and Landing Analysis (TOLA) Computer Program. Part III: User's Manual. AFFDL-TR-71-155, 1974.
7. McGehee, John R.; and Carden, Huey D.: A Mathematical Model of an Active Control Landing Gear for Load Control During Impact and Rollout. NASA TN D-8080, 1976.
8. Young, Fay O.; and Dueweke, John J.: Takeoff and Landing Analysis (TOLA) Computer Program; Part IV - Programmer's Manual. AFFDL-TR-71-155, Part 4, 1975.
9. McGehee, John R.; and Dreher, Robert C.: Experimental Investigation of Active Loads Control for Aircraft Landing Gears. NASA TP 2042, 1982.
10. Ross, Irving; and Edson, Ralph: An Electronic Control for an Electrohydraulic Active Control Aircraft Landing Gear. NASA CR-3113, 1979.

Appendix A
Program Modifications

```
*IDENT      081877
*DELETE      PLTDAT.3,PLTDAT.3
             DIMENSION TITLE(18), BUF(400), NDIL(28), TBUF(400),
*DELETE      PLTDAT.125,PLTDAT.125
             50 DO 51 I=1,18
*IDENT      C82477
*INSERT      TOLA.6
             COMMON/NWSTER/CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE
             1 ,EA,ETAMAX,ETAMIN,ETANOS,ETART1,INDNWS,KANWS,KPNWS,PCTETA
             2 ,PSIDES,RPSI,RYR,VPOWR,YRDES
             3 ,DELN,ETADES,FGPY
*INSERT      TOLA.9
             1 ,CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE,EA,ETAMAX
             2 ,ETAMIN,ETANOS,ETART1,INDNWS,KANWS,KPNWS,PCTETA,PSIDES,RPSI
             3 ,RYR,VPOWR,YRDES
*INSERT      TOLA.19
             READ(5,5004)CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE,EA
             1 ,ETAMAX,ETAMIN,ETANOS,ETART1,PCTETA,PSIDES,RPSI,RYR,VPOWR
             2 ,YRDES
*INSERT      TOLA.20
             READ(5,5005)INDNWS,KANWS,KPNWS
             5005 FORMAT(3I5)
*INSERT      LGEA3C.2
             EXTERNAL ATAN2
*INSERT      LGEA3C.34
             COMMON/NWSTER/CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE
             1 ,EA,ETAMAX,ETAMIN,ETANOS,ETART1,INDNWS,KANWS,KPNWS,PCTETA
             2 ,PSIDES,RPSI,RYR,VPOWR,YRDES
             3 ,DELN,ETADES,FGPY
*INSERT      LGEA3C.59
             EQUIVALENCE (DM18(72),PSIPD)
*INSERT      LGEA3C.61
             DATA RADDEG,DEGRAD/57.2957795,.01745329/
*DELETE      LGEA3C.143,LGEA3C.143
             IF(I.EQ.1 .AND. INDNWS.EQ.1)GO TO 200
             GO TO 7
             200 VTX(I)=VTX(I)-TMP(2)
             VTY(I)=RDYG(I)
             7 VTZ(I)=RG31*RDYG(I)+RG33*RDZG(I)
*INSERT      LGEA3C.167
             IF(DELTA1 .EQ. 0.0) FGPY=0.0
             IF(DELTA1 .EQ. 0.0) GO TO 41
             IF(INDNWS.EQ.0)GO TO 41
             IF(I.GT.1)GO TO 41
             ETAVE= ATAN2(VTY(I),VTX(I))
             DELN=PSIPD+ETADES-ETAVE*RADDEG
             IF(DELN.EQ.0.0.OR.XG77F1.EQ.0.0)GO TO 110
             SIDEMU=CONFRI*(ABS(DELN))*DELPWR*XG77F1**VPOWR
             GO TO 180
```



```

110 SIDEMU=1.0E-6
180 IF((DELTA1/DTIRE).GT.0.0875)GO TO 130
    CRNPWR=CONE*DELTA1-CTWO*DELTA1**2.
    GO TO 170
130 CRNPWR=CTHREE-CFOUR*DELTA1
170 YAWPRM=ABS(CRNPWR*DELN/(SIDEMU*FTRZ(I)))
    IF(YAWPRM.LE.1.5)GO TO 150
    FGPY=SIDEMU*FTRZ(I)*COS(DELN*DEGRAD)
    GO TO 160
150 FGPY=((YAWPRM-(4.0/27.0)*YAWPRM**3.0)*SIDEMU*FTRZ(I))
    1 *COS(DELN*DEGRAD)
160 FGPY=PCTETA*FGPY
    IF(DELN.GT.0.0)FGPY=ABS(FGPY)
    IF(DELN.LT.0.0)FGPY=-ABS(FGPY)
    DFTRX=FGPY*SIN(ETAVE)
    DFTRY=FGPY*COS(ETAVE)
    FTRX(I)=FTRX(I)+DFTRX
    FTRY(I)=FTRY(I)+DFTRY
*INSERT      LGEA3C.170
    FGPY = 0.0
*INSERT      SDFLGP.34
    COMMON/NWSTER/CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE
    1 ,EA,ETAMAX,ETAMIN,ETANDS,ETART1,INDNWS,KANWS,KPNWS,PCTETA
    2 ,PSIDES,RPSI,RYP,VPOWR,YRDES
    3 ,DELN,ETADES,FGPY
*DELETE      SDFLGP.59,SDFLGP.60
    DIMENSION OP16(18),OP17(8),OP18(8),OP19(7),OP20(8),OP21(8),
    1 DAT2(18),DAT3(6),DAT4(15)
*DELETE      SDFLGP.67,SDFLGP.67
    *(OP19(I),I=1,7)/3HFZM,2HLM,2HMM,2HNM,4HFGPY,4HDELN,6HETADES/,
*DELETE      SDFLGP.73,SDFLGP.73
    DATA DAT1/4HTIME/, (DAT2(I),I=1,18)/2HLM,2HMM,2HNM,5HQI77R,
*DELETE      SDFLGP.75,SDFLGP.75
    *5HXG77F,5HYG77F,5HAX77F,6HXG77F1,4HFGPY,4HDELN,6HETADES/,
*INSERT      SDFLGP.80
    DATA N1/1/,N15/15/,N14/14/,N18/18/
*DELETE      SDFLGP.95,SDFLGP.95
    IF(ISDF.NE.0)WRITE(13)N18,N1,DAT2
*INSERT      SDFLGP.102
    2 ,FGPY,DELN,ETADES
*DELETE      SDFLGP.164,SDFLGP.165
    CALL STFL(2,7,OP19)
    CALL STOVAR(7,FZM,LM,MM,NM,FGPY,DELN,ETADES,DU)
*INSERT      AUTS.74
    COMMON/NWSTER/CFOUR,CONE,CONFRI,CTHREE,CTWO,DELPWR,DTIRE
    1 ,EA,ETAMAX,ETAMIN,ETANDS,ETART1,INDNWS,KANWS,KPNWS,PCTETA
    2 ,PSIDES,RPSI,RYP,VPOWR,YRDES
    3 ,DELN,ETADES,FGPY
*INSERT      AUTS.81

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DATA ILIM, INOSE/0,0/
*INSERT      AUTS.102
  IF(DELTA1 .EQ. 0.0) GO TO 8
  IF(INDNWS.EQ.0)GO TO 8
  IF(KANWS.EQ.1)GO TO 7
C      CURRENTLY, NOSEWHEEL STEERING IS FOR DISTANCE ERROR
C      NOT ANGLE ERROR
  IF(KPNWS.EQ.1)GO TO 9
  ETADES=ETANOS
  GO TO 8
7  ERROR=(PSIPD-PSIDES)+RPSI*PSIPD1*RADDEG
  GO TO 3
9  ERROR=(YR-YRDES)+RYR*YRD1
3  IF(ILIM .EQ. 1 .AND. ETADES .EQ. 0.0) INOSE=0
  IF(INOSE .GT. 0 .AND. ETADES .EQ. 0.0) GO TO 8
  IF(ABS(ERROR) .GT. EA) GO TO 30
  GO TO 32
30 IF(ERROR .GT. 0.0) GO TO 37
  ETADES=ETADES+ETART1*DELTS
  INOSE=1
  GO TO 31
37 ETADES=ETADES-ETART1*DELTS
  INOSE=2
31 IF(ETADES .GT. ETAMAX) ETADES=ETAMAX
  IF(ETADES .LT. ETAMIN) ETADES=ETAMIN
  GO TO 8
32 IF(INOSE .EQ. 2) GO TO 38
  GO TO 39
38 ETADES=ETADES+ETART1*DELTS
  ILIM=1
  GO TO 99
39 ETADES=ETADES-ETART1*DELTS
  ILIM=1
99 IF(ABS(ETADES) .LT. 1.0) ETADES = 0.0
8  CONTINUE
*IDENT      C11778
*DELETE     TOLA.4,C82477.4
*DELETE     TOLA.9,C82477.7
*DELETE     TOLA.15,TOLA.21
*INSERT     EXE.32
      *,DM64(30),GAMA,DM65(15),PCTETA,DM66(13),AH(5),DM67(35)
*DELETE     EXE.54,EXE.54
      34 DO 35 I=1,4029
*INSERT     EXE.68
      DO 30 I=1,5
      30 AH(I)=1.0
*INSERT     EXE.79
      PCTETA=1.0
      GAMA=1.0

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*INSERT      INUPD.4
      1,DM1(100)
*INSERT      UPDAT.3
      1,DM2(100)
*INSERT      MIMIN.6
      *,DM8(100)
*INSERT      LGDET.8
      C,DM9(100)
*DELETE      LGDET.33,LGDET.33
      49 FORMAT(58X,4H-ES(,I1,19H) EXCEEDED IN LGDET/
*INSERT      LGDET.34
      Y(J)=-.5*ES(I)
      Y(JJ)=-1.0E-10
      P(JJ)=-1.0E-10
*DELETE      LGDET.47,LGDET.50
      51 IF(P(JJ).LT.0.)GO TO 30
      IF(Y(JJ).LT.0.)Y(JJ)=0.
      GO TO 55
      30 P(J)=0.
      P(JJ)=0.
*INSERT      STGTSI.7
      *,DM8(100)
*INSERT      STGTST.6
      4,DM8(100)
*INSERT      DEF.5
      *,DM5(100)
*INSERT      LINES.4
      C,D4(100)
*INSERT      ERROR.4
      C,DM3(100)
*INSERT      EXERR.3
      C,DM3(100)
*INSEPT      ATMS.10
      4,DM8(100)
*INSERT      TFFS1.16
      *,DM16(100)
*INSERT      TFFS8.16
      *,DM16(100)
*DELETE      VPCS1.23,VPCS1.23
      9XCGBF      ,XCGRF      ,DM19( 930),DM20(2068),DM21(100)
*INSERT      SACS1.63
      *,DDM78(100)
*INSERT      OPT1.72
      *,DM158(100)
*INSERT      LGEAR1.30
      *,DDM21(60),AH(5),PH(5),DDM22(30)
*INSERT      LGEAR1.203
      IF(P2(I).GT.0.)GO TO 1000
      PH(I)=(P(I)*AH(I)-FC2(I))/AH(I)

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      IF(PH(I).LE.-1600.)GO TO 1003
*DELETE      LGEAR1.204,LGEAR1.204
      1000 SF(I)=-P(I)*(A(I)-A2(I))-P2(I)*A2(I)+FC2(I)-S2D1(1,I)*
*INSERT      LGEAR1.205
      GO TO 1002
      1003 SF(I)=-P(I)*(A(I)-AH(I))+1600.*AH(I)-FF(I)*TMP(2)
*DELETE      LGEAR1.206,LGEAR1.206
      1002 IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(SF(I)))SF(I)=-FT(I)
*DELETE      LGEAR1.222,LGEAR1.222
      49 FORMAT(58X,4H-ES(,I1,20H) EXCEEDED IN LGEAR1/
*INSEPT      LGEAR1.223
      S(1,I)=-0.5*ES(I)
      SD1(1,I)=-1.0E-10
      SD2(1,I)=-1.0E-10
*INSERT      LGEAR1.229
      S(1,I)=0.5*ES(I)
*DELETE      LGEAR1.234,LGEAR1.235
      51 IF(SD2(1,I).LT.0.)GO TO 30
      IF(SD1(1,I).LT.0.)SD1(1,I)=0.
      GO TO 55
      30 SD2(1,I)=0.
      SD1(1,I)=0.
*INSERT      LGEAR1.240
C      RE-CHECK SHOCK STRUT FORCE FOR RE-CONDITIONED FULLY EXTENDED STATE
      IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(SF(I)))SF(I)=-FT(I)
*INSERT      LGEA3C.30
      *,DDM30(20),SLEN1(5),SLEN2(5),GAMA,CFOUR,CONE,CONFRI,CTHREE,CTWO,
*DELPWR,DTIRE,DDM31(5),INDNWS,DDM32(2),PCTETA,DDM33(3),VPOWR,
*DDM34,FGPY,DELN,ETADES,DDM35(45)
*DELETE      LGEA3C.32,LGEA3C.33
*DELETE      C82477.14,C82477.17
*INSERT      LGEA3C.59
      EQUIVALENCE (DM18(31),PA77P)
*DELETE      LGEA3C.105,LGEA3C.106
      P(I) = (PZERO(I)+PA77P)*(VZERO(I)/(VZERO(I)+A2(I)*S2(1,I)-
      *S(1,I)*A(I)))*GAMA-PA77P
*DELETE      LGEA3C.110,LGEA3C.110
      P2(I) = (P20(I)+PA77P)*(V20(I)/TMP(1))*GAMA-PA77P
*DELETE      C82477.22,C82477.23
      200 IF(ABS((OMET(1,I)*TMP(1))*COS((PSIPD+ETADES)*DEGRAD)*RI(2,2,I))
      1 .GE. (RG11*RDYG(I)+RG13*RDZG(I)))GO TO 201
      VTX(I)=VTX(I)-TMP(2)+(OMET(1,I)*TMP(1))*COS((PSIPD+ETADES)*
      1 DEGRAD)*RI(2,2,I)
      VTY(I)=RDYG(I)-(OMET(1,I)*TMP(1))*SIN((PSIPD+ETADES)*DEGRAD)*
      1 RI(2,1,I)
*INSERT      C82477.24
      GO TO 203
      201 VTX(I)=1.E-10
      OMET(1,I)=-((RG11*RDYG(I)+RG13*RDZG(I))/(RZERO(I)-DELTA(I)))

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      VTY(I)=RDYG(I)-(DMET(1,I)*TMP(1))*SIN((PSIPD+ETADES)*DEGRAD)*
1  RI(2,1,I)
      VTZ(I)=RG31*RDYG(I)+RG33*RDZG(I)
*DELETE      LGEA3C.145,LGEA3C.145
      203 TMP(1)=RG11*RDYG(I)+RDZG(I)*RG13
*DELETE      C82477.25,C82477.27
*DELETE      C82477.29,C82477.29
      ETAVE=ATAN2(RDYG(I),TMP(1))
*INSERT      C82477.30
      IF(DELTA1 .EQ. 0.0) FGPY = 0.0
      IF(DELTA1 .EQ. 0.0)GO TO 41
*INSERT      FLEX1.17
      *,GDAMP(20),DDM25(80)
*DELETE      FLEX1.18,FLEX1.18
*INSERT      SDFLGP.31
      C,DM18(52),FGPY,DELN,ETADES,DM19(45)
*DELETE      C82477.53,C82477.56
*DELETE      STORE.5,STORE.5
      COMMON/DIRCOM/DATA(4029)
*DELETE      DSERCH.3,DSERCH.3
      COMMON/FIXDIR/ANAME(1000),LOC(1000),NCOUNT
*DELETE      DIRODA.3,DIRODA.3
      COMMON/FIXDIR/NAME(1000),LOC(1000),NCOUNT
*DELETE      DIR1DA.3,DIR1DA.3
      COMMON/FIXDIR/NAME(1000),LOC(1000),NCOUNT
*DELETE      DIR2DA.3,DIR2DA.4
      COMMON/FIXDIR/NAME(1000),LOC(1000),NCOUNT
      DATA NCOUNT/929/
*DELETE      DIR3DA.3,DIR3DA.4
      COMMON/FIXDIR/NAME(1000),LOC(1000),NCOUNT
      DATA (NAME(K3),K3=876,929)/
*DELETE      DIR3DA.7,DIR3DA.8
      *      6HPF      ,6HGQ      ,6HGQD1      ,6HGQD2      ,6HIFLX      ,6HGDAMP      ,6HSLEN1      ,
      *      6HSLEN2      ,6HGAMA      ,6HCFOUR      ,6HCONE      ,6HCONFRI      ,6HCTHREE      ,6HCTWO      ,
      *      6HDELPWR      ,6HDTIRE      ,6HEA      ,6HETAMAX      ,6HETAMIN      ,6HETANOS      ,6HETART1      ,
      *      6HINDNWS      ,6HKANWS      ,6HKPNWS      ,6HPCTETA      ,6HPSIDES      ,6HRPSI      ,6HRYR      ,
      *      6HVPOWR      ,6HYRDES      ,6HFGPY      ,6HDELN      ,6HETADES      ,6HILIM      ,6HINOSE      ,
      *      6HISTER      ,6HIRUDD      ,6HINWRUD      ,6HAH      ,6HPH      /
      DATA (LOC(K4),K4=876,929)/
*DELETE      DIR3DA.11,DIR3DA.11
      *      3760      ,3880      ,3900      ,3920      ,3940      ,3960      ,3980      ,
      *      3985      ,3990      ,3991      ,3992      ,3993      ,3994      ,3995      ,
      *      3996      ,3997      ,3998      ,3999      ,4000      ,4001      ,4002      ,
      *      4003      ,4004      ,4005      ,4006      ,4007      ,4008      ,4009      ,
      *      4010      ,4011      ,4012      ,4013      ,4014      ,4015      ,4016      ,
      *      4017      ,4018      ,4019      ,4020      ,4025      /
*INSERT      AUTS.43
      *,DDM41(38),EA,ETAMAX,ETAMIN,ETANOS,ETART1,INDNWS,KANWS,KPNWS,
      *DDM42,PSIDES,RPSI,RYR,DDM43,YRDES,DDM44(2),ETADES,ILIM,INOSE,

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*ISTER,IRUDD,INWRUD,DDM45(40)
*DELETE      C82477.67,C82477.70
*DELETE      C82477.71,C82477.71
*DELETE      C82477.72,C82477.72
      IF(DELTA1.EQ.0.0)GO TO 130
*INSERT      C82477.77
130 IF(ISTER.EQ.1)GO TO 29
      IF(INWRUD.EQ.1)GO TO 82
      GO TO 8
82 IF(DELRDE.GT.0.0)GO TO 86
      ETADES = DELRDE*(ETAMAX/DELRD)
      GO TO 8
86 ETADES = DELRDE*(ETAMIN/DELRD)
*DELETE      C82477.78,C82477.78
*INSERT      C82477.79
29 IF(TR.LE.TST)GO TO 35
      ETADES=ETANOS+ETART1*(TR-TST)
      GO TO 31
35 ETADES=ETANOS
      GO TO 31
*DELETE      C82477.93,C82477.95
      GO TO 31
*INSERT      C82477.103
31 IF(ETADES .GT. ETAMAX) ETADES=ETAMAX
      IF(ETADES .LT. ETAMIN) ETADES=ETAMIN
*DELETE      AUTS.447,AUTS.448
34 IF(IAP.GT.2)GO TO 500
      DELRN=DELRNN
*DELETE      AUTS.454,AUTS.454
500 IF(IRUDD.EQ.1)GO TO 5
      GO TO 150
5 IF(INWRUD.EQ.1)GO TO 6
      IF(TR.LE.TST)GO TO 27
      DELRDE=DELRDI+DELRD*(TR-TST)
      GO TO 101
27 DELRDI=DELRD
      GO TO 101
150 DELRN=DELRNN
      PSIE=PSIPD
*INSERT      AUTS.457
      GO TO 101
6 IF(TR.LE.TST)GO TO 110
      DELRDE=DELRDI+DELRD*(TR-TST)
      GO TO 101
110 IF(ABS(DELRDE).GT.0.0)GO TO 120
      DELRDI=DELRD
      GO TO 101
120 DELRDI=DELRDE
*INSERT      AUTS.460

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106 CONTINUE
*DELETE      AUTS.582,AUTS.583
      IF(IRUDD.EQ.1)GO TO 49
      IF(DELRDE.LT.DELRD)DELRD1=-DELRD
49 DELPD1=DELA
*DELETE      AUTS.607,AUTS.607
61 IF(IAP.EQ.4)GO TO 62
      IF(ABS(DELRD1*DELTS).GE.ABS(DELRDE-DELRD))GO TO 62
*INSERT      FLARE1.38
      *,DM21(100)
*INSERT      AUTPR1.35
      *,DM41(100)
*INSERT      THAUTS.32
      *,DDM9(100)
*INSERT      ENGFL.29
      *,DM8(100)
*INSERT      CENGL.9
      *,DM5(100)
*IDENT      SDDMODS
*DELETE      C11778.2,C11778.2
34 DO 35 II=1,4036
*DELETE      C11778.32,C11778.32
      *,DDM21(43),INDNWS,DDM22(10),ETADES,DDM23(5),AH(5),PH(5),DDM24(30)
*INSERT      LGEAR1.61
      EQUIVALENCE (INDSTE(73),PSIPD)
      DATA RADDEG,DEGRAD/57.2957795,.01745329/
*INSERT      LGEAR1.242
      IF(INDNWS.EQ.1.AND.I.EQ.1)GO TO 301
*INSERT      LGEAR1.245
301 MA(I)=-FTRY(I)*TMP(1)*SIN((PSIPD+ETADES)*DEGRAD)+
1 FTRX(I)*TMP(1)*COS((PSIPD+ETADES)*DEGRAD)
GO TO 201
*DELETE      C11778.54,C11778.54
      *DDM34,FGPY,DELN,ETADES,DDM35(15),CONFRM,DELPPM,VPOWRM,CONEM,CTWOM,
      *CTHREM,CFOUPM,DDM36(23)
*INSERT      LGEA3C.142
      TMPETA = ETADES
*DELETE      C82477.21,C82477.21
      TMPETA = 0.0
*DELETE      C11778.59,C11778.59
200 IF(ABS(OMET(1,I)*TMP(1)*COS((PSIPD+TMPETA)*DEGRAD))
*DELETE      C11778.61,C11778.64
      VTX(I)=VTX(I)-TMP(2)+OMET(1,I)*TMP(1)*COS((PSIPD+TMPETA)*DEGRAD)
      VTY(I) = RDYG(I)+OMET(1,I)*TMP(1)*SIN((PSIPD+TMPETA)*DEGRAD)
*DELETE      C11778.67,C11778.69
      OMET(1,I)=-((RG11*RDYG(I)+RG13*RDZG(I))/(COS((PSIPD+TMPETA)*
1 DEGRAD)))/(RZERO(I)-DELTA(I))
      VTY(I) = RDYG(I)+OMET(1,I)*TMP(1)*SIN((PSIPD+TMPETA)*DEGRAD)
*DELETE      C11778.74,C11778.74

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IF(DELTA1.EQ.0.0)GO TO 48
*DELETE      C82477.32,C82477.32
  SIDMU = CONFRI*(ABS(DELN))**DELPWR*VAXLE(I)**VPOWR
*DELETE      LGEA3C.168,LGEA3C.168
  GO TO 48
*DELETE      C82477.52,C82477.52
  IF(I.GT.1)GO TO 48
  FGPY=0.0
*DELETE      LGEA3C.171,LGEA3C.171
  GO TO 48
41  ETAVEM=ATAN2(RDYG(I),TMP(1))
  DELNM=PSIPD-ETAVEM*RADDEG
  IF(DELTA(I).EQ.0.0)FGPYM=0.0
  IF(DELTA(I).EQ.0.0)GO TO 48
  IF(DELNM.EQ.0.0.OR.XG77F1.EQ.0.0)GO TO 1100
  SIDMUM =CONFIRM*(ABS(DELNM))**DELPRM*VAXLE(I)**VPOWRM
  GO TO 1800
1100 SIDMUM =1.0E-6
1800 IF((DELTA(I)/2.*RZERO(I)).GT.0.0875)GO TO 1300
  CRNPRM = CONEM*DELTA(I)-CTWOM*DELTA(I)**2.
  GO TO 1700
1300 CRNPRM = CTHREM-CFOURM*DELTA(I)
1700 YAWPM = ABS(CRNPRM*DELM/(SIDMUM*FTRZ(I)))
  IF(YAWPM.LE.1.5)GO TO 1500
  FGPYM = SIDMUM*FTRZ(I)*COS(DELM*DEGRAD)
  GO TO 1600
1500 FGPYM = ((YAWPM-(4.0/27.0)*YAWPM**3.0)*SIDMUM*FTRZ(I))
1    *COS(DELM*DEGRAD)
1600 IF(DELNM.GT.0.0)FGPYM=ABS(FGPYM)
  IF(DELNM.LT.0.0)FGPYM=-ABS(FGPYM)
  DFTRXM=FGPYM*SIN(ETAVEM)
  DFTRYM=FGPYM*COS(ETAVEM)
  FTRX(I)=FTRX(I)+DFTRXM
  FTRY(I)=FTRY(I)+DFTRYM
48  CONTINUE
*DELETE      C11778.77,C11778.77
  COMMON/DIRCOM/DATA(4036)
*DELETE      C11778.82,C11778.82
  DATA NCOUNT/936/
*DELETE      C11778.84,C11778.84
  DATA(NAME(K3),K3=876,936)/
*DELETE      C11778.90,C11778.90
  * 6HISTER,6HIRUDD,6HINWRUD,6HAH,6HPH,6HCONFIRM,6HDELPWM,
  * 6HVPOWRM,6HCONEM,6HCTWOM,6HCTHREM,6HCFOURM/
*DELETE      C11778.91,C11778.91
  DATA(LDC(K4),K4=876,936)/
*DELETE      C11778.97,C11778.97
  * 4017,4018,4019,4020,4025,4030,4031,
  * 4032,4033,4034,4035,4036,

```


*IDENT ACOBLK

*INSERT TOLA.27

*COMDECK ACOBLK

C***** ACTIVE VARIABLES *****

| COMMON | /ACTIVE/ | AMUH | ,ACON | ,BCON | ,CCON | ,DCON |
|-------------|-------------|-------------|------------|------------|------------|-----------|
| 1,APINT(5) | ,AP2T0(5) | ,AREA1(5) | ,AREA2(5) | ,AREA3(5) | | |
| 2,AREMO(5) | ,AREO3(5) | ,RLFORT(5) | ,RLMU(5) | ,BUFORT(5) | ,RUMU(5) | |
| 3,BETA | ,CDMOC(5) | ,CDMOE(5) | ,CDSV(5) | ,CFFOR(5) | ,CD3(5) | |
| 4,COEF | ,COEFO(5) | ,COEF3(5) | ,CDPA | | | |
| 5,CSV1(5) | ,CSV3(5) | ,DELT | ,DELTX(5) | ,DELTX1(5) | ,DLTX1D(5) | |
| 6,DF(5) | ,DP1(5) | ,DSV(5) | ,DSTOP | ,DIOTA | ,DMTANH(5) | |
| 7,ENUP(5) | ,EPSILO(5) | ,EPSROL(5) | ,EPSSLP | ,ETASV | ,FSTOPK | |
| 8,FFORT(5) | ,FOAHST(5) | ,FQNHST(5) | ,FORCHT(5) | ,FORSST(5) | ,FSTOP(5) | |
| 9,GAMAH(5) | ,GAMAN | ,GNR | ,HMM(5) | ,ICOSV(5) | ,IDEACT | |
| *,IFRI(5) | ,IIXSVH(5) | ,IIXSVL(5) | ,IOPCO(5) | ,IPASS(5) | ,ISET(5) | |
| 1,IXSVH(5) | ,IXSVL(5) | ,IFSTOP(5) | ,ISTROK(5) | ,KAPT(5) | | |
| 2,NAC(5) | ,NITER | ,OMRUN | ,PATM | | | |
| 3,PERCNT(5) | ,PGAHAC(5) | ,PGALAC(5) | ,PGA1I(5) | ,PGA1T(5) | | |
| 4,PGA2I(5) | ,PGA2T(5) | ,PGA3I(5) | ,PGA3T(5) | | | |
| 5,PR(5) | ,PS(5) | ,P1(5) | ,QC(5) | ,QO(5) | ,QPUMPS(5) | |
| 6,QSV(5) | ,QSVCU(5) | ,QSVN(5) | ,OSV1(5) | ,OSV3(5) | ,QS1(5) | |
| 7,QS3(5) | ,QTOLER | ,RCLSV(5) | ,REDSLP(5) | ,RHOM | ,SBFOT | |
| 8,SIPA | ,SA(5) | ,RESA(5) | ,TAUF | ,TC1 | ,TC2 | |
| COMMON | /ACTIVE/ | TC3 | ,TC4 | ,VCUM(5) | ,VELDEC | ,VOL1I(5) |
| 1,VOL1T(5) | ,VOL2I(5) | ,VOL2T(5) | ,VOL3I(5) | ,VOL3T(5) | ,WC | |
| 2,WC1 | ,WFORT(5) | ,WLFOR(5) | ,WLFORR | ,XBIAS(5) | ,XVALVE(5) | |
| 3,XKA(5) | ,XKF(5) | ,XKSV(5) | ,XLPSV1(5) | ,XLPSV3(5) | ,XSTHR | |
| 4,XMA(5) | ,XMA1(5) | ,XMA2(5) | ,XMA3(5) | ,XMA4(5) | ,XMA5(5) | |
| 5,XMA6(5) | ,XMA7(5) | ,XMA8(5) | ,XMA9(5) | ,XMA10(5) | ,XMA11(5) | |
| 6,XMU | ,XSCOM(5) | ,XSTOT(5) | ,XSV(5) | ,XSVDOT(5) | ,XSVDOD(5) | |
| 7,XSVDOD(5) | ,XSVDOMN(5) | ,XSVDOMX(5) | ,XSVMAX(5) | ,XSVMIN(5) | ,XDDMAX(5) | |
| 8,XDDMIN(5) | ,VOLACI(5) | ,VOLAH(5) | ,VOLANI(5) | ,VOLANT(5) | | |
| 9,WSV | ,WSV1 | ,WSV3 | ,ZETAC1 | ,ZETAC2 | ,ZSSC | |

C*****

*IDENT CSCMODS

*INSERT EXE.42

*CALL ACOBLK

*INSERT EXE.253

IF (IABS(INDLG).NE.3) GO TO 594

CALL ALGEAR1

GO TO 596

594 CONTINUE

*INSERT EXE.254

596 CONTINUE

*INSERT INUPD.5

NNUM = NUM+N

WRITE(6,600) NNUM

600 FORMAT(5X,32HNUMBER OF INTEGRATED VARIABLES =,I3)

*DELETE LGDET.4, LGDET.4

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COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
*DELETE      LGDET.39,LGDET.39
  53 FORMAT(58X,4H ES(,I1,20H) EXCEEDED IN LGDET /
*DELETE      LINES.6,LINES.6
  IF(LONG.LE.41)RETURN
*INSERT      OPT1.75
*CALL ACOBLK
*INSERT      OPT1.77
  EQUIVALENCE (DM155(133), INDLG)
*INSERT      OPT1.354
  IF (IABS(INDLG).NE.3) GO TO 2070
  CALL ALGEAR1
  GO TO 2075
2070 CONTINUE
*INSERT      OPT1.355
2075 CONTINUE
*DELETE      LGEAR1.34,LGEAR1.34
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
*INSERT      LGEAR1.121
  IF (IABS(INDLG).EQ. 3) NDEQ=10*NSTRUT
*INSERT      LGEAR1.122
  IF (IABS(INDLG).EQ.3) CALL ACTINIT
*DELETE      LGEAR1.228,LGEAR1.228
  53 FORMAT(58X,4H FS(,I1,20H) EXCEEDED IN LGEAR1/
*DELETE      LGEA3C.35,LGEA3C.35
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
*INSERT      LGEA3C.53
  EQUIVALENCE (DM14(155),INDLG)
*INSERT      LGEA3C.104
  IF (IABS(INDLG).EQ.3) GO TO 31
*INSERT      LGEA3C.196
  IF (IABS(INDLG).EQ.3) GO TO 46
*DELETE      SDFLGP.36,SDFLGP.36
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
*INSERT      SDFLGP.47
*CALL ACOBLK
*INSERT      C82477.58
  DIMENSION ACOVAR1(8), ACOVAR2(8), ACOVAR3(8), ACOVAR4(8),
  * ACOVAR5(8), ACOVAR6(8), ACOVAR7(8), ACOVAR8(8),ACOVAR9(8)
*INSERT      SDFLGP.78
  DATA(ACOVAR1(I),I=1,8 ) /5HVOL1T,5HVOL2T,5HVOL3T,5HPGA1T,5HPGA2T,
  * 5HPGA3T,5HGAMAH,5HCOEFO/
  DATA(ACOVAR2(I),I=1,8 ) /2HQD,4HVCUM,6HBLFORT,6HBUFORT,5HFFORT,
  * 6HFORCHT,6HFOAHST,6HFONHST/
  DATA(ACOVAR3(I), I=1,8) /5HCFFOR,5HFSTOP,6HFORSST,6HXVALVE,3HXSIV,
  * 5HDELTX,6HDELTX1,6HDLTX1D/
  DATA(ACOVAR4(I), I=1,8) /2HDF,3HDP1,2HPS,2HP1,3HQSIV,4HQSVN,4HOSV1,
  * 4HOSV3/
  DATA(ACOVAR5(I),I=1,8)/6HEPSILO,3HHMM,6HDMTANH,5HXSTOT,5HICOSV,

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* 4HENUP,5HWLFOR,5HIOPCO/
  DATA(ACOVAR6(I), I=1,8) /6HIIXSVH,6HIIXSVL,5HIXSVH,5HIXSVL,5HIPASS
* ,3HNAC,4HRESA,2HSA/
  DATA(ACOVAR7(I), I=1,8) /6HVELDEC,6HWLFORR,6HIDEACT,4HZSSC,4HCOPA,
* 4HSIPA,5HDSTOP,6HFSTOPK/
  DATA(ACOVAR8(I), I=1,8) /5HWFORT,6HVOLAHT,6HVOLANT,5HQSVCU,3HXMA,
* 4HXMA5,4HXMA8,5HXMA11/
  DATA(ACOVAR9(I), I=1,8) /6HXSVDOD,5HXSVDOD,6HXSVDOT,2HPR,4HIFRI,
* 6HIFSTOP,6HISTROK,4HISET/
*INSERT      C82477.66
  IF(IABS(INDLG).NE.3) GO TO 50
  CALL STFL(2,8,ACOVAR1)
  DO 33 I=1,NSTRUT
    CALL STOVAR(8,VOL1T(I),VOL2T(I),VOL3T(I),PGA1T(I),PGA2T(I),
* PGA3T(I),GAMAH(I),CDEFO(I))
33 CONTINUE
  CALL STFL(2,8,ACOVAR2)
  DO 34 I=1,NSTRUT
    CALL STOVAR(8,QO(I),VCUM(I),BLFORT(I),BUFORT(I),FFORT(I),
* FORCHT(I),FOAHST(I),FONHST(I))
34 CONTINUE
  CALL STFL(2,8,ACOVAR3)
  DO 35 I=1,NSTRUT
    CALL STOVAR(8,CFFOR(I),FSTOP(I),FORSST(I),XVALVE(I),XSV(I),
* DELTX(I),DELTIX1(I),DLTX1D(I))
35 CONTINUE
  CALL STFL(2,8,ACOVAR4)
  DO 36 I=1,NSTRUT
    CALL STOVAR(8,DF(I),DP1(I),PS(I),P1(I),QSV(I),QSVN(I),QSV1(I),
* QSV3(I))
36 CONTINUE
  CALL STFL(2,8,ACOVAR8)
  DO 37 I=1,NSTRUT
    CALL STOVAR(8,WFORT(I),VOLAHT(I),VOLANT(I),QSVCU(I),XMA(I),
* XMA5(I),XMA8(I),XMA11(I))
37 CONTINUE
  CALL STFL(2,8,ACOVAR9)
  DO 38 I=1,NSTRUT
    CALL STOVAR(8,XSVDDD(I),XSVDD(I),XSVDOT(I),PR(I),FLOAT(IFRI(I)),
* FLOAT(IFSTOP(I)),FLOAT(ISTROK(I)),FLOAT(ISET(I)))
38 CONTINUE
  CALL STFL(2,8,ACOVAR5)
  DO 39 I=1,NSTRUT
    CALL STOVAR(8,EPSILO(I),HMM(I),DMTANH(I),XSTOT(I),FLOAT(ICOSV(I)),
* ENUP(I),WLFOR(I),FLOAT(IOPCO(I)))
39 CONTINUE
  CALL STFL(2,8,ACOVAR6)
  DO 40 I=1,NSTRUT
    CALL STOVAR(8,FLOAT(IIXSVM(I)),FLOAT(IIXSVM(I)),FLOAT(IXSM(I)),

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* FLOAT(IXSVL(I)),FLOAT(IPASS(I)),
* FLOAT(NAC(I)),RESA(I),SA(I))
40 CONTINUE
  CALL STFL(2,8,ACDVAR7)
  CALL STVAR(8,VELDEC,WLFORR,FLOAT(IDEACT),ZSSC,COPA,SIPA,DSTOP,
* FSTOPK)
50 CONTINUE
*INSERT      SDFLGP.183
  IF(IABS(INDLG) .NE. 3) GO TO 6
  I1 = I+3*NSTRUT
  I2 = I+4*NSTRUT
  I3 = I+5*NSTRUT
  CALL UPDAT(1,LA(I1),PGA1T(I),DU,DU,DU,DU)
  CALL UPDAT(1,LA(I2),VCUM(I),DU,DU,DU,DU)
  CALL UPDAT(1,LA(I3),OSVCU(I),DU,DU,DU,DU)
  I1 = 3*I+6*NSTRUT-2
  I2 = I+9*NSTRUT
  CALL UPDAT(3,LA(I1),XSVD(I),XSVDOT(I),XSV(I),DU,DU)
  CALL UPDAT(1,LA(I2),DELTX1(I),DU,DU,DU,DU)
*INSERT      READ.35
  SLTSYM = 0.0
  IBC = 0
  INXQ = 0
  JBC = 0
*INSERT      AUTS.87
  DELPI = 0.0
  TR = 0.0
*IDENT      ACTINIT
*INSERT      LGEA3C.227
*DECK ACTINIT
  SUBROUTINE ACTINIT
C
C***** FATOLA VARIABLES *****
COMMON/DIPCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,
CDCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FXB7P,
CDUM4(3),FYB7P(4),FZB7P,DM5(17),GXB7F,DM6(8),GZB7F,
CDM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,
CINDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),
CQI77R(2),QI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),
CXG77F(2),XG77F1(12),YG77F(2),YG77F1(12),
CZG77F(2),ZG77F1(12),DUM13(52),
CNSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEF,GR,
CNTIRES(5),RZERO(5),W(5),DELTAM(5),MOMENT(5),
CRF(5),VZ ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),
CA2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),
CCC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),RLT,NDELTA,
CES(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)
COMMON/DIRCOM/
CSD11(2),SD12(2),SD13(2),SD14(2),SD15(2),

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CS1(2),SS2(2),S3(2),S4(2),S5(2),
CS2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),
CS2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),
CS21(2),S22(2),S23(2),S24(2),S25(2),
COMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),
COMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),
CAI(5),BI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,
CDELTA1,DDELTA2,DDELTA3,DDELTA4,DDELTA5,ISTAGE,
CPRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,
CDM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLX,
*NMODE,DM18(40),SXMOD(100),SYM0D(100),SZMOD(100),DM19(1686),
*GQD2(20),DDM20(20)
*,DDM21(20),SLEN1(5),SLEN2(5),DUM15(13),INDNWS,DDM22(10),ETADES,
*DDM23(5),AH(5),PH(5),DDM24(30)
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT

C

*CALL ACOBLK

DIMENSION CD3(5),SD1(2,5)

EQUIVALENCE(SD11(1),SD1(1,1)), (DM5(16),GREFF)

C

C

C

DATA STATEMENTS TO SET VALUES OF INPUT VARIABLES

DATA AMUH/0.00018/, BETA/14400000./
DATA BLMU,BUMU/10*0.15/
DATA AREA1,AREA2,AREA3/5*0.21139,5*0.32167,5*0.00698/
DATA AREMO,AREO3,APINT/5*0.0066,5*0.19635,5*0.0/
DATA CDMOC,CDMOE,CD3/5*0.9,5*0.14197,5*0.6/, CFFOR/5*50.0/
DATA CDSV,DELT,DIOTA/5*0.62,0.0001,0.0/
DATA EPSILO,EPSSLP,EPSROL,ETASV/5*200.0,0.0,5*2000.0,0.436/
DATA GAMAN,GAMAH,GNR/1.06,5*52.36,0.0/
DATA ICOSV,IFSTOP/10*0/
DATA IDEACT,II,KAPT,OMPRUN,PERCNT/7*0,-0.07575,5*1.66115/
DATA PATM,PGAHAC,PGALAC/2116.8,5*432000.0,5*0.0/
DATA PGA1I,PGA2I,PGA3I/15*40320.0/
DATA TAUF,TC1,TC2,TC3,TC4/0.1,0.281,0.141,0.001,0.0001/
DATA VOL1I,VOL2I,VOL3I/5*0.36379,5*0.47164,5*0.0/
DATA RHOH,WLFOR,WLFORR/1.626,5*500.0,0.0/
DATA DSV,RCLSV,WSV1,WSV3/5*1.1875,5*0.0000475,2*3.480/
DATA WC,WCl,WSV,XKSV,XSTHP/1263.0,251.3,655.5,5*0.00250,0.020833/
DATA XKA,XKF,XLPSV1,XLPSV3/5*0.04,5*60.0,5*0.0,5*0.0/
DATA XBIAS,XSCOM,XSVDMM,XSVDMMX/5*-0.00014,5*1.206,5*-30.1,5*30.1/
DATA XDDMAX,XDDMIN/5*1.0E20,5*-1.0E20/
DATA XSVMAX,XSVMIN,ZETAC1,ZETAC2/5*0.100,5*-0.100,5.1,0.1/
DATA VOLACI,VOLANI,QPUMPS/5*1.33333,5*0.26667,5*0.02005/

IF (TIME .GT. DELT) RETURN

C

ACON=1./(WSV*WSV*WC)

BCON=1./(WSV*WSV)+2*ETASV/(WSV*WC)

CCON=2.*ETASV/WSV+1./WC
DCON=1.
XMU=AMUH/.0000209
QTOLER=.0001
SBFOT=0.0
VELDEC=0.
OMRUN=OMRUN*0.01745329
II = 0

C

DO 100 I=1,NSTRUT
REDSLP(I) = 100000.0
ISTROK(I)=0
NAC(I)=0
VOLANT(I) = VOLANI(I)
VOL1T(I)=VOL1I(I)
VOL2T(I)=VOL2I(I)
VOL3T(I)=VOL3I(I)
OSVCU(I)=0.0
DF(I)=0.
DELTX(I)=0.
DELTX1(I)=DELTX(I)*XKF(I)
DLTX1D(I)=0.
XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
XMA5(I)=XMA(I)
XMA8(I)=XMA5(I)
XMA11(I)=XMA8(I)
XMA1(I)=0.
XMA2(I)=0.
XMA3(I)=0.
XMA4(I)=0.
XMA6(I)=0.
XMA7(I)=0.
XMA9(I)=0.
XMA10(I)=0.
XSVDD(I)=0.
XSVDDD(I)=0.
XSV(I)=XKSV(I)*XMA11(I)+XBIAS(I)
XSVDDT(I)=0.0
DP1(I)=0.
CDEF3(I)=CD3(I)*SQRT(2.*GREFF/GAMAH(I))

C

C

C

NOTE: SUBROUTINE 'PHLOZ2' COMPUTES INITIAL PRESSURES AND FLOWS IN
UNITS OF INCHES.
COEF=CDSV(I)*SQRT(2.*GREFF/GAMAH(I))*144.
CSV1(I)=COEF*WSV1
CSV3(I)=COEF*WSV3
PS(I)=PGAHAC(I)/144.
PR(I)=PGALAC(I)/144.
QC(I)=0.

```
CALL PHLOZ2(PS(I),PR(I),XSV(I),QC(I),XLPSV1(I),XLPSV3(I),RCLSV,DSV
& ,CSV1(I),CSV3(I),XMU,QTOLR,NITER,P1(I),QS1(I),QS3(I))
PGA1I(I)=P1(I)*144.
PGA2I(I)=PGA1I(I)
PGA3I(I)=PGA2I(I)
QSV1(I)=QS1(I)/1728.
QSV3(I)=QS3(I)/1728.
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C

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PGA1T(I)=PGA1I(I)
PGA2T(I)=PGA2I(I)
PGA3T(I)=PGA3I(I)
QQ(I)=0.
QSV(I)=QSV1(I)-QSV3(I)
VCUM(I)=0.
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C

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AP2TO(I)=PGA2I(I)+PATM
FFORT(I) = 0.0
FORSST(I)=0.0
WFORT(I)=0.0
FONHST(I)=0.0
FORCHT(I)=0.0
XVALVE(I)=XSV(I)
```

100 CONTINUE

C

```
CALL SETUP
RETURN
END
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*IDENT ALGEAR
*INSERT ACTINIT.145
*DECK ALGEAR
SUBROUTINE ALGEAR1
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C

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***** FATOLA VARIABLES *****
COMMON/DIRCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,
CDCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FXB7P,
CDUM4(3),FYB7P(4),FZB7P,DM5(17),GX87F,DM6(8),GZB7F,
CDM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,
CINDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),
COI77R(2),QI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),
CXG77F(2),XG77F1(12),YG77F(2),YG77F1(12),
CZG77F(2),ZG77F1(2),DUM13(52),
CNSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEG,RGR,
CNTIRES(5),RZERO(5),W(5),DELTAM(5),MOMENT(5),
CRF(5),VZ ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),
CA2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),
CCC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),RLT,NDELTA,
CES(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)
COMMON/DIRCOM/
CSD11(2),SD12(2),SD13(2),SD14(2),SD15(2),
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CS1(2),SS2(2),S3(2),S4(2),S5(2),
CS2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),
CS2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),
CS21(2),S22(2),S23(2),S24(2),S25(2),
COMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),
COMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),
CAI(5),BI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,
CDELT1,DDELT2,DDELT3,DDELT4,DDELT5,ISTAGE,
CPRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,
CDM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLX,
*NMODE,DM18(40),SXMOD(100),SYM0D(100),SZMOD(100),DM19(1686),
*G0D2(20),DDM20(20)
*,DDM21(20),SLEN1(5),SLEN2(5),DUM15(13),INDNWS,DDM22(10),ETADES,
*DDM23(5),AH(5),PH(5),DDM24(30)
REAL MASS,MOMENT, MASS2,MUS,NTIRES,MB
DIMENSION DLGAUT(14),DLGDE(47),DLG(7),DLGE(299)
COMMON/LGAUTS/ARG11,ARG13,ARG31,ARG33,AMA(5),VAXLE(5)
COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
COMMON/LG/FXM,FYM,FZM,LM,MM,NM,EPSLO2
COMMON/LGE/A11(5),A13(5),A31(5),A33(5),RRCGX(5),
CRL(3,3),RI(3,3,5),RAX(5),RAY(5),RAZ(5),THP(3),ZZERO(5),
CXR(5),YR(5),EPSLON(5),PA(5),FDELTA(5),
CFTRZ(5),RDX(5),RDY(5),RDZ(5),PDXG(5),RDYG(5),RDZG(5),
CVTX(5),VTY(5),VTZ(5),GZ(5), VGPT(5),FTRX(5),FTRY(5),
CDX(5),DY(5),DZ(5),FT(5),FDX(5),FDY(5),FF(5),AA(5),C2(5),
CSR(5),SF(5),PSKD(5),MUV(5),MTRX(5),MTRY(5),
CMTRZ(5),MA(5),RG11,RG13,RG31,RG33,IPRT,
CMTX,MTY,MTZ,SFTRX,SFTRY,SFTRZ,FTRA,
CFTRB,FTRC,SMTRX,SMTRY,SMTRZ
COMMON/FLXOP/GFORC2(100),GFORC3(100),GFORC4(100),BH1(308)
EQUIVALENCE (DLGAUT(1),ARG11),(DLGDE(1),LA(1)),
*(DLG(1),FXM),(DLGE(1),A11(1))
COMMON/TABSRC/DUMM1(103),LOC(7)
COMMON/HTCOM/HT,HT1,HT2

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C

*CALL AC0BLK

C

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REAL MUV(5),MTRX,MTRY,MTRZ,MA,
CMTX,MTY,MTZ,LM,MM,NM
DIMENSION DELTA(5),DDELTA(5),P(5),
C SD2(2,5),SD1(2,5),S(2,5),S2D2(2,5),S2D1(2,5),S2(2,5),
COMETD1(2,5),OMET(2,5)
EQUIVALENCE (P(1),PRES(1))
EQUIVALENCE (DELTA(1),DELTA1),(DDELTA(1),DDELT1)
EQUIVALENCE
C(SD21(1),SD2(1,1)),(SD11(1),SD1(1,1)),(S1(1),S(1,1)),
C(S2D21(1),S2D2(1,1)),(S2D11(1),S2D1(1,1)),(S21(1),S2(1,1)),
C(OMTD11(1),OMETD1(1,1)),(OMT1(1),OMET(1,1))

```

C


```
C*****ACTIVE CODE*****
EQUIVALENCE (DM5(16),GREFF)
DATA IFRI/5*0/,FSTOPK/0.0/,FSTOP/5*0.0/,DSTOP/0.004/
DATA SA,RESA,HMM,IXSVL,IXSVH,IIXSVL,IIXSVH,IPASS
1 /15*0.,25*0/
DATA ENUP/5*0.0/
DATA ISET,IDPCD,OSVN/10*1,5*0.0/
Q1(T1,T2)=SIGN(1.,(T1-T2))*SORT(ABS(T1-T2))
```

C*****

```
C
DATA RADDEG,DEGRAD/57.2957795,.01745329/
```

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C
C MAIN COMPUTATIONAL AREA
C RL MATRIX ELEMENTS
RL(1,1)=DCL1*RG11+DCL3*RG13
RL(1,2)=DCL2
RL(1,3)=DCL1*RG31+DCL3*RG33
RL(2,1)=DCM1*RG11+DCM3*RG13
RL(2,2)=DCM2
RL(2,3)=DCM1*RG31+DCM3*RG33
RL(3,1)=DCN1*RG11+DCN3*RG13
RL(3,2)=DCN2
RL(3,3)=DCN1*RG31+DCN3*RG33
CALL LGEA3C
```

C

C***** ACTIVE CODE *****

```
C START ACTIVE GEAR CALCULATIONS
COPA = COS((THTPD+DIOTA)*DEGRAD)
SIPA = SIN((THTPD+DIOTA)*DEGRAD)
IF(TIME .GT. DELT)GO TO 18
15 WRITE(6,1013)
1013 FORMAT (1H020H ACTIVE CONTROL GEAR)
18 IF(IDEACT.EQ.1) GO TO 56
19 IF(IDEACT.EQ.2) GO TO 80
IF(HMM(I) .EQ. 0.) GO TO 90
IF(OMRUN .GT. 0.0)GO TO 25
IF(ZG77F1(1) .GE. VELDEC) GO TO 90
GO TO 40
25 IF(ZG77F1(1) .GE. VELDEC+XG77F1(1)*TAN(OMRUN))GO TO 90
40 WRITE(6,1014)TIME
1014 FORMAT (1H036H REDUCE CONTROL LIMIT FORCE AT TIME=,E16.8)
IDEACT=1
56 WLFOR(I)=WLFOR(I)-REDSLP(I)*DELT
EPSILO(I)=EPSILO(I)+EPSSLP*DELT
IF(WLFOR(I) .GT. WLFORR) GO TO 90
60 WRITE(6,1015)TIME
1015 FORMAT (1H027H CONTROL AT WLFORR AT TIME=,E16.8)
IDEACT=2
80 WLFOR(I)=WLFORR
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      EPSILO(I)=EPSROL(I)
90  CONTINUE
C*****
      DO 65 I=1,NSTRUT
      WFORT(I) = (SQRT(FXB7P*FXB7P+FYB7P*FYB7P+FZB7P*FZB7P))/(NSTRUT-1)
      E      +(FORSST(I)*COPA)
C***** ACTIVE CODE *****
      IF(KAPT(I).NE.0)GO TO 210
      APINT(I)=0.0
210  CONTINUE
      IF(PGA1T(I) .LE. -1600.0) PGA1T(I) = -1600.0
      VOL1T(I)=VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
      VOL3T(I)=VOL3I(I)+AREA3(I)*S(1,I)
      VOL2T(I)=VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)+(VOL3T(I)
X -VOL3I(I))-VCUM(I)
      PGA2T(I)=AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMAN)-PATM
      IF(SD1(1,I) .EQ. 0.0)GO TO 104
      PGA3T(I)=((COEF3(I)*AREO3(I))*2*PGA2T(I)-SD1(1,I)/ABS(SD1(1,I))
X *(SD1(1,I)*AREA3(I))*2)
      E/((COEF3(I)*AREO3(I))*2)
      GO TO 105
104  PGA3T(I)=PGA2T(I)
105  IF(PGA1T(I) .GE. PGA2T(I))GO TO 106
      GO TO 107
106  GAMAH(I)=RHOH*GREFF*(1.0+(PGA1T(I)*3.04E-08)-
      * (PGA1T(I)**2*2.72E-15))
      GO TO 108
107  GAMAH(I)=RHOH*GREFF*(1.0+(PGA2T(I)*3.04E-08)-
      * (PGA2T(I)**2*2.72E-15))
108  IF(PGA1T(I) .GE. PGA2T(I))COEFO(I)=
      * CDMOC(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
      IF(PGA2T(I) .GT. PGA1T(I))COEFO(I)=
      * CDMOE(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
      IF(SD1(1,I).LE.0.0) GO TO 109
      QD(I)=COEFO(I)*(AREMO(I)-APINT(I))*Q1(PGA1T(I),PGA2T(I))
109  IF(PGA2T(I) .LE. -1600.0)PGA2T(I)=-1600.0
      IF(PGA3T(I) .LE. -1600.0)PGA3T(I)=-1600.0
100  IF(DELTA(I) .LE. 0.0 .AND. TIME .GT. DELT)GO TO 101
      GO TO 110
-101  FFORT(I)=0.0
      GO TO 140
110  CONTINUE
C  COMPUTE STRUT AXIAL BINDING FRICTION FORCE
      BLFORT(I)=FONHST(I)*((SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))+1.0)
      BUFORT(I)=FONHST(I)*(SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))
      FFORT(I)=BUMU(I)*ABS(BUFORT(I))+BLMU(I)*ABS(BLFORT(I))
140  CONTINUE
C  COMPUTE SHOCK STRUT CHARGING FORCE
      IF(S(1,I) .GT. 0.0)GO TO 142

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141 FORCHT(I)=PGA1T(I)*AREA1(I)+PGA2T(I)*((AREA2(I)-AREA1(I))-PGA3T(I)
X *AREA3(I))+FFORT(I)+CFFOR(I)
C   COMPUTE NORMAL AND AXIAL HUB TO SHOCK STRUT FORCES AT HUB
142 FONHST(I)=SQRT(FDX(I)**2+FDY(I)**2)-MASS(I)*GREFF*SIPA+SBFOT
IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 150
GO TO 801
150 CONTINUE
FORSST(I)=FT(I)
SD1(1,I)=0.0
IF(I.EQ.1) GO TO 450
ISTROK(I)=1
C***** 289 CHANGED TO 295 FOR DEBUGGING PURPOSES:
GO TO 295
C   COMPRESSION VELOCITY OF SHOCK STRUT IS POSITIVE
801 IF(SD1(1,I) .LE. 0.8 .AND. IFRI(I) .EQ. 0)GO TO 2
GO TO 3
2   DMTANH(I)=1.0
GO TO 284
3   DMTANH(I)=ABS(TANH(2.0*SD1(1,I)))
IFRI(I)=1
284 IF(S(1,I) .LE. DSTOP .AND. SD1(1,I) .LT. 0.0)GO TO 900
GO TO 902
900 IF(S(1,I) .LE. 0.0001)GO TO 903
GO TO 904
903 SD2(1,I)=0.0
SD1(1,I)=0.0
S(1,I)=0.0
DPI(I)=0.0
FFORT(I)=0.0
VCUM(I)=0.0
QD(I)=0.0
II=0
C   CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C   1          ELT,ERRVAL,DERSUB,CHSUB,ITEXT)
GO TO 902
904 CONTINUE
IF(IFSTOP(I) .NE. 0)GO TO 906
905 DSTOP=S(1,I)
FSTOPK=2.0*MASS(I)*SD1(1,I)**2/DSTOP**2
906 IF(S(1,I) .LE. DSTOP/2.0)GO TO 908
907 FSTOP(I)=-FSTOPK*(DSTOP-S(1,I))
GO TO 909
908 FSTOP(I)=-FSTOPK*S(1,I)
909 IFSTOP(I)=1
IFRI(I)=0
GO TO 901
902 FSTOP(I)=0.0
901 IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 500
IF(SD1(1,I) .LT. 0.0)GO TO 470

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      GO TO 471
470  FFORT(I)=-FFORT(I)
      CFFOR(I)=-CFFOR(I)
471  FORSST(I)=-((PGA1T(I)-PGA2T(I))*(AREA1(I)-APINT(I))
      & +PGA2T(I)*AREA2(I)
      X -PGA3T(I)*AREA3(I)+(FFORT(I)
      1 +CFFOR(I))*DMTANH(I)+FSTOP(I))
500  IF(INDFLX.GE. 1)GO TO 295
      IF(I.EQ.1) GO TO 450
      ISTROK(I)=1
C***** BRANCH FOR DEBUGGING PURPOSES:
      GO TO 295
289  IF(S(1,I) .LE. 0.0)290,295
290  IF(IOPCO(I) .EQ. 1)GO TO 295
      IF((PGA1I(I)-1000.0).LT.PGA1T(I).AND.
      & PGA1T(I).LT.(PGA1I(I)+1000.0))299,298
299  IF(XVALVE(I) .NE. 0.0)GO TO 311
      IF(IPASS(I) .EQ. 1)GO TO 296
      XVALVE(I)=XKSV(I)*XMA11(I)+XBIAS(I)
      IPASS(I)=1
      GO TO 294
298  IF(ICOSV(I) .EQ. 1)GO TO 291
      IOPCO(I)=0
      IF(XSV(I) .LT. 0.002 .AND. XSV(I) .GT. -0.002)291,295
291  IF(SD2(1,I) .LE. 0.0 .AND. ICOSV(I) .EQ. 1)GO TO 311
      IF(IOPCO(I) .EQ. 1)GO TO 295
      IF(PGA1T(I) .GT. PGA1I(I))292,293
292  IF(IXSVL(I) .EQ. 1)GO TO 294
      XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .LE. -0.1)300,294
300  XVALVE(I)=-0.1
      IXSVL(I)=1
      GO TO 294
293  IF(IXSVH(I) .EQ. 1)GO TO 294
      XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .GE. 0.1)302,294
302  XVALVE(I)=0.1
      IXSVH(I)=1
294  CONTINUE
      II=0
C  CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C  1  ELT,ERRVAL,DESUB,CHSUB,ITEXT)
      DLTX1D(I)=0.0
      ICOSV(I)=1
296  IF(WFORT(I) .GT. 0.0 .AND. S(1,I) .LE. 0.0)GO TO 410
311  IF(NAC(I) .EQ. 1)GO TO 307
      IF(IIIXSVH(I) .EQ. 1)GO TO 305
      XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .LE. 0.0)305,400

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305  XVALVE(I)=0.0
      IIXSVH(I)=1
      GO TO 400
307  IF(IIXSVL(I) .EQ. 1)GO TO 308
      XVALVE(I)=XVALVE(I)+XSVDMX(I)*DELT*PERCNT(I)
      IF(XVALVE(I) .GE. 0.0)308,400
308  XVALVE(I)=0.0
      IIXSVL(I)=1
400  II=0
C    CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
C    1      ELT,ERRVAL,DESUB,CHSUB,ITEXT)
410  IF(XVALVE(I) .NE. 0.0)GO TO 295
      ICOSV(I)=0
      DELTX1(I)=DELT*(I)*XKF(I)
      XMA(I)=(DF(I)+DELT*(I))*XKA(I)
      XMA1(I)=XMA(I)
      XSV(I)=XKSV(I)*XMA1(I)+XBIAS(I)
      CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
      IPASS(I)=0
      IXSVL(I)=0
      IXSVH(I)=0
      IIXSVL(I)=0
      IIXSVH(I)=0
      IOPCO(I)=1
      CALL PHLOZ2(PS(I),PR(I),XSV(I),QC(I),XLPSV1(I),XLPSV3(I),RCLSV(I),
      & DSV(I),CSV1(I),CSV3(I),XMU,QTOLER,NITER,P1(I),QS1(I),QS3(I))
C
295  IF(ISTROK(I) .EQ. 1 .AND. S(1,I) .GT. 0.0)IOPCO(I)=0
C
      ENUP(I)=.5*AMASS*ZG77F1(1)**2
      ENUP(I)=ENUP(I)/(NSTRUT-1)
      IF(HMM(I) .EQ. 1.0)GO TO 130
      SA(I)=0.
      IF(WFORT(I).LT.0.) XSTOT(I)=ENUP(I)/((-WFORT(I))*COPA)
      IF(WFORT(I).GE.0.) XSTOT(I)=1.E20
C    ZSSC IS A PERCENTAGE OF SB(I) FOR ACTIVATING CONTROL-COMOC(I) IS US
      ZSSC=0.6*COMOC(I)*SB(I)
      IF(XSTOT(I) .LE. (ZSSC-S(1,I)) .OR. RESA(I) .EQ. 1.0)SA(I)=1.0
      RESA(I)=SA(I)
      IF(SA(I).EQ.0. .OR. HMM(I).EQ.1.) GO TO 130
      WLFOR(I)=-WFORT(I)
      VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/(AMASS*REDSLP(I))
      WRITE(6,121)TIME,WLFOR(I),VELDEC
121  FORMAT(50H ACTIVE CONTROL INITIATED...TIME, WLFOR, VELDEC = ,
      1      3E13.5)
      HMM(I)=1.
130  IF(S(1,I) .GT. 0.0)ISET(I)=0
      IF(HMM(I).EQ.0.) GO TO 451
      IF(-WFORT(I).GT.(WLFOR(I)+EPSILO(I))) DF(I)=(WLFOR(I)+

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8 EPSILO(I))-(-WFORT(I))
  IF(-WFORT(I).LT.(WLFOR(I)-EPSILO(I))) DF(I)=(WLFOR(I)-
8 EPSILO(I))-(-WFORT(I))
  IF(-WFORT(I).LE.(WLFOR(I)+EPSILO(I)).AND.
8 -WFORT(I).GE.(WLFOR(I)-EPSILO(I)))
: 457,456
457 IF(S(1,I) .LE. 0.0)GO TO 456
453 IF(WFORT(I) .GT. 0.0 .AND. QSVCU(I) .LT. 0.0)454,455
454 DF(I)=WLFOR(I)-(-WFORT(I))
  GO TO 456
455 DF(I)=0.0
456 DELTX(I)=S(1,I)-XSCOM(I)
  IF(S(1,I) .LE. 0.0 .AND. ISET(I) .EQ. 0)GO TO 451
  GO TO 452
451 DF(I)=0.
  DELTX(I)=0.
452 XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
  IF(GNR.EQ.1. .AND. XMA(I).GT.0.) XMA(I)=
: XMA(I)*SQRT((PGA1T(I)-PGALAC(I))
X /(PGAHAC(I)-PGA1T(I)))

C
C NOTE: SUBROUTINE 'FLOZE2' COMPUTES THE FLOWS FROM THE PRESSURES
C IN UNITS OF INCHES.
C P1(I)=PGA1T(I)/144.
C COMPUTATION OF HIGH PRESSURE ACCUMULATOR NITROGEN VOLUME
C AND ACCUMULATOR PPESSURE
VOLANT(I)=VOLANT(I)+OSVN(I)*DELT-QPUMPS(I)*DELT
PS(I)=(((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMAN)-PATM)/144.0
IF(PS(I) .GE. 3000.0)464,465
464 PS(I)=3000.0
VOLANT(I)=VOLANI(I)
465 VOLAHT(I)=VOLACI(I)-VOLANT(I)
  IF(VOLAHT(I) .LE. 0.0)466,467
466 WRITE(6,1050)TIME
1050 FORMAT(1H0//45H ACCUMULATOR OIL VOLUME INSUFFICIENT AT TIME=,E16.8
1 //)
  CALL LGEAR6
  STOP 500
467 CONTINUE
  CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),OSV(I),
8 XSV(I),OS1(I),QS3(I),CSV1(I),CSV3(I),XMU)
  OSV1(I)=OS1(I)/1728.
  OSV3(I)=QS3(I)/1728.

C
450 CONTINUE
  QSV1(1) = 0.0
  QSV3(1) = 0.0
  QSV(I)=OSV1(I)-QSV3(I)
  IF(QSV(I) .LT. 0.0)NAC(I)=1

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      IF(QSV(I) .GT. 0.0)NAC(I)=2
      IF(NAC(I) .EQ. 2)461,462
461  QSVN(I)=QSV(I)
      GO TO 463
462  QSVN(I)=0.0
463  IF(SD1(1,I).LT. 0.0 .AND. PGA1T(I).LE. -1600.0)PGA1T(I)=-1600.0
274  CONTINUE
      I1 = I+3*NSTRUT
      I2 = I+4*NSTRUT
      I3 = I+5*NSTRUT
      CALL INTEG(LA(I1),DP1(I))
      CALL INTEG(LA(I2),QO(I))
      CALL INTEG(LA(I3),QSV(I))
C*****
      IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(FORSST(I)))FORSST(I)=-FT(I)
      AA(I) = (FT(I)+FORSST(I))/MASS(I)
      SD2(1,I) = SR(I)+AA(I)-GZ(I)
      HT1=HT
      IF(SD1(1,I))76,77,78
76   TTIME=S(1,I)/ABS(SD1(1,I))
79   IF(TTIME.GE.HT)GO TO 77
      HT1=TTIME
      GO TO 77
78   TTIME=(SB(I)-S(1,I))/SD1(1,I)
      GO TO 79
77   CONTINUE
      IF(S(1,I).GT.(-ES(I)))GO TO 50
      WRITE(6,49)I,I,S(1,I)
49  FORMAT(58X,4H-ES(,I1,20H) EXCEEDED IN ALGEAR/
      C58X,2HS(,I1,4H) = E15.7)
      S(1,I)=-0.5*ES(I)
      SD1(1,I)=-1.0E-10
      SD2(1,I)=-1.0E-10
50   IF(S(1,I).LE.ES(I))GO TO 51
      IF(S(1,I).LE.(SB(I)-ES(I)))GO TO 55
      IF(S(1,I).LE.(SB(I)+ES(I)))GO TO 52
      WRITE(6,53)I,I,S(1,I)
53  FORMAT(58X,4H ES(,I1,20H) EXCEEDED IN ALGEAR/
      C58X,2HS(,I1,4H) = E15.7)
      S(1,I)=0.5*ES(I)
52   IF(SD1(1,I).GT.0.)SD1(1,I)=0.
      IF(SD2(1,I).LT.0.)GO TO 55
      SD2(1,I)=0.
      GO TO 55
51   IF(SD2(1,I).LT.0.)GO TO 30
      IF(SD1(1,I).LT.0.)SD1(1,I)=0.
      GO TO 55
30   SD2(1,I)=0.
      SD1(1,I)=0.

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55  CONTINUE
    I2=2*I+NSTRUT-1
    I1=I2+1
    CALL INTEG(LA(I2),SD2(1,I))
    CALL INTEG(LA(I1),SD1(1,I))
C   RE-CHECK SHOCK STRUT FORCE FOR RE-CONDITIONED FULLY EXTENDED STATE
    IF(SD1(1,I).EQ.0.0.AND.FT(I).LE.ABS(FORSST(I)))FORSST(I)=-FT(I)
    TMP(1)=RZERO(I)-DELTA(I)
    IF(CASK(I).GT.1.E-10)GO TO 200
    IF(INDNWS.EQ.1.AND.I.EQ.1)GO TO 301
    MA(I)=-FTRY(I)*TMP(1)*RI(2,1,I)+FTRX(I)*TMP(1)
C*RI(2,2,I)
    GO TO 201
301  MA(I)=-FTRY(I)*TMP(1)*SIN((PSIPD+ETADES)*DEGRAD)+
    1  FTRX(I)*TMP(1)*COS((PSIPD+ETADES)*DEGRAD)
    GO TO 201
200  MA(I)=TMP(1)*SQRT(FTRY(I)*FTRY(I)+FTRX(I)*FTRX(I))
    MA(I)=SIGN(MA(I),-VAXLE(I)-OMET(1,I)*TMP(1))
201  AMA(I)=MA(I)
    IF(IB(I).NE.(-1))GO TO 48
    OMETD1(1,I)=0.
    OMET(1,I)=0.
    GO TO 21
48   TMP(1)=0.
    IF(OMET(1,I).NE.0.)TMP(1)=OMET(1,I)/ABS(OMET(1,I))
    OMETD1(1,I)=(MA(I)-MB(I)*TMP(1))/(NTIRES(I)*MOMENT(I))
21   CALL INTEG(LA(I),OMETD1(1,I))
C*****
65  CONTINUE
C   CALCULATION OF FTRA, FTRB, AND FTRC
    SFTRX=0.
    SFTRY=0.
    SFTRZ=0.
    DO 70 I=1,NSTRUT
    SFTRX=SFTRX+FTRX(I)
    SFTRY=SFTRY+FTRY(I)
70   SFTRZ=SFTRZ+FTRZ(I)
    FTRA=RL(1,1)*SFTRX+RL(1,2)*SFTRY+RL(1,3)*SFTRZ
    FTRB=RL(2,1)*SFTRX+RL(2,2)*SFTRY+RL(2,3)*SFTRZ
    FTRC=RL(3,1)*SFTRX+RL(3,2)*SFTRY+RL(3,3)*SFTRZ
C   CALCULATION OF MTX, MTY, AND MTZ
    SMTRX=0.
    SMTRY=0.
    SMTRZ=0.
    DO 75 I=1,NSTRUT
    SMTRX=SMTRX+MTRX(I)
    SMTRY=SMTRY+MTRY(I)
75   SMTRZ=SMTRZ+MTRZ(I)
    MTX=RL(1,1)*SMTRX+RL(1,2)*SMTRY+RL(1,3)*SMTRZ

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C

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MTY=RL(2,1)*SMTRX+RL(2,2)*SMTPY+RL(2,3)*SMTRZ
MTZ=RL(3,1)*SMTRX+RL(3,2)*SMTRY+RL(3,3)*SMTRZ
CALCULATION OF FXM, FYM, FZM, LM, MM, AND NM
FYM=0.
FZM=0.
FXM=0.
LM=0.
MM=0.
NM=0.
BFX=0.
BFY=0.
BFZ=0.
BLM=0.
BMM=0.
BNM=0.
DO 82 I=1,NSTRUT
DFXM=0.
DFYM=0.
DFZM=0.
DLM=0.
DMM=0.
DNM=0.
DO 14 IL4=1,NMODE
NBR=(IL4-1)*NSTRUT+I
DFXM=DFXM-MASS(I)*SXMOD(NBB)*GQD2(IL4)
DFYM=DFYM-MASS(I)*SYMOD(NBB)*GQD2(IL4)
DFZM=DFZM-MASS(I)*SZMOD(NBB)*GQD2(IL4)
NBH=(I-1)*NMODE+IL4
DLM=DLM-MASS(I)*GFORC2(NBH)*GQD2(IL4)
DMM=DMM-MASS(I)*GFORC3(NBH)*GQD2(IL4)
14 DNM=DNM-MASS(I)*GFORC4(NBH)*GQD2(IL4)
BFX=BFX+DFXM
BFY=BFY+DFYM
BFZ=BFZ+DFZM
BLM=BLM+DLM
BMM=BMM+DMM
BNM=BNM+DNM
TMP(1)=MASS(I)*SD2(1,I)
FXM=FXM+TMP(1)*A31(I)
FYM=FYM
FZM=FZM+TMP(1)*A33(I)
LM=LM+TMP(1)*A11(I)*RY(I)
MM=MM-TMP(1)*RRCGX(I)
82 NM=NM+TMP(1)*A13(I)*RY(I)
FXM=FXM+BFX+FTRA
FYM=FYM+BFY+FTRB
FZM=FZM+BFZ+FTRC
LM=LM+BLM+MTX
MM=MM+BMM+MTY

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NM=NM+BNM+MTZ

ENTRY SETUP

DO 28 I=1,NSTRUT

IF(SD1(1,I) .LE. 0.0) GO TO 20

DP1(I)=(-QD(I)+QSV1(I)-QSV3(I)+(AREA1(I)-APINT(I))*SD1(1,I))

*BETA/VOL1T(I)

IF(PGA1T(I) .LE. -1600.0) 10,20

10 PGA1T(I) = -1600.0

IF (DP1(I) .LT. 0.0) DP1(I)=0.0

20 CONTINUE

II=0

CALL VIRK4(II,N,NT,CI,SPEC,CIMAX,IERR,VAR,CUVAR,DER,ELE1,ELE2,
1 ELT,ERRVAL,DESUB,CHSUB,ITEXT)

IF(ICOSV(I) .EQ. 1)GO TO 26

XMA5(I)=XMA(I)+(2.*ZETAC2*WC1)*XMA1(I)+(WC1**2)*XMA2(I)

6 -(2.*ZETAC1*WC1)*XMA3(I)-(WC1**2)*XMA4(I)

XMA8(I)=(TC1/TC2)*XMA5(I)+(1./TC2)*XMA6(I)-(1./TC2)*XMA7(I)

XMA11(I)=(TC3/TC4)*XMA8(I)+(1./TC4)*XMA9(I)-(1./TC4)*XMA10(I)

IF(S(1,I).LE.XSTHR) XMA11(I)=0.

XSVDDD(I)=(XKSV(I)*XMA11(I)-BCON*XSVDD(I)-CCON*XSVDDOT(I)-

* DCON*(XSV(I)-XBIAS(I)))/ACON

DLTX1D(I)=(XKF(I)*DELTX(I)-DELTX1(I))/TAUF

26 CONTINUE

CALL LIMITS(XSV(I),XSVDDOT(I),XSVMAX(I),XSVMIN(I))

CALL LIMITS(XSVDDOT(I),XSVDD(I),XSVDMX(I),XSVDMN(I))

CALL LIMITS(XSVDD(I),XSVDDD(I),XDDMAX(I),XDDMIN(I))

I1 = 3*I+6*NSTRUT-2

I2 = I1+1

I3 = I1+2

CALL INTEG(LA(I1),XSVDDD(I))

CALL INTEG(LA(I2),XSVDD(I))

CALL INTEG(LA(I3),XSVDDOT(I))

I1 = I+9*NSTRUT

CALL INTEG(LA(I1),DLTX1D(I))

28 CONTINUE

RETURN

END

*IDENT PHLOZ2

*INSERT PHLOZ2.1

*DECK PHLOZ2

SUBROUTINE PHLOZ2(PS,PR,X,OC,LAP1,LAP3,RCL,D,COEF1,COEF3,MU,QTOLER
6,NITER,P1,Q1,Q3)

'PHLOZ2'.....R. D. EDSON

THIS SUBROUTINE CALCULATES THE STEADY-STATE CHAMBER PRESSURE (P1)
AND FLOW RATES (Q1 & Q3) FOR A TWO-WAY NONSYMMETRICAL SPOOL VALVE
WITH RECTANGULAR WINDOW SLOTS, GIVEN THE STROKE (X) AND THE LOAD

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C FLOW (QC). THE PARAMETERS REQUIRED IN THE 'CALL' STATEMENT ARE
C THE SAME AS DESCRIBED IN SUBROUTINE 'FLOZE2', WITH THE FOLLOWING
C ADDITIONAL PARAMETERS:
C   QC      = THE FLOW RATE TO THE LOAD
C   QTOLER  = THE TOLERANCE ALLOWED IN CALCULATING FLOW RATES, FOR
C             DETERMINING WHETHER OR NOT THE SOLUTION HAS CONVERGED
C             (.0001 IS TYPICAL)
C   NITER   = THE NUMBER OF ITERATIONS REQUIRED TO CONVERGE TO A
C             SOLUTION (INTEGER)
C
C
C   IMPLICIT REAL(L,M)
C   PFN(X1,X2,Y1,Y2,Y3,Y4)=X1+(X2-X1)*(Y2-Y1)/(-Y1+Y2-Y3+Y4)
C   NITER=0.
C   FLAG=-1.
C   P1FLAG=-1.
C
C   P1A=PR
C   P1B=PS
C   CALL FLOZE2(PS,PR,P1A,LAP1,LAP3,RCL,D,X,Q1A,Q3A,COEF1,COEF3,MU)
C   CALL FLOZE2(PS,PR,P1B,LAP1,LAP3,RCL,D,X,Q1B,Q3B,COEF1,COEF3,MU)
C   IF(Q1A.EQ.0. .AND. Q3B.EQ.0.) GO TO 51
C   Q3A=Q3A+QC
C   Q3B=Q3B+QC
C   GO TO 50
51  P1=(PS+PR)/2.
C   Q1=0.
C   Q3=0.
C   GO TO 400
C
C   50  P1=PFN(P1A,P1B,Q3A,Q1A,Q1B,Q3B)
C   CALL FLOZE2(PS,PR,P1,LAP1,LAP3,RCL,D,X,Q1,Q3,COEF1,COEF3,MU)
C   Q3=Q3+QC
C   IF(FLAG.LT.0.) GO TO 55
C   IF(P1.EQ.P1I) GO TO 100
55  P1I=P1
C   IF(Q1.EQ.0. .AND. Q3.EQ.0.) GO TO 100
C   IF(ABS(Q1) .GE. ABS(Q3)) QDEN=Q1
C   IF(ABS(Q3) .GT. ABS(Q1)) QDEN=Q3
C   IF(ABS((Q1-Q3)/QDEN) .LT. QTOLER) GO TO 100
C   IF(Q1.LT.Q3) GO TO 90
C   P1A=P1
C   Q1A=Q1
C   Q3A=Q3
C   GO TO 150
90  P1B=P1
C   Q1B=Q1
C   Q3B=Q3
C   GO TO 150

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100 P1FLAG=1.
150 FLAG=1.
    NITER=NITER+1
    IF(P1FLAG.GT.0.) GO TO 300
    GO TO 50
300 CONTINUE
    Q3=Q3-QC
400 RETURN
    END
*IDENT    FLOZE2
*INSERT    PHLOZ2.66
*DECK FLOZE2
    SUBROUTINE FLOZE2(PS,PR,P1,LAP1,LAP3,RCL,D,X,Q1,Q3,COEF1,COEF3,MU)
    'FLOZE2'.....R. D. EDSON

    THIS SUBROUTINE CALCULATES THE STEADY-STATE FLOW RATES (Q1 AND
    Q3) FOR A TWO-WAY NONSYMMETRICAL SPOOL VALVE WITH RECTANGULAR
    WINDOW SLOTS, GIVEN THE LOAD CHAMBER PRESSURE (P1) AND STROKE
    (X). THE PARAMETERS REQUIRED IN THE 'CALL' STATEMENT ARE AS
    FOLLOWS:
    X      = VALVE STROKE
    P1     = PRESSURE IN CHAMBER 1 (TO LOAD)
    Q1     = FLOW RATE FROM SUPPLY LINE TO CHAMBER 1
    Q3     = FLOW RATE FROM CHAMBER 1 TO RETURN LINE
    PS     = SUPPLY PRESSURE
    PR     = RETURN PRESSURE
    LAP1   = OVERLAPPED OR UNDERLAPPED LENGTH BETWEEN THE SPOOL
             AND SLEEVE AT NULL, FOR FLOW Q1. A POSITIVE NUMBER
             IS USED FOR OVERLAP, A NEGATIVE NUMBER FOR UNDERLAP.
    LAP3   = OVERLAPPED OR UNDERLAPPED LENGTH BETWEEN THE SPOOL
             AND SLEEVE AT NULL, FOR FLOW Q3. A POSITIVE NUMBER
             IS USED FOR OVERLAP, A NEGATIVE NUMBER FOR UNDERLAP.
    RCL    = RADIAL CLEARANCE BETWEEN THE SPOOL AND SLEEVE
    D      = DIAMETER OF SPOOL
    COEF1  = FLOW COEFFICIENT OF ORIFICE 1 (SUPPLY TO CHAMBER 1)
             =  $CD*W1*SQRT(2.*GC/RHO)$ 
    COEF3  = FLOW COEFFICIENT OF ORIFICE 3 (CHAMBER 1 TO RETURN)
             =  $CD*W3*SQRT(2.*GC/RHO)$ 
             WHERE W1 = TOTAL WINDOW WIDTH OF ORIFICE 1
                   W3 = TOTAL WINDOW WIDTH OF ORIFICE 3
                   CD = DISCHARGE COEFFICIENT
                   GC = GRAVITATIONAL ACCELERATION CONSTANT
                   RHO = DENSITY OF HYDRAULIC FLUID
    MU     = VISCOSITY OF HYDRAULIC FLUID, CENTIPOISE

    THE METHOD OF SOLUTION UTILIZES THE TURBULENT ORIFICE EQUATION
    AND THE EQUATION FOR FULLY-DEVELOPED LAMINAR FLOW THROUGH AN
    ANNULUS, WITH FULL ECCENTRICITY ASSUMED. FOR ORIFICE OPENINGS
    WHERE SOME OVERLAPPED LENGTH EXISTS, THE PROCEDURE IS TO CALC-

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C      ULATE THE FLOW RATE BY BOTH EQUATIONS, AND THEN USE THE ONE
C      THAT GIVES THE SMALLEST ABSOLUTE VALUE AS THE ANSWER.  FOR
C      OPENINGS WHERE NO OVERLAPPED LENGTH EXISTS, ONLY THE TURBULENT
C      ORIFICE EQUATION APPLIES.
C
C      IMPLICIT REAL(L,M)
C      Q12(T1,T2)=SIGN(1.,(T1-T2))*SQRT(ABS(T1-T2))
C      Q34(T3,T4)=4.5E06*(T3-T4)*D*RCL**3/MU
C
C      ***** CALCULATE Q1 *****
C      X2=LAP1-X
C      X4=SQRT(X2**2+RCL**2)
C      IF(LAP1 .LE. 0.) GO TO 99
C
C      POSITIVE LAPS:
C      IF(X .GE. LAP1) GO TO 65
C      Q1=RCL*COEF1*Q12(PS,P1)
C      Q1L=Q34(PS,P1)/X2
C      IF(ABS(Q1L) .LT. ABS(Q1)) Q1=Q1L
C      GO TO 20
65    Q1=X4*COEF1*Q12(PS,P1)
C      GO TO 20
C
C      NEGATIVE LAPS:
C      99  IF(X .LT. LAP1) GO TO 10
C      Q1=X4*COEF1*Q12(PS,P1)
C      GO TO 20
10    Q1=RCL*COEF1*Q12(PS,P1)
C      Q1L=Q34(PS,P1)/X2
C      IF(ABS(Q1L) .LT. ABS(Q1)) Q1=Q1L
C
C      ***** CALCULATE Q3 *****
C      20  X1=LAP3+X
C      X3=SQRT(X1**2+RCL**2)
C      IF(LAP3 .LE. 0.) GO TO 199
C
C      POSITIVE LAPS:
C      IF(X .LE. -LAP3) GO TO 165
C      Q3=RCL*COEF3*Q12(P1,PR)
C      Q3L=Q34(P1,PR)/X1
C      IF(ABS(Q3L) .LT. ABS(Q3)) Q3=Q3L
C      GO TO 120
165   Q3=X3*COEF3*Q12(P1,PR)
C      GO TO 120
C
C      NEGATIVE LAPS:
C      199 IF(X .GT. -LAP3) GO TO 110
C      Q3=X3*COEF3*Q12(P1,PR)
C      GO TO 120

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110  Q3=RCL*COEF3*Q12(P1,PR)
      Q3L=Q34(P1,PR)/X1
      IF(ABS(Q3L) .LT. ABS(Q3)) Q3=Q3L
120  RETURN
      END
*IDENT  LIMITS
*INSERT  FLOZE2.92
*DECK LIMITS
      SUBROUTINE LIMITS(X,XDOT,XMAX,XMIN)
C
C      STATEMENTS FOR THIS SUBROUTINE OBTAINED BY PHONE BY
C      JOHN R. MCGEEHEE ON 2/1/77.
C
      IF(X .GE. XMAX) GO TO 10
      IF(X .LE. XMIN) GO TO 20
      GO TO 30
10  X=XMAX
      IF(XDOT .GT. 0.0) XDOT = 0.0
      GO TO 30
20  X=XMIN
      IF(XDOT .LT. 0.0) XDOT = 0.0
30  RETURN
      END
*IDENT  GMMODS
*DELETE  D81877.1,D81877.1
      DIMENSION TITLE(20),BUF(550),NDIL(28),TBUF(550)
*DELETE  PLTDAT.6,PLTDAT.6
      1 BHM(2,550),CMMDS(6),DEPVAR(5),LINE(7),NDVA(5)
*IDENT  JMMODS
*DELETE  ACTINIT.40,ACTINIT.65
      DATA ICOSV,IFSTOP,II/11*0/
*INSERT  SDFLGP.80
      DATA N19/19/
*DELETE  SDFLGP.96,SDFLGP.96
      IF(ISUM1.NE.0) WRITE(13) N19,ISUM1,DAT3,OP17,ACOVAR8(1),
      1 ACOVAR8(4),ACOVAR4(5),ACOVAR1(4),ACOVAR1(5)
*DELETE  SDFLGP.105,SDFLGP.105
      1 OMET(1,1),WFORT(1),OSVCU(1),OSV(1),PGA1T(1),PGA2T(1)
*DELETE  SDFLGP.108,SDFLGP.108
      1 OMET(1,2),WFORT(2),OSVCU(2),OSV(2),PGA1T(2),PGA2T(2)
*DELETE  SDFLGP.111,SDFLGP.111
      1 OMET(1,3),WFORT(3),OSVCU(3),OSV(3),PGA1T(3),PGA2T(3)
*DELETE  SDFLGP.114,SDFLGP.114
      1 OMET(1,4),WFORT(4),OSVCU(4),OSV(4),PGA1T(4),PGA2T(4)
*DELETE  SDFLGP.117,SDFLGP.117
      1 OMET(1,5),WFORT(5),OSVCU(5),OSV(5),PGA1T(5),PGA2T(5)
*INSERT  READ.14
      DATA ACTIVE/6HACTIVE/
*DELETE  READ.42,READ.42

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19 IF(SYM.EQ.ACTIVE) GO TO 805
   CALL DIPLAC(RA1,INC,BLANK)
*INSERT      READ.183
805 CALL ACTIN
   GO TO 100
810 CALL ACTIN
   GO TO 802
*DELETE      READ.184,READ.184
26 IF(SYM.EQ.ACTIVE) GO TO 810
   IF(SYM.EQ.STCASE) GO TO 21
*IDENT      DIRACT
*INSERT      DIR3DA.12
*DECK DIRACT
BLOCK DATA DIRACT
COMMON/ACTDIR/NAME(71),LOC(71)
DATA NAME/ 6HAMUH , 6HAPINT , 6HAREA1 , 6HAREA2 ,
1 6HAREA3 , 6HAREMO , 6HAREO3 , 6HBETA , 6HBLMU ,
2 6HBUMU , 6HCDMOC , 6HCDMOE , 6HCDSV , 6HCD3 ,
3 6HCFFOR , 6HDELT , 6HDIOTA , 6HDSV , 6HEPSILO,
4 6HEPSROL , 6HEPSSLP, 6HETASV , 6HGAMAH , 6HGAMAN ,
5 6HGNR , 6HIDEACT, 6HKAPT , 6HOMRUN , 6HPATM ,
6 6HPGAHAC, 6HPGALAC, 6HPGA1I , 6HPGA2I , 6HPGA3I ,
7 6HPERCNT, 6HQPUMPS, 6HRCLSV , 6HRHOH , 6HTAUF ,
8 6HTC1 , 6HTC2 , 6HTC3 , 6HTC4 , 6HVOLACI,
9 6HVOLANI, 6HVOL1I , 6HVOL2I , 6HVOL3I , 6HWC ,
1 6HWC1 , 6HWSV , 6HWSV1 , 6HWSV3 , 6HWLFOR ,
2 6HWLFORR, 6HXBIA , 6HXDDMAX, 6HXDDMIN, 6HXKA ,
3 6HXKF , 6HXKSV , 6HXLPSV1, 6HXLPSV3, 6HXSOM ,
4 6HXSTHR , 6HXSVDNM, 6HXSVDMX, 6HXSVMAX, 6HXSVMIN,
5 6HZETAC1, 6HZETAC2/
DATA LOC/ 1, 6, 16, 21, 26, 31, 36, 61, 46,
1 56, 62, 67, 72, 82, 77, 109, 141, 135, 152,
2 157, 162, 163, 195, 200, 201, 212, 263, 274, 275,
3 281, 286, 291, 301, 311, 276, 346, 387, 397, 410,
4 411, 412, 413, 414, 621, 631, 421, 431, 441, 451,
5 452, 641, 642, 643, 458, 463, 464, 611, 616, 474,
6 479, 484, 489, 494, 561, 499, 591, 596, 601, 606,
7 644, 645/
END
*IDENT      ACTIN
*INSERT      READ.194
*DECK ACTIN
SUBROUTINE ACTIN
COMMON/ACTDIR/XNAME(71),LOC(71)
COMMON/ACTIVE/DATA(646)
DIMENSION IRA(55),MSG(58),IDATA(646)
EQUIVALENCE (MSG(1),SYM),(MSG(2),OP)
EQUIVALENCE (MSG(3),IRA(1)),(MSG(58),INC)
EQUIVALENCE (DATA(1),IDATA(1))

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```
INTEGER COMMA,POINT,E,BLANK
DATA REMARK,COMMA,BLANK,POINT,E,ENDACT,MINUS,AINT/
1 3HREM,1R,,1R ,1R.,1RE,6HENDACT,1R-,3HINT/
100 CONTINUE
1 FORMAT(A6,1X,A3,1X,55R1,I1)
2 FORMAT(18X,A6,1X,A3,1X,55R1,I6)
3 FORMAT(20HOERROR.THE SYMBOL **,A6,
1 26H** IS NOT IN THE DIRECTORY/1H )
4 FORMAT(44HOERROR.ILLEGAL CHARACTER IN NUMERIC FIELD **,1R1,2H**/)
READ(5,1) SYM,OP,IRA,INC
CALL LINES(1)
J = 58
IF(INC.LE.0) J = 57
WRITE(6,2) (MSG(I),I=1,J)
IF(SYM.EQ.REMARK) GO TO 100
IF(SYM.EQ.ENDACT) GO TO 999
DO 110 I=1,71
IF(SYM.EQ.XNAME(I)) GO TO 120
110 CONTINUE
CALL LINES(3)
WRITE(6,3) SYM
GO TO 100
120 CONTINUE
INDEX = LOC(I)
IF(INC.EQ.0) INC = 1
INDEX = INDEX + INC - 1
NUMEXP = 0
NEXP = 0
IEXP = 0
NL = 0
NR = 0
NUML = 0
NUMR = 0
ISIGN = 0
JSIGN = 0
LEFT = 1
DO 210 I=1,56
IF(I.EQ.56) GO TO 140
IF(IRA(I).EQ.BLANK) GO TO 210
IF(IRA(I).EQ.COMMA) GO TO 140
IF(IRA(I).EQ.POINT) GO TO 170
IF(IRA(I).EQ.E) GO TO 180
IF(IRA(I).EQ.MINUS) GO TO 200
IF(IRA(I).GT.36) GO TO 130
IF(IRA(I).LT.27) GO TO 130
NUM = IRA(I) - 27
IF(IEXP.EQ.1) GO TO 190
IF(LEFT.GT.0) NUML = 10*NUML + NUM
IF(LEFT.GT.0) NL = NL + 1
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IF(LEFT.LT.0) NUMR = 10*NUMR + NUM
IF(LEFT.LT.0) NR = NR + 1
GO TO 210
130 CALL LINES(3)
WRITE(6,4) IRA(I)
GO TO 210
140 CONTINUE
IF(NL.EQ.0.AND.NR.EQ.0) GO TO 210
IF(NR.EQ.0) GO TO 160
X = FLOAT(NUML) + FLOAT(NUMR)/10.**NR
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
DATA(INDEX) = X
150 NUML = 0
NUMR = 0
NL = 0
NR = 0
LEFT = 1
ISIGN = 0
JSIGN = 0
IEXP = 0
NEXP = 0
NUMEXP = 0
INDEX = INDEX + 1
GO TO 210
160 CONTINUE
X = NUML
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
NUML = X
IF(OP.EQ.AINT) IDATA(INDEX) = NUML
IF(OP.NE.AINT) DATA(INDEX) = X
GO TO 150
170 CONTINUE
LEFT = -1
GO TO 210
180 CONTINUE
IEXP = 1
GO TO 210
190 CONTINUE
NUMEXP = 10*NUMEXP + NUM
NEXP = NEXP + 1
GO TO 210
200 CONTINUE
IF(IEXP.EQ.0) ISIGN = 1
IF(IEXP.NE.0) JSIGN = 1
210 CONTINUE

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GO TO 100
999 RETURN
END
*IDENT      CSCMOD2
*DELETE      ACOBLK.11,ACOBK.11
             8,FFORT(5) ,FWORK(5) ,FONHST(5) ,FORCHT(5) ,FORSST(5) ,FSTOP(5)
*DELETE      ACOBLK.16,ACOBK.16
             3,PERCNT(5) ,PGAHAC(5) ,PGALAC(5) ,PGA1I(5) ,PGA1T1(5)
*DELETE      ACOBLK.31,ACOBK.31
             9,WSV      ,WSV1      ,WSV3      ,ZETAC1      ,ZETAC2      ,ZSSC(5)
             *,IMODE(5) ,CMASNG      ,VMAS(5)
*DELETE      EXE.41,EXE.41
             COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*INSERT      EXE.280
             STOP "FLIGHT TIME LIMIT"
*DELETE      MIMIN.8,MIMIN.8
             COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*INSERT      MIMIN.10
C
DO 10 I=1,5
10 INDINT(I) = 1
*DELETE      LGDET.9,LGDET.9
             COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*DELETE      LGEAR1.50,LGEAR1.50
             COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*INSERT      LGEAR1.81
DO 6 I=1,5
P(I) = 0.0
6 P2(I) = 0.0
*INSERT      ACTINIT.35
             COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*DELETE      ACTINIT.38,ACTINIT.39
*DELETE      JMMODS.1,JMMODS.1
             EQUIVALENCE (DM15(16),GREFF), (DM1(37),AIYYBS)
C
DATA DSTOP,FSTOPK /0.004,0.0/
DATA ENUP,FSTOP,HMM,QSVN,RESA,SA /30*0.0/
DATA ICOSV,IFRI,IFSTOP,IPASS /20*0/
DATA IOPCO,ISET /10*1/
DATA IXSVH,IXSVL,IIXSVH,IIXSVL /20*0/
C
*INSERT      ACTINIT.75
             OMRUN=ERDEG*0.01745329
*DELETE      ACTINIT.76,ACTINIT.77
*INSERT      ACTINIT.80
             INDINT(I) = 1
*DELETE      ACTINIT.127,ACTINIT.127
             PGA1T1(I)=PGA1I(I)
*INSERT      ACTINIT.140

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      VMASS(I)=AMASS
*DELETE      ALGEAR.32,ALGEAR.33
      *GQD2(20),DDM20(20),DDM21(20),SLEN1(5),SLEN2(5),
      *GAMA,DUM15(12),INDNWS,DDM22(10),ETADES,
*DELETE      ALGEAR.54,ALGEAR.54
      COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*DELETE      ALGEAR.71,ALGEAR.76
      EQUIVALENCE (DM5(16),GREFF), (DM2(27),AXP7F), (DM2(28),AX77F),
      *(DM2(29),AYP7F), (DM2(30),AY77F), (DM2(31),AZP7F), (DM2(33),AZ77F),
      *(DM1(37),AIYYBS)
      DIMENSION INDEACT(5),IPSTOP(5),AIC(5),PGA1T(5)
      DATA AIC,INDEACT,IPSTOP /5*0.0,10*0/
*INSERT      ALGEAR.96
C      CMASNG = AIYYBS/AMASS+AMASS*RX(1)*RX(1)/AIYYBS
      CMASNG = 7.9677
      VMASS(1) = AMASS/CMASNG
      ENCG = 0.5*AMASS*ZG77F1(1)*ZG77F1(1)
      ZDANT = ZG77F1(1)-QI77P*RX(1)
*DELETE      ALGEAR.102,ALGEAR.103
C
      18 DO 90 I=1,NSTRUT
      IF(INDEACT(I).EQ.1) GO TO 56
      IF(INDEACT(I).EQ.2) GO TO 80
*INSERT      ALGEAR.105
      IF(I.EQ.1 .AND. ZDANT.GE.VELDEC) GO TO 90
*INSERT      ALGEAR.108
      IF(I.EQ.1 .AND. ZDANT.GE.VELDEC+XG77F1(1)*TAN(OMRUN)) GO TO 90
*DELETE      ALGEAR.111,ALGEAR.113
      INDEACT(I)=1
      56 IF(INDINT(I).EQ.0) GO TO 58
      WLFOR(I)=WLFOR(I)-REDSLP(I)*HT
      EPSILO(I)=EPSILO(I)+EPSSLP*HT
      58 CONTINUE
      INDINT(I)=0
*DELETE      ALGEAR.117,ALGEAR.117
      INDEACT(I)=2
*DELETE      ALGEAR.122,ALGEAR.125
C*** CALCULATION OF THE WING-GEAR INTERFACE FORCE (WFORT)
      UNSPRNG = 0.0
      DO 66 J=1,NSTRUT
      IF(OMETD1(J).NE. 0.0) UNSPRNG = UNSPRNG+MASS(J)
      66 CONTINUE
      DWFORT = (-SQRT(AXP7F*AXP7F+AYP7F*AYP7F+
      8      AZP7F*AZP7F)+GREFF)*(AMASS-UNSPRNG)
      WFORT(1) = DWFORT+FORSS1(1)
      DO 67 I=2,NSTRUT
      WFORT(I) = DWFORT/(NSTRUT-1)
      67 CONTINUE

```

C

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DO 65 I=1,NSTRUT

C

C***

```
*DELETE      ALGEAR.129,ALGEAR.129
      PGA1T(I)=PGA1T1(I)
      IF(PGA1T1(I).LE.-1600.0) PGA1T(I)=-1600.0
*DELETE      ALGEAR.134,ALGEAR.134
      PGA2T(I)=AP2T0(I)*((VOL2I(I)/VOL2T(I))*GAMA)-PATM
*DELETE      ALGEAR.152,ALGEAR.153
      IF(IMODE(I).EQ.0 .AND. DDELTA(I).LE.0.0) 112,113
112 QO(I) = 0.0
      GO TO 109
```

C

```
113 QO(I)=COEFO(I)*(AREMO(I)-APINT(I))*Q1(PGA1T1(I),PGA2T(I))
      IF(QO(I).LT.0.0 .AND. VCUM(I).LE.0.0) GO TO 102
      AIC(I)=0.0
      GO TO 103
102 IF(PGA1T1(I).LT.PGA2T(I)) GO TO 103
      GO TO 111
111 QO(I)=0.0
      VCUM(I)=0.0
      AIC(I)=1.0
103 IF(QO(I).GT.0.0) AIC(I)=0.0
*DELETE      ALGEAR.177,ALGEAR.177
      IF(IMODE(I).EQ.0) GO TO 297
*DELETE      ALGEAR.179,ALGEAR.180
      GO TO 289
*DELETE      ALGEAR.188,ALGEAR.188
284 IF(S(1,I).LT.0.0) GO TO 160
      GO TO 161
160 IPSTOP(I)=1
161 IF(S(1,I).LE.DSTOP .AND. IPSTOP(I).EQ.1) GO TO 900
*DELETE      ALGEAR.190,ALGEAR.190
900 IF(S(1,I).LE.0.005) GO TO 903
*DELETE      ALGEAR.195,ALGEAR.199
      IPSTOP(I)=0
*DELETE      ALGEAR.215,ALGEAR.215
901 IF(PGA1T(I).LE.(PGA1I(I)+500.0) .AND.
      : PGA1T(I).GT.(PGA1I(I)-500.0)) GO TO 158
      GO TO 159
158 IF(ABS(FT(I)).LE.FORCHT(I) .AND. S(1,I).EQ.0.0) GO TO 500
159 IF(S(1,I).GE.0.0) GO TO 470
*DELETE      ALGEAR.225,ALGEAR.225
      IF(IMODE(I).EQ.0) GO TO 297
*DELETE      ALGEAR.227,ALGEAR.228
*INSERT      ALGEAR.293
297 I2 = 2*I+NSTRUT-1
      I1 = I2+1
      CALL INTEG(LA(I2),SD2(1,I))
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      CALL INTEG(LA(I1),SD1(1,I))
C
      IF(IMODE(I).EQ.0) GO TO 450
*DELETE      ALGEAR.295,ALGEAR.296
      IF(I.NE.1) GO TO 119
      ENUP(1) = 0.5*AIYYBS*QI77R*QI77R+(ENCG/CMASNG)*
      *      (ZG77F1(1)/ABS(ZG77F1(1)))
      GO TO 120
      119 ENUP(I) = ENCG/(NSTRUT-1)
      120 CONTINUE
*DELETE      ALGEAR.300,ALGEAR.300
      IF(WFORT(I).GE.0.0 .OR. DDELTA(I).LT.0.0) XSTOT(I)=1.E20
*DELETE      ALGEAR.302,ALGEAR.303
      ZSSC(I)=FWORK(I)*SB(I)
      IF(XSTOT(I).LE.(ZSSC(I)-S(1,I)) .OR. RESA(I).EQ.1.0) SA(I)=1.0
*DELETE      ALGEAR.307,ALGEAR.307
C
      VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/
      &      (AMASS*REDSLP(I))
*DELETE      ALGEAR.342,ALGEAR.342
      PS(I)=(((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMA)-PATM)/144.0.
*DELETE      ALGEAR.360,ALGEAR.361
*DELETE      ALGEAR.371,ALGEAR.376
      I4 = I+4*NSTRUT
      I5 = I+5*NSTRUT
      CALL INTEG(LA(I4),OO(I))
      CALL INTEG(LA(I5),QSV(I))
      I9 = I+9*NSTRUT
      CALL INTEG(LA(I9),DLTX1D(I))
*DELETE      ALGEAR.414,ALGEAR.417
*DELETE      ALGEAR.516,ALGEAR.516
*DELETE      ALGEAR.519,ALGEAR.521
      IF(IMODE(I).EQ.0 .AND. DDELTA(I).LE.0.0) GO TO 19
      IF(S(1,I).NE.0.0 .OR. AIC(I).NE.1.0) GO TO 20
      PGA1T1(I)=PGA1I(I)
      PGA1T(I) =PGA1I(I)
      19 DP1(I)=0.0
*DELETE      ALGEAR.539,ALGEAR.546
      I3 = I+3*NSTRUT
      CALL INTEG(LA(I3),DP1(I))
      I6 = 3*I+6*NSTRUT-2
      I7 = I6+1
      I8 = I6+2
      CALL INTEG(LA(I6),XSVDDD(I))
      CALL INTEG(LA(I7),XSVDD(I))
      CALL INTEG(LA(I8),XSVDDOT(I))
*DELETE      FLEX1.25,FLEX1.25
      COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
*DELETE      CSCMODS.32,CSCMODS.32

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DATA(ACOVARI(I),I=1,8 ) /5HVOL1T,5HVOL2T,5HVOL3T,6HPGA1T1,5HPGA2T,
*DELETE      CSCMODS.35,CSCMODS.35
*      6HFORCHT,5HVMAS,6HFONHST/
*DELETE      JMMODS.5,JMMODS.5
      1 OMET(1,1),WFORT(1),QSV(1),PGA1T1(1),PGA2T(1)
*DELETE      JMMODS.6,JMMODS.6
      1 OMET(1,2),WFORT(2),QSV(2),PGA1T1(2),PGA2T(2)
*DELETE      JMMODS.7,JMMODS.7
      1 OMET(1,3),WFORT(3),QSV(3),PGA1T1(3),PGA2T(3)
*DELETE      JMMODS.8,JMMODS.8
      1 OMET(1,4),WFORT(4),QSV(4),PGA1T1(4),PGA2T(4)
*DELETE      JMMODS.9,JMMODS.9
      1 OMET(1,5),WFORT(5),QSV(5),PGA1T1(5),PGA2T(5)
*DELETE      CSCMODS.53,CSCMODS.53
      CALL STOVAR(8,VOL1T(I),VOL2T(I),VOL3T(I),PGA1T1(I),PGA2T(I),
*DELETE      CSCMODS.59,CSCMODS.59
      * FORCHT(I),VMAS(I),FONHST(I))
*DELETE      CSCMODS.100,CSCMODS.100
      CALL UPDAT(1,LA(I1),PGA1T1(I),DU,DU,DU,DU)
*DELETE      ACTIN.4,ACTIN.5
      COMMON /ACTIVE/ DATA(656)
      DIMENSION IRA(55), MSG(58), IDATA(656)
*DELETE      PACK.3,PACK.3
      DIMENSION I1(6)
*DELETE      DIRACT.8,DIRACT.9
      4 6HEPSROL, 6HEPSSLP, 6HETASV , 6HFWORK ,6HGAMAH ,
      5 6HGNR , 6HKAPT , 6HOMRUN , 6HPATM ,
*DELETE      DIRACT.18,DIRACT.18
      5 6HZETAC1, 6HZETAC2, 6HIMODE /
*DELETE      DIRACT.21,DIRACT.21
      2 157, 162, 163, 170, 195, 201, 263, 274, 275,
*DELETE      DIRACT.26,DIRACT.26
      7 644, 645, 651/
*IDENT      EOR
*IDENT      EORPL
*INSERT      CTENGL.65
*DECK EORPL
*WEOR
*IDENT      CSCMOD3
*DELETE      CSCMOD2.4,CSCMOD2.4
      *,IMODE(5) ,CMASNG ,VMAS(5) ,ZDANT
*INSERT      EXE.42
      COMMON/TABDIR/TABLE(800)
      COMMON/READ1/DUM1(64),JBC,INX0
      COMMON /UPDCAL/ NUM , P( 90),Y( 90)
      COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
      COMMON/STGT/ICOUNT,KCOUNT,LOCAIN(4),LOCADE(4)
      COMMON/TABCOM/LOCS(115),STABLE(115)
      COMMON/CLEAUP/I2,CLEAN,INTEG

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COMMON/STORA/ARG(48),ALIST(8),GETARG(8),NENT,LENT,K
COMMON/LGE/A11(5),A13(5),A31(5),A33(5),RRCGX(5),
CRL(3,3),RI(3,3,5),RAX(5),RAY(5),RAZ(5),TMP(3),ZZERO(5),
CXR(5),YR(5),EPSLON(5),PA(5),FDELTA(5),
CFTRZ(5),RDX(5),RDY(5),RDZ(5),RDXG(5),RDYG(5),RDZG(5),
CVTX(5),VTY(5),VTZ(5),GZ(5), VGPT(5),FTRX(5),FTRY(5),
CDX(5),DY(5),DZ(5),FT(5),FDX(5),FDY(5),FF(5),AA(5),C2(5),
CSR(5),SF(5),PSKD(5),MUV(5),MTRX(5),MTRY(5),
CMTRZ(5),MA(5),RG11,RG13,RG31,RG33,IPRT,
CMTX,MTY,MTZ,SFTRX,SFTRY,SFTRZ,FTRA,
CFTRB,FTRC,SMTRX,SMTRY,SMTRZ
COMMON/ACTDIR/NAME(71),LOC(71)

```
*INSERT      C11778.22
DATA NCASE /10H      /
*INSERT      SDDMODS.4
DATA PGA1T1,PGA2T,OSV,OSVCU,WFORT /25*0.0/
*INSERT      LGEA3C.47
*CALL ACDBLK
*DELETE      CSCMOD2.44,CSCMOD2.45
      CMASNG = 1+(AMASS*RX(1)*RX(1))/AIYYBS
*INSERT      CSCMOD2.62
      UNSPR = 0.0
*DELETE      CSCMOD2.65,CSCMOD2.65
      IF(J.EQ.1 .AND. OMETD1(1).NE.0.0) UNSPR = MASS(1)
      IF(J.GT.1 .AND. OMETD1(J).NE.0.0) UNSPRNG = UNSPRNG+MASS(J)
*DELETE      CSCMOD2.69,CSCMOD2.69
      WFORT(1) = SR(1)*(VMASS(1)-UNSPR)
*DELETE      CSCMOD2.125,CSCMOD2.125
      6      (VMASS(1)*REDSLP(I))
*DELETE      CSCMODS.44,CSCMODS.44
      DATA(ACOVAR7(I), I=1,8) /6HVELDEC,6HWFORR,5HZDANT,4HZSSC,4HCOPA,
*DELETE      SDFLGP.80,SDFLGP.80
*INSERT      C82477.63
      IF(IABS(INDLG).NE.3) 110,115
110 IF(ISUM1.NE.0) WRITE(13) N14,ISUM1,DAT3,OP17
      GO TO 120
115 CONTINUE
*INSERT      JMMODS.4
120 CONTINUE
*INSERT      C82477.64
      IF(IABS(INDLG).NE.3) 210,220
210 CONTINUE
      IF(ISTPL1.NE.0) WRITE(13) FT(1),SF(1),DELTA(1),P(1),P2(1),MA(1),
*SD2(1,1),SD1(1,1),S(1,1),S2D2(1,1),S2D1(1,1),S2(1,1),OMETD1(1,1),
*OMET(1,1)
      IF(ISTPL2.NE.0) WRITE(13) FT(2),SF(2),DELTA(2),P(2),P2(2),MA(2),
*SD2(1,2),SD1(1,2),S(1,2),S2D2(1,2),S2D1(1,2),S2(1,2),OMETD1(1,2),
*OMET(1,2)
      IF(ISTPL3.NE.0) WRITE(13) FT(3),SF(3),DELTA(3),P(3),P2(3),MA(3),
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*SD2(1,3),SD1(1,3),S(1,3),S2D2(1,3),S2D1(1,3),S2(1,3),OMETD1(1,3),
*OMET(1,3)
  IF(ISTPL4.NE.0) WRITE(13) FT(4),SF(4),DELTA(4),P(4),P2(4),MA(4),
*SD2(1,4),SD1(1,4),S(1,4),S2D2(1,4),S2D1(1,4),S2(1,4),OMETD1(1,4),
*OMET(1,4)
  IF(ISTPL5.NE.0) WRITE(13) FT(5),SF(5),DELTA(5),P(5),P2(5),MA(5),
*SD2(1,5),SD1(1,5),S(1,5),S2D2(1,5),S2D1(1,5),S2(1,5),OMETD1(1,5),
*OMET(1,5)
  GO TO 230
220 CONTINUE
*INSERT      CSCMOD2.153
230 CONTINUE
*DELETE      CSCMODS.93,CSCMODS.93
  CALL STOVAR(8,VELDEC,WLFORR,ZDANT,ZSSC,COPA,SIPA,DSTOP,
*IDENT      CSCMOD4
*DELETE      ACOBLK.12,ACOBK.12
  9,GAMAH(5) ,GAMAN ,GNR ,HMM(5) ,ICOSV(5) ,INITSW
*DELETE      ACOBLK.13,ACOBK.13
  *,IFR(5) ,IIXSVH(5) ,IIXSVL(5) ,IOPCO(5) ,IPASS(5) ,ISET(5)
*DELETE      ACOBLK.22,ACOBK.22
  COMMON /ACTIVE/ TC3 ,TC4 ,VCUM(5) ,ENCG ,VOL11(5)
*DELETE      CSCMOD3.1,CSCMOD3.1
  *,IMODE(5) ,CMASNG ,VMAS(5) ,ZDANT ,INDEACT(5),VELDEC
*DELETE      EXE.175,EXE.175
  IF(TIME .LT. TMAX) GO TO 413
*DELETE      EXE.286,EXE.286
  STOP "EXECUTIVE ROUTINE"
*INSERT      JMMODS.1
  DIMENSION S(2,5)
  EQUIVALENCE (S(1),S(1,1))
  EQUIVALENCE (DM15(1),ITO)
*DELETE      CSCMOD2.21,CSCMOD2.21
  DATA ICOSV,IFR,IFSTOP,IPASS /20*0/
*INSERT      CSCMOD2.25
  INITSW = 1
*INSERT      CSCMOD2.28
  INDEACT(I) = 0
  IF(ITO.EQ.1) INDEACT(I) = 2
*DELETE      ACTINIT.143,ACTINIT.143
  CALL ALGEAR
*DELETE      ALGEAR.70,ALGEAR.70
  EQUIVALENCE (DM15(1),ITO)
*DELETE      ALGEAR.93,ALGEAR.93
C
  IF(INITSW.EQ.1) GO TO 16
*INSERT      ALGEAR.98
C
  CALL LGEA3C
.C

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*DELETE      CSCMOD2.50,CSCMOD2.50
      18 CALL ACTNG
C
      DO 274 I=2,NSTRUT
*DELETE      CSCMOD2.53,CSCMOD2.53
*DELETE      CSCMOD2.54,CSCMOD2.54
*DELETE      CSCMOD2.62,CSCMOD2.72
*DELETE      CSCMOD2.74,CSCMOD2.74
*DELETE      ALGEAR.182,ALGEAR.182
      801 IF(SD1(1,I).LE.0.8 .AND. IFR(I).EQ.0) GO TO 2
*DELETE      ALGEAR.187,ALGEAR.187
      IFR(I) = 1
*DELETE      ALGEAR.200,ALGEAR.201
*DELETE      ALGEAR.212,ALGEAR.212
      IFR(I) = 0
*DELETE      ALGEAR.257,ALGEAR.258
*DELETE      ALGEAR.275,ALGEAR.276
*DELETE      CSCMOD2.114,CSCMOD2.117
*DELETE      CSCMOD2.119,CSCMOD2.119
*DELETE      CSCMOD3.29,CSCMOD3.29
      6 VMASS(I)*REDSLP(I))
*INSERT      ALGEAR.370
C
      DO 65 I=1,NSTRUT
*DELETE      ALGEAR.513,ALGEAR.513
      16 CONTINUE
      INITSW = 0
*DELETE      ALGEAR.523,ALGEAR.525
*DELETE      CSCMODS.48,CSCMODS.48
      DATA(ACOVAR9(I), I=1,8) /6HXSVD00,5HXSVD0,6HXSVD0T,2HPR,3HIFR,
*DELETE      JMMODS.2,JMMODS.2
      DATA N20 /20/
*DELETE      JMMODS.3,JMMODS.3
      IF(ISUM1.NE.0) WRITE(13) N20,ISUM1,DAT3,OP17,ACOVAR8(1),
*INSERT      JMMODS.4
      2,ACOVAR3(3)
*INSERT      CSCMOD2.149
      2,FORSST(1)
*INSERT      CSCMOD2.150
      2,FORSST(2)
*INSERT      CSCMOD2.151
      2,FORSST(3)
*INSERT      CSCMOD2.152
      2,FORSST(4)
*INSERT      CSCMOD2.153
      2,FORSST(5)
*DELETE      CSCMODS.78,CSCMODS.78
      CALL STOVAR(8,XSVD00(I),XSVD0(I),XSVD0T(I),PR(I),FLOAT(IFR(I)),
*IDENT      ACTNG

```

*INSERT ALGEAR.549
*DECK ACTNG
 SUBROUTINE ACTNG

C

***** FATOLA VARIABLES *****

C

```
COMMON/DIRCOM/DM1(115),ALPHD,DM1A(20),AMASS,DM2(147),DCL1,DCM1,  
C DCN1,DCL2,DCM2,DCN2,DCL3,DCM3,DCN3,DM3(99),FXB7P,  
C DUM4(3),FYB7P(4),FZB7P,DM5(17),GX87F,DM6(8),GZB7F,  
C DM7(218),INDSTE(48),PHIPD,INDSTE1(23),PSIPD,INDSTE2(156),THTPD,  
C INDSTE3(5),TIME,DM8(287),PI77R(2),PI77R1(2),DM9(4),  
C QI77R(2),QI77R1(2),DM10(4),RI77R(2),RI77R1(2),DM11(48),  
C XG77F(2),XG77F1(12),YG77F(2),YG77F1(12),  
C ZG77F(2),ZG77F1(2),DUM13(52),  
C NSTRUT,MASS(5),RX(5),RY(5),RZ(5),THETAD(5),ERDEG,RGR,  
C NTIRES(5),RZERO(5),W(5),DELTAM(5),MOMENT(5),  
C RF(5),VZ,IFD,PZERO(5),VZERO(5),A(5),P20(5),V20(5),  
C A2(5),IL,S2T(5),ES2(5),C2L(5),MASS2(5),MUS(5),  
C CC(5),CE(5),C2C(5),C2E(5),NVGPT,NPP,MB(5),RLT,NDELTA,  
C ES(5),SB(5),SD21(2),SD22(2),SD23(2),SD24(2),SD25(2)
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C

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COMMON/DIRCOM/  
C SD11(2),SD12(2),SD13(2),SD14(2),SD15(2),  
C S1(2),SS2(2),S3(2),S4(2),S5(2),  
C S2D21(2),S2D22(2),S2D23(2),S2D24(2),S2D25(2),  
C S2D11(2),S2D12(2),S2D13(2),S2D14(2),S2D15(2),  
C S21(2),S22(2),S23(2),S24(2),S25(2),  
C OMTD11(2),OMTD12(2),OMTD13(2),OMTD14(2),OMTD15(2),  
C OMT1(2),OMT2(2),OMT3(2),OMT4(2),OMT5(2),  
C AI(5),BI(5),DELTA1,DELTA2,DELTA3,DELTA4,DELTA5,  
C DDELT1,DDELT2,DDELT3,DDELT4,DDELT5,ISTAGE,  
C PRTMIN,IPLT,ISDF,ISTPL1,ISTPL2,ISTPL3,ISTPL4,ISTPL5,  
C DM14(22),IB(5),DM15(127),INDLG,DM16(107),CASK(44),INDFLX,  
* NMODE,DM18(40),SXMOD(100),SYM0D(100),SZMOD(100),DM19(1686),  
* GQD2(20),DDM20(20),DDM21(20),SLEN1(5),SLEN2(5),  
* GAMA,DUM15(12),INDNWS,DDM22(10),ETADES,  
* DDM23(5),AH(5),PH(5),DDM24(30)
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C

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COMMON/LGDE/LA(50),FC2(5),P2(5),PRES(5),C(5),IPPT,LTPT
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C

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COMMON/LGE/A11(5),A13(5),A31(5),A33(5),RRCGX(5),  
C RL(3,3),RI(3,3,5),RAX(5),RAY(5),RAZ(5),TMP(3),ZZERO(5),  
C XR(5),YR(5),EPSLON(5),PA(5),FDELTA(5),  
C FTRZ(5),RDX(5),RDY(5),RDZ(5),RDXG(5),RDYG(5),RDZG(5),  
C VTX(5),VTY(5),VTZ(5),GZ(5),VGPT(5),FTRX(5),FTRY(5),  
C DX(5),DY(5),DZ(5),FT(5),FDX(5),FDY(5),FF(5),AA(5),C2(5),  
C SR(5),SF(5),PSKD(5),MUV(5),MTRX(5),MTRY(5),  
C MTRZ(5),MA(5),RG11,RG13,RG31,RG33,IPRT,  
C MTX,MTY,MTZ,SFTRX,SFTRY,SFTRZ,FTRA,
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C FTRB,FTRC,SMTRX,SMTRY,SMTRZ
C
C COMMON /HTCOM/ HT, HT1, HT2, INDINT(5)
C
*CALL ACOBLK
C
REAL MASS,MOMENT,MASS2,MUS,NTIRES,MB
REAL MUVP,MTRX,MTRY,MTRZ,MA,MTX,MTY,MTZ
C
DIMENSION IPSTOP(5),AIC(5),PGA1T(5)
DIMENSION DELTA(5),DDELTA(5),DLGDE(47),
* SD2(2,5),SD1(2,5),S(2,5)
C
EQUIVALENCE (DLGDE(1),LA(1))
EQUIVALENCE (DELTA(1),DELTA1),(DDELTA(1),DDELT1)
EQUIVALENCE (SD21(1),SD2(1,1)),(SD11(1),SD1(1,1)),(S1(1),S(1,1))
EQUIVALENCE (DM15(1),ITO)
EQUIVALENCE (DM5(16),GREFF),
* (DM1(37),AIYYBS)
C
DATA AIC,IPSTOP /5*0.0,5*0/
C
O1(T1,T2)=SIGN(1.,(T1-T2))*SORT(ABS(T1-T2))
C*****
C
I = 1
C
IF(INDEACT(I).EQ.1) GO TO 56
IF(INDEACT(I).EQ.2) GO TO 80
IF(HMM(I) .EQ. 0.) GO TO 90
IF(DMRUN .GT. 0.0)GO TO 25
IF(ZDANT.GE.VELDEC) 90,40
25 IF(ZDANT.GE.VELDEC+XG77F1(1)*TAN(DMRUN)) GO TO 90
40 WRITE(6,1014)TIME
1014 FORMAT (1H036H REDUCE CONTROL LIMIT FORCE AT TIME=,E16.8)
INDEACT(I)=1
56 IF(INDINT(I).EQ.0) GO TO 58
WLFOR(I)=WLFOR(I)-REDSLP(I)*HT
EPSILO(I)=EPSILO(I)+EPSSLP*HT
58 CONTINUE
INDINT(I)=0
IF(WLFOR(I) .GT. WLFORR) GO TO 90
WRITE(6,1015)TIME
1015 FORMAT (1H027H CONTROL AT WLFORR AT TIME=,E16.8)
INDEACT(I)=2
80 WLFOR(I)=WLFORR
EPSILO(I)=EPSROL(I)
90 CONTINUE
C*****

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C
  IF(KAPT(I).NE.0)GO TO 210
  APINT(I)=0.0
210  CONTINUE
  PGA1T(I)=PGA1T1(I)
  IF(PGA1T1(I).LE.-1600.0) PGA1T(I)=-1600.0
  VOL1T(I)=VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
  VOL3T(I)=VOL3I(I)+AREA3(I)*S(1,I)
  VOL2T(I)=VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)+(VOL3T(I)
X -VOL3I(I))-VCUM(I)
  PGA2T(I)=AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMA)-PATM
  IF(SD1(1,I) .EQ. 0.0)GO TO 104
  PGA3T(I)=((COEF3(I)*AREO3(I))*2*PGA2T(I)-SD1(1,I)/ABS(SD1(1,I))
X *(SD1(1,I)*AREA3(I))*2)
  E/((COEF3(I)*AREO3(I))*2)
  GO TO 105
104  PGA3T(I)=PGA2T(I)
105  IF(PGA1T(I) .GE. PGA2T(I))GO TO 106
  GO TO 107
106  GAMAH(I)=RHOH*GREFF*(1.0+(PGA1T(I)*3.04E-08)-
  * (PGA1T(I)**2*2.72E-15))
  GO TO 108
107  GAMAH(I)=RHOH*GREFF*(1.0+(PGA2T(I)*3.04E-08)-
  * (PGA2T(I)**2*2.72E-15))
108  IF(PGA1T(I) .GE. PGA2T(I))COEFO(I)=
  * CDMOC(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
  IF(PGA2T(I) .GT. PGA1T(I))COEFO(I)=
  * CDMOE(I)*SQRT(ABS(2.*GREFF/GAMAH(I)))
  IF(IMODE(I).NE.0 .OR. DDELTA(I).GT.0.0) GO TO 113
  QO(I) = 0.0
  GO TO 109
C
113  QO(I)=COEFO(I)*(AREMO(I)-APINT(I))*Q1(PGA1T1(I),PGA2T(I))
  IF(QO(I).LT.0.0 .AND. VCUM(I).LE.0.0) GO TO 102
  AIC(I)=0.0
  GO TO 103
102  IF(PGA1T1(I).LT.PGA2T(I)) GO TO 103
  QO(I)=0.0
  VCUM(I)=0.0
  AIC(I)=1.0
103  IF(QO(I).GT.0.0) AIC(I)=0.0
109  IF(PGA2T(I) .LE. -1600.0)PGA2T(I)=-1600.0
  IF(PGA3T(I) .LE. -1600.0)PGA3T(I)=-1600.0
  IF(DELTA(I) .LE. 0.0 .AND. TIME .GT. DELT)GO TO 101
  GO TO 110
101  FFORT(I)=0.0
  GO TO 140
110  CONTINUE
C  COMPUTE STRUT AXIAL BINDING FRICTION FORCE

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BLFORT(I)=FONHST(I)*((SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))+1.0)
BUFORT(I)=FONHST(I)*((SLEN2(I)-S(1,I))/(SLEN1(I)+S(1,I))
FFORT(I)=BUMU(I)*ABS(BUFORT(I))+BLMU(I)*ABS(BLFORT(I))
140  CONTINUE
C    COMPUTE SHOCK STRUT CHARGING FORCE
    IF(S(1,I) .GT. 0.0)GO TO 142
    FORCHT(I)=PGA1T(I)*AREA1(I)+PGA2T(I)*((AREA2(I)-AREA1(I))-PGA3T(I)
X   *AREA3(I)+FFORT(I)+CFFOR(I)
C    COMPUTE NORMAL AND AXIAL HUB TO SHOCK STRUT FORCES AT HUB
142  FONHST(I)=SQRT(FDX(I)**2+FDY(I)**2)-MASS(I)*GREFF*SIPA+SBFOT
    IF(ABS(FT(I)) .LE. FORCHT(I) .AND. S(1,I) .EQ. 0.0)GO TO 150
    GO TO 801
150  CONTINUE
    FORSST(I)=FT(I)
    SD1(1,I)=0.0
    IF(IMODE(I).EQ.0) GO TO 297
    ISTROK(I)=1
    GO TO 289
C    COMPRESSION VELOCITY OF SHOCK STRUT IS POSITIVE
801  IF(SD1(1,I).LE.0.8 .AND. IFR(I).EQ.0) GO TO 2
    GO TO 3
    2  DMTANH(I)=1.0
    GO TO 284
    3  DMTANH(I)=ABS(TANH(2.0*SD1(1,I)))
    IFR(I) = 1
284  IF(S(1,I).LT.0.0) GO TO 160
    GO TO 161
160  IPSTOP(I)=1
161  IF(S(1,I).LE.DSTOP .AND. IPSTOP(I).EQ.1) GO TO 900
    GO TO 902
900  IF(S(1,I).LE.0.005) GO TO 903
    GO TO 904
903  SD2(1,I)=0.0
    SD1(1,I)=0.0
    S(1,I)=0.0
    IPSTOP(I)=0
    GO TO 902
904  CONTINUE
    IF(IFSTOP(I) .NE. 0)GO TO 906
    DSTOP=S(1,I)
    FSTOPK=2.0*MASS(I)*SD1(1,I)**2/DSTOP**2
906  IF(S(1,I) .LE. DSTOP/2.0)GO TO 908
    FSTOP(I)=-FSTOPK*(DSTOP-S(1,I))
    GO TO 909
908  FSTOP(I)=-FSTOPK*S(1,I)
909  IFSTOP(I)=1
    IFR(I) = 0
    GO TO 901
902  FSTOP(I)=0.0

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901 IF(PGA1T(I).LE.(PGA1I(I)+500.0) .AND.
: PGA1T(I).GT.(PGA1I(I)-500.0)) GO TO 158
GO TO 159
158 IF(ABS(FT(I)).LE.FORCHT(I) .AND. S(1,I).EQ.0.0) GO TO 500
159 IF(S(1,I).GE.0.0) GO TO 470
IF(SD1(1,I) .LT. 0.0)GO TO 470
GO TO 471
470 FFORT(I)=-FFORT(I)
CFFOR(I)=-CFFOR(I)
471 FORSST(I)=-((PGA1T(I)-PGA2T(I))*(AREA1(I)-APINT(I))
E +PGA2T(I)*AREA2(I)
X -PGA3T(I)*AREA3(I)+(FFORT(I)
1 +CFFOR(I))*DMTANH(I)+FSTOP(I))
500 IF(INDFLX.GE. 1)GO TO 295
IF(IMODE(I).EQ.0) GO TO 297
ISTROK(I)=1
289 IF(S(1,I) .GT. 0.0) GO TO 295
IF(IOPCO(I) .EQ. 1) GO TO 295
IF((PGA1I(I)-1000.0).LT.PGA1T(I).AND.
E PGA1T(I).LT.(PGA1I(I)+1000.0))299,298
299 IF(XVALVE(I) .NE. 0.0)GO TO 311
IF(IPASS(I) .EQ. 1)GO TO 296
XVALVE(I)=XKSV(I)*XMA11(I)+XBIAS(I)
IPASS(I)=1
GO TO 294
298 IF(ICOSV(I) .EQ. 1)GO TO 291
IOPCO(I)=0
IF(XSV(I) .LT. 0.002 .AND. XSV(I) .GT. -0.002)291,295
291 IF(SD2(1,I) .LE. 0.0 .AND. ICOSV(I) .EQ. 1)GO TO 311
IF(IOPCO(I) .EQ. 1)GO TO 295
IF(PGA1T(I) .LE. PGA1I(I)) GO TO 293
IF(IXSVL(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
IF(XVALVE(I) .GT. -0.1) GO TO 294
XVALVE(I)=-0.1
IXSVL(I)=1
GO TO 294
293 IF(IXSVH(I) .EQ. 1)GO TO 294
XVALVE(I)=XVALVE(I)+XSVDMX(I)*DELT*PERCNT(I)
IF(XVALVE(I) .LT. 0.1) GO TO 294
XVALVE(I)=0.1
IXSVH(I)=1
294 CONTINUE
DLTX1D(I)=0.0
ICOSV(I)=1
296 IF(WFORT(I) .GT. 0.0 .AND. S(1,I) .LE. 0.0)GO TO 410
311 IF(NAC(I) .EQ. 1)GO TO 307
IF(IIIXSVH(I) .EQ. 1)GO TO 305
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)

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305 IF(XVALVE(I) .LE. 0.0)305,400
XVALVE(I)=0.0
IIXSVH(I)=1
GO TO 400
307 IF(IIXSVL(I) .EQ. 1)GO TO 308
XVALVE(I)=XVALVE(I)+XSVDPMX(I)*DELT*PERCNT(I)
IF(XVALVE(I) .GE. 0.0)308,400
308 XVALVE(I)=0.0
IIXSVL(I)=1
400 CONTINUE
410 IF(XVALVE(I) .NE. 0.0)GO TO 295
ICDSV(I)=0
DELTX1(I)=DELTX(I)*XKF(I)
XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
XMA11(I)=XMA(I)
XSV(I)=XKSV(I)*XMA11(I)+XBIAS(I)
CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
IPASS(I)=0
IXSVL(I)=0
IXSVH(I)=0
IIXSVL(I)=0
IIXSVH(I)=0
IOPCO(I)=1
CALL PHLOZ2(PS(I),PR(I),XSV(I),QC(I),XLPSV1(I),XLPSV3(I),RCLSV(I),
& DSV(I),CSV1(I),CSV3(I),XMU,QTOLER,NITER,P1(I),QS1(I),QS3(I))
C
295 IF(ISTROK(I) .EQ. 1 .AND. S(1,I) .GT. 0.0)IOPCO(I)=0
297 I2 = 2*I+NSTRUT-1
I1 = I2+1
CALL INTEG(LA(I2),SD2(1,I))
CALL INTEG(LA(I1),SD1(1,I))
C
IF(IMODE(I).EQ.0) GO TO 450
C
ENUP(1) = 0.5*AIYYBS*QI77R(1)*QI77R(1)+(ENCG/CMASNG)*
* (ZG77F1(1)/ABS(ZG77F1(1)))
IF(HMM(I) .EQ. 1.0)GO TO 130
SA(I)=0.
IF(WFORT(I).LT.0.) XSTOT(I)=ENUP(I)/((-WFORT(I))*COPA)
IF(WFORT(I).GE.0.0 .OR. DDELTA(I).LT.0.0) XSTOT(I)=1.E20
C
ZSSC IS A PERCENTAGE OF SB(I) FOR ACTIVATING CONTROL-CDMOC(I) IS U
ZSSC(I)=FWORK(I)*SB(I)
IF(XSTOT(I).LE.(ZSSC(I)-S(1,I)) .OR. RESA(I).EQ.1.0) SA(I)=1.0
RESA(I)=SA(I)
IF(SA(I).EQ.0. .OR. HMM(I).EQ.1.) GO TO 130
WLFOR(I)=-WFORT(I)
C
VELDEC=((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/
& (VMASS(1)*REDSLP(I))

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WRITE(6,121)TIME,WLFOR(I),VELDEC
121  FORMAT(50H ACTIVE CONTROL INITIATED...TIME, WLFOR, VELDEC = ,
1      3E13.5)
HMM(I)=1.
130  IF(S(1,I) .GT. 0.0) ISET(I)=0
      IF(HMM(I).EQ.0.) GO TO 451
      IF(-WFOR(I).GT.(WLFOR(I)+EPSILO(I))) DF(I)=(WLFOR(I)+
6      EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LT.(WLFOR(I)-EPSILO(I))) DF(I)=(WLFOR(I)-
6      EPSILO(I))-(-WFOR(I))
      IF(-WFOR(I).LE.(WLFOR(I)+EPSILO(I)).AND.
6      -WFOR(I).GE.(WLFOR(I)-EPSILO(I)))
      : 457,456
457  IF(S(1,I) .LE. 0.0) GO TO 456
      IF(WFOR(I) .GT. 0.0 .AND. QSVCU(I) .LT. 0.0) 454,455
454  DF(I)=WLFOR(I)-(-WFOR(I))
      GO TO 456
455  DF(I)=0.0
456  DELTX(I)=S(1,I)-XSCOM(I)
      IF(S(1,I) .LE. 0.0 .AND. ISET(I) .EQ. 0) GO TO 451
      GO TO 452
451  DF(I)=0.
      DELTX(I)=0.
452  XMA(I)=(DF(I)+DELTX1(I))*XKA(I)
      IF(GNR.EQ.1. .AND. XMA(I).GT.0.) XMA(I)=
      : XMA(I)*SQRT((PGA1T(I)-PGALAC(I))
X /(PGAHAC(I)-PGA1T(I)))

C
C      NOTE: SUBROUTINE 'FLOZE2' COMPUTES THE FLOWS FROM THE PRESSURES
C      IN UNITS OF INCHES.
C      P1(I)=PGA1T(I)/144.
C      COMPUTATION OF HIGH PRESSURE ACCUMULATOR NITROGEN VOLUME
C      AND ACCUMULATOR PRESSURE
VOLANT(I)=VOLANT(I)+QSVN(I)*DELT-QPUMPS(I)*DELT
PS(I)=(((PGAHAC(I)+PATM)*(VOLANI(I)/VOLANT(I))*GAMA)-PATM)/144.0
IF(PS(I) .GE. 3000.0) 464,465
464  PS(I)=3000.0
      VOLANT(I)=VOLANI(I)
465  VOLAHT(I)=VOLACI(I)-VOLANT(I)
      IF(VOLAHT(I) .LE. 0.0) 466,467
466  WRITE(6,1050)TIME
1050  FORMAT(1H0//45H ACCUMULATOR OIL VOLUME INSUFFICIENT AT TIME=,E16.8
1      //)
      CALL LGEAR6
      STOP 500
467  CONTINUE
      CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
6      XSV(I),QS1(I),QS3(I),CSV1(I),CSV3(I),XMU)
      QSV1(I)=QS1(I)/1728.

```


QSV3(I)=QSV(I)/1728.

C

450 CONTINUE

QSV(I)=QSV1(I)-QSV3(I)

IF(QSV(I) .LT. 0.0)NAC(I)=1

IF(QSV(I) .GT. 0.0)NAC(I)=2

IF(NAC(I) .NE. 2) GO TO 462

QSVN(I)=QSV(I)

GO TO 463

462 QSVN(I)=0.0

463 IF(SD1(1,I).LT. 0.0 .AND. PGA1T(I).LE. -1600.0)PGA1T(I)=-1600.0

C

RETURN

END

*IDENT CSCMOD5

*DELETE ACORLK.7,ACORLK.7
4,ALGDUM1 ,COEF0(5) ,COEF3(5) ,COPA

*DELETE CSCMOD4.1,CSCMOD4.1

9,GAMAH(5) ,ALGDUM2 ,GNR ,HMM(5) ,ICOSV(5) ,INITSW

*DELETE CSCMOD4.4,CSCMOD4.4

*,IMODE(5) ,CHASNG ,VMAS(5) ,ZDANT ,INDEACT(5),VELDEC(5)
1,COEF1(5) ,LMODE(5)

*DELETE CSCMOD4.5,CSCMOD4.5

IF(TPD.LT.TMAX) GO TO 413

*DELETE MIMIN.114,MIMIN.114

IF((XF-X0).GT.1.E-10) GO TO 211

*DELETE CSCMOD3.21,CSCMOD3.21

*DELETE ACTINIT.33,ACTINIT.33

*,DDM21(20),SLENDUM(10),GAMA,DUM15(12),INDNWS,DDM22(10),ETADES,

*DELETE ACTINIT.35,ACTINIT.35

*DELETE CSCMOD2.22,CSCMOD2.22

DATA IOPCO,ISET,ISTROK,NAC /10*1,10*0/

*INSERT CSCMOD2.23

DATA INDEACT,INDINT,INITSW /5*0,5*1,1/

DATA FFORT,FONHST,FORCHT,FORSST,WFORT /25*0.0/

DATA QC,QD,QSV1,QSV3,QSVCU,QTOLER /25*0.,0.0001/

DATA DCON,DMTANH,DF,DP1,DELTX,DELTX1,DLTX1D /6*1.0,25*0.0/

DATA XSVDDOT,XSVDD,XSVDDD /15*0.0/

DATA REDSLP,SBFOT,VCUM,VELDEC /5*100000.,0.,10*0.0/

DATA XMA1,XMA2,XMA3,XMA4,XMA6,XMA7,XMA9,XMA10 /40*0.0/

DATA XMA,XMA5,XMA8,XMA11 /20*0.0/

*DELETE ACTINIT.71,ACTINIT.71

*DELETE ACTINIT.73,ACTINIT.75

*DELETE CSCMOD4.11,CSCMOD4.11

*DELETE ACTINIT.80,ACTINIT.82

*DELETE ACTINIT.87,ACTINIT.105

*DELETE ACTINIT.107,ACTINIT.108

*DELETE ACTINIT.110,ACTINIT.115

COEF1(I) = CDSV(I)*SQRT(2.*GREFF/GAMAH(I))*144.

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      CSV1(I) = COEF1(I)*WSV1
      CSV3(I) = COEF1(I)*WSV3
*DELETE      ACTINIT.118,ACTINIT.118
      QSV1(I) = 0.0
      QSV3(I) = 0.0
      IF(ITO.EQ.1) GO TO 60
      IF(IMODE(I).EQ.0) GO TO 80

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```

C
*INSERT      ACTINIT.126
      GO TO 80

```

```

C
60 CONTINUE
      HMM(I) = 1.0
      INDEACT(I) = 2
      LMODE(I) = IMODE(I)
      IMODE(I) = 0
      AP2TO(I) = PGA2I(I)+PATH
      VOL1T(I) = VOL1I(I)-(AREA1(I)-APINT(I))*S(1,I)
      VOL3T(I) = VOL3I(I)+AREA3(I)*S(1,I)
      VOL2T(I) = VOL2I(I)-(AREA2(I)-AREA1(I)+APINT(I))*S(1,I)
      + (VOL3T(I)-VOL3I(I))-VCUM(I)
      PGA2I(I) = AP2TO(I)*((VOL2I(I)/VOL2T(I))*GAMA)-PATH
      PGA1I(I) = PGA2I(I)
      PGA3I(I) = PGA2I(I)
      FORSST(I) = -(PGA2I(I)*AREA2(I)-PGA3I(I)*AREA3(I)
      +DMTANH(I)*(FFORT(I)+CFFOR(I))+FSTOP(I))
      WFORT(I) = FORSST(I)

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```

C
80 CONTINUE
*DELETE      ACTINIT.130,ACTINIT.130
*DELETE      ACTINIT.132,ACTINIT.132
*DELETE      ACTINIT.135,ACTINIT.139
*DELETE      CSCMOD4.12,CSCMOD4.13
*DELETE      CSCMOD4.14,CSCMOD4.14
      CALL ALGEAR1
*INSERT      ALGEAR.69
      EQUIVALENCE (DM8(79),XCGRF)
*DELETE      CSCMOD2.42,CSCMOD2.43
      DIMENSION IPSTOP(5),AIC(5),PGA1T(5)
      DATA AIC,IPSTOP /5*0.0,5*0/
*DELETE      CSCMOD3.24,CSCMOD3.24
      RXCG1 = RX(1)-XCGRF
      CMASNG = 1+(AMASS*RXCGL1*RXCGL1)/AIYYBS
*DELETE      CSCMOD2.48,CSCMOD2.48
      ZDANT = ZG77F1(1)-QI77R(1)*RX(1)
*DELETE      CSCMOD4.21,CSCMOD4.21
      18 CONTINUE
*INSERT      CSCMOD4.22
      UNSPRNG = 0.0

```

```

DO 22 I=1,NSTRUT
  IF(ITO.EQ.1 .AND. TIME.GE.2.0) IMODE(I) = LMODE(I)
  IF(OMETD1(I).NE.0.0) UNSPRNG = UNSPRNG+MASS(I)
22 CONTINUE
C
  DWFORT = (-SQRT(AXP7F*AXP7F+AYP7F*AYP7F+AZP7F*AZP7F)+GREFF)*
  .          (AMASS-UNSPRNG)
C
  CALL ACTNG
C
*INSERT      CSCMOD4.23
C
  WFORT(I) = DWFORT/(NSTRUT-1)
C
*DELETE      ALGEAR.106,ALGEAR.106
  IF(ZG77F1(1).GE.VELDEC(I)) GO TO 90
*DELETE      ALGEAR.108,ALGEAR.108
  25 IF(ZG77F1(1).GE.VELDEC(I)+XG77F1(1)*TAN(OMRUN)) GO TO 90
*DELETE      ALGEAR.256,ALGEAR.256
*DELETE      ALGEAR.261,ALGEAR.261
  296 IF(WFORT(I).GT.0.0 .AND. S(1,I).LE.0.0) GO TO 400
*DELETE      ALGEAR.274,ALGEAR.274
  400 CONTINUE
*DELETE      ALGEAR.277,ALGEAR.277
  IF(XVALVE(I).NE.0.0) GO TO 295
*DELETE      CSCMOD2.113,CSCMOD2.113
  IF(IMODE(I).NE.0.0) GO TO 119
  OSV1(I) = 0.0
  OSV3(I) = 0.0
  GO TO 450
*DELETE      CSCMOD2.124,CSCMOD2.124
  VELDEC(I) = ((WLFOR(I)+WLFORR)/2.+(WLFOR(I)-WLFORR))/
*DELETE      CSCMOD4.27,CSCMOD4.27
  .          (VMAS(I)*REDSLP(I))
*DELETE      ALGEAR.308,ALGEAR.308
  WRITE(6,121) TIME,WLFOR(I),VELDEC(I)
*DELETE      ACTNG.60,ACTNG.60
  .          SD2(2,5),SD1(2,5),S(2,5),OMETD1(2,5)
*DELETE      ACTNG.62,ACTNG.62
  EQUIVALENCE (DLGDE(1),LA(1)), (OMTD1(1),OMETD1(1,1))
*DELETE      ACTNG.80,ACTNG.80
  IF(ZDANT.GE.VELDEC(I)) 90,40
*DELETE      ACTNG.81,ACTNG.81
  25 IF(ZDANT.GE.VELDEC(I)+XG77F1(1)*TAN(OMRUN)) GO TO 90
*INSERT      ACTNG.97
  UNSPRNG = 0.0
  IF(OMETD1(1).NE.0.0) UNSPRNG = MASS(1)
  WFORT(1) = SR(1)*(VMAS(1)-UNSPRNG)
*DELETE      ACTNG.241,ACTNG.241

```

```

296 IF(WFORT(I).GT.0.0 .AND. S(1,I).LE.0.0) GO TO 400
*DELETE      ACTNG.255,ACTNG.255
      IF(XVALVE(I).NE.0.0) GO TO 295
*DELETE      ACTNG.277,ACTNG.277
      IF(IMODE(I).NE.0.0) GO TO 119
      OSV1(I) = 0.0
      OSV3(I) = 0.0
      GO TO 450
119 CONTINUE
*DELETE      ACTNG.292,ACTNG.292
      VELDEC(I) = ((WLFOR(I)+WLFORR)/2.*(WLFOR(I)-WLFORR))/
*DELETE      ACTNG.294,ACTNG.294
      WRITE(6,121) TIME,WLFOR(I),VELDEC(I)
*DELETE      DECOMP.4,DECOMP.4
      COMMON /IPSCOM/ IPS
*DELETE      SOLVE.4,SOLVE.4
      COMMON /IPSCOM/ IPS
*INSERT      ACTIN.11
C
C-----
C  THE 100 LOOP PROCESSES DATA CARDS UNTIL ENDACT IS ENCOUNTERED
C-----
C
*INSERT      ACTIN.17
C
C----- READ A DATA CARD -----
*INSERT      ACTIN.22
C
C----- IGNORE REMARKS -----
*INSERT      ACTIN.23
C
C----- TEST FOR END OF ACTIVE DATA -----
*INSERT      ACTIN.24
C
C----- LOOK FOR A MATCHING MNEUNONIC -----
*INSERT      ACTIN.28
C
C----- NO MATCH FOUND -----
*INSERT      ACTIN.30
C
C----- FOUND A MATCH -----
*INSERT      ACTIN.31
C
C-----
C  SET INDEX TO ADDRESS IN ACTIVE AND CORRECT FOR OFFSET
C  (INC ON DATA CARD)
C-----
C
*INSERT      ACTIN.34

```

C
C----- INITIALIZE ALL FLAGS AND COUNTERS -----
*INSERT ACTIN.44
C
C-----
C THE 210 LOOP SCANS DATA COLUMNS ON A SINGLE CARD
C-----
C
*INSERT ACTIN.46
C
C----- IGNORE BLANKS -----
*INSERT ACTIN.47
C
C----- COMMA SEPARATES NUMBERS -----
*INSERT ACTIN.48
C
C----- REAL NUMBER -----
*INSERT ACTIN.49
C
C----- SCIENTIFIC NOTATION -----
*INSERT ACTIN.50
C
C----- NEGATIVE -----
*INSERT ACTIN.51
C
C----- NON-NUMERIC CHARACTER -----
*INSERT ACTIN.52
C
C----- ALPHABETIC CHARACTER -----
*INSERT ACTIN.53
C
C----- CORRECT FROM DISPLAY CODE TO INTEGER DIGIT -----
*INSERT ACTIN.55
C
C----- LEFT POSITIVE INDICATES NO DECIMAL POINT HAS BEEN FOUND
*INSERT ACTIN.57
C
C----- LEFT NEGATIVE INDICATES A DECIMAL POINT HAS BEEN FOUND
*INSERT ACTIN.63
C
C-----
C A COMMA OR END OF DATA CARD SIGNALS END OF NUMBER
C-----
C
*INSERT ACTIN.66
C
C----- COMBINE LEFT AND RIGHT PORTIONS OF REAL NUMBER -----
*INSERT ACTIN.70
C

```

C----- STORE IN ACTIVE BLOCK -----
*INSERT      ACTIN.71
C
C----- RESET FLAGS AND COUNTERS -----
*INSERT      ACTIN.83
C
C----- NO DECIMAL POINT FOUND -----
*INSERT      ACTIN.86
C
C----- DATA GIVEN IN SCIENTIFIC NOTATION -----
*INSERT      ACTIN.89
C
C----- INTEGER DATA -----
*INSERT      ACTIN.90
C
C----- REAL DATA -----
*INSERT      ACTIN.93
C
C----- SET FLAG TO SHOW A DECIMAL POINT WAS FOUND ----
*INSERT      ACTIN.96
C
C----- SET FLAG TO SHOW SCIENTIFIC NOTATION -----
*INSERT      ACTIN.99
C
C----- ACCUMULATE EXPONENT -----
*INSERT      ACTIN.103
C
C----- NEGATIVE NUMBER -----
*INSERT      ACTIN.104
C
C----- NEGATIVE EXPONENT -----
*INSERT      DIRACT.3
C
C-----
C  ACTIVE CONTROL DATA IDENTIFIED BY THE MTH MNEUMONIC
C  IN ARRAY NAME IS STORED IN THE ACTIVE COMMON BLOCK
C  AT THE ADDRESS CONTAINED IN THE MTH WORD IN ARRAY LOC.
C-----
C
*IDENT      PINARY
*DELETE      CSCMOD5.4,CSCMOD5.4
             1,COEF1(5),LMODE(5),PINN(30),PINM(30),STRON(30),
             2 STROM(30)
*DELETE      CSCMOD3.20,CSCMOD3.20
             COMMON/ACTDIR/NAME(75),LOC(75)
*INSERT      CSCMOD5.21
             IF(KAPT(I).EQ.1) GO TO 50
             APINT(I) = 0.
             IF(KAPT(I).EQ.0) GO TO 50

```

```

      IF(I.GT.1) GO TO 25
      STROK = -1.
      DO 10 J=1,29
      IF(STRON(J).LT.STROK) GO TO 15
      STROK = STRON(J)
      IF(S(1,I).GE.STRON(J).AND.S(1,I).LE.STRON(J+1)) GO TO 20
10    CONTINUE
15    J = J - 1
20    APINT(I) = PINN(J)
      GO TO 50
25    STROK = -1.
      DO 30 J=1,29
      IF(STROM(J).LT.STROK) GO TO 35
      STROK = STROM(J)
      IF(S(1,I).GE.STROM(J).AND.S(1,I).LE.STROM(J+1)) GO TO 40
30    CONTINUE
35    J = J - 1
40    APINT(I) = PINM(J)
50    CONTINUE
*DELETE      ALGEAR.126,ALGEAR.128
*DELETE      ACTNG.99,ACTNG.101
*DELETE      ACTIN.3,ACTIN.3
      COMMON/ACTDIR/XNAME(75),LOC(75)
*DELETE      CSCMOD2.34,CSCMOD2.35
      COMMON/ACTIVE/DATA(802)
      DIMENSION IRA(55),MSG(58),IDATA(802)
*INSERT      ACTIN.11
C
C      ZERO OUT PIN AND STROKE ARRAYS
C
      DO 10 J=683,802
      DATA(J) = 0.
10    CONTINUE
*DELETE      ACTIN.13,ACTIN.13
      1 FORMAT(A6,1X,A3,1X,55R1,I2)
*DELETE      ACTIN.25,ACTIN.25
      DO 110 I=1,75
*DELETE      DIRACT.3,DIRACT.3
      COMMON/ACTDIR/NAME(75),LOC(75)
*DELETE      CSCMOD2.31,CSCMOD2.31
      5 6HZETAC1, 6HZETAC2, 6HIMODE , 6HPINN , 6HPINM ,
      6 6HSTRON , 6HSTROM /
*DELETE      CSCMOD2.33,CSCMOD2.33
      7 644, 645, 651, 683, 713, 743, 773/
*IDENT      KLUGEZ
*INSERT      LGEAR1.155
C
C*****
C

```

C***** TEMPORARY FIX TO FREEZE NOSE SECONDARY

C

IF(I.EQ.1) GO TO 59

C

C*****

C

*IDENT SECFIX

*INSERT LGEAR1.173

S2(1,I) = -0.5*ES2(I)

S2D1(1,I) = -1.E-10

S2D2(1,I) = -1.E-10

*DELETE LGEAR1.183,LGEAR1.184

61 IF(S2D2(1,I).LT.0.) GO TO 140

IF(S2D1(1,I).LT.0.) S2D1(1,I) = 0.

*INSERT LGEAR1.187

S2D1(1,I) = 0.

*INSERT LGEA3C.109

IF(TMP(1).LT.0.) WRITE(6,1234) I,S2(1,I),S2D1(1,I),

1 S2D2(1,I)

1234 FORMAT(1X,7H*--*--*,I5,3E16.8)

IF(TMP(1).LT.0.) GO TO 31

*IDENT TABFIX

*DELETE LGEA3C.205,LGEA3C.205

CALL HIHO(3,LOC(7),NS2NDY,NSTRUT,DU,DU,S2(1,I),FI,DU,DU,C2(I))

*DELETE LGEA3C.208,LGEA3C.208

CALL HIHO(3,LOC(6),NS2NDY,NSTRUT,DU,DU,S2(1,I),FI,DU,DU,C2(I))

*IDENT XSVMOD

*DELETE ALGEAR.224,ALGEAR.224

500 CONTINUE

*INSERT ALGEAR.230

XVALVE(I) = XSV(I)

*DELETE ALGEAR.233,ALGEAR.233

299 CONTINUE

IF(XVALVE(I).NE.0.) GO TO 311

*INSERT CSCMOD5.70

XSV(I) = XVALVE(I)

*DELETE ACTNG.209,ACTNG.209

500 CONTINUE

*INSERT ACTNG.213

XVALVE(I) = XSV(I)

*DELETE ACTNG.216,ACTNG.216

299 CONTINUE

IF(XVALVE(I).NE.0.) GO TO 311

*INSERT ACTNG.254

XSV(I) = XVALVE(I)

*IDENT GENFIX

*DELETE REST.32,REST.32

COMMON/XACTNG/AICX(5),IPSTOX(5),DM35(3)

*DELETE EXE.24,EXE.24


```

/*      DM43      ,INDVPC      ,DM44 ( 7),NCASE      ,DM45
*DELETE      SDDMODS.1,SDDMODS.1
      34 DO 35 II=1,4059
*INSERT      EXE.55
      NCASE = 1H
*INSERT      XMAFIX.7
      REWIND 7
*DELETE      CSCMOD6.4,CSCMOD6.5
*INSERT      MIMIN.113
      DO 45 I=1,5
      45 INDINT(I) = 1
*DELETE      DELTFIX.11,DELTFIX.11
      COMMON/XACTNG/AIC,IPSTOP,UNSPRNG,NIN,TIMEL
*IDENT      DELTFXX
*INSERT      CSCMOD5.48
      DATA TIMEL,NIN/0.,0/
*INSERT      ALGEAR.83
      IF(NIN.EQ.0) TIMEL = TIME
      NIN = 1
      DELT = TIME - TIMEL
      TIMEL = TIME
*DELETE      DELTFIX.3,DELTFIX.5
*INSERT      ACTNG.67
      DATA TIMEL,NIN/0.,0/
*DELETE      DELTFIX.13,DELTFIX.15
      IF(NIN.EQ.0) TIMEL = TIME
      NIN = 1
      DELT = TIME - TIMEL
      TIMEL = TIME
*IDENT      MOD282
*DELETE      REST.31,REST.31
      COMMON/XALGEA/AIC(5),DM34(14),IPSTOP(5),DM34X(12)
*DELETE      XMAFIX.7,XMAFIX.7
      READ(7) (DMD1(IJ),IJ=1,11081)
*DELETE      XMAFIX.8,XMAFIX.8
      WRITE(7) (DMD1(IJ),IJ=1,11081)
*DELETE      REST.82,REST.82
      2 PGALT,RXCG1,TTIME,UNSPRNG,NIN,TIMEL
*DELETE      CSCMOD2.106,ALGEAR.217
      159 IF(SD1(1,I).LT.0.) GO TO 471
*DELETE      ACTNG.200,ACTNG.202
      159 IF(SD1(1,I).LT.0.) GO TO 471
*IDENT      MOD296
*DELETE      CSCMOD2.80,CSCMOD2.80
      IF(IMODE(I).EQ.0.AND.S(1,I).EQ.0.) 112,113
*DELETE      CSCMOD2.133,CSCMOD2.133
*DELETE      ACTNG.126,ACTNG.126
      IF(S(1,I).GT.0..OR.IMODE(I).EQ.1) GO TO 113
*IDENT      REST1

```

```

*DELETE      REST.35,REST.35
COMMON/XAUTS/DM38(11)
*DELETE      MOD282.2,MOD282.2
READ(7) (DMD1(IJ),IJ=1,11084)
*DELETE      MOD282.3,MOD282.3
WRITE(7) (DMD1(IJ),IJ=1,11084)
*DELETE      REST.93,REST.93
1 TMP2,TMP3,TMP5,DELTS,ERROR,IPR
*IDENT      MOD329
*INSERT      MIMIN.17
H = AMIN1(HT,HT1,HT2,HT3)
*DELETE      HTRY.2,HTRY.2
*DELETE      CSCMOD2.131,CSCMOD2.132
*INSERT      ALGEAR.439
DP1(I) = (-QD(I)+QSV1(I)-OSV3(I)+(AREA1(I) - APINT(I))
1 *SD1(1,I))*BETA/VOL1T(I)
IF(S(1,I).NE.0..OR.AIC(I).NE.1.) GO TO 64
DP1(I) = 0.
64 CONTINUE
I3 = I + 3*NSTRUT
CALL INTEG(LA(I3),DP1(I))
*INSERT      ALGEAR.515
IF(IMODE(I).EQ.0) GO TO 28
*DELETE      ALGEAR.517,ALGEAR.518
*DELETE      CSCMOD2.138,CSCMOD2.139
I9 = I + 9*NSTRUT
CALL INTEG(LA(I9),DLTX1D(I))
*INSERT      CSCMODS.102
IF(IMODE(I).EQ.0) GO TO 6
*IDENT      MOD351
*INSERT      MOD329.5
PGA1T1(I) = PGA1I(I)
PGA1T(I) = PGA1I(I)
*IDENT      MOD1029
*DELETE      REST.14,REST.14
COMMON/UPDCAL/DM14(201)
*DELETE      REST.24,REST.24
COMMON/XMIMIN/DM26(1008)
*DELETE      CSCMOD3.4,CSCMOD3.4
COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      REST1.2,REST1.2
READ(7) (DMD1(IJ),IJ=1,11204)
*DELETE      REST1.3,REST1.3
WRITE(7) (DMD1(IJ),IJ=1,11204)
*DELETE      INUPD.5,INUPD.5
COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      INUPD.6,INUPD.6
IF(NUM+N .LE. 100) GO TO 5
*DELETE      INUPD.9,INUPD.9

```

```

700 FORMAT(*NUMBER OF INTEGRATION VARIABLES EXCEEDS MAX LIMIT 100*)
*DELETE      LNUPD.3,LNUPD.3
      COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      INPUZ.3,INPUZ.3
      COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      INPUZ.5,INPUZ.5
      9 DO 10 I=1,100
*DELETE      INTEG.3,INTEG.3
      COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      UPDAT.4,UPDAT.4
      COMMON/UPDCAL/NUM,P(100),Y(100)
*DELETE      MIMIN.7,MIMIN.7
      COMMON/UPDCAL/N,P(100),Y(100)
*DELETE      MIMIN.9,MIMIN.10
      DIMENSION YMAX(100),YO(100),PO(100),S(100),YP(100),Y1(100),
      1 Z(100),XK(100,3)
*DELETE      LGDET.3,LGDET.3
      COMMON/UPDCAL/NUM,P(100),Y(100)
*IDENT      MOD1040
*INSERT      AUTS.406
      DELQI = DELQDE
*IDENT      MODMIM
*DELETE      MIMIN.173,MIMIN.173
      WRITE(6,701) HT,H
      701 FORMAT(* INTEG RTN.      HT = *,E15.8,* H = *,E15.8)
*IDENT      MOD1048
*DELETE      ALGEAR.326,ALGEAR.326
      456 DELTX(I) = S(1,I)*12.0 - XSCOM(I)
*DELETE      ACTNG.312,ACTNG.312
      456 DELTX(I) = S(1,I)*12.0 - XSCOM(I)
*IDENT      MOD1103
*INSERT      LGEAR1.156
C  MODIFICATION TO ACCOMODATE SECONDARY PISTON OF F4 MAIN GEAR
      IF(S(1,I).LE.0.) GO TO 59
      IF(S(1,I).GE.SB(I)-(S2T(I)-S2(1,I))) 83,85
C  SECONDARY PISTON IN CONTACT WITH DRIFICE TUBE
      83 IF(SD1(1,I)+1.E-4.GE.S2D1(1,I)) 84,85
      84 S2D1(1,I) = SD1(1,I)
      S2D2(1,I) = SD2(1,I)
      GO TO 60
      85 CONTINUE
*IDENT      AERAT
*INSERT      REST.37
      COMMON/AEROCO/DM41(8)
      COMMON/XAERO/DM42(2)
*DELETE      MOD1029.4,MOD1029.4
      READ(7) (DMD1(IJ),IJ=1,11214)
*DELETE      MOD1029.5,MOD1029.5
      WRITE(7) (DMD1(IJ),IJ=1,11214)

```

```

*INSERT      OPT1.204
      CALL AERO4
*INSERT      OPT1.545
      SUBROUTINE AERO4
      COMMON/DIRCOM/DM1(2),X
      COMMON/TABCOM/LOCS(115),ST(115)
      COMMON/TABDIR/TABLE(800)
      COMMON/AEROC/RTAB10(2),RTAB80(2),LTAB10(2),LTAB80(2)
      COMMON/XAERO/NIN,TIMEL
      REAL LTAB10,LTAB80
      DATA TIMEL,NIN/0.,0/
      DATA RTAB10,RTAB80,LTAB10,LTAB80/8*0./
      IF(NIN.EQ.0) TIMEL = X
      NIN = 1
      DELT = X - TIMEL
      TIMEL = X
      IND1 = LOCS(45)
      IND2 = LOCS(114)
      ATAB11 = TABLE(IND1)
      ATAB12 = TABLE(IND1+1)
      ATAB81 = TABLE(IND2)
      ATAB82 = TABLE(IND2+1)
      ATAB11 = ATAB11 + RTAB10(1)*DELT
      ATAB12 = ATAB12 + RTAB10(2)*DELT
      ATAB81 = ATAB81 + RTAB80(1)*DELT
      ATAB82 = ATAB82 + RTAB80(2)*DELT
      IF(RTAB10(1).GT.0..AND.ATAB11.GE.LTAB10(1)) GO TO 40
      IF(RTAB10(1).LT.0..AND.ATAB11.LE.LTAB10(1)) GO TO 40
10  IF(RTAB10(2).GT.0..AND.ATAB12.GE.LTAB10(2)) GO TO 50
      IF(RTAB10(2).LT.0..AND.ATAB12.LE.LTAB10(2)) GO TO 50
20  IF(RTAB80(1).GT.0..AND.ATAB81.GE.LTAB80(1)) GO TO 60
      IF(RTAB80(1).LT.0..AND.ATAB81.LE.LTAB80(1)) GO TO 60
30  IF(RTAB80(2).GT.0..AND.ATAB82.GE.LTAB80(2)) GO TO 70
      IF(RTAB80(2).LT.0..AND.ATAB82.LE.LTAB80(2)) GO TO 70
      GO TO 80
40  ATAB11 = LTAB10(1)
      RTAB10(1) = 0.
      GO TO 10
50  ATAB12 = LTAB10(2)
      RTAB10(2) = 0.
      GO TO 20
60  ATAB81 = LTAB80(1)
      RTAB80(1) = 0.
      GO TO 30
70  ATAB82 = LTAB80(2)
      RTAB80(2) = 0.
80  TABLE(IND1) = ATAB11
      TABLE(IND1+1) = ATAB12
      TABLE(IND2) = ATAB81

```

```
TABLE(IND2+1) = ATAB82
RETURN
END
*INSERT      JMMODS.11
      IF(SYM.EQ.6HRTAB10) GO TO 905
      IF(SYM.EQ.6HRTAB80) GO TO 905
      IF(SYM.EQ.6HLTAB10) GO TO 905
      IF(SYM.EQ.6HLTAB80) GO TO 905
*INSERT      JMMODS.16
      905 CALL AEROIN(SYM,RA)
      GO TO 100
*INSERT      READ.194
      SUBROUTINE AEROIN(SYM,RA)
      DIMENSION RA(55)
      COMMON/AEROCO/DATAX(8)
      DATA DATAX/8*0./
      CALL LINES(1)
      WRITE(6,1) SYM,RA
1  FORMAT(18X,A6,5X,55A1)
2  FORMAT(*OERROR.ILLEGAL CHARACTER IN NUMERIC FIELD*,1R1,2H**/)
      IF(SYM.EQ.6HRTAB10) INDEX = 1
      IF(SYM.EQ.6HRTAB80) INDEX = 3
      IF(SYM.EQ.6HLTAB10) INDEX = 5
      IF(SYM.EQ.6HLTAB80) INDEX = 7
      NUMEXP = 0
      NEXP = 0
      IEXP = 0
      NL = 0
      NR = 0
      NUML = 0
      NUMR = 0
      ISIGN = 0
      JSIGN = 0
      LEFT = 1
      DO 210 I=1,56
      IF(I.EQ.56) GO TO 140
      IF(RA(I).EQ.1H ) GO TO 210
      IF(RA(I).EQ.1H,) GO TO 140
      IF(RA(I).EQ.1H.) GO TO 170
      IF(RA(I).EQ.1HE) GO TO 180
      IF(RA(I).EQ.1H-) GO TO 200
      NUM = SHIFT(RA(I),6)
      NUM = NUM.AND.00000000000000000077B
      IF(NUM.GT.36) GO TO 130
      IF(NUM.LT.27) GO TO 130
      NUM = NUM - 27
      IF(IEXP.EQ.1) GO TO 190
      IF(LEFT.GT.0) NUML = 10*NUML + NUM
      IF(LEFT.GT.0) NL = NL + 1
```

```

IF(LEFT.LT.0) NUMR = 10*NUMR + NUM
IF(LEFT.LT.0) NR = NR + 1
GO TO 210
130 CALL LINES(3)
WRITE(6,2) RA(I)
GO TO 210
140 IF(NL.EQ.0.AND.NR.EQ.0) GO TO 210
IF(NR.EQ.0) GO TO 160
X = FLOAT(NUML) + FLOAT(NUMR)/(10.)**NR
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
150 DATAX(INDEX) = X
NUML = 0
NUMR = 0
NL = 0
NR = 0
LEFT = 1
ISIGN = 0
JSIGN = 0
IEXP = 0
NEXP = 0
NUMEXP = 0
INDEX = INDEX + 1
GO TO 210
160 X = NUML
IF(JSIGN.EQ.1) NUMEXP = -NUMEXP
IF(IEXP.EQ.1) X = X*(10.)**NUMEXP
IF(ISIGN.EQ.1) X = -X
GO TO 150
170 LEFT = -1
GO TO 210
180 IEXP = 1
GO TO 210
190 NUMEXP = 10*NUMEXP + NUM
NEXP = NEXP + 1
GO TO 210
200 IF(IEXP.EQ.0) ISIGN = 1
IF(IEXP.NE.0) JSIGN = 1
210 CONTINUE
RETURN
END

```

*IDENT MOD2056

*INSERT REST.42

DATA DM41/8*0./

*DELETE CSCMOD2.17,CSCMOD2.17

EQUIVALENCE (DM5(16),GREFF),(DM1(37),AIYYBS)

*DELETE AERAT.64,AERAT.64

*DELETE PINARY.26,PINARY.26

```

COMMON/ACTDIR/XNAME(77),LOC(77)
*DELETE      PINARY.36,PINARY.36
DO 110 I=1,77
*DELETE      PINARY.37,PINARY.37
COMMON/ACTDIR/NAME(77),LOC(77)
*DELETE      PINARY.39,PINARY.39
6 6HSTRON , 6HSTROM , 6HPEDSLP, 6HDSTOP /
*DELETE      PINARY.40,PINARY.40
7      644, 645, 651, 683, 713, 743, 773, 392, 140/
*IDENT      MOD2075
*DELETE      MOD282.5,ALGEAR.219
159 SIGX = 0.
IF(SD1(1,I).EQ.0.) GO TO 471
SIGX = SD1(1,I)/ABS(SD1(1,I))
*DELETE      ALGEAR.223,ALGEAR.223
1 + CFFOR(I))*SIGX*DMTANH(I) + FSTOP(I))
*DELETE      MOD282.6,ACTNG.204
159 SIGX = 0.
IF(SD1(1,I).EQ.0.) GO TO 471
SIGX = SD1(1,I)/ABS(SD1(1,I))
*DELETE      ACTNG.208,ACTNG.208
1 + CFFOR(I))*SIGX*DMTANH(I) + FSTOP(I))
*IDENT      MOD2203
*DELETE      MOD282.1,MOD282.1
*DELETE      GENFIX.1,GENFIX.1
COMMON/XALGEA/AIC(5),DM34(14),IPSTOP(5),DM34X(12),DM34Y(5),
1 DM34Z(5),DM34A(65)
COMMON/XACTNG/AICX(5),IPSTOX(5),DM35(3),DM35X(5),DM35Y(5),
1 DM35Z(65)
*INSERT      REST.41
DATA DM34Y,DM34Z,DM34A/5*0.,5*31000.,65*0./
*INSERT      REST.42
DATA DM35X,DM35Y,DM35Z/5*0.,5*31000.,65*0./
*DELETE      AERAT.3,AERAT.3
READ(7) (DMD1(IJ),IJ=1,11364)
*DELETE      AERAT.4,AERAT.4
WRITE(7) (DMD1(IJ),IJ=1,11364)
*INSERT      MOD282.4
3 ,IGE,RDSLP,ITRIP,ICU,IQCU,IXS,IA1,IA2,IA3,IA4,
4 IA6,IA7,IA9,IA10,IGD
*DELETE      CSCMOD5.47,CSCMOD5.48
DIMENSION IPSTOP(5),AIC(5),PGA1T(5),IGE(5),RDSLP(5),ITRIP(5),
1 ICU(5),IQCU(5),IXS(5),IA1(5),IA2(5),IA3(5),IA4(5),
2 IA6(5),IA7(5),IA9(5),IA10(5),IGD(5)
*DELETE      CSCMOD2.78,CSCMOD2.78
IF(SD1(1,I).LT.0.0.AND.PGA1T1(I).LE.-1600.0)PGA1T(I)=-1600.0
IF(ITRIP(I).EQ.1)GO TO 101
*DELETE      ALGEAR.172,ALGEAR.172
IF(ABS(FT(I)).LE.FORCHT(I).AND.S(1,I).LE.E5(I))GO TO 150

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*DELETE      ALGEAR.175,ALGEAR.175
      FORSST(I)=-FT(I)
*DELETE      CSCMOD2.94,CSCMOD2.94
      IF(IMODE(I).EQ.0.AND.S(1,I).LT.ES(I))GO TO 421
*DELETE      ALGEAR.178,ALGEAR.178
*DELETE      CSCMOD2.95,CSCMOD2.95
*DELETE      CSCMOD4.24,CSCMOD4.24
      801 IF(S(1,I).GT.ES(I))ISTROK(I)=1
      IF(SD1(1,I).LE.0.8.AND.IFR(I).EQ.0)GO TO 2
*INSERT      ALGEAR.184
      IFR(I)=0
*DELETE      CSCMOD2.105,CSCMOD2.105
      158 IF(ABS(FT(I)).LE.FORCHT(I).AND.S(1,I).LE.ES(I))GO TO 500
*DELETE      CSCMOD2.107,CSCMOD2.107
      IF(IMODE(I).EQ.0)GO TO 421
*DELETE      ALGEAR.226,ALGEAR.226
      IF(S(1,I).GT.ES(I))287,289
      287 ISTROK(I)=1
      ICU(I)=0
      IQCU(I)=0
      IXS(I)=0
C      THE FOLLOWING LOGIC RETURNS THE GEARS,DURING REBOUND, TO INITIAL
C      CONDITIONS IN THE EVENT THE GEAR CONTACTS THE SURFACE BEFORE
C      THE LOGIC BETWEEN STATEMENTS 226 AND 421 IS FULLY EFFECTIVE
      289 IF(ISTROK(I).EQ.1.AND.DDELTA(I).LT.0.0)IGO(I)=1
      IF(IGE(I).EQ.1.OR.ITRIP(I).EQ.1)227,297
      227 IF(IGO(I).EQ.0)GO TO 226
      IF(DELTA(I).GT.0.0.AND.WLFOR(I).EQ.0.0)220,226
      220 IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)221,222
      221 QQ(I)=-VCUM(I)/DSTOP
      GO TO 226
      222 PGA2T(I)=PGA1I(I)
      QQ(I)=0.0
      IGO(I)=0
      226 IF(IGE(I).EQ.1)290,297
*DELETE      ALGEAR.229,ALGEAR.230
      290 IF(ITRIP(I).EQ.1.AND.IOPCD(I).EQ.1)GO TO 297
      IOPCD(I)=0
      IF(ABS(FORSST(I)).LE.FT(I))FORSST(I)=-FT(I)
      XSVDD(I)=0.0
      XSVDD(I)=0.0
      IF(DELTX1(I).GT.0.00001.OR.DELTX1(I).LT.-0.00001)GO TO 210
      DLTXD(I)=0.0
      GO TO 202
      210 DLTXD(I)=-DELTX1(I)/TAUF
      202 XMA(I)=0.0
      XMA1(I)=0.0
      XMA2(I)=0.0
      XMA5(I)=0.0

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XMA3(I)=0.0
XMA4(I)=0.0
XMA8(I)=0.0
XMA11(I)=0.0
XMA1DT(I)=0.0
XMA2DT(I)=0.0
XMA3DT(I)=0.0
XMA4DT(I)=0.0
XMA6DT(I)=0.0
XMA7DT(I)=0.0
XMA9DT(I)=0.0
XMA10D(I)=0.0
*DELETE      ALGEAR.231,ALGEAR.232
      IF((PGA1I(I)-2000.0).LT.PGA1T(I).AND.
1  PGA1T(I).LT.(PGA1I(I)+2000.0))299,298
*DELETE      XSVMOD.4,XSVMOD.4
      IF(XVALVE(I).EQ.XBIAS(I).AND.ICOSV(I).EQ.1)400,312
*DELETE      ALGEAR.234,ALGEAR.273
312  IF(ICOSV(I).NE.1)313,314
314  IF(XVALVE(I).NE.XBIAS(I))GO TO 311
298  IF(ICOSV(I).EQ.1)GO TO 311
313  IF(PGA1T(I).GT.PGA1I(I)+2000.)292,293
292  IF(IXSVL(I).EQ.1)GO TO 311
      XSVDOT(I)=XSVDMN(I)*PERCNT(I)
      IPASS(I)=1
      XVALVE(I)=XVALVE(I)+XSVDMN(I)*DELT*PERCNT(I)
      IF(XVALVE(I).LE.-0.13)300,303
303  XSV(I)=XVALVE(I)
      GO TO 297
300  XVALVE(I)=-0.13
      XSVDOT(I)=0.0
      IXSVL(I)=1
      GO TO 294
293  IF(PGA1T(I).LT.PGA1I(I)-2000.)295,294
295  IF(IXSVH(I).EQ.1)GO TO 311
      XSVDOT(I)=XSVDMX(I)*PERCNT(I)
      IPASS(I)=2
      XVALVE(I)=XVALVE(I)+XSVDMX(I)*DELT*PERCNT(I)
      IF(XVALVE(I).GE.0.13)302,304
304  XSV(I)=XVALVE(I)
      GO TO 297
302  XVALVE(I)=0.13
      IXSVH(I)=1
294  CONTINUE
      XSVDOT(I)=0.0
      ICOSV(I)=1
311  IF(PGA2T(I).GT.PGA1I(I)+4000.0)316,315
316  IF(QSV(I).LT.QO(I).AND.ICOSV(I).EQ.1)317,318
317  XSVDOT(I)=0.0

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XSV(I)=XVALVE(I)
GO TO 297
318 XSVDOT(I)=XSVDNM(I)*PERCNT(I)
XSV(I)=XVALVE(I)
GO TO 297
315 CONTINUE
IF(NAC(I).EQ.1)GO TO 307
320 IF(II XSVH(I).EQ.1)GO TO 305
XSVDOT(I)=XSVDNM(I)
IPASS(I)=3
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT
IF(XVALVE(I).LE.XBIAS(I))305,306
306 XSV(I)=XVALVE(I)
GO TO 297
305 XVALVE(I)=XBIAS(I)
XSVDOT(I)=0.0
II XSVH(I)=1
GO TO 400
307 IF(II XSVL(I).EQ.1)GO TO 308
XSVDOT(I)=XSVDNM(I)
IPASS(I)=4
XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT
IF(XVALVE(I).GE.XBIAS(I))308,309
309 XSV(I)=XVALVE(I)
GO TO 297
308 XVALVE(I)=XBIAS(I)
XSVDOT(I)=0.0
II XSVL(I)=1
*DELETE CSCMOD5.71,ALGEAR.278
*INSERT ALGEAR.279
DELT X1(I)=0.0
*INSERT ALGEAR.283
ICOSV(I)=0
QD(I)=0.0
QSVCU(I)=0.0
QSV1(I)=0.0
QSV3(I)=0.0
VCUM(I)=0.0
VOL1T(I)=VOL1I(I)
VOL2T(I)=VOL2I(I)
VOL3T(I)=VOL3I(I)
ITRIP(I)=1
PGA1T1(I)=PGA1I(I)
PGA1T(I)=PGA1I(I)
PGA2T(I)=PGA1I(I)
PGA3T(I)=PGA1I(I)
*DELETE ALGEAR.290,CSCMOD2.108
297 IF(ITRIP(I).EQ.1)430,421
430 IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)GO TO 431

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      QQ(I)=0.0
      ICU(I)=1
      GO TO 440
431  IF(VCUM(I).LT.-0.00001)GO TO 432
      QQ(I)=-VCUM(I)/DSTOP
      GO TO 440
432  QQ(I)=-VCUM(I)/DSTOP
440  IF(QSVCU(I).GT.0.00001.OR.QSVCU(I).LT.-0.00001)GO TO 433
      QSV(I)=0.0
      IQCU(I)=1
      GO TO 420
433  IF(QSVCU(I).LT.-0.00001)GO TO 434
      QSV(I)=-QSVCU(I)/DSTOP
      GO TO 420
434  QSV(I)=-QSVCU(I)/DSTOP
420  IF(XSV(I).GT.XBIAS(I)+.000001)422,423
422  IF(XSV(I).GT. 0.0)XSVDOT(I)=-XSV(I)/DSTOP
      IF(XSV(I).LT. 0.0)XSVDOT(I)=XSV(I)/DSTOP
      GO TO 600
423  IF(XSV(I).LT.XBIAS(I)-.000001)424,425
424  XSVDOT(I)=-XSV(I)/DSTOP
      GO TO 600
425  XSVDOT(I)=0.0
      IXS(I)=1
600  IF(XMA1(I).GT.0.00001 .OR.XMA1(I).LT.-0.00001)GO TO 601
      XMA1DT(I)=0.0
      IA1(I)=1
      GO TO 602
601  XMA1DT(I)=-XMA1(I)/DSTOP
602  IF(XMA2(I).GT.0.00001 .OR.XMA2(I).LT.-0.00001)GO TO 603
      XMA2DT(I)=0.0
      IA2(I)=1
      GO TO 604
603  XMA2DT(I)=-XMA2(I)/DSTOP
604  IF(XMA3(I).GT.0.00001 .OR.XMA3(I).LT.-0.00001)GO TO 605
      XMA3DT(I)=0.0
      IA3(I)=1
      GO TO 606
605  XMA3DT(I)=-XMA3(I)/DSTOP
606  IF(XMA4(I).GT.0.00001 .OR.XMA4(I).LT.-0.00001)GO TO 607
      XMA4DT(I)=0.0
      IA4(I)=1
      GO TO 608
607  XMA4DT(I)=-XMA4(I)/DSTOP
608  IF(XMA6(I).GT.0.00001 .OR.XMA6(I).LT.-0.00001)GO TO 609
      XMA6DT(I)=0.0
      IA6(I)=1
      GO TO 610
609  XMA6DT(I)=-XMA6(I)/DSTOP

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610 IF(XMA7(I).GT.0.00001 .OR.XMA7(I).LT.-0.00001)GO TO 611
    XMA7DT(I)=0.0
    IA7(I)=1
    GO TO 612
611 XMA7DT(I)=-XMA7(I)/DSTOP
612 IF(XMA9(I).GT.0.00001 .OR.XMA9(I).LT.-0.00001)GO TO 613
    XMA9DT(I)=0.0
    IA9(I)=1
    GO TO 614
613 XMA9DT(I)=-XMA9(I)/DSTOP
614 IF(XMA10(I).GT.0.00001 .OR.XMA10(I).LT.-0.00001)GO TO 615
    XMA10D(I)=0.0
    IA10(I)=1
    GO TO 421
615 XMA10D(I)=-XMA10(I)/DSTOP
C   THESE SWITCHES ARE EITHER ZERO OR ONE
421 IF(ICU(I)+IQU(I)+IXS(I)+IA1(I)+IA2(I)+IA3(I)+IA4(I)+IA6(I)
    1   +IA7(I)+IA9(I)+IA10(I) .EQ. 11)ISTROK(I)=0
    I2=2*I+NSTRUT-1
*DELETE      ALGEAR.301,ALGEAR.301
C   ZSSC IS A PERCENTAGE OF SB(I) FOR ACTIVATING CONTROL
*DELETE      CSCMOD5.77,CSCMOD5.77
    1   (VMAS(I)*RDSLP(I))
*DELETE      ALGEAR.313,ALGEAR.313
    IF(IGE(I).EQ.0)GO TO 131
    WLFOR(I)=WLFORR
    INDEACT(I)=2
    EPSILO(I)=EPSROL(I)
    GO TO 451
131 IF(S(1,I).LE.ES(I))GO TO 451
    IF(HMM(I).EQ.0)GO TO 451
*INSERT      ALGEAR.330
    IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)GO TO 458
*DELETE      ALGEAR.338,ALGEAR.338
458 P1(I)=PGA1T(I)/144.
459 VOLANT(I)=VOLANT(I)+QSVN(I)*DELT-QPUMPS(I)*DELT
*DELETE      ALGEAR.354,ALGEAR.354
    IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)462,410
410 CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
    CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
*INSERT      ALGEAR.397
    IGE(I)=0
    ITRIP(I)=0
*INSERT      ALGEAR.412
    IF(IMODE(I).EQ.0)GO TO 55
    IF(ISTROK(I).EQ.1)IGE(I)=1
*INSERT      ALGEAR.439
    IF(IGE(I).EQ.1.AND.ICOSV(I).EQ.1)GO TO 63
    IF(DELTA(I).GT.0..AND.IGD(I).EQ.1)GO TO 63

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*DELETE      MOD329.4,MOD329.5
      IF(S(1,I).NE.0.0.OR.AIC(I).NE.1.)64,63
63  DP1(I)=0.0
*INSERT      MOD329.9
      IF(S(1,I).LE.ES(I))GO TO 29
*DELETE      ALGEAR.526,ALGEAR.526
      IF(IGE(I).EQ.1)GO TO 29
*DELETE      ALGEAR.535,ALGEAR.538
*DELETE      MOD329.10,MOD329.10
      29 I9=I+9*NSTRUT
*DELETE      CSCMOD2.140,CSCMOD2.140
      27 I6=3*I+6*NSTRUT-2
*INSERT      CSCMOD2.145
      CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
      CALL LIMITS(XSVDD(I),XSVDD(I),XSVDMX(I),XSVDMN(I))
      CALL LIMITS(XSVDD(I),XSVDD(I),XDDMAX(I),XDDMIN(I))
*INSERT      GENFIX.8
      2 ,IGE,RDCLP,ITRIP,ICU,IQCU,IXS,IA1,IA2,IA3,IA4,
      3 IA6,IA7,IA9,IA10,IGO
*DELETE      ACTNG.58,ACTNG.58
      DIMENSION IPSTOP(5),AIC(5),PGA1T(5),IGE(5),RDCLP(5),ITRIP(5),
      1 ICU(5),IQCU(5),IXS(5),IA1(5),IA2(5),IA3(5),IA4(5),
      2 IA6(5),IA7(5),IA9(5),IA10(5),IGO(5)
*DELETE      ACTNG.103,ACTNG.103
      IF(SD1(1,I).LT.0.0.AND.PGA1T1(I).LE.-1600.0)PGA1T(I)=-1600.0
      IF(ITRIP(I).EQ.1)GO TO 101
*DELETE      ACTNG.157,ACTNG.157
      IF(ABS(FT(I)).LE.FORCHT(I).AND.S(1,I).LE.ES(I))GO TO 150
*DELETE      ACTNG.160,ACTNG.160
      FORSST(I)=-FT(I)
*DELETE      ACTNG.162,ACTNG.162
      IF(IMODE(I).EQ.0.AND.S(1,I).LT.ES(I))GO TO 421
*DELETE      ACTNG.163,ACTNG.164
*DELETE      ACTNG.166,ACTNG.166
      801 IF(S(1,I).GT.ES(I))ISTROK(I)=1
      IF(SD1(1,I).LE.0.8.AND.IFR(I).EQ.0)GO TO 2
*INSERT      ACTNG.168
      IFR(I)=0
*DELETE      ACTNG.199,ACTNG.199
      158 IF(ABS(FT(I)).LE.FORCHT(I).AND.S(1,I).LE.ES(I))GO TO 500
*DELETE      ACTNG.210,ACTNG.210
      IF(IMODE(I).EQ.0)GO TO 421
*DELETE      ACTNG.211,ACTNG.211
      IF(S(1,I).GT.ES(I))287,289
      287 ISTROK(I)=1
      ICU(I)=0
      IQCU(I)=0
      IXS(I)=0
C      THE FOLLOWING LOGIC RETURNS THE GEARS,DURING REBOUND, TO INITIAL

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C      CONDITIONS IN THE EVENT THE GEAR CONTACTS THE SURFACE BEFORE
C      THE LOGIC BETWEEN STATEMENTS 226 AND 421 IS FULLY EFFECTIVE
289    IF(ISTROK(I).EQ.1.AND.DDELTA(I).LT.0.0)IGO(I)=1
      IF(IGE(I).EQ.1.OR.ITRIP(I).EQ.1)227,297
227    IF(IGO(I).EQ.0)GO TO 226
      IF(DELTA(I).GT.0.0.AND.WLFOR(I).EQ.0.0)220,226
220    IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)221,222
221    QO(I)=-VCUM(I)/DSTOP
      GO TO 226
222    PGA2T(I)=PGA1I(I)
      QO(I)=0.0
      IGO(I)=0
226    IF(IGE(I).EQ.1)290,297
*DELETE      ACTNG.212,ACTNG.213
290    IF(ITRIP(I).EQ.1.AND.IOPCO(I).EQ.1)GO TO 297
      IOPCO(I)=0
      IF(ABS(FORSST(I)).LE.FT(I))FORSST(I)=-FT(I)
      XSVDDD(I)=0.0
      XSVDD(I)=0.0
      IF(DELTIX1(I).GT.0.00001.OR.DELTIX1(I).LT.-0.00001)GO TO 210
      DLTX1D(I)=0.0
      GO TO 202
210    DLTX1D(I)=-DELTIX1(I)/TAUF
202    XMA(I)=0.0
      XMA1(I)=0.0
      XMA2(I)=0.0
      XMA5(I)=0.0
      XMA3(I)=0.0
      XMA4(I)=0.0
      XMA8(I)=0.0
      XMA11(I)=0.0
      XMA1DT(I)=0.0
      XMA2DT(I)=0.0
      XMA3DT(I)=0.0
      XMA4DT(I)=0.0
      XMA6DT(I)=0.0
      XMA7DT(I)=0.0
      XMA9DT(I)=0.0
      XMA10D(I)=0.0
*DELETE      ACTNG.214,ACTNG.215
      IF((PGA1I(I)-2000.0).LT.PGA1T(I).AND.
1    PGA1T(I).LT.(PGA1I(I)+2000.0))299,298
*DELETE      XSVMOD.9,XSVMOD.9
      IF(XVALVE(I).EQ.XBIAS(I).AND.ICOSV(I).EQ.1)400,312
*DELETE      ACTNG.217,ACTNG.253
312    IF(ICOSV(I).NE.1)313,314
314    IF(XVALVE(I).NE.XBIAS(I))GO TO 311
298    IF(ICOSV(I).EQ.1)GO TO 311
313    IF(PGA1T(I).GT.PGA1I(I)+2000.)292,293

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292 IF(IXSVL(I).EQ.1)GO TO 311
   XSVDDOT(I)=XSVDNM(I)*PERCNT(I)
   IPASS(I)=1
   XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
   IF(XVALVE(I).LE.-0.13)300,303
303 XSV(I)=XVALVE(I)
   GO TO 297
300 XVALVE(I)=-0.13
   XSVDDOT(I)=0.0
   IXSVL(I)=1
   GO TO 294
293 IF(PGA1T(I).LT.PGA1I(I)-2000.)295,294
295 IF(IXSVH(I).EQ.1)GO TO 311
   XSVDDOT(I)=XSVDNM(I)*PERCNT(I)
   IPASS(I)=2
   XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT*PERCNT(I)
   IF(XVALVE(I).GE.0.13)302,304
304 XSV(I)=XVALVE(I)
   GO TO 297
302 XVALVE(I)=0.13
   IXSVH(I)=1
294 CONTINUE
   XSVDDOT(I)=0.0
   ICOSV(I)=1
311 IF(PGA2T(I).GT.PGA1I(I)+4000.0)316,315
316 IF(QSV(I).LT.QO(I).AND.ICOSV(I).EQ.1)317,318
317 XSVDDOT(I)=0.0
   XSV(I)=XVALVE(I)
   GO TO 297
318 XSVDDOT(I)=XSVDNM(I)*PERCNT(I)
   XSV(I)=XVALVE(I)
   GO TO 297
315 CONTINUE
   IF(NAC(I).EQ.1)GO TO 307
320 IF(IIIXSVH(I).EQ.1)GO TO 305
   XSVDDOT(I)=XSVDNM(I)
   IPASS(I)=3
   XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT
   IF(XVALVE(I).LE.XBIAS(I))305,306
306 XSV(I)=XVALVE(I)
   GO TO 297
305 XVALVE(I)=XBIAS(I)
   XSVDDOT(I)=0.0
   IIIXSVH(I)=1
   GO TO 400
307 IF(IIIXSVL(I).EQ.1)GO TO 308
   XSVDDOT(I)=XSVDNM(I)
   IPASS(I)=4
   XVALVE(I)=XVALVE(I)+XSVDNM(I)*DELT

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309 IF(XVALVE(I).GE.XBIAS(I))308,309
XSV(I)=XVALVE(I)
GO TO 297
308 XVALVE(I)=XBIAS(I)
XSVDOT(I)=0.0
IIXSVL(I)=1
*DELETE CSCMOD5.87,ACTNG.256
*INSERT ACTNG.257
DELTX1(I)=0.0
*INSERT ACTNG.261
ICOSV(I)=0
QD(I)=0.0
QSVCU(I)=0.0
QSV1(I)=0.0
QSV3(I)=0.0
VCUM(I)=0.0
VOL1T(I)=VOL1I(I)
VOL2T(I)=VOL2I(I)
VOL3T(I)=VOL3I(I)
ITRIP(I)=1
PGA1T1(I)=PGA1I(I)
PGA1T(I)=PGA1I(I)
PGA2T(I)=PGA1I(I)
PGA3T(I)=PGA1I(I)
*DELETE ACTNG.268,ACTNG.272
297 IF(ITRIP(I).EQ.1)430,421
430 IF(VCUM(I).GT.0.00001.OR.VCUM(I).LT.-0.00001)GO TO 431
QD(I)=0.0
ICU(I)=1
GO TO 440
431 IF(VCUM(I).LT.-0.00001)GO TO 432
QD(I)=-VCUM(I)/DSTOP
GO TO 440
432 QD(I)=-VCUM(I)/DSTOP
440 IF(QSVCU(I).GT.0.00001.OR.QSVCU(I).LT.-0.00001)GO TO 433
QSV(I)=0.0
IQCUI(I)=1
GO TO 420
433 IF(QSVCU(I).LT.-0.00001)GO TO 434
QSV(I)=-QSVCU(I)/DSTOP
GO TO 420
434 QSV(I)=-QSVCU(I)/DSTOP
420 IF(XSV(I).GT.XBIAS(I)+.000001)422,423
422 IF(XSV(I).GT. 0.0)XSVDOT(I)=-XSV(I)/DSTOP
IF(XSV(I).LT. 0.0)XSVDOT(I)=XSV(I)/DSTOP
GO TO 600
423 IF(XSV(I).LT.XBIAS(I)-.000001)424,425
424 XSVDOT(I)=-XSV(I)/DSTOP
GO TO 600

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425 XSVDDT(I)=0.0
    IXS(I)=1
600 IF(XMA1(I).GT.0.00001 .OR.XMA1(I).LT.-0.00001)GO TO 601
    XMA1DT(I)=0.0
    IA1(I)=1
    GO TO 602
601 XMA1DT(I)=-XMA1(I)/DSTOP
602 IF(XMA2(I).GT.0.00001 .OR.XMA2(I).LT.-0.00001)GO TO 603
    XMA2DT(I)=0.0
    IA2(I)=1
    GO TO 604
603 XMA2DT(I)=-XMA2(I)/DSTOP
604 IF(XMA3(I).GT.0.00001 .OR.XMA3(I).LT.-0.00001)GO TO 605
    XMA3DT(I)=0.0
    IA3(I)=1
    GO TO 606
605 XMA3DT(I)=-XMA3(I)/DSTOP
606 IF(XMA4(I).GT.0.00001 .OR.XMA4(I).LT.-0.00001)GO TO 607
    XMA4DT(I)=0.0
    IA4(I)=1
    GO TO 608
607 XMA4DT(I)=-XMA4(I)/DSTOP
608 IF(XMA6(I).GT.0.00001 .OR.XMA6(I).LT.-0.00001)GO TO 609
    XMA6DT(I)=0.0
    IA6(I)=1
    GO TO 610
609 XMA6DT(I)=-XMA6(I)/DSTOP
610 IF(XMA7(I).GT.0.00001 .OR.XMA7(I).LT.-0.00001)GO TO 611
    XMA7DT(I)=0.0
    IA7(I)=1
    GO TO 612
611 XMA7DT(I)=-XMA7(I)/DSTOP
612 IF(XMA9(I).GT.0.00001 .OR.XMA9(I).LT.-0.00001)GO TO 613
    XMA9DT(I)=0.0
    IA9(I)=1
    GO TO 614
613 XMA9DT(I)=-XMA9(I)/DSTOP
614 IF(XMA10(I).GT.0.00001 .OR.XMA10(I).LT.-0.00001)GO TO 615
    XMA10DT(I)=0.0
    IA10(I)=1
    GO TO 421
615 XMA10DT(I)=-XMA10(I)/DSTOP
C   THESE SWITCHES ARE EITHER ZERO OR ONE
421 IF(ICU(I)+IQU(I)+IXS(I)+IA1(I)+IA2(I)+IA3(I)+IA4(I)+IA6(I)
    1 +IA7(I)+IA9(I)+IA10(I) .EQ. 11)ISTROK(I)=0
    IZ=2*I+NSTRUT-1
*DELETE      ACTNG.285,ACTNG.285
C   ZSSC IS A PERCENTAGE OF SB(I) FOR ACTIVATING CONTROL
*DELETE      ACTNG.293,ACTNG.293

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```
      1      (VMASS(I)*RDSLP(I))
*DELETE      ACTNG.299,ACTNG.299
      IF(IGE(I).EQ.0)GO TO 131
      WLFOR(I)=WLFORR
      INDEACT(I)=2
      EPSILO(I)=EPSROL(I)
      GO TO 451
131  IF(S(1,I).LE.ES(I))GO TO 451
      IF(HMM(I).EQ.0)GO TO 451
*INSERT      ACTNG.316
      IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)GO TO 458
*DELETE      ACTNG.324,ACTNG.324
458  P1(I)=PGA1T(I)/144.
459  VOLANT(I)=VOLANT(I)+QSVN(I)*DELT-QPUMPS(I)*DELT
*DELETE      ACTNG.340,ACTNG.340
      IF(S(1,I).LE.ES(I).AND.ITRIP(I).EQ.1)462,410
410  CALL LIMITS(XSV(I),XSVDOT(I),XSVMAX(I),XSVMIN(I))
      CALL FLOZE2(PS(I),PR(I),P1(I),XLPSV1(I),XLPSV3(I),RCLSV(I),DSV(I),
*IDENT      MOD2235
*DELETE      MOD2203.41,MOD2203.41
      IF(DELTA(I).GT.0.) IGO(I) = 0
*DELETE      MOD2203.235,MOD2203.235
      IF(ITRIP(I).EQ.1) GO TO 63
*DELETE      MOD2203.277,MOD2203.277
      IF(DELTA(I).GT.0.) IGO(I) = 0
*IDENT      MOD2342
*DELETE      MOD2203.216,MOD2203.216
      1      (VMASS(I)*REDSLP(I))
*IDENT      MOD3028
*DELETE      MOD2203.452,MOD2203.452
      1      (VMASS(I)*REDSLP(I))
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| 16. Abstract Modifications to a multi-degree-of-freedom flexible aircraft take-off and landing analysis (FATOLA) computer program, including a provision for actively controlled landing gears to expand the programs simulation capabilities, are presented. Supplemental instructions for preparation of data and for use of the modified program are included. | | | | | |
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